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MCR 78-1323

10 MWe SOLAR THERMAL CENTRAL RECEIVER PILOT PLANT

CONCEPTUAL DESIGN REVIEW PACKAGE

Dated October 16, 1978

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Prepared under Contract No. ET-78-C-03-2182 by Martin Marietta Corporation for Department of Energy

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SYSTEM PARAMETERS

• HELIOSTAT GENERAL REQUIREMENTS

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- o OPERATIONAL
- o CONFIGURATION
- o SURVIVAL

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- HELIOSTAT WEIGHTS
- **o PHASE I DIFFERENCES**

HELIOSTAT GENERAL REQUIREMENTS

HELIOSTAT CONFIGURATION

REFLECTIVE AREA SIZE FIELD CLEARANCE RADIUS REFLECTING SURFACE STOWAGE ROTATIONAL REQUIREMENTS 430 SQ. FT. 22.6 x 22.6 FT. 16 FT. GLASS, SECOND SURFACE MIRROR FACE DOWN AZ $\pm 270^{\circ}$, $\pm 0^{\circ}$ EL $\pm 0^{\circ}$

HELIOSTAT OPERATIONAL REQUIREMENTS

SLEW

OPERATIONAL TEMPERATURE OPERATIONAL WINDS INITIATE STOWAGE DESIGN WIND RISE RATE WIND SURVIVAL, ANY ATTITUDE WIND SURVIVAL, STOWED 19.7°/MIN, MINIMUM +32°F TO +120°F 20 MPH WITH GUSTS TO 27 36 MPH, INCLUDING GUSTS 0.02 MPH/SEC² 50 MPH, INCLUDING GUSTS GUSTS TO 90 MPH @ + 10° ANGLE OF ATTACK

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PART DESCRIPTION	WEIGHT EACH	QUANTITY PER HELIOSTAT	TOTAL IN	WEIGHT LBS.
				27/0
HELIOSTAT RACK ASSEMBLY	115	12	1860	2/42
MIRKUR ASSEMBLI	110	2	220	
BAR JUISIS - UUI	90	2	180	
TODOUE TURE	412	1	412	
	25	2	50	
SUPPORT ARTS MOINTING STIDS	1	36	18	
MOUNTING HARDWARE	-	1 LOT	2	
				819
DEIVE UNII ASSEMBLI	670	1	670	017
DRIVE RECHANISH	90	1	90	
OTI	30	15 OUART	30	
MOTOR GEARHEADS	10	2	20	
FNCODER	3	2	6	
MOUNTING HARDWARE	-	1 LOT	3	
DEDECTAL ASSEMBLY				552
PEDESTAL	550	1	550	
COVER (ACCESS HOLE)	1	1	1	
MOUNTING HARDWARE	1	1 LOT	1	
CABLE AND ELECTRONICS				6.0
CABLE	40	1	40	62
HELIOSTAT CONT. ELECTRONICS	20	1	20	
MOUNTING HARDWARE	-	1 LOT	2	

HELIOSTAT WEIGHTS

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DIFFERENCES IN PHASE I AND PHASE II APPROACHES

PHASE I DESIGN DIFFERENCES

- 1. HAC MINIMUM CONFIGURATION (NO REDUNDANCY AND NOT COMPLETE SOFTWARE)
- 2. HC CONTAIN ERASABLE REPRO-GRAMMABLE MEMORIES

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3. ALIGNMENT WILL BE OPTICAL VIA A LASER BEAM COST SAVINGS

RATIONALE

FLEXIBILITY IN HARDWARE AND SOFTWARE DESIGN AND DEBUGGING

COST SAVINGS BY USE OF EXISTING EQUIPMENT FOR THESE SMALL QUANTITIES

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HELIOSTAT STRUCTURES AND MECHANISMS

- o 10 MWe COLLECTOR SYSTEM HELIOSTAT CONFIGURATION
 - O VARIATION FROM BASELINE SUBMITTED IN PROPOSAL
- HELIOSTAT WINDLOADING ANALYSIS
 - HELIOSTAT COORDINATE SYSTEM
 - WINDLOADING COMPARISONS
 - o HELIOSTAT WINDLOADS MAXIMUM FROM ASCE APPROACH
- o HELIOSTAT/FOUNDATION INTERFACE REQUIREMENTS
 - **o** ACCEPTABLE ALTERNATIVES

• HELIOSTAT DRIVE MECHANISM

- o HELIOSTAT DRIVE MANUFACTURER
- WORKING AGREEMENT
- DRIVE MECHANISM DESIGN
- STTF DRIVE MECHANISM TESTS
- PROGRAM WORK PLAN



10 MWe COLLECTOR SYSTEM HELIOSTAT



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HELIOSTAT COORDINATE SYSTEM DEFINITION



ELEV. ANGLE	AZIMUTH ANGLE	MOMENT	MOMENT		FORCES (1bf)	ť
a (deg)	β (deg)	M _{yy} (ft-1bf)	M_{zz} (ft-lbf)	DRAG	LIFT	SIDE
10	0	4930	0	272	878	0
20	0	7473	о	693	2029	0
90	0	0	0	2771	0	0
90	70	0	7473	693	0	2029

HELIOSTAT WIND LOADS - ASCE APPROACH

 $V_{ref} = 50 \text{ mph HORIZONTAL}$

THIN FLAT PLATE: ASPECT RATIO = 1AREA = 506 ft² SOLIDITY RATIO = 1

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HELIOSTAT ELEVATION MOMENT - CSU WIND TUNNEL DATA



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HELIOSTAT/FOUNDATION INTERFACE REQUIREMENTS

Elevation Encoder - 011 Fill Leveling Surface-Lifting Hook Ø \bigcirc Motor Gearhead · 🚯 6) Elevation Axis Shaft Azimuth Fixed Shaft Oil-Level Plug Pedestal Interface Support 0 Azimuth Encoder MARTINMARIETTA

HELIOSTAT COMBINED AZIMUTH AND ELEVATION DRIVE

Drive Mechanism Assembly

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STTF DRIVE MECHANISM - TEST DATA



STRUCTURES AND MECHANISMS

WEEK OF	DRIVE MECHANISM	STRUCTURAL
10/2	DRIVE MANUFACTURER SELECTION	MIRROR MODULE DRAWINGS TO BALTIMORE
10/9	PRELIMINARY MEETING BETWEEN MMC/HUB CITY	TENTATIVE INTERFACE DRIVE MECHANISM, PEDESTAL, CONTROL ARMS
10/16	DRIVE MECHANISM PROGRAM PLAN	·
10/23	ESTABLISH DRIVE MECHANISM INTERFACES	CONCEPTUAL DESIGN REVIEW
10/30		PEDESTAL I/F ADAPTER - PRELIMINARY ENCODER BRACKETS - PRELIMINARY
11/6	DRIVE MECHANISM FINAL DESIGN LAYOUT - PRELIMINARY DESIGN REVIEW	CONTROL ARMS - PRELIMINARY BAR JOISTS - PRELIMINARY
11/13	DRIVE MECHANISM DETAILING START	WINDLOADING ANALYSIS PRELIMINARY DESIGN REVIEW
11/20		PEDESTAL, ELEVATION BEAM DETAILS - PRELIMINARY
11/27		FOUNDATION REQUIREMENTS
12/4	RELEASE HOUSING DRAWINGS - ORDER PATTERNS, TOOLS, ETC.	
12/11		PEDESTAL ASSEMBLY - PRELIMINARY
12/18		

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STRUCTURES AND MECHANISMS

WEEK	DETUE MECHANI SM	Γ STRICTIRAT
12/25	DRIVE MECHANISMS DRAFTING COMPLETE	CHRISTMAS HOLIDAYS
12/23		· · · · · · · · · · · · · · · · · · ·
1/1	DRIVE MECHANISM CRITICAL DESIGN REVIEW	
1/8		
1/15		PEDESTAL I/F ADAPTER (MOCKUP) TESTING
1/22		CONTROL ARM (MOCKUP) TESTING
1/29		
2/5		FACILITY CRITERIA CRITICAL DESIGN REVIEW
2/12	•	
2/19		
2/26	DRIVE MECHANISM 1ST ARTICLE AVAILABLE	
3/5		PREPRODUCTION AND PRODUCTION DESIGN DRAWINGS
3/12		· · ·

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$\ensuremath{\,\text{MW}_{e}}$ heliostat mirror assembly

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MIRROR ASSEMBLY STATUS

o DESIGN DRAWING RELEASED TO BALTIMORE DIVISION 10/9/78

- o LONG-LEAD PROCUREMENT INITIATED: STEEL 10/5/78, H/C CORE 10/9/78
- o SILVERING PROCUREMENT INITIATED 10/10/78, GARDNER MIRROR CORPORATION
- o SEALANT EVALUATION TEST INITIATED 10/12/78
- O NASTRAN MIRROR ASSEMBLY DETAIL DESIGN ANALYTICAL VERIFICATION IN PROCESS

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EXPLOYED VIEW OF HONEYCOMB-CORE MIRROR ASSEMBLY



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MIRROR ASSEMBLY CROSS SECTION



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MIRROR MODULE MOUNTING CONCEPT





