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MDAC/ROCKETDYNE Solar Receiver Design Review

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MDAC/ROCKETDYNE SOLAR RECEIVER Design Review

FINAL REPORT C-E Contract 17677

ASA 78-06

H. M. Payne Project Leader August 1978

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Combustion Engineering, Inc. Power Systems Group Windsor, Connecticut 06095

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SECTION 1

INTRODUCTION AND SUMMARY

1.1 Introduction

This report presents the results of a review of the MDAC/Rocketdyne solar central receiver designs for both the 100 MWe commercial plant and the 10 MWe pilot plant. The major objective of this design review was to assess the adequacy of the design in meeting the requirements of the solar central receiver (boiler) over a commercial lifetime of 30 years.

The MDAC/Rocketdyne design consists of an external solar heated receiver, composed of a multiple of modular panels arranged in parallel and operating on the once-through steam generation principle. Each panel is composed of welded tangent tubes, connected between inlet and outlet headers. Subcooled water enters the bottom headers, flows upward, absorbs heat, produces saturated steam throughout the two phase region, and exits at the top as superheated steam. Tube size and material is the same for both the commercial and pilot plants. Panel sizes are different between the two plants. Commercial plant heat flux is approximately 2.8 times that of the pilot plant. Structural supports and attachments of both designs are similar. Control of final superheat temperature is maintained by varying the water flow to each of the panels, according to the thermal absorption of each panel. The pilot and commercial plant receiver designs are therefore similar in construction and mode of operation. They differ significantly, however, in thermal loading (heat flux).

1.2 Summary

1.2.1 MDAC and Rocketdyne design documents were examined initially, and potential problem areas were identified, in order to determine the analysis procedures required in the design review.

These initial observations and potential problem areas are listed in Reference 24. The tentative problem areas were identified as:

- 1. Lack of CHF test data at the high flux levels.
- 2. Tube wall thickness is greater than necessary.
- 3. A start-up rate of temperature rise may be too high.
- 4. ASME Code Case 1592 should be used for fatigue data.
- 5. Daily cyclic operation is only part of the potential fatigue stress problem to be addressed. CHF oscillations and cloud cover will contribute to fatigue damage, since Incoloy 800 does not show an endurance limit.
- The weld joining tangent tubes may contribute stress concentration and is a potential source of cracking.

1.2.2 Thermal-Hydraulic Analysis

An existing C-E computer program was modified for the solar receiver configuration. This program performs an energy and mass balance on the receiver panels by a finite difference type integration of the heat flux curve. Outputs include tube metal temperatures, enthalpy, absorbed heat flux and pressure drop, as a function of panel length.

The design parameters of the solar receiver lie outside the ranges of existing C-E in-house standards for predicting CHF and the minimum film boiling coefficient. A literature search was conducted for suitable correlations of CHF and film boiling. It was determined that the MacBeth correlation was suitable for CHF prediction, and the Groeneveld correlation for film boiling. The Groeneveld correlation is conservative, compared to that used by Rocketdyne; i.e., it predicts higher metal temperatures.

Test data from Rocketdyne was analyzed for minimum film boiling coefficients. These were compared to values calculated by the two correlations above, and a wide variation was found. Generally speaking, the test data lies somewhere between the two correlations mentioned above. The Rocketdyne tests were limited in heat flux and mass flux. Further testing is needed, particularly at the higher mass and heat flux levels of the commercial plant.

Static stability analysis indicated that orificing of individual boiler tubes was not needed. Header sizing appears to be adequate as a check showed virtually no flow imbalance due to header and piping orientation.

Tube temperature calculations, using a two-dimensional steadystate heat conduction program, indicated that the maximum tube crown temperature for the pilot plant was $553^{\circ}C$ ($1027^{\circ}F$), in the superheater region. Maximum tube temperature for the commercial plant was $597^{\circ}C$ ($1106^{\circ}F$) at the minimum film coefficient and $626^{\circ}C$ ($1158^{\circ}F$) in the superheater region.

1.2.3 Stress Analysis

Stress analysis of the receiver panels indicated no problem with the pilot plant. Fatigue life in excess of 30 years should be achieved. The commercial plant design is subject to a reduced fatigue life due to the high panel temperature difference as a result of the CHF and film boiling problem. This causes the panel to tend to assume a "mattress" shape. Calculated panel stresses were well beyond yield (in compression) and an elastic analysis was conducted to simulate an inelastic analysis.

Results, while very conservative, indicate a serious fatigue problem as a result of the daily start-up and shut-down cycles of the receiver. Fatigue cracking would be expected on the outer surface of the tubes. High cycle fatigue due to CHF oscillations and cloud cover was not addressed in this review. Changing tubing size and material shows some improvement in fatigue stress but not enough to assure 30 years life. An inelastic analysis is recommended to further refine and clarify the fatigue life of the unit. The single most important factor in improving fatigue life would be to eliminate the CHF and film boiling in the high flux commercial plant. This might be accomplished by the use of rifled tubing in the evaporator region. An alternative would be a lower flux, but at some loss of efficiency.

A review of the CHF temperature oscillation problem indicated that further testing will be needed to establish the magnitude and frequency data needed to perform a fatigue analysis.

1.2.4 Water Chemistry Specifications

Water chemistry specifications and procedures were reviewed. In general, the specifications established for the receiver water quality control are satisfactory. Additional recommendations on pH control are included, along with C-E procedures for lay-up of once-through boilers.

1.2.5 Major Problems and Proposed Solutions

Major problem areas identified include the lack of accurate DNB and film boiling data for this design. Further testing at the heat flux and mass flux levels of the commercial design is

recommended. Analysis indicates that the pilot plant will not experience CHF, whereas the commercial plant will experience CHF and temperature oscillations, which may affect fatigue life. The pilot plant will give an optimistic view of the fatigue life expected in the commercial plant design, due to the CHF problem.

Preliminary review of a "turbulator" as a solution to the CHF problem in the commercial design was undertaken. Indications are that a rifled tubing evaporator section and a separate superheater section could be adapted to this external receiver without any adverse impact on receiver efficiency. Success of this method is predicated on results of testing of rifled tubing in this high flux environment.

SECTION 2

THERMAL-HYDRAULICS ANALYSIS AND WATER CHEMISTRY

2.1 Introduction

This section includes three major sub-sections. The first is a description of the thermal-hydrualics analysis of the pilot plant and commercial plant receiver designs. Secondly, an evaluation of the proposed 5-tube panel SRE retest is presented. Thirdly, the proposed plant water quality control is reviewed.

The method employed for the thermal analysis consisted of modifying an existing computer program to conform to the geometry of the MDAC/ Rocketdyne receiver configuration. This thermal analysis program is a finite difference calculation of the energy and mass balance on small increments axially along the length of a receiver panel tube. It is a one-dimensional calculation with a correction to account for the 2-D radial heat flow in the tube. Output consists of tube crown temperature profile, fluid enthalpy, and pressure losses. As existing standard correlations did not cover the ranges of parameters in this design, a literature review was made to select applicable correlations for the critical heat flux and the film boiling coefficients. Output of the thermal program served to generate input to the 2-D temperature and stress finite element programs, MARC-Heat, and MARC-Stress. The thermal analysis program was also employed to evaluate the proposed retest of the 5-tube panel. Each sub-section is discussed in detail below.

2.2 Thermal/Hydraulic Analysis of Pilot and Commercial Plants

2.2.1 <u>Review of Correlations for Critical Heat Flux and Film Boiling</u> A review was made of available correlations for the calculation of the film boiling heat transfer coefficient. References are

listed at the end of the report. The review was conducted with the objective of selecting a correlation applicable to the range of design parameters of the MDAC receiver design:

Pressure	11.03 M Pa (1600 psia)
Tube Diameter (Inside)	6.858 mm (.27 in)
Inside Heat Flux	1.57 MW/M ² (498,900 BTU/hr ft ²)
Mass Flux	2712 Kg/M ² -s $(2 \times 10^6 \text{ lbm/hr ft}^2)$
Flux Distribution	Heated 180 ⁰ Circumferentially (Approximate cosine shading)

Existing C-E standards for CHF and film boiling are not applicable to the above ranges. In addition to a correlation for film boiling, a correlation for predicting the critical quality (X_c) is also needed to establish the point of minimum film coefficient (X_{min}) .

2.2.1.1 Correlation for X_c

Results of the review indicated that the Thompson-MacBeth correlation would be satisfactory for this analysis. It represents a large amount of world data in this pressure range. The MacBeth correlation is in two parts: one for low velocity, and another for high velocity. The transition from low to high velocity was determined as $G > 406.8 \text{ Kg/M}^2$ -S (0.3 x 10⁶ lbm/hr ft²) for the high velocity range. A description of this correlation is given in Appendix A.

The high mass flow correlation is shown graphically in Figure 2.1, for a range of CHF and mass flow rates

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FIGURE 2.1

2.4

encountered in the solar receiver design. Figure 2.1 indicates the "inverse mass flow effect" in which an increase in mass flux results in a lower critical quality, q/A, being constant. This is the reverse of the trend at higher pressures.

The Thompson-MacBeth correlation is limited to a maximum pressure of 13.79 MPa, (2000 psia). The "change-over" to inverse mass flow effect is seen in other data at about the 13.79 MPa (2000 psia) pressure level. Constants in this correlation are listed for discrete pressure levels. For this analysis, these constants are interpolated lineraly between 10.68 MPa (1550 psia) and 13.79 MPa (2000 psia). Figure 2.2 shows a typical low mass flow correlation, used for pilot plant analysis.

2.2.1.2 Correlation for h and X hmin

Having a method of predicting the critical quality above, it is now possible to predict the point of minimum film boiling, the minimum coefficient (h_{min}) and the coefficients existing from X_{hmin} to full saturated steam. Of the various film boiling correlations reviewed, practically all are variations on the Dittus-Boelter equation:

 $Nu = .023 R_e^{.8} Pr^{1/3}$

The correlation selected was a modified version of the Groeneveld correlation shown in Appendix B. It assumes equilibrium between the liquid and vapor phases.



FIGURE 2.2

Values of h calculated for the commercial plant design conditions for the North panel are plotted on Figure 2.3. The value of h, at the calculated minimum film quality $(X_{h_{min}})$, is the minimum coefficient (h_{min}) . Note that the Pr_w , evaluated at the wall temperature, requires an iterative procedure to determine the exact value. A $Pr_w = 1.0$ was assumed in Figure 2.3

This correlation is applicable for the following conditions: Geometry Tube Flow Direction Vertical and Horizontal D, 5 to 25.4 mm (0.2 to 1.0 in) P, 6.89 to 21.5 MPa (1000 to 3124 psia) $(.21 \times 10^6 \text{ to } 3 \times 10^6 \text{ lbm/ft}^2)$ G, 284 to 4069 Kg/m^2-S X. % wt. 10 to 90 $(35 \times 10^3 \text{ to } 650 \times 10^3 \text{ BTU/hr ft}^2)$ Ø, .11 to 2 MW/m² 6.6×10^4 to 1.3×10^6 R_{_}Factor .88 to 2.21 Pr Y .706 to .976

Accuracy of this correlation is \pm 30% against experimental values, the majority of which were obtained with uniform 360° circumferential heating. The effects of nonuniform 180° heating on the solar boiler will be discussed in a later section.

2.2.2.2 <u>Review of Experimental Heat Transfer Data as Reported by</u> <u>Rocketdyne/MDAC</u>

Certain experiments were conducted by Rocketdyne during





Quality FIGURE 2.3 the solar receiver design development. These tests were reported in References 1 and 3.

Data as reported above, in tabular form and in the form of plots of tube temperatures, were analyzed with the objective of determining the minimum film boiling coefficient (h_{min}) and comparing these values with those calculated by the Groeneveld correlation, and with the correlation recommended by Rocketdyne. The tests analyzed are given in Table 2.1. These tests came from 3 sources: 1) electric resistance heated SS tubing, 2) 1- and 5-tube panels heated with radiant heaters (horizontal), and 3) SRE tests of 5-tube and 70-tube actual size panels mounted vertically and heated with radiant heaters.

Input data was taken from tables and graphs of measured tube and fluid temperatures as presented in Reference | . Values of h_{min} were back-calculated using heat flux values determined to exist at the apparent location of the minimum heat transfer coefficient. Heat absorption to the fluid was calculated based on increase in fluid enthalpy. Heat flux locally was prorated according to the reported axial distribution of heater power for each test, in terms of power normalized against the average. Inside tube metal temperatures were calculated assuming radial flow based on measured outside crown temperatures, for the local calculated inside heat flux (radial flow). Film coefficient from the test data was then determined by:

2.0

TABLE 2.1

Rocketdyne Test Data

				Inputs			Measu	red	P	redicted	by Groeneve	ld	Predicted Dittus-Boe	by lter
		Tube L					Tube O.D.		1		Tube O.D.		Tube O.D.	
No.	0.D./I.D.	ft.	Tube K	Ppsia	$q/Aix10^{-3}$	$Gx10^{-6}$	Temp. ^o F	hmin	X _{DNB}	X _{F.B.}	Temp.	hmin	Temp.	hmin
Rocke	etdyne Base	Tests, 36	0° Resist	ance Hea	ted (Single	Tube)								
	.25/.12	15.75	150	1750	244	1.55	1100	594	.27	.36	965	906	760	3809
	.25/.12	15.75	150	2000	244	1.55	1022	777	.28	- 38	997	843	774	4103
	.25/.12	15.75	150	2400	244	1.55	824	2711	.80	.92	861	1920	787	5245
Horiz	i contal Radia	nt Heated	Tests, 1	80 ⁰ , Sin	gle Tube Tes	ts								
11	.375/.305	65	150	1900	149.8	.59	820	942	.55	.64	980	467	754	1580
12	.375/.305	65	150	1515	175.8	.56				No Film	m Boiling			
13	.375/.305	65	150	1590	210	.75	680	6774	.55	.64	1023	558	779	1586
14	.375/.305	65	150 ·				Two Pha	ase Inl	et - Mos	st of Tu	be SH			
20	.375/.305	65	150	1965	145.2	. 59	850	789	.56	.66	968	475	774	1580
21	.375/.305	65	150	2250	239	.56	1000	683	.48	. 59	1204	476	836	1792
22	.375/.305	65	150	1100	166.3	.62	825	618	.91	.99	960	451	757	1001
23	.3/5/.305	65	150	1775	143.3	.62	890	600	.58	.67	957	465	747	1465
SRE T	'ests, 5-Tub 	e Panel To	est No. 6	-15, 70-2	Tube 12-Max.									
6	.5/.27	56	150	1500	15.36	.054	630	614	1.0	1.0	1095	31.5	691	181
15	.5/.27	56	150	1500	163.2	.254	1030	474	.88	.96	1437	218	934	663
12	.5/.27	56	150	1395	96.0	.04	1220	165	.93	1.0	3051	40	1181	181
13	.5/.27	56	150	1400	124.8	.043	1100	281	.91	.99	3674	41.5	1356	181
14	.5/.27	56	150	1485	134.4	.053	1070	336	.90	.98	3879	41.9	1415	181
16	.5/.27	56	150	1565	144.0	.043	1060	378	.89	.97	4105	42.2	1473	181
17	.5/.27	56	150	1550	124.8	.041			.91	.99	3674	41.5	1356	181
Max.	.5/.27	56	150	1550	182.4	.40	1080	480	.87	.95	1221	349	889	954
Exten	ded SRE Test	s												
1	.5/.27	56	150	1543	115.2	.062	1000	342	.92	1.0	2933	50.7	1199	214
2	.5/.27	56	150	1543	110.4	.062	975	352	.92	1.0	2844	50.5	1175	214
3	.5/.27	56	150	1543	115.2	.061	910	468	.92	1.0	2933	50.7	1199	214
4	.5/.27	56	150	1543	115.2	.058	965	383	.92	1.0	2933	50.7	1199	214
5	.5/.27	56	150	1543	124.8	.063	920	497	.91	.99	3099	51.3	1250	214
6	.5/.27	56	150	1543	144.0	.063	870	754	.89	.97	3455	52.2	1350	214
	1					1		1						

2.10

 $h = \frac{(q/A)i}{\Delta t_{film}}$

These values are therefore the "measured" values of h min.