1-2 1-1 MIRROR ASSEMBLY - CHANGES FROM PROPOSAL DESIGN

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ITEM	ORIGINAL PROPOSAL	NEW DESIGN	BASIS FOR CHANGE
HONEYCOMB CORE	ACG-3/4-003 3.2 IN. THICK	ACG-1-003 2.6 IN. THICK	MATERIAL AVAILABILITY COMMERCIALLY, COST REDUCTION 30% IN QUANTITY, ACCEPTABLE PERFORMANCE
ADHESIVE	MP 347, HIGH- TEMPERATURE-CURING EPOXY	HEXABOND 3 HIGH-TEMPERATURE- CURING EPOXY	EQUIVALENT ADHESIVE - LOWER COST - INCREASED PRODUCTION EFFICIENCY
SEALANT	PPC 2000 BUTYL CALKING COMPOUND	TBD - TEST IN PROGRESS	ORIGINAL SEALANT HAS UNACCEPTABLE CURING TIME FOR PRODUCTION RATES

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MIRROR ASSEMBLY THERMAL BENDING VS CORE THICKNESS



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EFFECT OF THERMAL DISTORTION ON BEAM QUALITY FOR SINGLE MIRROR ASSEMBLY

MIRROR ASSEMBLY LIFE

CRITICAL LIFE COMPONENTS:

STEEL:

THREE LAYER LONG LIFE FINISH

- o ELECTROGALVANIZE: 0.10 OZ. ZINC CHEMICAL TREATMENT
- O EPOXY PRIMER
- **o** BAKED SILICONE POLYESTER PAINT

SEALANT: SELECTION TO BE BASED ON:

• VENDOR LIFETIME PERFORMANCE DATA

• MARTIN MARIETTA TEST PROGRAM

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MIRROR ASSEMBLY SEALANT TEST PROGRAM

• PURPOSE - EVALUATE COMMERCIAL SEALANTS CAPABILITIES FOR SEALING GLASS

MIRROR, EPOXY PRIMER AND STEEL EDGE INTERFACE.

o TEST CONDITIONS - 150°F, 95-100% RELATIVE HUMIDITY, TIME - TBD.

o CANDIDATE MATERIALS -

PRODUCTS RESEARCH AND CHEMICAL PR-365 1) 2) PRODUCTS RESEARCH AND CHEMICAL PR-380M UNITED GILSONITE LABORATORIES BUTYL CAULK 3) UNITED GILSONITE LABORATORIES ACRYLIC LATEX CAULK 4) UNITED GILSONITE LABORATORIES TUB AND TILE CAULK 5) 3M WEATHERBAN 101 6) SCOTCH - SEAL 1792 7) 8) SCOTCH - SEAL 2084 9) ESSEX CHEMICAL PRO-SEAL 860 10) 3M SCOTCH SEAL 1792 11) NATIONAL ADHESIVE'S DURIBBON 4050

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12) GE SILPROOF WEATHERPROOFING SEALANT

13) SILICONE SEALANT RTV 108

o SAMPLE SIZE 2" X 2" MIRROR ON 4" X 4" EPOXY PRIMED STEEL.

MIRROR ASSEMBLY ADHESIVE COMPATIBILITY TEST

HEXABOND ADHESIVE TESTS

O HEXCEL CORPORATION: MATERIAL COMPATIBILITY TESTS OF HONEYCOMB-

ADHESIVE-MIRROR BACK INTERFACE.

o MARTIN MARIETTA CORPORATION - BALTIMORE:

SAMPLE NUMBER: 55 - 2" X 2"

SAMPLE STRUCTURE: 0.125" FLOAT GLASS - 3.23" ACG-3/4-.003 HEXABOND COATED

ALUMINUM HONEYCOMB CORE - 0.017" PREFINISHED STEEL.

TEST: A. FLATWISE TENSION PULL TESTS

B. FLATWISE TENSION LOAD TEST

C. ENVIRONMENTAL SOAK (150°F, 100% RH) THEN FLATWISE TENSION PULL TESTS.

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MIRROR ASSEMBLY - PRODUCTION AREA PLAN



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MIRROR ASSEMBLY PRODUCTION TOOL

- HOT BOND TOOL STABILIZED FOR HOT BOND PROCESS
- TOOL CURVED TO SHAPE FOR CORRECT MIRROR CURVE AT BOND TEMPERATURE ($\sim 245^{\circ}$ F) FOR FLAT MIRROR SHAPE AT 32° F
- VACUUM HOLD DOWN FOR MIRROR
- FAST ACTING VACUUM BAG FOR UNIFORM PRESSURE DURING BONDING
- o TOOL FLATNESS TOLERANCE GOAL + 0.2 MRAD MAX SLOPE ERROR

MIRROR ASSEMBLY PRODUCTION TOOL (CONCEPTUAL)



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MIRROR ASSEMBLY DESIGN VERIFICATION ANALYSIS

NASTRAN COMPUTER ANALYSES

o THERMAL DEFLECTION ANALYSIS

MIRROR DEFLECTION

GLASS STRESS

o STRUCTURAL LOADS ANALYSIS

MIRROR DEFLECTION

SUPPORT LOCATION AND SIZE

GLASS STRESS

STRUCTURAL STIFFNESS

• WIND LOADS ANALYSIS MIRROR DEFLECTION GLASS STRESS SUPPORT LOCATION AND SIZE STRUCTURAL STIFFNESS

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NASTRAN MODEL - MIRROR MODULE



2 STRUCTURAL SUPPORTS

16 PLANE OF SYMMETRY CONSTRAINTS

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С С
NASTRAN PLATE ELEMENT STIFFNESS

 $TG_1 T^2G_4 0$ $T^2G_4 IG_2 0$ 0 T_sG₃. 0

WHERE:

T = TOTAL THICKNESS OF PLATE I = UNIT BENDING INERTIA OF PLATE $T_s = CORE THICKNESS$



E = YOUNG'S MODULUS

 γ = POISSON'S RATIO

 $G_{1} = \frac{1}{T} \int_{-\frac{T}{2}}^{\frac{T}{2}} \left[G_{e} \right] dZ$ $G_{2} = \frac{1}{T} \int_{-\frac{T}{2}}^{\frac{T}{2}} Z^{2} \left[G_{e} \right] dZ$ $\frac{2}{\left[\begin{array}{cc}
G_{L} & 0\\
0 & G_{w}
\end{array}\right]}$ $G_{4} = \frac{1}{T^{2}} \int_{-\frac{T}{2}}^{\frac{T}{2}} (-Z) \left[G_{e}\right] dZ$ $\stackrel{\neg E \text{ SHEAR MODULUS J}}{\qquad W \text{ DIREC}}$

= CORE SHEAR MODULUS IN L DIRECTION

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NASTRAN MIRROR MODULE MODEL



SUPPORT CONSTRAINTS:

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- A) NODES IN THE PLANE OF SYMMETRY ARE CONSTRAINED AGAINST ROTATION ALONG THE X AXIS AND AGAINST TRANSLATION IN THE Y DIRECTION.
- B) THE STRUCTURAL SUPPORTS ARE MODELED BY CONSTRAINING THE APPROPRIATE NODES AGAINST TRANSLATION IN THE Z DIRECTION. ONE OF THE SUPPORTS IS ALSO CONSTRAINED AGAINST TRANSLATION IN THE X DIRECTION.

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UNIFORM NORMAL WIND LOAD VERIFICATION



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LINEAR THERMAL EXPANSION VERIFICATION



CORE DEPTH

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ANALYSIS APPROACH TO MIRROR ASSEMBLY SUPPORT LOCATIONS

HELIOSTAT ALIGNMENT, POINTING AND REALIGNMENT

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MIRROR ASSEMBLY ALIGNMENT CONCEPT - SIMPLIFIED VIEW



o image size for aligned heliostat (FLAT MIRROR ASSYS.) At 1200 FT \approx size of heliostat.

o image size for any heliostat aligned for 1200 ft \approx size of heliostat.

O ALL HELIOSTAT IMAGES ARE UNIFORMLY SPREAD OVER RECEIVER.

o GIVES DESIRABLE IMAGE SIZE AND UNIFORMITY (NO HOT SPOTS).



MIRROR ASSEMBLY ALIGNMENT - PHASE II

- o ALIGNMENT IS LAST STEP ON ASSEMBLY LINE.
- O OPTIMIZED FOR PRODUCTION LINE EFFICIENCY.
- o MIRROR SURFACE DIRECT READOUT MAXIMUM ACCURACY.
- o DIGITAL READOUT TO MINIMIZE READING ERROR.
- o STANDARD SIMPLE CALIBRATION SURFACE PLATE REFERENCE.

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- o ALLOWS ALIGNMENT VARIATION WITH NO HARDWARE CHANGE.
- o LOW-COST ALIGNMENT FIXTURE.

REFLECTIVE ASSEMBLY ALIGNMENT FIXTURE





HELIOSTAT POINTING - ENCODER BIAS DETERMINATION

CONSTRAINTS

• MUST BE PERFORMED AFTER FIELD INSTALLATION.

APPROACH

- o LASER ORIGINAL/REFLECTED BEAM AUTOCOLLIMATION.
- o HELIOSTAT OPERATION BY HC STIMULATOR.
- o MANUAL LASER OPERATION AND POINTING.
- O ENCODER INPUT BY PREPROGRAMED HAND-HELD CALCULATOR OR STIMULATOR.
- MANUAL REFLECTED BEAM POINTING CORRECTION.
- o REQUIRES KNOWLEDGE OF HELIOSTAT AND TARGET CENTER LOCATION COORDINATES.
- o FIELD TIME 10 MINUTES PER HELIOSTAT PLUS SET UP.
- o NO CONTROL COMPUTER INTERFACE.



HELIOSTAT POINTING - ENCODER BIAS DETERMINATION





HELIOSTAT POINTING - OPTICAL DIAGRAM



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HELIOSTAT REALIGNMENT

- PRE-COMPUTE TWO AXIS SLOPE DIFFERENCE BETWEEN THE REFERENCE MIRROR ASSEMBLY AND OTHER MIRROR ASSEMBLIES.
- MANUALLY POSITION THE HELIOSTAT TO PRODUCE DESIRED SLOPE RANGE OF REFERENCE MIRROR AS MEASURED WITH A 2 AXIS INCLINOMETER.
- MEASURE 2 AXIS SLOPE DIFFERENCE BETWEEN REFERENCE MIRROR AND MIRROR BEING ALIGNED WITH DIFFERENTIAL INCLINOMETER SYSTEM.
- ADJUST MIRROR BEING ALIGNED TO CORRECT SLOPE RELATIVE TO REFERENCE MIRROR.

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DIFFERENTIAL INCLINOMETER SYSTEM FOR HELIOSTAT REALIGNMENT



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MIRROR SIDE UP

PREPRODUCTION HELIOSTAT ALIGNMENT - PHASE I (OPTION A)

o ALIGNMENT WILL BE DONE USING STTF LASER/COLLIMATOR (L/C) AND TARGET.

o ALIGNMENT WILL USE STTF ALIGNMENT COMPUTER PROGRAM MODIFIED FOR 10 MWe HELIOSTAT.

O COMPUTER PROGRAM RUN ON DENVER DDC TO MINIMIZE STTF COMPUTER REQUIREMENTS.

o ALIGNMENT WILL HAVE BETTER THAN 13-BIT ACCURACY.

O HELIOSTAT POSITIONING DURING ALIGNMENT BY HFC STIMULATOR.

o ALIGNMENT WILL BE VERIFIED USING THE SUN.

O REQUIRES EXACT LOCATION COORDINATES OF THE HELIOSTATS, L/C AND TARGET CENTER.



PREPRODUCTION HELIOSTAT ALIGNMENT - PHASE I - (OPTION B)

- o SPHERICAL ALIGNMENT
- o 50 MW HeNe LASER (STTF SPARE)
- o LOCATE LASER AT TARGET CENTER 2f (2400 FT.) FROM HELIOSTAT
- O STATIC MANUAL ALIGNMENT NOT DEPENDENT ON ENCODER ACCURACY

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• VERIFY ALIGNMENT WITH THE SUN

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HELIOSTAT ALIGNMENT AT THE CENTER OF CURVATURE



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INTERPRETATION OF PERFORMANCE SPECIFICATION

BEAM POINTING ERRORS - ≤ 1.5 MRAD STANDARD DEVIATION (REFLECTED BEAM) WITH SUN ELEV. $\geq 15^{\circ}$.

O DOES NOT INCLUDE STRUCTURAL DEFLECTIONS DUE TO WIND OR GRAVITY.

BEAM QUALITY - ≥90% OF REFLECTED ENERGY WITHIN AREA 1.4 MRAD LARGER THAN THEORETICAL BEAM SHAPE.

- INCLUDES SPECULAR DISPERSION, MIRROR IRREGULARITIES, MIRROR THERMAL DISTORTIONS, AND MISALIGNMENT BETWEEN FACETS.
- DOES NOT INCLUDE STRUCTURAL DEFLECTIONS OF THE MIRROR MODULE OR SUPPORT STRUCTURE.

REFLECTIVE SURFACE STATIC AND DYNAMIC DEFLECTIONS - \leq 1.7 MRAD STANDARD DEVIATION (MIRROR NORMAL DEFLECTION)

- INCLUDES ALL WIND-INDUCED STRUCTURAL DEFLECTIONS (ON DRIVE MECHANISM, STRUCTURE, AND MIRROR MODULE).
- WIND VELOCITY: 9.23 M/S STEADY/12 M/S STEADY WITH GUSTS.
- STANDARD DEVIATIONS DETERMINED FROM DISTRIBUTIONS OF WIND DIRECTION, HELIOSTAT GIMBAL ANGLES, AND DEFLECTIONS OVER THE HELIOSTAT SURFACE.



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TEST UNDER (APPROXIMATELY) NO WIND CONDITIONS. TEST AT (APPROXIMATELY) THE ELEVATION GIMBAL ANGLE USED FOR ALIGNMENT TO EXCLUDE STRUCTURAL DEFLECTION EFFECTS.

BEAM QUALITY

OF THE HELIOSTAT.

TEST UNDER (APPROXIMATELY) NO-WIND CONDITIONS. USE A PRECISION REFERENCE MIRROR MOUNTED NEAR THE CENTER

BEAM POINTING

TESTING IMPLICATIONS OF PERFORMANCE SPEC INTERPRETATION

BEAM POINTING ERROR ALLOCATIONS

		REFLECTED BEAM	
ERROR SOURCE	RESOLUTION	AZI	ELEV
ENCODER RESOLUTION	13 BIT WORDS	0.44	0.44
SUN POSITION PREDICTION	SEE NEXT TABLE	0.08	0.08
LIGHT PATH REFRACTION	22 ARC SEC	0.11	0.25
HELIOSTAT GEOMETRY ERRORS			
PAD LOCATION	<u>+</u> 0.50 INCH	0.13	0.10
REF. FACET CENTER LOCATION	<u>+</u> 0.25 INCH	0.06	0.06
PEDESTAL TILT	1.5 MRAD (3 🖝)	0.71	0.50
DRIVE UNIT AZI. AXIS TILT	0.5 MRAD (3 or)	0	0.24
DRIVE UNIT ELEV. AXIS TILT	0.5 MRAD (3 c-)	0.24	0.02
MIRROR SUPPORT MISALIGNMENT	1.2 MRAD (3 r)	0.56	0.02
ALIGNMENT ERRORS			
ENCODER ERROR	<u>+</u> 0.2 BIT	0.18	0.18
REFLECTED BEAM CENTROID	\pm 2 INCHES	0.50	0.32
NORTH/SOUTH REF. DETERMINATION	0.01 DEGREE	0.17	0
RSS TOTALS RSS BEAM POINTING TOTAL	$\sqrt{(EL)^2 + (AZ \ COS \ 13.8^{\circ})^2} =$	1.19	0.85

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տ Ծ BEAM POINTING ERROR CALCULATION ASSUMPTIONS

ERROR SOURCE	ASSUMPTIONS
ENCODER RESOLUTION	EQUAL PROBABILITY WITHIN + 1/2 BIT '
SUN POSITION PREDICTION	CODED EPHEMERIS ALGORITHM (9 ARC SEC ACCURACY) CUBIC FIT ET TO UTC CONV. (0.3 SEC ACCURACY) LATITUDE/LONGITUDE UNCERTAINTIES (0.06 MRAD) 16-BIT SUN VECTOR WORDS SENT EVERY SECOND.
LIGHT PATH REFRACTION	SUN TO HELIOSTAT AND HELIOSTAT TO TARGET ALGORITHM CORRECTION ERRORS TO 22 ARC SEC.
HELIOSTAT GEOMETRY ERRORS	<pre>PAD AND REF. FACET LOCATIONS ON CLOSEST HELIOSTAT TILTS AND MISALIGNMENTS OF PEDESTAL AND AXES BASED ON MAXIMUM 3 VALUE FOR 2000 HELIOSTATS. VALUES CONVERTED FROM MIRROR NORMAL TO REFLECTED BEAM.</pre>
ALIGNMENT ERRORS	ENCODER CAN BE CONTROLLED DURING ALIGNMENT TO \pm 0.2 BIT. REFLECTED BEAM CENTROID FOR CLOSEST HELIOSTAT.

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THEORETICAL BEAM SHAPE



HELIOS PROGRAM

KUIPER DISTRIBUTION SUNSHAPE EXPONENTIAL DECAY FOR SUN EDGE NO MIRROR ERROR DISTRIBUTION RECTANGULAR FLAT FACETS FACET REFLECTIVITY = 88% HELIOSTAT LOCATION = (0, 357m,0) HELIOSTAT SLANT RANGE = 366m CALCULATED FOR SUMMER SOLSTICE NOON CALIBRATED FOR VERNAL EQUINOX NOON BARSTOW LATITUDE

1.4 MRAD FRINGE

 $-3.75 \le X \le 3.75$ METERS $-3.00 \le Z \le 3.00$ METERS



BEAM QUALITY PARAMETRICS



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BEAM QUALITY ERROR ALLOCATIONS

ERROR SOURCE	REFLECT ERROR (<u>AZI</u>	ED BEAM 1 J) ELEV
SPECULAR DISPERSION	0.8 [#]	0.8*
MIRROR IRREGULARITIES	1.4	1.4
MIRROR THERMAL DISTORTION	0*	0*
RELATIVE FACET MISALIGNMENT	1.4	1.4
RSS TOTALS	2.2	2.2
RSS BEAM POINTING TOTAL	3	.1

† DEPENDENT UPON QUALITY OF GFP GLASS.

* THERMAL DISTORTION NEGLIGIBLE BECAUSE FACETS ARE DESIGNED TO BE FLAT AT O^OC (32^OF) AND THERMAL EFFECTS UP TO 50^OC (122^OF) WILL RESULT IN A DECREASE IN BEAM SIZE.

