Using the same test data inputs, an h_{min} was calculated using both the Groeneveld and Dittus-Boelter (MDAC recommended correlation). Note that the mass flux for the SRE tests falls below the lower limit for the Groeneveld correlation, thus invalidating this correlation for SRE tests.

A comparison of the calculated values and the measured values is presented in Figure 2.4 and Figure 2.5. Figure 2.4 presents the results of comparing the 360[°] uniform heated data. The 180[°], one-side heated values are given in Figure 2.5.

For the 360° heated data, Figure 2.4, the Groeneveld equation predicts the coefficient better than the Dittus-Boelter equation, but only one point lies within the \pm 30% error band. Dittus-Boelter predicts coefficients much higher than measured.

For the 180° heated data at the higher mass flux, the Dittus-Boelter predicts high coefficients, and the Groeneveld correlation predicts low values. The data mostly lie outside the \pm 30% error band. In terms of metal temperatures, the Groeneveld correlation is conservative while the Dittus-Boelter is optimistic.







In the SRE tests, the mass flux was very low, $67.8 \text{ Kg/m}^2 \text{ S}$ (.05 x 10^6 lbm/hr ft^2). The Groeneveld correlation is not valid below $271 \text{ Kg/m}^2 \text{ s} (.2 \times 10^6 \text{ lbm/hr ft}^2)$. The Dittus-Boelter equation now predicts low coefficients for these low G rates. This analysis points out the wide variation existing between these two correlation methods. It is recommended that further testing be carried out, with higher mass flux and heat flux. If possible, the range of variables for both the pilot plant and the commercial plant should be tested with enough accuracy to determine the CHF, transition, and film boiling performance.

2.2.3 Thermal-Hydraulic Analysis Program

2.2.3.1 General Description

The computer program used to determine the thermalhydraulic performance of an MDAC design solar panel employs a step-wise integrative calculational method. That is, the program performs an energy balance on a small element of the fluid, calculates heat absorption, the increase in enthalpy at the element outlet, the tube metal temperature and the fluid quality. This then becomes the input for the next element.

The program also calculates the inside film coefficients for each element, along with the pressure drop, using the homogeneous model, in the two-phase region. The homogeneous model for ΔP calculation was selected because of its simplicity. It can be shown, that in the case where small increments of flow are considered, the

homogeneous model agrees with the Martinelli-Nelson two-phase multiplier approach at the high mass flows. At lower flows, the agreement is within 20%.

The program is designed mainly for the axisymmetrical heat flux case, with a correction factor to account for the 2-D heat flow to the rear of the tube.

The program incorporates the flexibility to allow the user to calculate:

1. Preheat Panel Performance

2. Boiler Panel Performance

- a. With variable outlet conditions.
- b. With specified outlet conditions.

3. Multiples of the above.

Such flexibility allows calculations of design and off-design parameters. A description of the program is given in Appendix C.

2.2.3.2 Receiver Performance Results

Program output for the pilot plant and commercial plant maximum heat flux cases (N panels) are shown in Appendix C.

Tables 2.2 and 2.3 show the performance distribution data for the pilot and the commercial plant receivers, respectively. Figure 2.6 is a key plan, identifying the location of the panels by number. The receivers are symmetrical about the N-S axis. Boiler panel Nos. 1-9 were analyzed for each design. Panels 10, 11, and 12

TABLE 2.2

Predicted Performance (Rated Steam) Pilot Plant

Full Power

	Solar Flux	Flow	ΔP			
Panel	MW/m ²	Kg/s	KPa	Stability		
No.	(BTU/hr ft ²)	(1b/hr)	(psi)	Index	Efficiency	
1	.31	1.06	111	1.294	.89	
	(98000)	(8419)	(16.06)			
3	.28	.936	94.7	1.198	.88	
	(87175)	(7437)	(13.73)			
5	.25	.843	83.9	1.149	.88	
	(79250)	(6697)	(12.18)			
7	.18	.59	59.6	1.016	.86	
101	(57060)	(4712)	(8.64)			
9	.13	.41	46.4	0.849	.82	
	(41210)	(3265)	(6.73)			
		Hal	Lf Load			
1	.15	.5	52.5	0.953	.84	
	(49000)	(3970)	(7.61)			
3	.14	.439	48	0.885	.83	
	(43587)	(3487)	(6.97)			
5	.13	. 39	45.3	0.835	.81	
	(39625)	(3114)	(6.57)			
7	.09	.26	38.9	0.711	.76	
	(28530)	(2095)	(5.64)			
9	.07	.17	35.3	0.559	.69	
	(20605)	(1372)	(5.13)			
		Min. Co	ontrol Load			
			1014 11-2		550341	
1	.06	.15	33.6	0.532	.63	
	(19600)	(1194)	(4.87)	2 222		
3	.055	.147	35.6	0.615	.70	
	(17435)	(1170)	(5.16)		-	
5	.05	.088	28.9	0.461	.46	
-	(15850)	(697.7)	(4.20)			
7	.036	*	*	*	* .	
	(11412)		120	1.00		
9	.026	*	*	*	*	
	(8242)					

*Not capable of producing rated steam.

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TABLE 2.3

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Predicted Performance (Rated Steam) Commercial Plant

Panel No.	Solar Flux MW/m ² (BTU/hr ft ²)	Flow Kg/s (lb/hr)	♪ P MPa (psi)	Stability Index	Efficiency
1	.85	15	2.06	0.526	.91
3	(269500) .76 (239600)	(119100) 13.4 (106300)	(299) 1.67 (242)	0.650	,91
5	.686	(100300)	(242) 1.41 (204)	0.654	.91
7	.52 (164840)	9.1 (72600)	.89 (129)	0.477	.91
9	.37 (117600)	6.4 (51180)	.54 (78)	0.918	.90
		Half	Load		
1	.425	7.4 (58890)	.65	0.720	.90
3	.38 (119800)	6.6 (52300)	.55 (80)	0.711	.90
5	.34 (108730)	5.9 (47250)	.48 (70)	0.711	. 89
7	.26 (82420)	4.4 (35170)	.34 (50)	0.736	.88
9	.185 (58800)	3.1 (24380)	.24 (35)	0.785	.85
		Min. Con	trol Load		
1	.17	2.8	·22 (32)	1.155	.85
3	.15 (47920)	2.4 (19360)	.19 (27)	1.260	.83
5	.14 (43492)	2.2 (17300)	.17 (24)	1.038	.82
7	.104 (32968)	1.6 (12420)	.12 (17)	1.010	.78
9	.074 (23520)	1.0 (7942)	.09 (13)	0.692	.69

Full Power

2 ...



FIGURE 2.6

are preheat panels in the pilot palnt. Panels 11 and 12 are preheat penals in the commercial plant. The incident flux used was as given in Reference 1 for the panels shown in the pilot plant. Only the north panel flux was given directly for the commercial plant. The others were back-calculated from heat absorption data Three loads were selected to given in Reference 1. give a distribution of panel performance. In addition to the full load at maximum incident flux, half-load and a minimum load were picked. The minimum control load for producing rated steam 515.6°C (960°F) was set at 20% of full load. This proved to be too low for the pilot plant south-east panels to make rated steam (Table 2.2).

For these runs the bulk fluid inlet temperature was held constant at $260^{\circ}C$ ($500^{\circ}F$) for the pilot plant and $293^{\circ}C$ ($560^{\circ}F$) for the commercial plant. This was determined by running the preheater panels for full load in each case, and using the output as boiler input. In practice, the input temperature to the boiler panels would decrease with load. In this respect, the minimum control load may be too low for some of the panels indicated in Table 2.2. The heat absorption may not be enough to produce $515.6^{\circ}C$ ($960^{\circ}F$) final steam temperature with lower feedwater inlet temperatures. Since the thermal program does not iterate for a specified outlet pressure, input pressure is determined approximately, and the outlet

2.1.15

pressure is allowed to vary with flow rate. The program iterates flow to produce $515.6^{\circ}C$ (960°F) final temperature ± 2.78°C (± 5°F).

The ΔP reported in Tables 2.2 and 2.3 is the sum of the friction, acceleration, and gravity components for the tubes only. These values do not include entrance and exit losses.

The panel efficiency is defined as the ratio of absorbed heat (W · AH) divided by the integrated incident heat flux profiles. This is a function of the back radiation and convection losses which are in turn dependent upon the tube metal temperature. This accounts for the greater losses, percentage-wise in the lightly-loaded panels. The loss model is the same for all cases. It can be seen that in cases of lightly loaded panels the superheater portion may actually experience a negative absorbed heat flux, resulting in a loss of fluid temperature in the top of the panels. In some cases, final temperature, 515.6°C (960°F), cannot be obtained, even with reduced flow. Such is the case with the pilot plant Panels 7 and 9 at minimum control load. As flow is reduced to increase enthalpy, the metal temperature increases, increasing the losses, which are then more than the incident heat flux.

Figures 2.7 and 2.8 show plots of the incident and absorbed flux profiles for the pilot and commercial plant's north panel. Figure 2.9 shows a minimum load for the pilot plant

2 . 2 3









100 MWe Commercial Plant North Panel



Pilot Plant Panel 7 Absorbed and Incident Heat Flux

1122
where the absorbed heat flux is largely negative in the superheater region.

2.2.3.3 Temperature Profiles

Figure 2.10 shows the temperature profile for the commercial plant N-panel. The tube temperatures are crown temperatures from the thermal analysis program. These are corrected for 2-D heat flow to the rear of the tube. As indicated, the CHF occurs at 17% quality. This puts the minimum film boiling coefficient at 27% quality. The portion of the curve in dashed line represents areas where there is uncertainty, or discrepancy in the correlations used. The peak temperature of 593.3°C (1100°F) at 9.14 m (30 feet) represents a point calculated from the Groeneveld correlation discussed previously. The distance from CHF to X hmin is the transition boiling region, and is subject to oscillating temperature. The stable film boiling point is reached at 9.14 m (30 feet). Beyond the X_{hmin}, the Groeneveld correlation should predict metal temperature However, there is a discontinuity up to x = 1.0. when switching to the single phase correlation for the superheater. Until this is resolved by experiment, this region was smoothed to the single phase correlation at X = 1.0. The highest metal temperature occurs in the superheater, just before the heat flux curve starts to decrease.

Actual calculated temperatures are listed in Appendix C, on the program output sheets. The significant fact about



Tube Length m (ft)

these temperature profiles is the difference between the tube crown temperature and the bulk fluid temperature. In 2-D heat flow, the rear of the tube is at a temperature close to the bulk temperature. The front-to-back ΔT therefore represents a potential thermal stress condition if the panel is restrained in a certain way. The magnitude of this ΔT is the cause of the fatigue stress problem in the commercial plant. These temperature profiles are indications of the relative magnitude of this problem. The high ΔT is created by a combination of high heat flux, low film coefficient, tube thickness, and poor metal conductivity. There are two critical locations. One is at film boiling where a high ΔT exists. The other is where the superheater metal is at a higher average temperature, with less favorable metal properties.

Figure 2.11 shows the temperature profile for the No. 9 commercial plant panel for the full load condition. The lower flux results in a crown temperature of 471°C (880°F) while the CHF point has moved up the panel to approximately 10.67 m (35 feet).

Figures 2.12 and 2.13 show temperature profiles predicted for the pilot plant. The CHF occurs at vey high quality and there is no film boiling point outside the single phase superheat region. Maximum temperatures at the superheater location are $560^{\circ}C$ (1040°F) and 526.7°C



Tube Length m (ft)

DIOTODE 0 11





FIGURE 2.13

(980[°]F), respectively. The Δ T is smaller in the pilot plant due to reduced heat flux. Consequently, the thermal stresses are smaller.

2.2.4 CHF Temperature Oscillations

The "boiling crisis" is invariably accompanied by temperature excursions at the point of the onset of CHF. Precise determination of the magnitude and frequency of these temperature excursions is virtually impossible without testing the particular design under consideration. A review of the literature reveals some general trends which may indicate relative orders of magnitude of the temperature oscillations. See Reference No. 21. For instance, if the crisis occurs at low quality, the temperature oscillations are likely to be more severe than at high quality. Thus it would be expected that the pilot plant would probably not experience temperature oscillations at dryout. On the other hand, the commercial design indicates a critical quality of 17%. Accompanied by high heat flux, the temperature excursions may be severe. The mass flux with commercial plant, however, is relatively high and the higher mass velocities will tend to reduce the magnitude of temperature oscillations (Reference 21).

Some typical temperature oscillations are shown in the above reference. For instance, Figure 6 of the above reference shows typical strip chart reproductions of tube metal temperatures measured mid-way through the tube wall of a high pressure boiler (P = 19 mPa (2755 psia)). At heat flux levels of .52 to .567 MW/m² (165,000 to 180,000 BTU/hr ft²), the peak-to-peak temperatures reduce from 18.75° C to 10° C as the mass flux

increases from 1220 to 1763 Kg/m^2 s (.9 x 10⁶ to 1.3 x 10⁶ lbm/hr ft²). This heat flux and mass flux are lower than those calculated for the commercial solar receiver. Frequency is difficult to determine from the charts. The traces are irregular, and a spectral density analysis would be needed. Figure 10 shows that the magnitude of temperature oscillations increases with heat flux.

The proposed 5-tube panel SRE retest has a major objective of measuring these temperature oscillations. This program is discussed further in the next sub-section.