	MINIMUM	· ·		
<u></u>	STEM CHARACTERI	STICS		
	6801	6802/6846	8048	9940
PROGRAM ROM (BYTES)	2К	2К	1К	2K
USER RAM (BYTES)	128	128	64	128
USEABLE I/O (PARALLEL)	24	8	24	32
ON CHIP UART	YES	NO	NO	NO
I/O W/MEMORY EXPANSION	8	8	12	32
TIMER	16 BIT	16 BIT	8 BIT	14 BIT
EXTERNAL PROGRAM ADDR. CAPABILITY	*65K	*6 5K	2К	NONE
EXTERNAL RAM ADDR. CAPABILITY	*65K	*65K	256	256 BIT
EPROM VERSION AVAILABILITY	4 QTR. '78	-	NOW	4 QTR '78

*ANY COMBINATION OF I/O , ROM & RAM



MINIMUM SYSTEM

COST COMPARISON

MAJOR SYSTEM PACKAGES 2000 QUANTITY TOTAL 6801 Microcomputer \$16.50 \$16.50 6802 Microprocessor, RAM, CLOCK \$ 8.55 \$29.85 6846 ROM, 110, TIMER \$13,00 6850 Asynchronous I/O \$ 3.30 Communications Clock \$ 5.00 8048 Microcomputer \$14.00 \$24.44 8251 Asynchronous I/O \$ 5.44 Communications Clock \$ 5.00 9940 Microcomputer \$ 7.00 \$16.10 9902 Asynchronous I/O \$ 4.10 Communications Clock \$ 5.00



MICROCOMPUTER SELECTION - MC6801

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- * Low Cost
- * Lowest Number of Chips
- * Ram and Rom Easily expanded
- * Will be used for both HFC and HC

DRIVE MOTOR TRADEOFF STUDY

OBJECTIVES

MEET PERFORMANCE REQUIREMENTS AT MINIMUM COST.

USE ONE MOTOR PER AXIS.

OBTAIN REDUCED-SPEED OPERATION (~ 1200 RPM OR LESS) USING SLEW MOTOR.

CANDIDATES

THREE PHASE A-C, SINGLE PHASE A-C, D-C.

RESULTS TO DATE

THREE-PHASE AND SINGLE-PHASE A-C MOTORS

CONTROLLER DESIGN

BREADBOARD TEST

SPEED-TORQUE CURVES

CONTINUING EFFORT

D-C MOTOR CONTROLLER DESIGN AND BREADBOARD TEST

D-C MOTOR BRUSH LIFE TEST



SPEED-TORQUE CURVES FOR SINGLE-PHASE MOTOR



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MOTOR TRADE OFF RESULTS

TYPE OF MOTOR	D.C. MOTOR	SINGLE PHASE	THREE PHASE
NUMBER OF PARTS REQUIRED	TO BE DETERMINED	8 ,	22
NUMBER OF EXTERNAL CONNECTIONS	10	12	18
NUMBER OF MICROPROCESSOR INTERFACE LINES	5	5	9
APPROXIMATE PARTS COST	TO BE DETERMINED	\$31.00	\$64.00
APPROXIMATE MOTOR COST	\$100.00	\$87.00	\$92.00

MARTINMARIET

ENCODER

* Incremental - 2048 Counts/Turn

- * Interface Electronics will have X4 Multiplier
- * Development will include extensive EMI and Thermal Testing

MARTIN MARIET

P/W BOARD DESIGN

- o PW BOARDS WILL BE DOUBLE SIDED WITH PLATED-THRU HOLES, .062 THICK MATERIAL.
- o LAYOUT AND DESIGN PER IPC-CM-770A
- o FABRICATION PER IPC-A-600A
- INCORPORATE SOCKETS FOR PARTS SUCH AS OPTO-COUPLERS AND LINE DRIVERS THAT MAY REQUIRE REPLACEMENT.

VARTIN MARIETTA

- o PROVIDE INTERFACE CONNECTIONS THAT UTILIZE SOLDERLESS TERMINALS.
- DESIGNED FOR FLOW SOLDERING

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- o BOARD SIZE APPROXIMATELY 6.0 X 7.0 INCHES
- O SEPARATE BOARD DESIGNS FOR HC AND HFC

HC/HFC CONTROL (AC MOTORS)



HC/HFC CONCEPTUAL PACKAGING DESIGN - AC MOTORS

- o STD COMMERCIAL STEEL BOX 8 X 10 X 6 W/HINGED COVER
- O OPEN FRAME 5V POWER SUPPLY, QUICK DISCONNECT TERMINATIONS
- o DOUBLE SIDED, FLOW SOLDERED, PRINTED WIRING BOARD
- o SOLID STATE RELAYS SOLDERED ON BOARD, QUICK DISCONNECT OUTPUT TERMINATIONS
- O ADDRESSING WILL UTILIZE A PREWIRED DIP PLUG-IN, ON PW BOARD
- O CONNECTORS CRIMP TYPE PLUG IN CONTACTS
- TWO + 5V ISOLATED SUPPLIES TO BE INTEGRATED INTO BASIC 5V POWER SUPPLY ALLOWING A COMMON DESIGN FOR HC AND HFC



HC/HFC CONTROL (DC MOTORS)



HC/HFC CONCEPTUAL PACKAGING DESIGN - DC MOTORS

- o STANDARD COMMERCIAL STEEL BOX 12.0 X 12.0 x 6, W/HINGED COVER
- O OPEN FRAME 5V POWER SUPPLY, QUICK DISCONNECT TERMINALS
- o DOUBLE SIDED, FLOW SOLDERED PRINTED WIRING BOARD
- o ADDRESSING WILL UTILIZE A PREWIRED DIP PLUG-IN ON PW BOARD.
- O CONNECTORS CRIMP TYPE PLUG IN CONTACTS
- O HC PW BOARD WOULD HAVE SOLID STATE RELAYS (AC MOTORS) REPLACED BY APPROX.
 16 SCRS, 8-PULSE TRANSFORMERS, 4-DIPS, 16 RESISTORS. ALL PARTS WILL BE
 MOUNTED ON PW BOARD.
- NO CHANGE TO HFC PW BOARD.
- TWO + 5V ISOLATED SUPPLIES TO BE INTEGRATED INTO BASIC 5V POWER SUPPLY ALLOWING A COMMON DESIGN FOR HC AND HFC CONTROLLERS.



ALTERNATE PACKAGING APPROACHES

- SEPARATE PACKAGING DESIGNS FOR HC AND HFC CONTROLLERS
- O SOLID STATE RELAYS MOUNTED DIRECTLY TO HC ENCLOSURE
- O SEPARATE PW BOARD FOR SOLID STATE RELAYS AND/OR MOUNT SOLID STATE

RELAYS ON SOCKETS

• PACKAGING CONCEPT BASED ON REPLACING PIG-TAIL CABLE WITH CONNECTORS

ARTIN MARIETTA

ELECTRICAL POWER AND DISTRIBUTION SYSTEM

o ELECTRICAL INTERFACE

O HELIOSTAT TO FIELD

o COMPUTER TO FIELD

o FOCUS AND ALIGNMENT TO SITE

o LIGHTNING PROTECTION

O GROUNDING COUNTER POISE

• POWER CIRCUIT PROTECTION

O DATA CIRCUIT PROTECTION

o ELECTRICAL POWER QUALITY AND QUANTITY

MARTIN MARIETT

• TEST REQUIREMENTS

O EMI PROTECTION

• WIRE ROUTING

B CONN - PLUG & RECEP INTERFACE WITH (4)MS 1/WI7 ENVRONMENTAL 2 AZ L STRAIN RELIEF-ELECTRICAL FACIL ITIES D CONN - PLUG · SPLICE (a)UNNT SW SYMBOLS: 2 EL PWR/ MANUAL CONT DATA (HC-27PS) (HFC-47PS) HELIOSTAT - CONCEPTUAL CABLE DESIGN E) MOTOR AZ $\widehat{\mathcal{E}}$ Ľ MOTOR アラ $\exists_{(3,7)}^{(2)}$ ENCODER $||_{(\mathcal{B})}^{(\mathcal{B})}$ Ċ アイ ENCODER HC/HFC $\widehat{(g)}$ Q AZ CONCEPT NO. GROUND BASELINE Q

HARDWARE - CONCEPT #1

o SWITCHES (ENVIRONMENTAL) - 4 EACH - MICROSWITCH P/N 1SE1

o SWITCH ROLLER - 4 EACH - MICROSWITCH P/N JE5

o STRAIN RELIEF - ENVIRONMENTAL - 2 EACH

o CABLES

O CABLE A

o CONNECTOR, 2 PLUGS - 11 CONTACTS

• CABLE - 11 #20 PVC INSULATION WITH TPR JACKET

o 13 FEET

o CABLE B (HC)

o CONNECTOR, 2 PLUGS - 6 CONTACTS

o CABLE - 2TPS #20, PVC INSULATION WITH TPR JACKET

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o 13 FEET

O CABLE B (HFC)

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o CONNECTOR, 2 PLUGS - 12 CONTACTS

HARDWARE - CONCEPT #1 (CONT)

o CABLE - 4 TPS #20, PVC INSULATION WITH TPR JACKET

o 13 FEET

O CABLE C

O CONNECTOR - 1 PLUG - 11 CONTACTS

1 RECEPT - 25 CONTACTS

o CABLE - 27 #20, 6 TPS #20, PVC INSULATION, TPR JACKET

o 15 FEET

O CABLE D

O CONNECTOR - 2 PLUGS - 3 CONTACTS

1 PLUG - 11 CONTACTS

1 PLUG - 25 CONTACTS

o CABLE - 16 #20, 3 TPS #20, PVC INSULATION, TPR JACKET

o 15 FEET

MARTIN MARIETTA CORPORATION HARDWARE MATING TO CABLES

o CONNECTORS - 2 RECEP - 3 CONTACTS (MOTORS)



HARDWARE - CONCEPT #1 (CONT)

3 RECEP - 11 CONTACTS (ENCODERS/PWR TO HC)

1 RECEP - 6 CONTACTS (HC DATA)/(12 CONTACTS (HFC)

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MISCELLANEOUS HARDWARE - TAPE, CLAMPS, STRING TIE, BOLTS, NUTS, WASHERS, STUDS (POWER DRIVEN), SPLICES, TOOLING, BONDING STRAPS, TERMINALS, FERRULES, ETC.