A means of avoiding this problem would be the use of flow "turbulators", such as twisted tapes, or rifled tubes. These devices have been shown to be effective in certain designs in delaying the onset of CHF to higher qualities, thereby minimizing or eliminating the temperature excursions. A design study would be needed to assess these devices and a test program to confirm their performance for this application. Such a program will be recommended for use in the commercial plant design.

2.2.5 Static Stability Analysis

Figure 2.14 illustrates the concept of static stability. Problems arise when the system pressure curve has a zero or negative slope. Interactions with the pump flow/pressure curve could then cause flow excursions to occur. These, if severe, could cause eventual burnout.

To insure stability under these conditions, the slope of the ΔP flow curve must be positive. The greater the positive slope, the

- -1

Curve Illustrating Static Instability



Flow

FIGURE 2.14

more stability is enhanced. Maximum slope is 2.0, the case if all single-phase substance is flowing.

The stability index is defined as follows:

$$SI = \frac{\log \frac{\Delta P_1}{\Delta P_0}}{\log \frac{G_1}{G_0}}$$

where: Subscript O refers to a particular point in question.

Subscript 1 denotes a flow increment change of +5% from

the 0 point.

In the analysis, the ΔP term includes friction, momentum, and gravity terms.

This stability index (SI) was calculated for the load conditions in Tables 2.2 and 2.3. These are plotted on Figures 2.15 and 2.16. The absolute values of the SI are not significant. The fact that for all the steady state load points where rated 515.6°C (960°F) steam is generated, the SI values are positive, indicates static stability and no orificing of the tubes is needed.

Figures 2.17 and 2.18 show simulations of start-up conditions in a quasi-steady state. These curves were generated by holding the flow constant at 20% of the maximum steady state load point, then varying the heat flux from the 20% down to zero. This would simulate a normal start-up where a pre-determined flow is established, and the heat flux increased until rated steam conditions are produced at the panel outlet. This causes the ΔP to experience all points between full liquid phase to full steam quality phase at the panel outlet. Although the pilot plant results indicate the need for









orificing, only a small negative is indicated, and this is at very low heat flux level (20% of the minimum control load). If tests in the 5-tube and 70-tube panels confirm no serious instability, then orificing would not be necessary.

In Tables 2.2 and 2.3 each panel efficiency is tabulated for the load points investigated. This thermal efficiency is calculated by:

$$\eta = \frac{W \cdot a H}{\int_{0}^{A} q \, da}$$

where: $W \cdot A H =$ heat absorbed

 $\int_{e}^{A} \dot{q} da = \text{total incident energy available}$

3.3 Evaluation of 5-Tube Panel Retest Program

An interim conclusion (Reference 24) was that further tests were needed to better define the CHF/film boiling region for this design. This subsection describes a thermal analysis of the MDAC/Rocketdyne 5-tube test panel with a recommended test matrix as input. The test matrix recommended is presented in Table 2.4.

The objective of the 5-tube panel retest is to obtain more accurate data in the CHF, transitional, and film boiling regions. In order to select the instrumentation, it is necessary to predict the locations on the panel where the CHF/film boiling region will occur under the test conditions specified in Table 2.4. Toward that end, the matrix points listed in Table 2.4 were inputted to the thermal analysis program. Program outputs gave the temperature profiles, locations, ΔP 's, etc. for each test point.

TABLE 2.4

Test Matrix for 5-Tube Test Panel DOE/MDAC Solar Receiver

Pressure In = 12.76, 11.72, 11 MPa (1850, 1700, 1600 psia)

Temperature In = 260, 293°C (500, 560°F)

Kg/m ² s	Mass Flow (1b/hr ft ² x 10 ⁻⁶) (Incident Heat Flux MW/m ² (BTU/hr ft ² x 10 ⁻³)					
		.6 (190.2)	.5 (160)	.41 (130)	.31 (98)	.15 (49)	
2306	(1.7)	x	x	x	x+	x+	
1763	(1.3)	x	Х	x	x+	x+	
1153	(.85)	x	x	x	x+	x+	
879	(.648)	x					
732	(.54)		x				
678	(.50)	Х*	X	x	x	x+	
597	(.44)			x			
434	(.32)				X		
407	(.3)		X*	X*	x	x+	
271	(.2)			Х*	X*	×	
207	(.153)					х	
68	(.05)				X*	Х*	

+ No DNB indicated.

* 649°C (1200°F) final temperature reached before panel exit.

Maximum heat flux available for the test was .6 MW/m^2 or 70% of the commercial plant maximum. A range of 5 flux levels was selected. A range of 12 mass flux values was selected so as to provide at least 4 or 5 flows for each heat flux. The matrix is to be repeated for 12.75, 11.72, and 11.03 MPa (1850, 1700, and 1600 psia), and panel inlet temperature of 260 or 29.3°C (500 or 560°F). The upper (superheat) region is not of interest here, so the tests need not be carried out with the outlet temperatures as high as the results show. The thermal analysis program was stopped at 649°C (1200°F). Sometimes this occurred before the panel outlet. Sometimes superheat was not obtained, depending on the particular combination of test parameters. Runs are marked in Table 2.4 where no CHF was indicated. It is still necessary to run these points, to confirm the CHF locations. Tube sizes, material, and all other parameters were the same as for the pilot plant receiver analysis discussed previously.

Figure 2.19 shows the temperature profile for the highest heat flux, low flow case. Note that a 649° C (1200 $^{\circ}$ F) final temperature was predicted. The flux profile at the upper end need not be used in the actual tests.

Figure 2.20 shows a similar profile for the same heat flux, but at maximum flow. Here the flow is so high that film boiling does not occur before the flux starts to drop. In this case, the flux profile should be shifted toward the top of the receiver, and the subcooling reduced. These two graphs serve to illustrate the large area of panel over which CHF may occur.

Results of the analysis are shown on the next three graphs. Figure 2.21 shows how the location of the CHF point is affected by heat flux as the parameter.



Max. Flux; Low Flow



FIGURE 2.19

5-Tube Panel Test Predicted Performance

Max. Flux; High Flow



FIGURE 2.20

An area between 4.57 and 7 m (15 and 23 feet) was seen as including most of the CHF data of interest (high flux data). The area from 7 to 10 m (23 to 33 feet) is of a lesser interest. Thermocouples were concentrated in the area 4.57 to 7 m (15 to 23 feet), with spacing every 50.8 mm (2 inches). In the lesser region, spacing was selected as 76.2 mm (3 inches). Outside these two regions, 304.8 mm (12 inches) spacing is satisfactory. Figure 2.22 shows that the area can be narrowed by reducing panel subcooling. Figure 2.23 indicates the pressure effect on CHF location. These curves are for an inlet temperature of 260° F (500° F). Results of this thermal analysis indicate a general region exists where thermocouples can be concentrated to respond to the CHF phenomena over a wide range of input parameters.

Due to the fact that a number of the test points are "off-design", the pressure drop curves indicate static instability in certain regions (Figure 2.24). In order to assure stability during these tests, it is recommended that orifices be installed in the 5-tube panel. Figure 2.25 shows the minimum correction necessary. This requires orifices with a diameter ratio of 0.45. Tests may be run without orificing to check for instability. In static instability, the pump characteristic curve plays an important role. The effect of static instability should be manifested by gross flow excursions, which can occur between parallel tubes. Without sufficient orificing, test points should be approached cautiously to avoid the possibility of an actual burn-out. Dynamic instabilities may also be present, which are manifested by pressure and flow oscillations.









5-Tube Test Panel Predicted Location of DNB, L Effect of Pressure



5-Tube Test Panel Pressure Drop W/o Orifices



2-.11

5-Tube Test Panel Pressure Drop with Orifices



FIGURE 2:25

3.4 Water Chemistry

Since the STPS has a once-through steam generator, the water chemistry concepts which apply to C-E Combined Circulation (supercritical) Boilers are applicable to these units. Most of these concepts which include volatile water chemistry, condensate polishing, and a philosophy of zero solids entering the boiler have been incorporated in the design of both the pilot and commercial STPS.

However, the proposed feedwater pH of 9.5 is too high. A pH this high will cause excessive corrosion in the condenser (90/10; copper/nickel). The pH at the preheat panel inlet of HP heater outlet should be maintained between 9.0 and 9.2. If the condensate demineralizers are operated on a hydrogen cycle, high feedwater pH's will result in a greatly increased regeneration frequency due to ammonia exhaustion of the cation resin. The condensate demineralizers must be operated on the hydrogen cycle if there is any condenser leakage.

By its very nature, the STPS will have large load fluctuations. This type of operation results in increased amounts of preboiler oxide formation which eventually forms deposits on the heat transfer surfaces of the steam generator. Since the STPS will probably have to be chemically cleaned at a much higher frequency than normal high pressure boilers, necessary connections should be provided.

In addition to the nighttime standby procedures which maintain blanketing steam on the high pressure heaters and deaerator, there should be layup procedures for the preheat and evaporator panels. For short term shutdowns this whould consist of applying a nitrogen overpressure (20.7-345.5 KPa (3-5 psig)) and maintaining normal operational chemical limits.

During long term shutdowns the unit should be filled with condensate quality water treated with 200 ppm of hydrazine and 10 ppm of ammonia (pH 10.0) and a nitrogen overpressure (20.7-34.5 KPa (3-5 psig)) maintained. A typical layup procedure for a combined circulation unit is shown in Appendix D.

SECTION 3

STRESS ANALYSIS

3.1 Introduction

A heat transfer and an elastic stress analysis were performed for both the DOE/MDAC 100 MW commercial plant and a 10 MW pilot plant solar receiver designs.

The maximum stress of the commercial plant is $472.3 \text{ MPa} (68,500 \text{ lb/in}^2)$ and occurs at the film boiling point at the tube crown. This stress results in an estimated fatigue life of less than 200 cycles.

The maximum stress of the pilot plant is 185 MPa (26,850 $1b/in^2$). The estimated fatigue life of the pilot plant is 10^5 cycles.

The results of the stress analysis indicate the flux loading on the commercial plant is too severe. The high flux loading causes a large temperature gradient through the tubes, thereby causing large axial stresses. Table 3.7 summarizes the thermal cases analyzed. The fatigue life of the commercial plant is very low (200 cycles). Table 3.8 summarizes the fatigue lives of each case analyzed.

Two methods of reducing the temperature gradient across the tube were investigated, a thinner wall thickness and use of a ferritic steel. In each case, the gradient and, consequently, stress were reduced. A thinner tube reduced the gradient $35.6^{\circ}C$ ($96^{\circ}F$) and the ferritic reduced the gradient $31.7^{\circ}C$ ($89^{\circ}F$). The fatigue life of each was increased from 1100 cycles to 4500 cycles for the thin tube and from 1100 cycles to 3500 cycles for the ferritic. A combination of both produced a temperature gradient reduction of $61.7^{\circ}C$ ($143^{\circ}F$) and a fatigue life of 10,000 cycles.

?-2

A thermal transient analysis was performed on the pilot plant panel headers. Based on fluid temperature rise of 482° C/hr (900°F/hr) for 20 minutes, a fatigue life of 10^{5} cycles was calculated.

The stresses on the pilot plant are low and the fatigue life of the pilot plant reflects this (10^5 cycles) . The pilot plant could operate with the indicated flux .244 MW/m² (77,230 BTU/hr ft²).

This stress analysis consisted of an elastic analysis only. The high flux loading and stresses should be analyzed inelastically. The fatigue life assessments would not be as conservative as the ones based on an elastic analysis. The scope of this review did not allow this type of analysis. However, a follow-on stress analysis should be conducted to determine the degree of conservation of the elastic analysis.

3.2 Analytical Procedures

A heat transfer and an elastic stress analysis were performed for the DOE/MDAC solar receiver design review. Loading conditions for the pilot plant and 100 MW commercial plant were analyzed. Design recommendations and fatigue life of the receiver were made based on these results.

A finite element model was generated based on panel geometry and constraint conditions from Rockwell International Corp. Drawing AP77-084. An 8-noded isoparametric element was used. Material properties for Incoloy 800H were obtained from Code Case 1592. Figures 3.1, 3.2, and 3.3 show the geometry and Tables 3.1, 3.2, 3.5 and Figure 3.4 show material properties. Four heat flux loading conditions were analyzed, three for the commercial plant (film boiling points, maximum crown temperature) and one for the pilot plant (maximum crown temperature).

2 - -



FIGURE 3.1



COORDINATE SYSTEM

FIGURE 3.2



FIGURE 3.3

3.6





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TABLE 3.1

Instantaneous Coefficient of Thermal Expansion vs. Temperature

Temp., F	Instantaneous Coefficient of Thermal Expansion, in./in ⁹ F X 10 ⁶						
that is a factory	304 SS and 316 SS	Ni-Fe-Cr Alloy 800H	2% Cr - 1 Mo	Ni-Cr-Fn-Cb Alloy 718			
70	9.11	DC	* **	24			
100	9.21		6.6	6.91			
200	9,50	8.8	6.9	7.43			
300	9,73	8.9	7.35	7.77			
-100	9.90	9.0	7.65	7.97			
500	10.20	9.1	7.9	8.09			
600	10.13	9.2	8.1	8.17			
700	10.66	9.3	8.25	8.26			
750	. 10,81	19. A	•				
800	10.90	9.5	8.1	8.42			
850	11.00	9.65					
900	11.11	9.13	8.5	8.69			
950.	11.23	10.0					
1000	11.35	10.2	8.6	9.13			
1050	11.16	10.4		9.46			
1 1 4111	11.58	10.6	8.65				
11.50	11.70	10.8	1914				
1.200	11.81	11.0	8.7				
1250	11.93	11.2					
1300	12.01	11.4					
1350	12.16	11.55					
1 100	12.20	11.7					
1150	12.39	11.8					
1500	12.50	11.9					
1.550		12.0					
1600		12.1					

TABLE 3.2

Modulus of Elasticity vs. Temperature

(Static) Modulus of Elasticity, pai × 10 6 Temp., 'F Ni-Fe-Cr Ni-Cr-Fe-Cb 304 SS and 316 SS Alloy 800H 2% Cr - 1 Mo Alloy 718 70 28.3 28.5 29.9 -29.0 100 200 27.7 27.8 29.5 28.38 $27.93 \\ 27.51$ 27.1 27.3 300 29.0 26.6 26.8 28.6 100 500 26.1 20.3 28.0 27.10 25.7 600 25.1 27.4 26.69 700 21.8 25.2 26.6 26.26 750 ---21.1 21.6 25.7 25.82 200 850 23.724.4 23.3 21.1 25.35 900 24.5 23.8 9.50 22.9 1000 22.5 23.5 23.0 24.84 22.1 21.7 21.56 23.3 1050 22.0 1100 20.4 22.7 21.3 11.50 1200 20.9 22.1 15.6 1250 20.5 22.1 20.1 1:000 21.7 13.50 19.7 21.4 19.2 1100 21.1 1150 18.7 20.7 1500 18.3 20.3 1550 19.8 1000 19.2

The overall procedure consisted of inputting the heat flux loading into MARC Heat Program and generating steady-state temperature distributions across the model for each flux loading. These temperature distributions were then input into MARC Stress via a post tape. An elastic stress analysis was done for each case using the appropriate boundary conditions, pressure loading, and material properties. Using the results of these stress runs, fatigue life of both plants was calculated.

The 4 flux loading conditions analyzed are shown in Table 3.3. The first 3 are flux loadings on the commercial plant and the last is the flux loading on the pilot plant. The flux distribution was input on the tubes using a tube shading program. The model temperatures were put on tape to be input into the stress program. Results of the heat transfer analysis were also plotted for each case. Figures 3.5 through 3.8 show the temperature profiles for each loading condition.

The generated temperature distributions, pressure loads, and boundary conditions were input into MARC Stress. The panel geometry and the large front to back temperature loading necessitated the use of a generalized plane strain finite element. This element allowed the element to grow axially (strained T_{MEAN}) without constraint while still calculating the thermal strains due to differences in temperature from front to back.

The boundary conditions placed on the model are illustrated in Figure 3.9. Symmetry between horizontal slide locations cause the middle to be placed on rollers in the y direction. The horizontal slide weld point allows only x direction movement and symmetry forces the center line of tube to move the same amount. The tube rotation is fixed to zero in the z direction due to axial constraints.
List	of	Loading	Conditions
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No.	Heat Flux	Fluid Temp.	Film Coeff.	Pressure
	MW/m ²	C	MW/m ² C	MPa
	(BTU/hr ft ²)	(F)	(BTU/hr ft ² °F)	(psia)
1	.779	328	.021	12.5
	(247,213)	(622)	(3759)	(1815)
2	.779	328	5.3 x 10 ⁻³	12.5
	(247,213)	(622)	(940)	(1815)
3	.757	426	.011	11.67
	(239,862)	(798)	(1907)	(1693)
4	.244 (77,230)	432 (809)	2.6×10^{-3} (460)	10.97 (1592)











Table 3.4 shows the results of the elastic stress analysis for the highest stressed element for each loading condition (tube crown where the temperature loading is the highest). Stress contour plots for the commercial plant film boiling point are shown in Figures 3.10 to 3.13. These figures illustrate the stress pattern which arises for each case. Figure 3.14 shows the displacement of the model.

3.3 Fatigue Life Assessment

An assessment of the fatigue life of the commercial and pilot plant was made. Since an elastic analysis was performed, cyclic life had to be determined by using the calculated elastic strains. Fatigue life based on high elastic strains is very conservative. To fully evaluate the fatigue life, an inelastic analysis is required to determine the inelastic strain range.

Due to the high elastic strains calculated, it was necessary to modify the elastic strains. J. L. Houtman in the Westinghouse report, "Structural Evaluation of the In-Vessel FFTF Plant Unit Instrument Tree", presented a method to approximate inelastic strain using elastic analysis (Reference 25).

The procedure is as follows:

- Step 1 Calculate the elastic stress.
- Step 2 Calculate the elastic VonMises effective strain range.
- Step 3 Determine the inelastic strain from the elastic strain by multiplying by the appropriate strain correction factor K from Figure 3.15.
- Step 4 Using this strain follow up the stress strain curve to a maximum stress value (Figures 3.16, 3.17). Drop down from

2.1.

Stress Components

Loading Condition

No.		Ox	σy	Gz	JXY	Von Mises
1	MPa (psi)	-30.97 (4,492)	8.07 (1171)	-328 (-47,580)	1.47 (214)	318.4 (46,180)
2	в	44.93 (6,516)	7.88 (1143)	-44.8 (-64,510)	-2.19 (-318)	472.3 (68,500
3	ţ.	-3.02 (-438)	7.75 (1124)	-363 (-52,670)	.11 (16)	365.6 (53,030)
4		28.17 (4086)	2.1 (305)	-168.6 (-24,450)	-1.27 (-184)	185.1 (26,850)

TABL	Æ	3.	.5
	_	_	

Temp., °F	304 SS	316 55	Ni-Fe-Cr Alloy 800H	2¼ Cr-1 Mo	Ni-Cr-Fe-Cb Alloy 718	
		(Stross	as in ksi Units)			
BT	.30.0	30.0	25.0	30.0	150.0	
100	23.8	29.2	24.3	29.4	148.4	
200	25.0	25.8	22.5	27.8	143.9	
300	22.5	23.3	21.1	26.8	140.7	
.100	20.7	21.1	20.0	26.6	138.3	
500	19.1	19,9	19.0	26.5	136.7	
600	i8.2	18.8	18.3	26.5	135.4	
700	17.7	18.1	17.5	26.5	134.3	
750	17.3	17.8	17.2	26.5	133.7	
\$500	10.8	17.6	16.8	26.5	133.1	
850	10.5	17.4	16.5	26.3	132.1	
900	16.2	17.3	16.3	25.6	131.5	
950	15.9	17.1	16.1	24.7	130.5	
1000	1.5,6	17.0	15.8	23.6	129.4	
1050	15.2	16.7	15.6	22.1	128.0	
1100	11.7	10.5	15.3	20.4		
1150	11.4	10.1	15.0	18.4		
1200	1.1.1	16.2	14.3	16.1		
1250	13.7	15.8	14.5			
1300	13.2	15.3	14.3			
1350	12.5	11.9	14.0			
1400	11.0	11.1	13.7			
1450	10.6	13.8	13.4			
1500	9,5	13.1	13.0			
15.50			12.3			
1400			11.0			

20

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NO

FIGURE 3.14

3-23

1. 6

STRAIN ODRECTION FACTOR



Sn - ELASTIC STREES Sy - YEILD STREES (SEE TABLE 5)

FIGURE 3.15



3.25



3-26

this point elastically until a comprehensive yield value is reached. Proceed along the plastic slope back to the stress axis at zero strain.

Step 5 Using the determined inelastic effective strain and the appropriate fatigue design curve (Figure 3.18) fatigue damage is evaluated.

Using this method, the fatigue life of the two plants are:

	Strain Range	Cycles
Commercial Plant	8.8×10^{-3}	150
Pilot Plant	1.4×10^{-3}	10 ⁵

This technique calculates the first cycle strain range. Relaxation of this stress with time has not been taken into account. This method produces a conservative fatigue life.

3.4 Thermal Transient Analysis

A thermal analysis was done on the panel headers to determine how the start-up rates would effect fatigue life of the headers. The analysis was done on CREPLACYL, an elastic-plastic-creep analysis program for long, thick-walled cylinders.

Input into the program includes header geometry, film coefficients, material properties and fluid heat-up rates. Stress concentration factors due to header tees were included in the analysis. Exact dimensions of the panel headers could not be obtained from the design reports. Based on dimensioned figures, approximate dimensions were used (inside radius 76.2 mm (3 inches), thickness 25.4 mm (1 inch). The entire start-up loading profile for the unit was not contained



2.22

Design Fatigue Strain Range ϵ_{t} for Incalloy 800

in the report, although the report stated that within the start-up cycle, the maximum fluid temperature rise was 482° C (900 $^{\circ}$ F/hr) for 20 minutes.

Results of the cycling indicated header fatigue life due to this loading was greater than 10⁵ cycles. However, this is only a small portion of the start-up cycle. A full start-up, shut-down loading history is needed in order to provide a better indication of header life.

3.5 Thin Tube Analysis

The tube thickness from the design report is significantly greater than the minimum wall thickness from the ASME Boiler and Pressure Vessel Code. Because of less metal resistance, a thin tube provides better heat flow to the fluid. An analysis was performed to determine what reduction in stress would occur if the minimum ASME wall thickness was used.

From <u>Section I - Power Boilers</u>, Pg 27, the minimum allowable tubing thickness is defined as:

$$t = \frac{PD}{2S+P} + 0.005D + e$$

where:

P = maximum allowable working pressure

D = outside diameter of cylinder

S = maximum allowable stress value at the operating temperature of the metal

e = thickness factor for expanded tube ends
For the operating conditions:

1. 1

P = 11.72 MPa (1700 psi) D = 12.7 mm (0.5 in.)

$$S = 47.6 \text{ MPa} (6900 \text{ psi}) (800 \text{H at } 1260^{\circ} \text{F})$$

e = 0

The calculated minimum wall thickness is

t = 1.448 mm (0.057 in.)

A thermal and stress analysis was performed using a heat flux of .757 MW/m² (239,862 BTU/hr ft²), film coefficient .011 MW/m² (1907 BTU/ hr ft² $^{\circ}$ F), and fluid temperature 425 $^{\circ}$ C (798 $^{\circ}$ F) and compared to the previous analysis with the same loading with the tube thickness of 2.54 mm (0.1 in.).

The tube crown temperature (and the front to back temperature gradient) was reduced by $41.7^{\circ}C$ ($107^{\circ}F$). The axial stress was reduced 30%. This reduced the effective stress from 365 MPa (53,000 psi) to 293 MPa (42,500 psi). Using the same method as outlined in Section 4.3, a fatigue life of 4500 cycles was calculated. This jump in life from 1100 to 4500 was due not only to the reduction in stress but also the reduction in tube crown temperature.

3.6 Ferritic Material Tube Analysis

Ferritic steels have a much better heat conductivity than stainless steels and consequently better thermal performance. However, at high temperatures ferritic steels have poor creep resistance. An analysis was done using the material properties for an advanced ferritic and compared to the Incoloy 800H analysis. Table 3.6 compares the material properties for the ferritic and Incoloy 800H,

Using ferritic material properties, a thermal and stress analysis was performed using a heat flux .757 MW/m^2 (239,862 BTU/hr ft²), film

Comparison of Ferritic and Incoloy

	<u>SC/510 (950)</u>	800H/510 (950)
Thermal Expansion		
c ⁻¹	1.404×10^{-5}	1.8×10^{-5}
(F ⁻¹)	(7.8×10^{-6})	(10×10^{-6})
Thermal Conductivity		
W/m·K	29.41	19.55
(BTU/hr ft F)	(17)	(11.3)
Specific Heat		
J/kg·K	669.9	669.9
(BTU/1bm ^O F)	(.16)	(.16)
Modules of Elasticity		~
MPa	$.16 \times 10^{6}$.16 x 10 ⁶
(psi)	(23.3×10^6)	(23.8 x 10 ⁶)
Yield Strength		
MPa	176.5	111
(psi)	(25,600)	(16,100)

Tube Crown Temperatures

Loading No.	Thick mm	iness	<u>Material</u>	Crow C	n Temp. (°F)	ÅT (Tube) (°F)
4 - Pilot Plt	2.54	(.1)	800H	557	(1035)	103	(186)
2 - Comm Plt	2.54	(.1)	800H	607	(1125)	258	(464)
1 - Comm Plt	2.54	(.1)	800H	497	(926)	166	(299)
3 - Comm Plt	2.54	(.1)	800H	628	(1162)	196	(353)
3 - Comm Plt	1.488	(.057)	800H	568	(1055)	143	(257)
3 - Comm Plt	2.54	(.1)	9Cr	580	(1077)	147	(264)

Fatigue Life Comparison - 100 MW

	Thic	kness	Effective Stress Fatigue Lif					
Loading	mm	(in.)	Material	MPa	(psi)	Cycles		
2	2.54	(.1)	800H	472	(68,500)	200		
1	2.54	(.1)	800H	318	(46,200)	5,000		
3	2.54	(,1)	800H	365	(53,000)	1,100		
3	1.448	(.057)	800H	293	(42,500)	4,500/		
3	2.54	(.1)	90r	273	(39,600)	3,500		
3	1.448	(.057)	9Cr	248	(36,000)	10,000		

coefficient .011 MW/m² $^{\circ}$ C (1907 BTU/hr ft² $^{\circ}$ F) and fluid temperature 426 $^{\circ}$ C (798 $^{\circ}$ F) and compared to the same loading using 800H material properties. The tube crown temperature was reduced by 29 $^{\circ}$ C (85 $^{\circ}$ F) The axial stress was reduced from -363 MPa (-52,700 psi) to -257 MPa (-37,300 psi). This reduced the effective stress 25%. A fatigue life of 3500 cycles was calculated. This fatigue calculation does not include effects of creep. Because of low creep resistance of ferritics at 581 $^{\circ}$ C (1077 $^{\circ}$ F) (tube crown temperature), the calculated fatigue life would be reduced.

SECTION 4

PROPOSED REDESIGN WITH RIFLED TUBING

----1

4.1 Introduction

The design review of the DOE/MDAC once-through central receiver for the 100 MWe commercial plant indicated that the anticipated service life for the receiver is severely restricted, due to the thermal stresses created mainly by the high heat fluxes incident on the north side receiver panels, coupled with a low film boiling coefficient. Because of the cyclic operating conditions expected, these high stresses have a large impact on the fatigue life. These thermal stresses are most severe at two locations on the panels, in the CHF-film boiling zone, and in the superheater zone of the once-through panels. Possible solutions to these problems areas are to: 1) lower the heat flux and thereby increase the receiver surface, and 2) rearrange the heat surface to take advantage of turbulators to enhance the evaporative heat absorption process. In the first solution thermal efficiency would be sacrificed, and the cost and weight would increase. In the second case, a preliminary analysis indicates that a rearrangement of circuits is possible, which will maintain the same heat flux and receiver size without significant loss of overall efficiency. This redesign of absorption circuitry assumes the efficient performance of rifled tubing in the evaporator region.

4.2 Description of Proposed Redesign

The basic arrangement of the external central receiver remains intact, as seen in Figure 4.1, with 24 separate panels. Assuming symmetry about the N-S axis, the 12 panels are divided as follows: 1-6 are evaporator panels, each producing a slight superheat temperature. Each is individually controlled with subcooled water entering at the bottom. This section of the receiver is identical with the DOE/MDAC design except that the panel tubes are replaced with rifled tubes of approximately 12.7 mm

4.2

KEY PLAN

EXTERNAL CENTRAL RECEIVER

Symmetrical about N-S Axis





(1/2") O.D. Performance of the rifled tubing is assumed to eliminate DNB and film boiling. If the rifled tubing performance is proven to be as efficient as is indicated, nucleate boiling may exist all the way to 100% quality. This would allow the high heat flux to be retained without the associated problems with high thermal stress.