INTERFACE WITH MS LIVYIZ • SPLICE \$ STRAIN RELEIF -ENVIRONMENTAL ELECTRICH FACILITIES 252 D CONN-PLUG 2 AZ CINIS SIV HELIOSTAT - CONCEPTUAL CABLE DESKAN SYMBOLS: MANUAL CONT (F) (HFC-4 TPS)] DATA (HC-2TPS) MOTOR POWER/ 27 NOTOR J Z ENCODER Ð 73 ENCODER HC/HFC (\mathcal{B}) АĦ GROUND CONCEPT 2

HARDWARE - CONCEPT #2

• SWITCHES - SAME AS CONCEPT #1

c STRAIN RELIEF - ENVIRONMENTAL - 3 TO 6 EA

o CABLES

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O CABLE A - SAME AS CONCEPT #1

O CABLE B (HC) - SAME AS CONCEPT #1

O CABLE B (HFC) - SAME AS CONCEPT #1

O CABLE E

O CONNECTOR - 1 PLUG - 8 CONTACTS

• CABLE - 8 #20, PVC INSULATION, TPR JACKET

o 30 FEET

MARTIN MARIETTA HARDWARE MATING TO CABLES

O CONNECTORS - PAIRS - HC/HFC TO MOTORS/ENCODERS

o 2 PAIR - 11 CONTACTS - ENCODERS

o 2 PAIR - 3 CONTACTS - MOTORS

MISCELLANEOUS HARDWARE - BASICALLY THE SAME AS CONCEPT #1



HELIOSTAT - CONCEPTUAL CABLE DESIGN



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HFC SOFTWARE FUNCTIONAL FLOW DIAGRAM



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HFC SUPERVISORY CONTROL EXECUTIVE

INPUTS

- O COMMANDS FROM THE HAC VIA THE HFC-HAC COMMUNICATIONS HANDLER COMPONENT.
- o TIMER INTERRUPT.

OUTPUTS

O ACTIVATION CONTROL OF OTHER HFC COMPONENTS.

PROCESSING

- O RECEIVE AND INTERPRET COMMANDS FROM THE HAC.
- CONTROL ACTIVATION SEQUENCE AND OPERATIONS OF THE HFC HELIOSTAT COMMANDING, HFC HELIOSTAT STATUS ACQUISITION, HFC HELIOSTAT GROUP STATUS MONITOR, AND HFC INITIALIZATION COMPONENTS.
- O MAINTAIN THE HFC SUN VECTOR WATCH-DOG TIMER.
- o COMMAND HELIOSTAT STOWAGE UPON HAC COMMUNICATIONS LOSS OR SUN VECTOR TIMEOUT.

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HFC-HAC COMMUNICATIONS HANDLER

INPUTS

- DATA BYTES RECEIVED FROM HAC.
- DATA MESSAGES TO BE OUTPUT TO HAC.
- O RECEIVER BUFFER FULL AND TRANMITTER BUFFER EMPTY INTERRUPTS.

OUTPUTS

- DATA BYTES TRANSMITTED TO HAC.
- O DATA MESSAGES RECEIVED FROM HAC.
- O ACTIVATION OF THE HFC SUPERVISORY CONTROL EXECUTIVE COMPONENT.

PROCESSING

- o RECEIVE INPUT BYTES AND STORE THEM INTO MESSAGE BUFFER.
- ACTIVATE THE HFC SUPERVISORY CONTROL EXECUTIVE COMPONENT WHEN INPUT BYTE COUNT EXHAUSTED.

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- ACCUMULATE INPUT CHECKSUM AND VERIFY.
- TRANSMIT OUTPUT BYTES FROM MESSAGE OUTPUT BUFFER WHEN ACTIVATED BY THE HFC HELIOSTAT GROUP STATUS MONITOR COMPONENT.
- O ACCUMULATE AND OUTPUT CHECKSUM AS LAST MESSAGE BYTE.
- O MAINTAIN MESSAGE SYNC BY UTILIZING A WATCHDOG TIMER ON EACH INPUT BYTE.

HFC HELIOSTAT COMMANDING

INPUTS

- ACTIVATION BY THE HFC SUPERVISORY CONTROL EXECUTIVE COMPONENT.
- o COMMANDS FROM THE HAC.
- O HC STATUS FROM THE DATA-BASE.
- O POINTING PARAMETERS (SUCH AS CLLP) FROM THE HFC DATA-BASE.

OUTPUTS

O REFORMATTED COMMANDS TO THE HC'S.

PROCESSING

- OUTPUT THE SUN VECTOR/SYNC MESSAGE TO THE HC'S UPON RECEIPT OF THE SUN VECTOR FROM THE HAC. REFORMATTED MODE COMMANDS FROM THE PREVIOUS SECOND ARE OUTPUT AT THIS TIME.
- REFORMAT HAC MODE COMMANDS INTO ACCEPTABLE HC FORMAT AND APPLY DATA-BASE PARAMETERS (SUCH AS CORRIDOR LIMIT POINTS) TO THE COMMAND WHERE APPLICABLE.
- PERFORM CORRIDOR WALK CONTROL BY APPLYING \triangle Z TO CURRENT CORRIDOR WALK TARGET.

MARTIN MARIETTA

• UPON RECEIPT OF "GO-UP" OR "GO-DOWN" COMMAND, ENABLE CORRECT HELIOSTATS INTO CORRIDOR WALK MODE.

HFC HELIOSTAT STATUS ACQUISITION

INPUTS

- O ACTIVATION BY THE HFC SUPERVISORY CONTROL EXECUTIVE COMPONENT.
- STATUS MESSAGES FROM THE HC'S VIA THE HFC-HC COMMUNICATIONS HANDLER COMPONENT.

OUTPUTS

- STATUS POLL COMMANDS TO THE HC'S VIA THE HFC-HC COMMUNICATIONS HANDLER COMPONENT.
- o STATUS DATA TO THE HFC DATA-BASE.

PROCESSING

- O POLL AND ACQUIRE HC STATUS AND STORE INTO THE HFC DATA-BASE.
- O APPLY A WATCHDOG TIMER TO STATUS RESPONSES TO DETECT NON-RESPONDING HC'S.

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HFC HELIOSTAT GROUP STATUS MONITOR

INPUTS

- ACTIVATION BY THE HFC SUPERVISORY CONTROL EXECUTIVE COMPONENT IN RESPONSE TO A HAC POLL OF THE HFC.
- O STATUS DATA FROM THE HFC DATA-BASE.

OUTPUTS

• STATUS RESPONSE MESSAGES FOR THE HAC.

PROCESSING

- O REFORMAT DATA FROM THE HC STATUS IN THE HFC DATA-BASE INTO THE HAC FORMAT.
- OUTPUT HELIOSTAT GROUP STATUS DATA (WITH HFC OPERATIONAL STATUS) TO THE HAC VIA THE HFC-HAC COMMUNICATIONS HANDLER COMPONENT.



HFC-HC COMMUNICATIONS HANDLER

INPUTS

- o DATA BYTES RECEIVED FROM HC.
- DATA MESSAGES TO BE OUTPUT TO HC.
- O RECEIVER BUFFER FULL AND TRANSMITTER BUFFER EMPTY INTERRUPTS.

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OUTPUTS

- O DATA BYTES TRANSMITTED TO HC.
- O DATA MESSAGES FROM HC.
- REACTIVATION OF THE HFC HELIOSTAT STATUS ACQUISITION AND HFC HELIOSTAT COMMANDING COMPONENTS.

PROCESSING

- o OUTPUT COMMAND BYTES FROM COMMAND MESSAGE BUFFER.
- o ACCUMULATE AND OUTPUT MESSAGE CHECKSUM AS FINAL BYTE.
- RECEIVE INPUT BYTES AND STORE THEM INTO MESSAGE BUFFER.
- o ACCUMULATE INPUT CHECKSUM AND VERIFY.
- REACTIVATE REQUESTING COMPONENT WHEN I/O BYTE COUNT IS EXHAUSTED OR IF BYTE TIMEOUT ON INPUT.
- MAINTAIN MESSAGE SYNC BY UTILIZING THE MC6801 P'S "WAKE-UP" FEATURE AND A WATCHDOG TIMER ON EACH INPUT BYTE.

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HFC TIMER INTERRUPT HANDLER

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INPUTS

• TOF & TOC INTERRUPTS

OUTPUTS

- HAC I/O WATCHDOG TIMER TIMEOUT.
- o HC I/O WATCHDOG TIMER TIMEOUT.
- o SUN VECTOR WATCHDOG TIMER TIMEOUT.
- O SYSTEM CLOCK.

PROCESSING

- MAINTAIN WATCHDOG TIMERS OF 833 µs GRANULARITY FOR THE HFC-HAC COMMUNICATIONS HANDLER AND HFC-HC COMMUNICATIONS HANDLER COMPONENTS. DRIVEN FROM THE TOC (TIMER OUTPUT CAPTURE) INTERRUPT OF THE MC6801 µP.
- o MAINTAIN A SYSTEM CLOCK AND A SUN VECTOR WATCHDOG TIMER, BOTH DRIVEN FROM THE TOF (TIMER OVERFLOW) INTERRUPT OF THE MC6801µP. 50 ms GRANULARITY.