Panels 1-6 discharge 343°C (650°F) steam into a collecting header, where it is routed to the superheater portion of the receiver, represented by panel Nos. 7-10. These panels have a lower average heat flux than Nos. 1-6, and are therefore more adapted for superheating. Two possible arrangements were analyzed. One was a series arrangement with 38 mm (1.5") 0.D. tubes and the other was a series/parallel arrangement using 25.4 mm (1") 0.D. tubes. Both produce approximately $510^{\circ}C$ (950°F) rated steam. See Figure 4.2 and Figure 4.3. Preliminary calculations indicate that the metal temperatures are not as high as in the present design, and the temperature differences are lower, resulting in apparent lower stress values. Again, the crux of this design is the potential of the rifled tubing for elimination of the CHF/film boiling phenomena.

The thermal analysis program was utilized in the prediction of the performance of each panel involved in the redesign. Table 4.1 shows the physical dimensions of the various panels, tubes, and selected values of conductivity. In anticipation of lowered metal temperatures, values of conductivity selected correspond to T-11 and C-steel boiler tubing. This assumption of carbon steel may be too optimistic. Conductivity decreases as temperatures rise, reflecting the conductivity of Incoloy 800H for the high temperature superheater.



FIGURE 4.2

Parallel Evaporator - Series/Parallel Superheater





TABLE 4.1

Alt. Design Using Rifled Tubing Evaporator Ext. Central Receiver - Physical Dimensions

Series S.H.

Panel	Hea Len	ited igth	Wi F	dth t.	No.	Tu	be OD	Tub	e ID	Tu	be Condition
No.	m	(ft)	m	(ft)	Tubes	mm	(in)	mm	(in)	w/m•k	(BTU/hr ft ² F)
1-6	25.6	(84)	2.158	(7.083)	170	12.7	(0.5)	10.67*	(0.42)	519.2	(300)
7	25.6	(84)	2.17	(7.125)	57	38.1	(1.50)	32.36	(1.27)	346	(200)
8	25.6	(84)	2.17	(7.125)	57	38.1	(1.50)	32.26	(1.27)	346	(200)
9	25.6	(84)	2.17	(7.125)	57	38.1	(1.50)	32.26	(1.27)	259.6	(150)
10	25.6	(84)	2.17	(7.125)	. 57	38.1	(1.50)	32.26	(1.27)	259.6	(150)
					Series/	Paralle	<u>1 S.H.</u>				
7	25.6	(84)	2.16	(7.083)	85	2.54	(1.0)	21.6	(0.85)	346	(200)
8	25.6	(84)	21.6	(7.083)	85	2.54	(1.0)	21.6	(0.85)	346	(200)
9	25.6	(84)	2.16	(7.083)	85	2.54	(1.0)	21.6	(0.85)	259.6	(150)
10	25.6	(84)	2.16	(7.083)	85	2.54	(1.0)	21.6	(0.85)	259.6	(150)

*Nominal ID - variable with rifled tube geometry.

Results of the performance analysis appear in Table 4.2. The incident heat flux profiles, projected heated surface area, and loss model were not changed from the MDAC/Rocketdyne commercial receiver. The panel efficiencies remain high, with some reduction in the superheater panels. Overall receiver efficiency, however, was 93.3%. This compares favorably with the original receiver. The series superheater pressure drop is probably too high. The serials/parallel superheater has a better ΔP , and the overall efficiency is not changed. In the series/parallel arrangement streams of different temperatures must be mixed. This should not be a problem, since the temperature difference is not great. This is the preferred arrangement.

Table 4.3 lists maximum metal temperatures, and temperature differences from the tube crown to the rear. These are representative of relative thermal stress levels, and they indicate a 70% reduction of stress over the original design. Maximum tube temperatures are somewhat lower in the superheater, and very much lower in the evaporator section. This is because of the assumption that rifled tubing will be effective in eliminating CHF, thus allowing nucleate boiling to exist to 100% quality.

Figure 4.4 shows typical dimensions of a rifled tube.

4.3 Conclusion

Based on this preliminary analysis, it is concluded that the use of rifled tubing in the evaporator has a potential for significantly increasing the fatigue life of the commercial plant receiver, by eliminating CHF and film boiling, thus reducing the panel thermal stress.

TABLE 4.2

Alternate Design Using Rifled Tubing Evaporator Predicted Performance Using MDAC Solar Flux Taken at Sum. Solsitice Noon (Series Superheater)

Panel No.	P _{in} MPa (psia)	P _{out} MPa (psia)	T _{in} C (°F)	T _{out} C (°F)	Max. Solar Flux MW/m ² (BTU/hr ft ²)	Mass Flow Kg/s (lb/hr)	G x 10 ⁻⁶ Kg/m ² s ₂ (1b/hr ft ²)	Heat Abs. MW (BTU/hr x 10 ⁻⁶)	Panel Eff. %
1	12.7 (1850)	11 (1599)	284 (543)	342 (648)	.85 (269,500)	20.6 (163,600)	1333 (0.983)	32.9 (112.23)	95.2*
2	12.7 (1850)	11.2 (1626)	284 (543)	343 (649)	.83 (254,550)	19.5 (154,700)	1261 (0.93)	31 (105.81)	95
3	12.7 (1850)	11.4 (1651)	284 (543)	344 (651)	.76 (239,600)	18.4 (145,900)	1189 (0.877)	29.2 (99.65)	95
4	12.7 (1850)	11.5 (1668)	284 (543)	344 (651)	.72 (228,530)	17.6 (139,700)	1138 (0.839)	27.8 (94.99)	95
5	12.7 (1850)	11.6 (1681)	284 (543)	342 (648)	.69 (217,460)	16.9 (134,100)	1093 (0.806)	26.5 (90.52)	95.2
6	12.7 (1850)	11.8 (1710)	284 (543)	344 (651)	.60 (191,150)	14.9 (118,000)	962 (0.709)	23.3 (79.53)	95.2
7	11 (1600)	10.6 (1536)	343 (650)	383 (722)	.52 (164,840)	108 (856,000)	2315 (1.707)	19.3 (65.91)	91.4
8	10.6 (1536)	10 (1458)	383 (722)	428 (802)	.45 (141,220)	108 (856,000)	2315 (1.707)	16.3 (55.64)	90.1
9	10 (1458)	9.4 (1365)	428 (802)	470 (877)	.37 (117,600)	108 (856,000)	2315 (1.707)	13 (44.51)	86.6
10	9.4 (1365)	8.7 (1257)	469 (877)	502 (936)	.30 (93,980)	108 (856,000)	2315 (1.707)	10 (34.24)	83.3

*Panel efficiency includes radiation and still air conv. losses only.

Total Abs. in Evap. - 582.73 x 10⁶ BTU/hr.

Total Abs. in SH = 200.3 x 10^6 BTU/hr. Overal unit efficiency = 93.3%.

TABLE 4.2 (Cont'd.)

Alt. Design Using Rifled Tubing With Series/Parallel Superheater

Performance of Evap. Panels No. 1-6 Same

Panel No.	P _{in} MPa (psia)	P _{out} MPa (psia)	Tin C (°F)	T _{out} C (°F)	Max. Solar Flux MW/m ² (BTU/hr ft ²)	Mass Flow Kg/s (1b/hr)	$G \ge 10^{-6}$ Kg/m ² s ₂ (1b/hr ft ²)	Heat Abs. MW (BTU/hr x 10 ⁻⁶)	Panel Eff. %
7	11 (1600)	10.6 (1532)	342 (648)	440 (824)*	.52 (164,840)	53.9 (428,000)	1741 (12.84)	19.2 (65.48)	90.8
8	11 (1600)	10.6 (1534)	342 (648)	422 (792)*	.45 (141,200)	53.9 (428,000)	1741 (1.284)	16.3 (55.64)	90.1
9	10.6 (1533)	9.9 (1443)	431 (808)	518 (964)	.37 (117,600)	53.9 (428,000)	1741 (1.284)	12.8 (43.66)	84.9
10	10.6 (1533)	9.96 (1445)	431 (808)	498 (928)	.296 (93,980)	53.9 (428,000)	1741 (1.284)	10 (34.24)	83.3

*Note: Outlet streams are mixed to produce an average temperature entering the next panel-pair. Overall unit efficiency = 93.2%.

TABLE 4.3

Panel No.	Max, Tube	e Temp. (°F)	۵ د	^{T*} (^o F)	<u>MW/m</u> 2 ^H	eat Flux (BTU/hr ft ²	<u>2)</u>
1	391 ((736)	69.4	(125)	.62	(197,518)	
2	390 ((734)	67.2	(121)	. 59	(186,595)	
3	389 ((732)	65	(117)	.55	(175,890)	
4	387 ((728)	62.2	(112)	.526	(167,023)	
5	382 ((720)	57.8	(104)	.48	(152,571)	
6	380 ((717)	54.8	(98)	.42	(133,736)	
7	484 ((904)	108	(194)	.47	(149,761)	
8	517 ((963)	96	(173)	.40	(126,176)	s SH
9	554 (1	.030)	90.5	(163)	. 32	(102,091)	erie
10	569 (1	.056)	71	(128)	.25	(79,125)	~~~
7	528 (982)	104	(187)	.47	(148,167)	
8	497 (927)	88	(158)	.40	(126,962)	r SH
9	587 (1	.088)	81	(145)	.32	(100,767)	r/Pa
10	553 (1	.027)	64	(115)	.25	(79,799)	Se

Alt. Design Using Rifled Tubing Evaporators Predicted Maximum Tube Temperatures

* T - Tube Crown - Bulk Liquid)

Note: Based on the assumption that rifled tubing allows nucleate boiling to exist to 100% quality.

Max. metal temps. occur in SH region.


CROSS SECTION OF 6 RIB RIFLED TUBE

NOTATIONS OUTSIDE DIAMETER A В MAJOR DIAMETER С MINOR DIAMETER DEF RIB WALL THICNESS ROOT WALL THICNESS RIB WIDTH G ROOT WIDTH RIFLE PITCH (180° TWIST) Н

FIGURE 4.4

SECTION 5

1

APPENDICES

5. 1

APPENDIX A

Thompson-MacBeth Correlation

I. The low velocity correlation for $G < 0.3 \times 10^6$ lbm/hr ft²: $\frac{q_{crit}}{10^6} = .00633 H_{fg} D^{-.1} \left(\frac{G}{10^6}\right)^{.51} (1-x)$

where: q_{crit} = critical heat flux (BTU/hr ft²) (inside surface)
H_{fg} = heat of evaporation (BTU/lbm)
D = tube I.D. (in.)
G = mass flux (lbm/hr ft²)
x = critical quality

II. The high velocity correlation for $G > 0.3 \times 10^6$ lbm/hr ft²:

$$\frac{q_{crit}}{10^6} = \frac{A^1 - \frac{1}{4D} (G \times 10^{-6}) \times H_{fg}}{C^1}$$

where:

 A^1 and C^1 contain Y constants based on pressure ranges. Constants (Y) were determined from Ref. 15 at 1550 psia.

...
$$A^{1} = 36D^{-509}(G \ge 10^{-6})^{-109}[1-.19D+0.24(G \ge 10^{-6})+.463D(G \ge 10^{-6})]$$

and $C^{1} = 41.7D^{-053}(G \ge 10^{-6})^{-0109}[1+.231D+.0767(G \ge 10^{-6})+.117D(G \ge 10^{-6})]$

NOTE: In the thermal analysis program, a linear interpolation is made between 1550 psia and 2000 psia for the above (Y) constants.

APPENDIX B

Groeneveld Film Boiling Correlation

The film boiling coefficient, (h), is given as a function of quality, (x).

$$h = a \frac{Kg}{D} \left[\text{Reg} \left(x + \frac{\rho g}{\rho 1} (1 - x) \right) \right]^{b} \Pr_{w}^{C} Y^{d} \phi^{e}$$
where:

$$h = \text{film coefficient, BTU/hr ft}^{2 \circ} F$$

$$a = 1.85 \times 10^{-4}, \text{ const.}$$

$$Kg = \text{sat. steam conductivity, BTU/hr ft}^{2 \circ} F$$

$$D = \text{tube diameter, feet.}$$

$$\text{Reg = Reynolds Number at sat. steam condition.}$$

$$x = \text{local quality}$$

$$\rho g = \text{density sat. steam}$$

$$\rho l = \text{density sat. liquid}$$

$$b = 1.0, \text{ const.}$$

$$\Pr_{w} = \text{Prandtl Number evaluated at wall temperature}$$

$$c = 1.57, \text{ const.}$$

$$Y = \text{see below}$$

$$d = -1.12, \text{ const.}$$

$$\phi = \text{heat flux, BTU/hr ft}^{2}, \text{ inside surface}$$

$$e = .131, \text{ const.}$$

$$Y = 1 - .1 \left(\frac{\rho 1}{\rho g - 1}\right)^{-4} (1 - x)^{-4}$$

and ${\rm X}_{\rm h_{min}}$ is given by

$$(X_{h_{min}} - X_{DNB}) = .045 + \frac{.048}{2.3 - .01 p}$$

where: p, pressure, is in bar.

2

APPENDIX C

Description of Thermal Analysis Program

An energy and mass balance is made for each increment of tube length. The heat conduction equations are written for the axisymmetric flow at the tube crown. The incident heat flux is assumed normal to the tube at this point. A correction factor is incorporated into the one-dimensional, axisymmetrical case to correct the crown temperature for the effect of 2-D heat flux and heat flow. The pressure drop is calculated assuming the homogeneous model.

The absorbed crown heat flux is given by:

$$\dot{q}_{abs} = \dot{q}_{ine} \cdot \alpha_{5} - \sigma \in (T_{51} + - T_{0} +) - h_{ext} (T_{51} - T_{0})$$
where:

$$\dot{q}_{abs} = \dot{q}_{ine} \cdot \alpha_{5} - \sigma \in (T_{51} + - T_{0} +) - h_{ext} (T_{51} - T_{0})$$
where:

$$\dot{q}_{abs} = absorbed heat flux, BTU/hr ft^{2}$$

$$\dot{q}_{int} = incident solar flux, BTU/hr ft^{2}$$

$$\alpha_{5} = solar absorptance$$

$$\sigma = Stefan-Baltzmann Constant$$

$$\epsilon = infrared emittance$$

$$h_{ext} = external convection coefficient$$

$$T_{51} = absolute surface temperature ^{0}R$$

$$T_{50} = absolute ambient temperature ^{0}R$$
Also, by:

$$\dot{q}_{abs} = (T_{51} - T_{f}) / [D_{e} \Psi / h_{i} D_{i} + (D_{0} / 2K) [h (\frac{D_{e}}{D_{i}} \Psi)]$$
where:

$$\dot{q}_{abs} = crown absorbed heat flux, BTU/hr ft^{2}$$

$$T_{51} = absolute surface temperature ^{0}R$$

$$T_{5f} = absolute fluid bulk temperature ^{0}R$$

$$D_{e} = tube outside diameter, inches$$

$$h_{i} = inside film convection coefficient, BTU/hr ft^{2} ^{0}F$$

$$K = tube conductivity BTU in/hr ft^{2} ^{0}F$$

$$\Psi = correction factor for 2-D heat flow.$$

· . .