INPUTS

- POWER-ON INTERRUPT.
- O INITIALIZATION COMMAND FROM THE HFC SUPERVISORY CONTROL EXECUTIVE COMPONENT.
- o HFC ADDRESS.

OUTPUTS

- o DATA-BASE INITIAL VALUES.
- o STOW COMMANDS.

PROCESSING

- o PERFORM POWER-UP INITIALIZATION INCLUDING SENSING THE HFC'S ADDRESS.
- PERFORM COMMANDED INITIALIZATION WITH NEW DATA-BASE.
- O COMMAND ALL HC'S ATTACHED TO THIS HFC TO ELEVATION STOW FIRST, THEN AZIMUTH STOW.
- DETERMINE ENCODER ZERO POINT REFERENCES TO INITIALIZE THE AZIMUTH AND ELEVATION POSITIONS.
- SUPPRESS 3 ABOVE IF HAC COMMUNICATIONS (SUN VECTOR) IS RESUMED IN A SPECIFIED
 AMOUNT OF TIME.



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HC SUPERVISORY CONTROL EXECUTIVE

INPUTS

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O COMMANDS FROM THE HFC VIA THE HC-HFC COMMUNICATIONS HANDLER COMPONENT.

- o SUN VECTOR WATCH DOG TIMER.
- O STATUS DATA FROM THE HC DATA BASE.

OUTPUTS

- STATUS DATA TO THE HC-HFC COMMUNICATIONS HANDLER COMPONENT.
- ACTIVATION OF THE HC BEAM POINTING ALGORITHM AND HC INITIALIZATION COMPONENTS.
- O COMMAND AZIMUTH AND ELEVATION TO THE HC HELIOSTAT MONITOR CONTROL COMPONENT.

PROCESSING

- o INTERPRET AND PERFORM INPUT HFC COMMANDS.
- ACTIVATE THE HC BEAM POINTING ALGORITHM COMPONENT WHEN NECESSARY TO CALCULATE AZIMUTH AND ELEVATION.
- O PREPARE THE STATUS RESPONSE TO THE HFC.
- O MONITOR THE SUN VECTOR WATCHDOG TIMER.
- ACTIVATE THE HC INITIALIZATION COMPONENT WHEN COMMANDED OR WHEN THE SUN VECTOR WATCHDOG TIMES OUT.



HC-HFC COMMUNICATIONS HANDLER

INPUTS

- O DATA BYTES FROM THE SERIAL I/O INTERFACE .
- RECEIVER BUFFER FULL INTERRUPT.
- INPUT WATCHDOG TIMER INTERRUPT.
- O TRANSMITTER BUFFER EMPTY INTERRUPT.
- O STATUS DATA FROM THE HC SUPERVISORY CONTROL EXECUTIVE COMPONENT.

OUTPUTS

- INPUT MESSAGES TO THE HC SUPERVISORY CONTROL EXECUTIVE COMPONENT.
- O DATA BYTES TO THE SERIAL OUTPUT INTERFACE.
- O ACTIVATION OF THE HC SUPERVISORY CONTROL EXECUTIVE COMPONENT.

PROCESSING

- O RECEIVE AND STORE INPUT BYTES FROM THE SERIAL I/O INTERFACE HARDWARE.
- ACCUMULATE THE MESSAGE CHECKSUM AND VERIFY.
- ACTIVATE THE HC SUPERVISORY CONTROL EXECUTIVE COMPONENT WHEN THE REQUESTED BYTE COUNT IS EXHAUSTED.
- O OUTPUT THE CURRENT STATUS WHEN POLLED BY THE HFC.
- O DETECT AN EARLY END OF MESSAGE BY A WATCHDOG TIMER TIMEOUT TO MAINTAIN MESSAGE SYNC.
- SET THE SERIAL 1/O WAKE-UP FUNCTION WHEN THE REQUESTED BYTE COUNT IS EXHAUSTED TO DETECT THE START OF THE NEXT MESSAGE.
- OUTPUT ANY DATA REQUESTED BY THE HC SUPERVISORY CONTROL EXECUTIVE COMPONENT.



HC BEAM POINTING ALGORITHM

INPUTS

- H, THE HELIOSTAT COORDINATES IN SCALED 16 BIT INTEGER.
- T, THE DESIRED TARGET COORDINATES IN SCALED 16 BIT INTEGER.
- \$, THE SUN UNIT VECTOR IN SCALED 16 BIT INTEGER.
- O RAZ, REL REQUESTED AZIMUTH AND ELEVATION TO WHICH BIASES ARE TO BE ADDED.

OUTPUTS

• CAZ, CEL - CALCULATED AZIMUTH AND ELEVATION ACCORDING TO THE FOLLOWING FORMULAS:

$$\hat{B} = (\overline{T} - \overline{H}) / |\overline{T} - \overline{H}| \qquad [REFLECTED BEAM VECTOR] \\ \hat{\nabla} = (\hat{B} + \hat{S}) / |\hat{B} + \hat{S}| \qquad [HELIOSTAT NORMAL VECTOR] \\ CAZ = TAN^{-1} (V_x / V_y) \\ CEL = TAN^{-1} (V_z / (V_x^2 + V_y^2)^{1/2})$$

• CMDAZ, CMDEL - COMMANDED AZIMUTH AND ELEVATION RESULTING FROM ADDING THE ENCODER BIASES TO THE CALCULATED AZIMUTH AND ELEVATION, CAZ/CEL OR TO AN ABSOLUTE AZ/EL REQUESTED BY THE HFC. THE HC SUPERVISORY CONTROL EXECUTIVE COMPONENT DETERMINES WHICH AZIMUTH AND ELEVATION PAIR.

PROCESSING

- CALCULATE CAZ AND CEL (CALCULATED AZIMUTH AND ELEVATION) FROM THE HELIOSTAT'S POSI-TION COORDINATES (H), THE CURRENT TARGET COORDINATES (T), AND THE SUN VECTOR (S).
- PRODUCE THE CMDAZ AND CMDEL (COMMAND AZIMUTH AND ELEVATION) FROM THE REQUESTED AZ/EL (OUTPUT OF BEAM POINTING CALCULATION OR ABSOLUTE AZ/EL COMMAND FROM HFC) BY ADDING THE AZIMUTH AND ELEVATION ENCODER BIASES.

HC HELIOSTAT MONITOR/CONTROL

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INPUTS

- AC WAVEFORM ZERO CROSSING DETECTOR INTERRUPT 120H_.
- AZIMUTH AND ELEVATION ZERO POINT ENCODER MARKS.
- COMMAND AZIMUTH AND ELEVATION.
- AZIMUTH AND ELEVATION ENCODES DELTAS 4-BITS EACH.

OUTPUTS

- MOTOR COMMANDS 15H_.
- AZIMUTH, ELEVATION, AND GIMBAL STATUS.
- SOFTWARE WATCHDOG TIMER.

PROCESSING

- DRIVE AZIMUTH AND ELEVATION MOTORS TO ZERO THE DIFFERENCE BETWEEN THE COMMANDED AND CURRENT GIMBAL POSITIONS.
- DETERMINE NOTOR RATE (SLEW/SLOW) TO USE TO ACCOMPLISH THE ABOVE.
- O CHANGE MOTOR COMMANDS ONLY AT THE ZERO CROSSINGS OF THE AC WAVEFORM.
- PERIODICALLY ($\sim 15H_z$) UPDATE THE AZIMUTH AND ELEVATION POSITIONS BY THE DELTAS OF THE ENCODERS. HANDLE > $\pm 270^{\circ}$ FOR AZIMUTH.
- ZERO OUT THE AZIMUTH AND ELEVATION POSITIONS WHEN THEIR RESPECTIVE ZERO MARKS ARE ENCOUNTERED ON THE ENCODER.
- CALCULATE THE DIRECTION AND MAGNITUDE OF THE AZIMUTH AND ELEVATION DELTAS IN SUPPORT OF 1 AND 2 ABOVE. HANDLE $>+ 270^{\circ}$ FOR AZIMUTH.
- o STORE AZIMUTH, ELEVATION, AND GIMBAL STATUS FOR OUTPUT TO THE HFC.
- O UPDATE A SOFTWARE WATCHDOG TIMER FROM THE 120Hz ZERO CROSSING DETECTOR INPUT.

MARTIN MARIETT

O MONITOR MOVEMENT RATES FOR ERRORS AND DETERMINE HELIOSTAT FAULTS.

HC INITIALIZATION

INPUTS

- o POWER-ON INTERRUPT.
- HC ADDRESS.
- INITIALIZATION DATA FROM HAC VIA HFC.

OUTPUTS

- PROGRAM INITIALIZATION AND START UP.
- HARDWARE (µP) INITIALIZATION.
- o STOW COMMANDS TO MOTOR CONTROLLER.

PROCESSING

- PERFORM POWER-UP INITIALIZATION INCLUDING SENSING THE HELIOSTAT'S ADDRESS.
- o PERFORM COMMANDED INITIALIZATION WITH NEW DATA-BASE.
- O COMMAND THE HELIOSTAT TO ELEVATION STOW FIRST AND THEN AZIMUTH STOW.
- FIND ENCODER ZERO POINT REFERENCES TO INITIALIZE THE AZIMUTH AND ELEVATION POSITIONS.
- SUPPRESS 3 ABOVE IF HFC COMMUNICATIONS (SUN VECTOR) IS RESUMED IN A SPECIFIED AMOUNT OF TIME.



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Department of Energy San Francisco Operations Office 1333 Broadway Oakland, California 94612

Reply to:

DOE Site Office P.O. Box 366 Daggett, CA 92327 ATTN: S. D. Elliott, Jr.

Melvin W. Frohardt Martin Marietta Aerospace P.O. Box 179 Denver, CO 80201

JUN 2 5 1983

Subj.: Request for patent clearance and TIC Distribution of Dòcuments from DOE Contracts ET21007 and SF10539 (Solar One Heliostats, Phases I & II)

Dear Mel:

We are about to come out, with the help of EPRI, with a bibliography of key Project documents. To cope with anticipated requests for copies, I would like to arrange for properly cleared documents to be filed with and distributed through the DOE Technical Information Center at Oak Ridge. A check with TIC shows that only MCR-80-1377 has been cleared by them to date. Can you provide me with signed-off Patent Clearance Requests for:

o The five indicated documents from the Phase I study (ET21007);

o The twelve indicated documents from Phase II (SF10539);

o The as-built drawing set provided via Sandia at the end of Phase II;

o Any other Project documents generated by MMC you think the utility/industry community should have ?

I'd also appreciate a check on the Phase I CDR handout; was it MCR-78-1325?

Your help is greatly appreciated; it will save me (and you) a lot of running about once the bibliography comes out. I will insure that you get a copy; it lists about 500 documents, not including drawings (these we will provide to TIC in aperture card form at a later date, with a full index). Please call me ((619) 254-2672/-2142) if you have any questions or concerns.

Sincerely yours,

S. D. Elliott, Jr. DOE Project Director



Department of Energy San Francisco Operations Office 1333 Broadway Oakland, California 94612

Reply to:

DOE Site Office Post Office Box 366 Daggett, CA 92327

Mr. Melvin T. Frohardt Martin Marietta Aerospace Post Office Box 179 Denver, CO 80201

DEC 0 6 1983

Subj.: Closeout Actions on Martin Marietta Contracts with DOE San Francisco Operations Office

Dear Mel:

Nearly six months ago, I requested your assistance in finalizing patent clearance on a number of the documents from the Collector Phase I and Phase II contracts which we wish to enter into the DOE Technical Information Center system. Thus far, I have not had any response to this request. We are about to issue the bibliography developed by Burns & McDonnell under the EPRI-funded "Lessons Learned and Project Documentation" study (I assume you have received a copy of Vol. 1, "Lessons Learned" - if not, let me know and I will send you one), and we and TIC anticipate a substantial number of requests for key documents, including yours.

In addition, SAN Contracts Closeout (Sonia Jackson) advises me that several of the final documents needed to complete closeout (and release final payment of withheld funds), is as yet lacking, not only on the above two contracts, but also on the old Preliminary Design contract. I would greatly appreciate your assistance (or your guidance as to who can assist us) in getting this wrapped up and off both of our desks. To recapitulate (adding the items needed by SAN) for the three contracts:

DE-ACO3-76ET20422 (Old Contract -1110), Central Receiver System Prel. Design:

- o A "Final Invoice", to be submitted to Sonia Jackson, with copy to me;
- o "Contractors Assignment of Refunds and Rebates", to Sonia;
- o "Contractors Release", to Sonia;
- o "Contractor Request for Patent Clearance" (send to me, only), for:
 - MCR-77-161, "System Safety Design Criteria for Central Receiver...System",
 - MCR-77-162, "System Safety Program Requirements for Solar Thermal Systems".

(These were done under an extension to the Preliminary Design contract, and are valuable background documents.)

DE-AC03-78ET21007 Collector System, Phase I:

- o "Final Invoice", to Sonia, copy to me;
- o "Assignment of Funds and Rebates", to Sonia;

Page 2

- o "Contractors Release", to Sonia;
- o "Contractor Request for Patent Clearance", to me, for:
 - MCR-78-1323, "10-MWe Solar Thermal Pilot Plant Conceptual Design Review";
 - MTR-78-1330, "10-MWe Solar Thermal Pilot Plant Preliminary Design Review";
 - MCR-79-1302, "10-MWe Solar Thermal Pilot Plant Final Design Review (2 Vols.)";
 - 40-0-500-4P, "10-MWe Solar Thermal Pilot Plant Phase II 0&M Equipment";
 - 40-0-500-6P, "10-MWe Solar Thermal Pilot Plant Phase II Planning."

DE-AC03-80SF10539, Collector System Phase II

- o "Final Invoice", to Sonia, copy to me;
- o "Assignment of Funds and Rebates", to Sonia;
- o "Contractors Release", to Sonia;
- o "Contractor Request for Patent Clearance", to me, for:
 - MCR-79-1352B; "Quality Assurance Plan for 10-MWe Phase II Collector..";
 - MCR-80-1304, "10-MWe Solar Pilot Plant Collector Subsystem Safety Plan";
 - MCR-81-1331B, "Hazard Analysis for 10-MWe ...Pilot Plant";
 - 40-0-500-2P, "10-MWe ... Pilot Plant Phase II Mfg. Plan, Rev. 2";
 - MCR-80-1341A, "10-MWe Collector Sybsystem Software/Firmware Functional Req'ts.";
 - MCR-80-1362, "System Description Document, Collector Subsystem...";
 - MCR-80-1376; "Heliostat Stimulator Operators' Manual";
 - MCR-81-1708, "Operation Instructions, Heliostat Field Subsystem...";
 - MCR-81-1709A, "Maintenance Instructions, Heliostat Field Subsystem...";
 - MTR-81-1769, "...Collector Subsystem Functional Test Report";
 - MCR-81-1770, "Supplemental Spares Plan, Heliostat Field...";
 - MCR-80-1377A, "Software/Firmware Design Specifications...";
 - MCR-82-1701, "Control System Theory of Operation";
 - Drawing Set, as Identified in "Drawing Tree 400500 5132701";
 - Source Listing of Code for Heliostat Controller ROM or EPROM*
 - Source Listing of Code for Heliostat Field Controller ROM/EPROM*

Our files do not have current copies of the following other items identified in the Drawing Tree (400500 5132701):

Documents: 40M500-2S, "Foundation Req'ts.", 40M500-1T, "Installation Instructions", 40M500-2M, "Canting Procedures", 40M500-5P, "Acceptance Plan", MCR-80-1361, "Collector System Functional Test Plan", and MCR-81-1715, "Collector System Integrated Acceptance Test Plan."

* Current copies of these four items are lacking from the Project files; your assistance in obtaining at least one copy of each will be most appreciated. Drawings: 40M500 5132788, "Adapter Plate/Control Arm Heat Tool", 40M500 5132771, "Field Canting Tool", and 40E500 5132776, "Drive Unit Checkout Console".

While these items are not carried in the current version of the Bibliography (none of the Plant as-built drawings have been entered as yet), many, if not all, of them may be expected to be of interest to the solar community. I would appreciate at least one copy of each, again with your release. To save you considerable effort in preparing the Patent Clearance Request forms (I am enclosing several copies of the form), you may combine many of the above by simply clearing the "Drawing Tree", with its contents.

If you need the other closeout forms cited above (your Contract Administration staff should have them in stock), please call Sonia Jackson at FTS 536-4179, or write her at:

Ms. Sonia Jackson (CM) Department of Energy 1333 Broadway Oakland, CA 94612

Finally, since we are required to forward two clean, reproducible copies of each document to DOE/TIC, as well as needing one clean copy for our on-site archives, any "extras" you can turn up around your offices would be greatly appreciated; certainly, rather than throw anything of possible interest out, send it to me.

Mel, I know (believe me!) that this is all a significant amount of work, and I wish I didn't have to ask you (or your staff) to go through it, but it will be to our mutual benefit in the end to get these three contracts all cleaned up, and a comprehensive package of Project documentation (currently, over 550 documents, plus drawings) into the archives. If there is anything further I can do to assist you in this effort, please call on me.

Encl.: DOE Proj. Ofc. 1tr. 6/25/83 Patent Clearance Req. Forms

Sonia Jackson, DOE/SAN (CM)

cc: H. C. Wroton, MMC

Sincerely yours,

3

S. D. Elliott, Jr., Director, DOE Project Office, Barstow

PS: I keep running across references to a document I can't identify: MCR-78-1325; what was it? **FIN MARIETTA AEROSPACE**

DENVER DIVISION POST OFFICE BOX 179 DENVER, COLORADO 80201 TELEPHONE (303) 977-3000

January 30, 1984

Mr. Doug Elliott DOE Site Office Post Office Box 366 Daggett, CA 92327

Subject: Closeout Actions on Martin Marietta Contracts with DOE San Francisco Operations Office

Reference: Letter of December 06, 1983, S.D. Elliott, Jr. to M. Frohardt, Closeout of Contracts

In regard to the referenced letter, following is the status and actions in process to close out these items:

1. Contract Closeout Status

In regard to the closeout of cost type contracts DE-AC03-76ET20422, Central Receiver Test Facility, and DE-AC03-78ET21007, Collector System Phase I, we include the "Contractors Assignment of Refunds and Rebates" and "Contractors Release" with our final invoice package. The final invoices for these two contracts will be submitted upon completion of final settlement negotiations for our 1979 overhead and G&A rates which is currently in progress. In reference to the closeout of contract DE-AC03-80SF10539, Collector System Phase II, please see Attachment 1, the letter to Ms. Joann Littlehales dated January 23, 1984, for the current status.

2. Patent Clearance

The following documents are in the process of being cleared by our Patent office. When this transmittal is available, I will send a copy to you.