These 2 equations can be combined into a quartic equation in T_{SI} . This is then solved by formula from a standard math reference:

$$-\left\{ \hat{q}_{ine} \alpha_{s} / \sigma \epsilon + h_{ext} T_{o} / \sigma \epsilon + T_{o}^{4} + T_{s} / \sigma \epsilon [D_{o} \ell / h_{i} D_{i} + (D_{o} / 2K) L_{h} (D_{o} \ell / D_{i})] \right\} = 0$$

The 2-D correlation, Ψ , is calculated by an equation of the form:

$$\Psi$$
 = B1 + B2 (BIOT)^{B3}

where:

: B1, B2, B3 are constants that depend on the diameter ratio. BIOT = $\frac{\text{HT}}{\text{K}}$

where: H = inside film conductance T = tube thickness K = tube conductivity

The incremental increase in fluid enthalpy is then calculated from the absorbed heat flux based on the following energy balance:

$$\dot{q}_{net in} = \frac{M * \Delta h_f}{\Delta A}$$

M = mass flow rate

where:

 Δh_f = change in enthalpy

 ΔA = incremental area of the element

The program then calculates the fluid quality. If this quality lies between 0 and 1, the critical heat flux is determined and compared to the absorbed heat flux as an indicator of the point at which CHF occurs.

Finally, the pressure drop (ΔP) through the increment of tube length is determined as a function of frictional, momentum, and gravitational components. The detailes of this calculation are as follows:

$$f = 0.46/Re^{2}$$

where: f = friction factor

Re = Reynold's Number

$$\Delta P_{\text{friction}} = 4 \times 10^{-10} \text{ f} \frac{\Delta 1}{\text{Di}} \text{ v } \text{ G}^2$$

where:

∆1 = incremental tube length
Di = inside tube diameter
v = average specific volume in increment.
G = mass flow rate per unit area per tube.

$$\Delta P_{\text{momentum}} = 1.667 \times 10^{-11} (v_2 - v_1) G^2$$

where:

 v_2 = element outlet sp. vol. v_1 = element inlet sp. vol.

$$\Delta P_{\text{gravity}} = \frac{\Delta 1}{144 \text{ v}}$$

 $\Delta P_{\text{total}} = \Delta P_{\text{friction}} + \Delta P_{\text{momentum}} + \Delta P_{\text{gravity}}$

Pressure at the increment's exit is then simply:

 $P = P_{in} - \Delta P_{total}$

The inside film coefficient for use in the above equations is determined by the following correlations, depending on the fluid state existing in a particular element.

1. Single phase, liquid and vapor: (Dittus-Boelter)

$$\frac{hD}{K}$$
 = .023 Re^{•8} Pr^{•4}

2. Two-phase, nucleate boiling region:

 $h = 5000 \text{ BTU/hr ft}^2 ^{\circ}\text{F}$

4. Transitional boiling region (X to Xhmin):

A linear interpolation is made between $h_i = 5000$ and $h_i = h_{min}$, from the Groeneveld Correlation.

The program recalculates the above variables for each increment of the tube length. If desired, the program will test the final increment output for proper steam temperature. The program will then automatically adjust the flow rate and recalculate the above parameters until the desired steam temperature is achieved.

	BO (US)	LER PA ING GRO	NEL NO ENEVEL	• 1 D CORREL	ATION			COMMER	qual Pa	ANT	
	PRFSS	S TEMP F	ENTHAL BTU/L	PY L1 B FT	L2 NC FT*	• TUB	ES TUBE OD IN	•TUBE ID IN	TUBE LG' FT	TH INCR	EMENT FT
	1850.	550.	560.	27.7	67.2	170	0.50	0.27	84.	0.5	0
	CONDU B-II	JCTIVIT V/H-SF-	Y EXT. F B/H	COEFF EM -SF-F	ISSIVI	TY A	BSORPTANCE	●AMB • TEMP F	MAX.SOL BTU/H	AR FLU	X
	1	150.		3.	0.89		0.95	100.	2695	5 00 .,	· · · *
	TUBE LGTH	PRESS	BULK	ENTHALPY	FILM	DE COEF	OUTSIDE HEAT FLUX	TUBE CRO TEMP	WN BULK Qual	CRIT QUAL	•••••
	F 1	PSIA	F	BIUVER	D/H-S	5r - r	B/H-Sr				
	0.50	1849.6	557.	560.	4	1992 .	2470.	559.	19	0.0	
	1.00	1849.3	557.	560.	4	992 .	7051.	564 .	19	0.0	
	1.50	1848.9	557.	561.	4	993 .	11474.	569.	19	0.0	
	2.00	1848.5	558.	561.	4	994.	16056.	5/5.	19	0.0	
	2.50	1848.2	558.	562 .	4	995.	21142.	581.	19	0.0	
	3.70	1947.8	559.	562.	4	1996 .	25624 •	586.	19	0.0	
	3.50	1847.4	559.	563.	4	998.	29924 .	591.	18	0.0	
	4.06	1847.1	560 .	564.	5	001.	54388 .	597.	18	0.0	
	4.50	1846.7	561.	566 .	5	003.	39268.	603.	18	0.0	
	5.00	1846.3	562 .	567.	5	006.	43584.	609.	18	0.0	
	5.50	1846.0	563 .	568.	5	009.	483/5.	615.	17	0.0	<u>.</u>
	6.00	1845.6	564 .	570.		013.	53010.	621.	1/	0.0	
	6.50	1845.2	566 .	572 .	5	01/.	5/558.	627.	1/	0.0	
	7.00	1844.9	567.	573.	5	022.	61891.	633 .	-•16	0.0	
	7.50	1844.5	569.	575.	5	01210	E6641 .	640.	-010	0.0	
	8.00	1844.1	570 .	5/8.	5	032	71116.	646.	15	0.0	- 36
	8.50	1843.8	572 .	589.	5	00010	10349.	653.	- 10	0.0	
	9.00	1843.4	574.	582 -	5	044.	19846.	659.	14	0.0	
	9.50	1843.0	5/6.	585.	5	.1001.	84642.	666.	- • 1 4	0.0	
1	0.00	1842.6	578.	587.	5	058.	89289.	613.	13	0.0	
1	0.53	1842.3	580 .	590.	5	066.	33573.	680.	13	0.0	
1	1.00	1841.9	1 583.	593.	5	0115.	98343.	687.	-012	0.0	11. 14
1	1.50	1841.5	585 .	596 .	5	.085.	102625.	694.	-012	0.0	1.
1	2.99	1841.2	.180	597+	5	1996.	107407.	7020	11	0.0	
1	2.50	1840.8	590.	603 .		10/0	111983.	709.	10	0.0	
1	3.00	184 3.4	593 .	616.		120.	1161/8.	710.0	10	0.0	
1	3.50	1840.1	576 .	610.	5	154 .	1207550	7290		0.0	
1	4. 50	1070 7	578 .	613.	C	1/0	1200713	770		0.0	22
	4.00	1070 0	601.	601	÷	105	134486	707.		0.0	- (w. 4
	5.00	103007	604 .	621.	2	007	1 70 00 1	754		0.0	
-	5.50	1030+0	6.00 *	620 .	2 E	203.	147769	769		0.0	
-	6 50	103002	610	630.		5244 .	147909	769.		0.0	
	7 00	1037.0	612.	634		0244	150366	777.		0.0	
1	7 6 5	1027 1	610	607.		000	154945	784		0.0	· 9
1	0 00	1070 7	610.	6434	E	2700	1009400	707.		0.0	
-	18.60	1876 7	621 .	252	5	362	165867	810-	0-00	0-0	0.57
	0.00	1.34.0	6 24	452		5.00-	170497-	810-	0.01	0 - 0	
1	5,57	1235.0	624 .	467	5	0000	174968	815-	0-02	0.0	
-	0.00	1634-6	6.94	669	F	0000	179418	820-	0.03	0.0	
4		1027 7	604	607.	5	0000	183870	824	0-04	0.0	
-	0.00	1930 0	604 .	60.4.	5	5000	198620-	829	0.05	0.0	
-	1.5	1921 7	607	600.	5	0003.	1930200	9270	0.07	0	
-	1.1.10	1930 6	623 .	600 *		00020	197731	939	0.00	0.0	1
4	2.50	18:20 0	6020	691.	6	0000	202043	844	0.00	0.0	
-	2.00	1997 -	607	707	0	0000	2020000	0 / 0	0.00	U & U	
4	3.00	105100	6230	103 .	. 0	uuue	2101000	0470	0.10	0.00	

5-7

	23.50	1826.2	623.	710.	5000.	211259.	854.	0.12	0.0
	24.00	1824.5	623 .	716.	5000.	215904.	859.	0.13	0.00
	24.50	1822.7	623 .	723.	5000.	220178.	863.	0.14	0.0
	25.00	1820.7	623 .	729.	5000.	224781.	868.	0.16	0.0
	25.50	1818.5	623.	736 .	5000.	229301.	873.	0.17	0.17
	26.00	1817.8	622.	743.	4983.	234024.	878.	0.18	0.18
	26.50	1817.0	622 .	750.	4352.	238367.	890.	0.20	0.20
	27.04	1816.2	622 .	757.	3710.	242581 .	906.	0.21	0.21
	27.50	1815.4	622 .	765.	3057.	246862.	926 .	0.23	0.23
	28.00	1814.6	622 .	772.	2392.	248189.	952.	0.24	0.24
	28.50	1813.7	622 .	780.	1724.	247327.	992 .	0.26	0.26
	29.00	1812.8	622 .	787.	1059.	245224.	1076.	0.27	0.27
	29.50	1811.9	622 .	794 .	944 .	244578.	1100.	0.29	0.29
	30.00	1811.0	622 .	801.	975 .	244800.	1093.	0.30	0.30
	30.50	1810.0	622 .	809.	1006.	244969.	1086.	0.32	0.32
	31.00	1809.0	622 .	816.	1037.	245107.	1080.	0.33	0.33
	31.50	1808.3	622 .	823.	1067.	245252.	1074.	0.35	0.35
	32.00	1807.0	622 .	831.	1098.	245358 .	1068.	0.36	0.36
	32.50	1805-9	622 .	838 .	1128.	245575.	1063.	0.38	0.38
	33.00	1804.8	621 .	845.	1158.	245632	1058.	0.39	0.39
	33.50	1803.7	621 .	852 .	1187.	245813.	1053.	0.40	0.40
	34.00	1802.6	621 .	860.	1217.	245902.	1048.	0.42	0.42
	34.50	1801.4	621 .	867.	1246.	246017.	1044.	0.43	0.43
	35.00	1800.2	621 .	874.	1274.	246125	1640.	0.45	0.45
	35.50	1799.0	621 .	882 .	1303.	246214 .	1036.	0.46	0.46
	36.00	1797.8	621 .	889.	1331.	246364 .	1032.	0.48	0.48
	36.50	1796.5	621 .	896.	1359.	246399.	1028.	0.49	0.49
	37.00	1795.2	621 .	964.	1386.	246481.	1025.	0.51	0.51
	37.50	1793. 9	621 .	911.	1413.	246659.	1021.	0.52	0.52
	38.00	1792.6	621 .	918.	1440.	246627.	1018.	0.54	0.54
	38.50	1791.2	620 .	926 .	1466.	246709.	1015.	0.55	0.55
	39.00	1789.8	620 .	933.	1492.	246775.	1012.	0.57	0.57
	39.50	1788.3	620 .	940.	1518.	246950.	1009.	0.58	0.58
	40.00	1786.9	620 .	948.	1544.	246914 .	1007.	0.59	0.59
	40.50	1785.4	620 .	955 .	1569.	246919.	1004.	0.61	0.61
	41.00	1783.9	520 ·	962 .	1593.	247108.	1002.	0.62	0.62
	41.50	1782.3	620 .	970.	1517.	247130.	999.	0.64	0.64
	42.00	1780.8	620 .	977.	1641.	247268.	997.	0.65	0.65
	42.50	1779.2	619 .	984 .	1665.	247229.	995.	0.67	0.67
	43.00	1777.6	619.	992 .	1688.	247277.	993.	0-68	0.68
	43.50	1775.9	619.	999.	1710.	247314.	991.	0.70	0.70
	44.00	1774.2	619.	1007.	1732.	247454.	989.	0.71	0.71
	44.50	1772.5	619.	1014.	1754.	247460.	987.	0.72	0.72
	45.00	1770.8	619.	1021.	1775.	247382.	985.	0.74	0 = 74
	45.50	1769.0	619.	1029.	1795.	247517.	983.	0.75	0.75
8	46.00	1767.2	619.	1036 .	1815.	247497.	982 .	0.77	0.77
	46.50	1765.4	618 .	1043.	1835 .	247572 .	980.	0.78	0.78
	47.00	1763.6	618.	1051.	1853.	247522.	978.	0.80	0.80
	47.50	1761.7	618.	1058.	1871.	247728.	977.	0.81	0.81
	48.00	1759.8	618 .	1065.	1889.	247678.	976 .	0.83	0.83
	48.50	1757.8	618 .	1073.	1906.	247720.	975.	0.84	0.84
	49.00	1755.9	618 .	1080.	1921.	247736.	973.	0.85	0.85
	49.50	1753.9	618 .	1088.	1936.	247757.	972.	0.87	0.87
	50.00	1751.5	617.	1095.	1950.	247842 .	971.	0.88	0.88
	50.50	1749.0	617.	1102.	1963.	247905.	970.	0.90	0.90
	51.00	1747.7	617 .	1110.	1975.	247834.	969.	0.91	0.91
	51.50	1745.6	617.	1117.	1984.	247926.	969.	0.92	0.92
	52.00	1743.5	617 .	1124.	1993.	247927.	968.	0.94	0.94
	52.50	1741.3	617.	1132.	1998.	247839.	967.	0.95	0.95
	53.00	1739.1	616 .	1139.	2091.	247819.	967.	0.97	0.97
	53.50	1736.9	616.	1147.	1999.	247910.	967.	0.98	0.98
	54.00	1734.6	616.	1154.	1996.	247819.	967.	1.00	1.00
					5-8				

MASS	S FLOW =	0.11918	C6L8/HR	G = 0 • 1	762E 07L 8/H	R-SQ FT	HMIN :	= 9	14.388,
84.00	1550.7	965 .	1467.	1892.	3341.	767.	1.00	1.00	× .
83.50	1554.5	965 .	1467.	1804.	10428.	980.	1.55	1.00	2
83.00	1558.3	964 .	1467.	1804.	17497.	989.	1.55	1.00	
82.50	1562.1	964 .	1466.	1805.	24655.	999.	1.55	1.00	
82.00	1565.9	963.	1465.	1806.	31671.	1008.	1.55	1.00	
81.50	1569.7	962.	1464.	1806.	38751.	1017.	1.55	1.00	
81.00	1573.4	960 .	1463.	1807.	45791 .	1025.	1.55	1.00	ę
80.50	1577.2	958 .	1462.	1808.	52833.	1033.	1.54	1.00	10 A 10 A
80.00	1581.0	955 .	1460.	1809.	59987.	1041.	1.54	1.00	
79.50	1584.7	953 .	1459.	1809.	67146.	1048.	1.54	1.00	1
79.00	1588.5	950 .	1457.	1810.	74273.	1055.	1.54	1.60	t
78.50	1592.2	946 .	1454.	1811.	81306.	1062.	1.54	1.00	terr ann te
78.00	1595.9	943 .	1452.	1812.	88386.	1068.	1.53	1.00	ż
77.50	1599.6	939	1449.	1813.	95562 .	1074.	1.53	1.00	2
77.00	1603-3	934	1446 .	1814	102794-	1080-	1.52	1.00	
76.50	1606-9	930 -	1443.	1816.	109861	1086.	1.52	1.00	
76.00	1610-5	925 -	1440.	1817-	116945	1091-	1.51	1.00	
75.50	1614-2	919.	1437.	1819-	124139-	1095-	1.51	1.00	- + +-
75 00	162103	910	1433.	1821-	1 31 24 9	1100-	1.50	1.00	
74.00	1624.9	901 •	1420.	1824-	1 38 496 -	1100.	1.50	1.00	
13.50	1628.4	895.	1421+	1830.	152/54 .	11110	1.40	1.00	····
73.00	1631.9	888 •	1416.	1033.	160046.	1114.	1 40	1.00	
72.50	1635.3	881.	1411.	1857.	16/140.	111/0	1.40	1.00	
72.00	1638.7	873 .	1406.	1842.	1/4424 •	1120.	1.47	1.00	12 4
71.50	1642.1	866 .	1401.	1848.	181561.	1122.	1.45	1.00	
71.00	1645.5	858 .	1396.	1854.	188892.	1124.	1.45	1.00	
70.50	1648.9	850 .	1390.	1861.	196058.	1126.	1.43	1.00	مد و مود د
70.00	1652.2	841 .	1384 .	1869.	203250.	1127.	1.42	1.00	
69.50	1655.4	833 .	1378.	1879.	210507.	1128.	1.41	1.00	
69.00	1658.6	824 .	1372.	1889.	217763.	1129.	1.40	1.00	
68.50	1661.8	815.	1366.	1901.	224996.	1130.	1.39	1.00	
68.00	1665.0	806 .	1359.	1915.	232271.	1130.	1.38	1.00	
67.50	1668.1	796 .	1352.	1930.	239621.	1130.	1.36	1.00	
67.00	1671.2	787.	1345.	1948.	244076.	1126.	1.35	1.00	
66.50	1674.2	778 •	1338.	1967.	244340.	1116.	1.34	1.00	
66.00	1677.2	768.	1330.	1988.	244535.	1105.	1.32	1.00	
65.50	1680-1	759.	1323.	2011.	244837.	1096.	1.31	1.00	,
65.00	1683.0	750.	1316.	2036.	245180.	1086.	1.30	1.00	
64.50	1685.9	742 .	1308.	2064.	245454.	1076.	1.28	1.00	
64.00	1688.6	733.	1301.	2094.	245607.	1066.	1.27	1.00	
63.50	1691.4	725.	1294.	2127.	245859.	1057.	1.26	1.00	
63.00	1694.1	717 .	1287.	2163.	246059.	1047.	1.24	1.00	
62.50	1696.7	709 .	1279.	2203.	246334 .	1038.	1.23	1.00	
62.00	1699.3	702.	1272.	2246.	246523.	1028.	1.22	1.00	• • •
61.50	17.1.9	694 .	1265.	2298.	246686 .	1019.	1.20	1.00	
61.00	1704-4	687 .	1257.	2369.	246942 .	1009.	1.19	1.00	
60.50	1706-9	681 -	1250.	2443.	247214.	1000.	1.18	1.00	
60-00	1709-3	674	1243.	2522.	247458.	990.	1.16	1.00	
59-50	1711.7	668 -	1235-	2607.	247662	981.	1.15	1.00	
59.00	1714-0	662 -	1228-	2695 -	247737-	972 .	1.14	1.00	N. N.
58.50	1716.9	656 -	1220.	2789.	248164	964	1.12	1.00	
5/+50	1719.4	650	1213.	2889-	248275	955	1.11	1.00	
57.50	1720 6	6400	1206.	2977-	248333	948.	1.09	1.00	
56+50	1729.0	633.	1192	3040	248602	940-	1.08	1.00	
56.00	1726.8	631.	1185.	31/7	2401140	920.	1.07	1.00	CONTRACTOR OF STREET
55.50	1728.9	626 .	11/6+	34930	240000 .	524	1.05	1.00	
55.00	1730.8	622 .	1169.	5541.	247100.	712.	1.04	1.00	
54.50	1732.8	618.	1161.	1959.	2411410	767.	1.02	1.00	
					0.67747	610	1 0.1	1 00	0/

. • :

COMMERCIAL PLANT PANEL No. 1

 $\gamma = \frac{(19,100)(1467 - 560)}{(269,500)(437.4)} = .916$

n = - 909

CUMULATI	VE PRESS	SURE DROP
FRICTION	MOMENTUN	GRAVITY
PSI	PSI	PSI
0.2083	0.0001	0.1577
0.4166	0.0004	0.3154
0.6250	0.0008	0.4730
0.8335	0.0014	0.6306
1.0420	0.0023	0.7881
1.2507	0.0033	0.9454
1.4595	0.0044	1.1026
1.6685	0.0058	1.2596
1.8776	0.0073	1.4164
2.0870	0.0090	1.5730
2.2967	0.0109	1.7294
2.5066	0.0130	1.8854
2.7167	0.0152	2.0412
2.9272	0.0176	2.1967
3-1381	0.0202	2.3518
3.3493	0-0230	2.5066
3-5609	0.0260	2.6609
3.7728	0.0291	2.8149
3.9852	0.0324	2.9685
4-1991	0-0359	3,1216
4-4114	0-0395	3.2742
4.6252	0-0434	3.4964
4.9395	0.0474	3.5791
5-0543	0.0516	3.7992
5-2696	0-0560	3.8798
5-4856	0.0605	4-0298
5.7020	0-0652	4.1792
5-9191	0-0701	4.3281
6.1369	0.0764	4.4762
6.3557	0.0843	4 . 6235
6-5756	0.0924	4.7699
6-7967	0-1007	4-9152
7.0191	0.1093	5.0596
7.2426	0-1182	5.2029
7.4674	0.1974	5.3453
7.6936	0-1372	5.4865
7-9213	0-1505	5-6265
8-1579	0-2637	5.7613
8-4037	0.4798	5-8910
8-6589	0.8016	6.0160
8-9237	1.2321	6-1364
9.1985	1.7743	6.2524
9.4834	2-4713	6-3644
9.7789	3.2661	6.4723
10.0850	4.1017	6-5765
10.4022	5-1213	6.6771
10.7307	6.2691	6.7740
11.0708	7.5453	6-8680
	ىر	100000

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	11-4228	8.9560	6.9586	
	11.7870	10.5036	7.0462	
	10 1636	12,1912	7.1309	
	1201030	1201712	7.1900	
	12.7035	12.0200	7 0005	
	13.2684	1204685	7 = 2950	
	13.8588	12.6110	7.3006	
	14.4752	12.7560	7 • 3523	
	15.1179	12.9019	7.4020	
	15.7870	13.0474	7.4497	
	16.4824	13.1916	7.4956	
	17.2040	13.3355	7.5398	
	17.9517	13.4797	7.5825	
	18 7256	13.6240	7.6237	
	19,5259	13.7684	7.6636	
	20 3525	13,9130	7.7022	
	121/2056	14.0577	7.7397	
	21 2008	14.2027	7.7760	
	200001	1402021	7 0110	
	22.9911	1403477	7 0454	
	23.9231	14.4929	1.0707	
	24.8830	14.6382	1.8181	
	25.8690	14.7857	7.9111	
	26.8818	14.9294	7.9426	
	27.9215	15.0751	7.9733	
	28.9881	15.2211	8.0033	
	30.0818	15.3671	8.0325	
	31.2027	15.5134	8.0610	
	32.3508	15.6598	8.0888	
	33.5262	15.8063	8.1160	
	34.7290	15.9530	8.1425	
	35.9592	16.0999	8.1685	
	37.2169	16.2469	8.1939	
	38.5023	16.3940	8.2187	
	39.8154	16.5413	8.2431	
	41.1562	16+6888	8.2669	
	42-5248	16.8364	8.2902	
	43-9214	16.9843	8-3131	
	45.3461	17.1322	8.3356	
	46.7988	17.2803	8.3576	
	48.2797	17.4285	8.3791	
	40.7050	17.5770	8.4003	
	4767007 E1 30(E	17 7050	0 4011	
	51.65265	17 0743	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	
	52+6725	11.0143	0 4(1)	
· · · · ·	54.4870	18.0232	8.4616	
	56.1102	18.1725	8 4813	
	57=7621	18.3215	8.5006	
	59.4429	18.4709	8.5196	
	61.1525	18.6205	8=5383	
	62.8912	18.7702	8 • 5567	
	64.6590	18.9201	8.5748	
	66.4559	19.0702	8.5926	
	68.2822	19.2205	8.6101	
	70.1379	19.3709	8.6274	
	72.0231	19.5216	8.6444	
	73.9380	19.6723	8.6611	
	75.8826	19.8233	8.6775	
	77.8569	19.9745	8.6937	
	79.8613	20.1258	8.7097	
	81.8956	20.2772	8 . 7254	
	83.9601	20.4289	8.7409	
	86.0549	20.5807	8.7562	
	87.6736	20.7972	8.7712	
	89.3259	21.0612	8.7860	
		-2-11		

01 0151	01 3084	8.8004	
9100101	2105204	0 0145	10000 A
9267420	21.0930	C+10+0	
94.5071	21.8538	8.8282	
96.3099	22.1201	8.8417	
98-1531	22.3812	8.8550	
100 0417	00 (47(0 0670	
100.0417	22.6476	0.00/7	
101.9773	22.9119	8.8806	
103.9604	23.1785	8.8931	
105-9923	23-4480	8.9053	
109.0742	23.7161	8-9172	
100:0742	2367101	0.0000	
110.2069	23.7859	8.9290	
112.3918	24.2567	8.9405	
114.6224	24.5284	8.9518	
116.8967	24 .8015	8.9629	(e)
110.0164	25 0762	0 0730	
11702104	25.0762	0.7130	
121.5792	25.3511	8 . 7845	
123.9885	25.6275	8 • 995 0	
126.4440	25.9039	9.0054	
128-9460	26.1817	9.0155	
121.4051	26 4599	9.0255	
131.49701	2004307	700200	
134.0916	26.7377	9.0353	
136.7361	27.0168	9.0449	
139.4290	27.2966	9.0544	
142.1706	27-5771	9-0637	
14201/00	2765771	0.0007	AT- 22.10P
144.7011	27.6530	700129	-DI- 254 F
147.7998	28.1210	9.0819	• /
150.6857	28.3809	9.0908	
153.6179	28.6329	9.0996	
156-5955	28-8766	9-1082	
159.6174	20.1125	9.1169	
107+01/4	27.1125	7.1100	
162.6827	29.3394	9.1252	
165.7901	29.5591	9.1335	
168.9388	29.7702	9.1417	
172.1275	29.9734	9.1498	
175.3550	30-1685	9.1579	
17565552	30.1863	0.1070	
178.6207	30.3549	9.1658	
181.9229	30.5336	9.1737	
185.2606	30.7041	9.1814	
188.6326	30.8656	9.1892	
192.0377	31.0194	9,1968	
105 4747	31 4/50	0 0000	
170.4/4/	31.1652	9.2044	
198.9423	31.3030	9.2119	
202.4393	31.4328	9.2194	
205.9646	31.5547	9.2268	
209-5168	31.6684	9.2341	
20703100	31 3730	0 0414	
21300740	31.1737	702919	
216.6982	31.8710	9 • 2487	
220.3258	31.9596	9.2559	
223.9763	32.0389	9.2631	
227-6482	32-1094	9.2702	
021.2407	20 1715	9.0777	
23103403	5201115	302113	
235.0512	32.2252	9.2843	
238.7796	32.2704	9.2914	
242.5241	32.3071	9.2984	
246.2833	32-3353	9.3053	
250 0550	· 20 2540	0 7407	
250.0559	32.3348	7.5123	
253.8404	32.3658	9.3192	
257.6353	32.3681	9.3261	299.34

/HR-SF-F Afrin = 290.0034

5-12

80 (US	ILER PAN ING GROU	NEL NO Eneveli	• 1 D CORREL	ATION)	PILOT 1	PLANT		124
PRES: PSIA	S TEMP I	ENTHALI BTU/LI	PY L1 I B FT F	L2 NO.TUB	ES TUBE OD IN	•TUBE ID IN	TUBE LGTH FT	INCREMENT
1600	• 505•	493.	9. E :	32.8 70	0.50	0 • 27	41.	0 • 50
CONDI B-II	UCTIVITY N/H-SF-	EXT.	COEFF EM	ISSIVITY A	BSORPTANCE	•AMB.TEMF F	HAX.SOLA BTU/HR-	R FLUX SQFT
	150.	12	3.	0.89	0.95	100.	9800	0.
TUBE	PRESS	BULK I		INSIDE FILM COEF	OUTSIDE HEAT FLUX	TUBE CRO	WN BULK Qual	CRIT QUAL
	FSIA				0760	500	.	
0.50	1599.8	505.	493 .	1181.	2759.	509.	24	0.0
1.00	1597.6	505 .	475.	1182.	19100	507.	- 24	0.0
2.00	1599 3	509	47/ «	1184.	16725-	537.	- 23	0.0
2.50	1599.1	512.	503.	1185.	21321	547.	22	0.0
3.00	1598.9	516 .	508.	1187.	25997.	558 .	21	0.0
3.50	1598.8	520 .	513.	1190.	30586.	569.	21	0.0
4.00	1598.6	525 .	519.	1193.	35167.	581.	19	0.0
4.50	1598.4	536 .	526.	1197.	39752 .	594 .	18	0.0
5.00	1598.2	536 .	534 .	1201.	44523.	608.	17	0.0
5.50	1598.1	542.	542.	1205.	48995.	621.	15	0 • 0
6.00	1597.9	550.	552.	1210.	53623.	636.	13	0.0
6.50	1597.7	557.	562.	1215.	58063.	650.	11	0.0
7.00	1597.6	566 .	573 -	1222.	62662.	666.	09	0
7.50	1597.4	575 .	584.	1229.	6/14/0	682.		0.0
8.00	1597.2	204 .	577.	1237.	76310	715		0.0
0.00	159/01	090.	610 -	1247.	20240.	733.	0.00	0.0
9.50	1596-8	605.	639.	5000-	26532	699.	0.03	0
10.00	1596-6	665.	654 -	50000	89685 .	702.	0.06	0.0
10.50	1596.5	665.	670.	5000.	89697.	702.	0.09	0.0
11.00	1596.3	605.	686 .	5000.	89709.	702.	0.11	0.0
11.50	1596.2	605.	701.	5000.	89720.	702.	0.14	0.0
12.00	1596.0	605 .	717.	. 5000.	89732.	702.	0.17	0.0
12.50	1595.9	605 .	732 .	5000.	89745.	702.	0.20	0.0
13.00	1595.7	605.	748 .	5000.	89757.	702.	0.23	0 • 0
13.50	1595.5	624 .	763.	5000.	89770.	702.	0.26	0 • 0
14.00	1595.3	604 .	779 .	5000.	89347.	702.	0.29	0.0
14.50	1595.1	604 -	794 .	5000.	89361.	702.	0.32	0.0
15.00	1594.9	604 .	810.	5000.	89376.	702.	0.54	0.0
15.50	1594.7	614 .	825 .	5000.	89391.	792.	0.57	0.0
16.10	1094+0	604 .	841 *	5080s 5040s	094010	702.	0.43	0 • 0
17.00	1594-0	604.	872.	5000.	29442-	702.	0.46	0
17.50	1593-8	604 -	887.	5000.	8946 0.	702.	0.49	0.0
18.00	1593.5	614.	903.	5000.	89479.	702.	0.52	0.0
18.50	1593.3	604.	918.	5000.	89498.	702.	0.54	0.0
19.00	1593.0	604 .	934.	5090.	89519.	702.	0.57	0.0 /
19.50	1592.7	604 .	949.	5006.	89540.	702.	0.60	0.0
20.00	1592.4	664 .	965.	5000.	89562 .	702.	0.63	0.0
20.50	1592.1	664 .	980.	5000.	89585.	702.	0.66	0.0
21.00	1591.8	6:4 .	996.	5000.	89609.	702.	0.69	0.0
21.50	1591.4	6:4.	1011.	5900.	89633.	702.	0.72	0.0
55.13	1591.1	624 .	1027.	5266.	89659.	762.	0.74	0.0
22.50	1590.7	694 .	1:42.	5900.	99685.	702.	0.77	0.0

5-13

23.00	1590.4	664 •	1058.	5000.	89712.	702.	0•8û	Û • Ū
23.50	1590.	664 •	1073.	5000.	89740.	702.	0.83	0.0
24.00	1589.6	604 .	1089.	5000.	89768 •	702.	0.86	0.0
24.50	1589.2	604 -	1104.	5000.	89358.	702.	0.89	0 • 0
25.00	1588.8	604 •	1120+	5606.	89388.	702•	0.92	0.0
25.50	1588.4	604 •	1135.	5000.	89420.	702 •	0.95	0.95
26.00	1588.3	604 .	1151.	5003.	89663.	702.	0.97	0.97
26.50	1588.2	6 05 .	1166.	2534 •	89120.	720.	1.00	0.97
27.00	1588.1	614 •	1182.	808.	87856 •	790.	1.03	0.97
27.50	1588.0	624 •	1197.	756 -	87512.	805.	1.06	0.97
28.00	1587.9	635.	1212.	704.	87297.	821•	1.09	0+97
28.50	1587.7	648.	1227.	665.	87002.	838.	1.11	0.97
29.00	1587.6	661.	1242.	629.	86723.	855.	1.14	0.97
29.50	1587.5	675.	1257.	590.	86373.	874.	1.17	0.97
30.00	1587.4	691.	1272.	556 .	86052.	894.	1.20	0+97
30.50	1587.2	767.	1287.	527•	85646.	914.	1.22	0.97
31.90	1587.1	724 .	1301.	599.	85261.	933•	1.25	0.97
31.50	1587.0	742 .	1316.	495.	84857.	953•	1.28	0.97
32.00	1586.8	761.	1331.	483*	84459.	972.	1.30	0.97
32.50	1586.7	781.	1345.	473 .	84081.	992•	1.33	0.97
33.00	1586.5	800.	1359.	465.	81648.	1007.	1.36	0.97
33.50	1586.4	819.	1373+	459.	76326.	1013.	1.38	0.97
34.00	1586+2	837.	1385.	454•	71023.	1019.	1+40	0.97
34.50	1586.1	854 •	1396.	450.	65792.	1023.	1 = 43	0.97
35.00	1585.5	871.	1,407+	448 •	60543.	1025.	1.44	0.97
35.50	1585.7	885.	1416.	446 •	55314.	1027.	1.46	0.97
36.00	1585.6	899.	1425 .	444 •	50147.	1027.	1•48	0.97
36.50	1585.4	911.	1433*	443.	45007.	1027.	1.49	0.97
37.00	1585.3	922 •	1440+	443.	39900.	1025.	1.51	0.97
37.50	1585.1	932•	1446 •	442•	34741.	1021.	1.52	0.•97
38.00	1584.9	940.	1451.	442 •	29707.	1017.	1.53	0.97
38.50	1584.8	947.	1455.	442 •	24657.	1010.	1.53	0.97
39.00	1584+6	953.	1459.	442 •	19688.	1003.	1.54	0.97
39.50	1584.4	¥57•	1461.	442•	14698.	995.	1 • 5 4	0.97
40.00	1584.3	960.	1463.	442•	9720.	985.	1.55	0.97
4u+50	1584.1	961•	1464 .	442 •	4785.	973.	1.55	0.97
41.00	1583.9	961.	1464.	442 -	-102.	961.	1.55	0.97

MASS FLOW = 0.8419E 04LB/HR G = 0.3025E 06LB/HR-SQ FT

265.06B

 $S = \log \frac{12.017}{11.11}$ $lug \frac{.3176}{.3025}$

HMIN =

PiLot PLANT PANEL NO. 1

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N = . 890

,	CUMULATT	VE	PRESS	URE	DROF	>
FRI	CTION	MOM	ENTUM	G	RAVI	TY
P	SI		PSI		PS1	[
·						-
0.00	82	0.0	000	0.	1704	•
0.01	65	0.0	001	0.	3407	7
0.02	48	0.0	002	0.	5105	5
0.03	30	0.0	003	0.	6798	3
0.04	13	0.0	004	0.	8485	5
0.04	97	0.0	006	1.	0162	2
0.05	81	0.0	008	1.	1829	•
0.06	65	0.0	010	1.	3485	5
0.07	5 0	0.0	913	1.	5127	
0.08	35	0.0	016	1.	6754	ł
0.09	21	9.0	020	1.	8366	5
0.10	07	0.0	023	1.	9960)
0.10	94	8.0	027	2.	1536	3
0.11	83	0.0	032	2.	3093	5
0.12	72	0.0	036	2.	4630	}
0.13	61	0.0	041	2.	6146	5
0.14	52	0.0	046	2.	7639	2
0.15	44	0.0	054	2.	9108	,
0.16	48	0.0	159	3.	0400]
6.17	66	0.0	366	3.	1549	}
0.18	96	0.0	675	3.	2584	•
0.20	40	n_1	087	3.	3525	5
0.21	96	0.1	600	3.	4388	3
0.23	65	0.2	215	3.	5185	5
0.25	48	0.2	932	3.	5925	5
0.27	43	0.3	751	3.	6616	
0.29	52	0.4	672	3.	7263	5
6.31	73	0.5	695	3.	7873	5
0+34	07	0.6	820	3.	8449)
0.36	55	0+8	046	3.	8994	ł
0.39	15	0.9	374	3.	9513	5
0.41	88	1.0	805	4.	0007	7
0.44	74	1.2	337	4.	0478	3
0.47	74	1.3	971	4.	0929	}
0.50	86	1.5	707	4.	1361	
0.54	11	1.7	545	4.	1776	
0.57	49	1.9	486	4.	2175	5
. 0.61	01	2.1	529	4.	2559	3
0.64	65	2.3	674	4.	2929	3
0+68	43	2.5	922	4.	3287	7
0.72	33	2.8	272	- 4.	3633	5
0.76	37	3.0	725	4.	3967	7
0+80	54	3.3	280	4.	4291	
0 . 8 4 4	84	3.5	939	4 .	4605	j i

0.007	7 0700	4 4010
0.8927	5.8700	4.4910
0.9383	4 . 1 364	4.5205
0.9852	4.4532	4.5495
1.0334	4.7602	4.5773
1.0830	5.0776	4.6045
1.1338	5.4052	4.6311
1.1860	5.7432	4.6569
1.2826	5.7534	4.6709
1.3576	5.7641	4.6845
1.4353	5.7810	4.6976
1.5164	5.7982	4.7102
1.6008	5.8155	4.7223
1.6889	5.8329	4.7340
1.7811	5.8503	4.7452
1.8778	5.8677	4.7560
1.9790	5.8852	4.7664
2.0839	5.9027	4.7765
2.1927	5.9203	4.7862
2.3052	5.9378	4.7957
2.4217	5.9553	4.8049
2.5421	5.9728	4.8137
2.6663	5.9898	4.8224
2.7943	6.0058	4.8308
2.9258	6.0206	4.8390
3.0605	6.0343	4.8470
3.1983	6.0469	4.8549
3.3388	6.0584	4.8626
3.4820	6.0688	4.8702
3.6274	6.0781	4.8777
3.7749	6.0864	4.8851
3.9242	6.0935	4.8925
4.0751	6.0996	4.8997
4.2273	6.1047	4.9069
4.3806	6.1087	4.9141
4.5347	6.1117	4.9212
4.6894	6.1137	4.9283
4.8445	6.1146	4.9354
4.9997	6-1146	4.9425

16.0568

'HR-SF-F

LP= 11.11 psi

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RECOMMENDED LAY-UP PROCEDURES C-E DRUM TYPE UTILITY UNITS

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	TYPE OF SHUTDOWN	PROCEDURE	NOTES
	Pre-Operational Period Post Hydro (See Note 1)	With the economizer, waterwalls, superheater, and reheater filled to overflowing pressurize the unit with nitrogen to 5 psig pressure (See Notes 4 & 5).	 All non-drainable sections to be hydrostatically tested should be filled with demineralized or condensate quality water containing 10 ppm of
	Pre-Operational Period Post Chemical Cleaning	 Introduce demineralized or condensate quality water containing 10 ppm of ammonia and 200 ppm of hydrazine into the superheater, reheater, feedwater heaters, (tube side) and associated piping, economizer and waterwalls (Refer to Notes 3 & 4). 	ammonia and 200 ppm of hydrazine. Inis should produce a solution pH of approximately 10.0. The superheater should be filled first, to overflow into the boiler drum. Then the economizer and waterwalls can be filled through normal fill connections (See Note 2) with demineralized or condensate quality water, or if not available, any source of clean, filtered water may be used. This water should also contain 10 ppm of ammonia
		 Nitrogen cap the superheater, feedwater heaters (shell side) and drum. Maintain 5 psig nitrogen pressure (See Note 5). 	and 200 ppm of hydrazine.
	Short Outage - 4 Days of Less	 Maintain the same hydrazine and ammonia concentrations as those present during normal operation. 	 Involution and annound should be added in a manner that results in a uniform concentration throughout. They may be added to the system in several ways, as for example:
		 Establish and maintain a 5 psig nitrogen cap on the superheater and the steam drum (See Note 5). 	a. By pumping concentrated solutions through the chemical feed equipment and blend filling to achieve the desired concentrations.
		3. Nitrogen cap the shell side of the feedwater heaters.	b. If condenser leakage is not a cause for shutdown, concentrated
	Short Outage - 4 Days or Less	1. Drain and open only those sections requiring repairs,	be mixed to achieve the desired concentrations. If condensate demineralizers are employed, they must be bypassed during this
	Unit Partially Drained for	 Isolate remainder of unit under 5 psig nitrogen pressure where possible (See Note 5). 	operation. It is important to have the fluid temperature in the curle below 400° E
	neparts	 Maintain the same hydrazine and ammonia concentrations for water remaining in the cycle as those present during normal operation. 	before addition of hydrazine. If this temperature is exceeded, the hydrazine will decompose.
		4. Nitrogen cap the shell side of the feedwater heaters.	3. The tube side of copper alloy feedwater heaters should be filled with
	Long Outage - Longer than 4 Days	 Fill the superheater and reheater with demineralized or condensate quality water containing 10 ppm of ammonia and 200 ppm of hydrazine. The pH of the solution should be approximately 10.0. Add the fill water to the outlet of the non-drainable sections (See Note 4). 	 If freezing is a problem, the water in drainable circuits can be displaced with nitrogen and the unit layed up under 5 psig nitrogen pressure. Auxiliary heat may be applied to keep the operationable sections from
	Wi	 Increase the hydrazine and ammonia concentration in the waterwalls, economizer and feedwater heaters (tube side) and associated piping to 200 ppm and 10 ppm respectively (See Notes 2, 3, and 4). 	freezing. 5. Nitrogen cap should be applied through the drum vent, superheater outlet
		 Establish and maintain a 5 psig nitrogen cap on the superheater and steam drum (See Note 5). 	header drain/vent, and reheater outlet header drain/vent, as the unit is cooled, when pressure drops to 5 psig. Admission of air through atmospheric vents should be avoided,
		4. Nitrogen cap the shell side of the feedwater heaters.	1 ²
•	Long Outage -	1. Drain and open only those sections requiring repairs.	1
	Unit Partially	2. Fill the superheater and reheater (if not requiring draining for repairs) with	
	Drained for Repairs	demineralized or condersate quality water containing 10 ppm of ammonia and 200 ppm of hydrazine. The pH of the solution should be approximately 10.0. Add the lift water to the outlet of the non-drainable sections (See Note 4).	
		 Increase the hydrazine and ammonia concentrations in the tube side of the feedwater heaters and the undrained circuits of the economizer and water- walls to 200 ppm and 10 ppm respectively (See Notes 2, 3 & 4). 	
		 Establish and maintain a 5 psig nitrogen cap on the undrained sections of the unit, where possible (See Note 5). 	
		5. Nitrogen cap the shull side of the feedwater heaters.	
		 After completion of the repair, fill the drained sections with demineralized or condensate quality water containing 10 ppm of ammonia and 200 ppm of hydrazine. Cap with nitrogen (See Notes 2, 3, & 4). 	·

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SECTION 6

References

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