MCR-78-1323, "10-MWe Solar Thermal Pilot Plant Conceptual Design Review"
MCR-78-1330, "10-MWe Solar Thermal Pilot Plant Preliminary Design Review"
MCR-79-1302, "10-MWe Solar Thermal Pilot Plant Final Design Review (2 Vols)"
40-0-500-4P, "10-MWe Solar Thermal Pilot Plant Phase II 0&M Equipment"
40-0-500-6P, "10-MWe Solar Thermal Pilot Plant Phase II Planning"

ريدة ود ميخية
Mr. Doug. Elliott January 27, 1984 Page 2

1.

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The remainder of the documents have been previously cleared by the following letters, copies of which are included in Attachment 2.

Letters from Phillip DeArment to Roger Gaither:

DAC-83-417, dated May 24, 1983 80-Y-15555, dated July 28, 1980 DAC-82-389, dated May 3, 1982 Letter dated March 11, 1982 Letter dated November 10, 1982

3. Documents

You requested copies of some documents and drawings in the referenced letter. Copies of the following drawings and documents are being submitted under Attachment 3.

MCR-78-1330, "Preliminary Design Review Package"
MCR-79-1352B, "Quality Assurance Plan for 10-MWe Phase II Collector"
MCR-80-1376, "Heliostat Stimulator Operators' Manual"
40M500-2S, "Foundation Requirements"
40M500-2M, "Canting Procedures"
40M500-1T, "Installation Instructions"
40M500-5P, "Acceptance Plan"
MCR-81-1715, "Collector System Integrated Acceptance Test Plan"
MCR-80-1361, "Collector System Functional Test Plan"
40M500 5132788, "Adapter Plate/Control Arm Heat Tool"
40M500 5132771, "Field Canting Tool"
Source Listing of Code for Heliostat Controller ROM or EPROM*

No drawing exists for 40E500 5132776, "Drive Unit Checkout Console" as this checkout console consisted of a stimulator to operate a production Drive Mechanism Assembly. Also MCR-78-1330 is the correct document number for the Preliminary Design Review Package rather than MCR-78-1325. MCR-78-1325 is the document number assigned to all the Monthly Progress Reports written during the Phase I contract.

Doug, I hope this will help in getting the documentation finalized. I will follow-up with the additional information identified. If you have any questions please call on me.

Sincerely yours,

MARTIN MARIETTA CORPORATION

Melvin W. Frohardt Solar Programs

Enclosures

cc: H. Wroton Sonia Jackson IN MARIETTA AEROSPACE

DENVER DIVISION POST OFFICE BOX 179 DENVER. COLORADO 80201 TELEPHONE (303) 977-3000

March 13, 1984

Mr. Doug Elliott DOE Site Office Post Office Box 366 Daggett, CA 92327

ET21007

Subject: Closeout Actions on Martin Marietta Contracts with DOE San Francisco Operations Office

Reference: Letter of January 30, 1984, M. W. Frohardt to S. D. Elliott, Jr., Closeout of Contracts

As per Item 2 of the referenced letter, attached is a copy of a letter from our Patent Counsel, Phillip L. DeArment, to the Department of Energy Patent Counsel. This letter includes the Patent Certification for the documents in question.

It is my belief that this completes the information you have requested from us except for the final closeout of the contracts which is being handled by our Contracts department.

Please contact me if you need more information.

Sincerely yours,

MARTIN MARIETTA DENVER AEROSPACE

ma Protecto

M. W. Frohardt Solar Systems

cc: H. Wroton C. Bolton N MARIETTA AEROSPACE

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DENVER AEROSPACE POST OFFICE BOX 179 DENVER, COLORADO 80201 TELEPHONE (303) 977-6008

OFFICE OF CHIEF COUNSEL

8 March 1984

Refer to: DAC-84-0211

То:

United States Department of Energy P. O. Box 808 Livermore, California 94550

Attn: Assistant Chief for Prosecution Office of Patent Counsel, L-376

Subj: Contract DE-ACO3-79ET21007 Revised Final Patent Certification

1. Attached is a revised Patent Certification on the subject contract.

2. If you have any questions, please contact Mr. Melvin W. Frchardt at (303) 977-0123.

Very truly yours,

MARTIN MARIETTA CORPORATION

Phillip L. De an

Phillip L. DeArment Associate Patent Counsel

PLD:jes

The following is a complete list of technical reports prepared during the course 1. of the work under this contract and the DOE office to which the reports were sent:

SEE ATTACHED SHEET

Technical data of this contract other than reports (i.e., notebooks, drawings, 2. etc.) are completely listed, as follows:

NONE

Each of the above-listed documents under paragraphs 1 and 2 has been examined for 3. invention subject matter by me and/or technical personnel under my direction; to the best of my knowledge and belief, no inventions or discoveries were made or conceived in the course of or under this contract other than the following:

CONTRACTOR NO. TITLE

DATE REPORTED DOE NO.

NONE

There were no subcontracts or purchase orders involving research and development, 4.

NONE

- The completion date of this contract is as follows: 5. 12/13/79
- The following period is covered by this certification: 9/5/78 12/31/79 6.

September 5, 1978				,, ,
Month Day	Year t	o December Month	31, 1979 Dav	Vean
Martin Marietta Corpora Contractor Denver Aerosp P. O. Box 179	ace	Phillip &	1. De arm	<u>int</u>
Denver, CO 80201 Address		<u>Associate I</u> Title	<u>Patent Counsel</u>	, .
Submit in duplicate to:				
Roger S. Gaither Assistant Chief for Prosecution California Patent Group, L-376 U. S. Department of Energy P. O. Box 808				

Livermore, California 94550

Date of Certification

ATTACHMENT

 MCR-79-1310-10 MWe Software/Firmware Functioning Segments Spec.
 MCR-79-1311-10 MWe Overall Plant Design MCR-79-1352-10 MWe Quality Assurance Plan for Phase II
 SENT TO: United States Department of Energy San Francisco Operations Office Solar Ten Megawatt Project Office 9550 Flair Drive, Suite 210 El Monte, California 91731
 MCR-78-1323, "10-MWe Solar Thermal Pilot Plant Conceptual Design Review";

MTR-78-1330, "10-MWe Solar Thermal Pilot Plant Preliminary Design Review";

MCR-79-1302, "10-MWe Solar Thermal Pilot Plant Final Design Review (2 Vols.)";

40-0-500-4P, "10 MWe Solar Thermal Pilot Plant Phase II O&M Equipment";

40-0-500-6P, "10-MWe Solar Thermal Pilot Plant Phase II Planning"; and

MCR-78-1325 "10-MWe Solar Thermal Pilot Plant Phase I Monthly Progress Reports."

DOE F 1325.8 (7-79)

DATE

U.S. DEPARTMENT OF ENERGY

MAY 0 8 1984

memorandum



S. D. Elliott, Jr., Director, DOE Solar One Project Office

SUBJECT: Submission of Five Reports under Contract DE-AC03-78ET21007 with Martin Marietta Aerospace Corporation

ro: Roger S. Gaither, DOE/SAN Office of Patent Counsel William D. Matheny, DOE/TIC Document Control

Enclosed are five documents prepared by the Martin Marietta Corporation, Denver Aerospace Division, for the Solar Ten-Megawatt Project Office, under Contract DE-AC03-78ET21007:

Primary Document No.	Secondary No.	Brief Title
DOE/ET/21007-1	(STMP0-283)	10 MWeConceptual Design Review
DOE/ET/21007-2	(STMPO-284)	10 MWePreliminary Design Review
DOE/ET/21007-3	(STMP0-285)	10 MWeFinal Design Review
DOE/ET/21007-4	(STMP0-286)	10 MWePhase II Operating & Mainten- ance Equipment
D0E/ET/21007-5	(STMP0-287)	10 MWePhase II Planning

One copy of each document, accompanied by a completed SAN Form 70, is provided to SAN/OPC for patent review and clearance; the Form 70's have been prepared by this office based upon the Patent Certification provided by the attached letter of March 13, 1984 from Martin Marietta. Please return the feedback copies of the Form 70's to this office; the documents may be returned to Mr. Mike Lopez at SAN/FGS.

Two copies of each document, accompanied by a completed DOE Form RA-426, are provided for archiving and announcement by the DOE Technical Information Center, and forwarding to the National Technical Information Service.

This action completes all documentation requirements under Contract DE-AC03-78 ET21007.

S. D. Elliott, Jr., Director, DOE Solar One Project Office

Attch.: Martin Marietta ltr. 3/13/84

Encls.: 5 Documents, w/transmittal forms

cc: Mike Lopez, DOE/SAN (FGS)
Don Holz, DOE/SAN (ISEA)
Sonia Jackson, DOE-SAN (CM)
Mary Soderstrum, Burns & McDonnell

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	SAN FRANCISCO OPERATIONS OFFICE	······································
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E CONTRACTOR	CONTRACTOR REQUEST FOR PATENT CLEARANCE	DE-AC03-78FT21007
	FOR RELEASE OF UNCLASSIFIED DOCUMENT	Subcontract No.
Boger S.G	Baither, Asst. Chief for Prosecution	
Office of P	Patent Counsel/Livermore Office	(N/A)
P.O. Box 8	808, L-376	Report No.
Livermore,	, California 94550	DOC/CT/21007 1 (STMD0 202)
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Post Off	fice Box 366	October, 1978
Daggett,	, CA 92327	Name & Phone No. of DOE
		Technical Representative
		(5. U. EIII0TT, Jr.)
1 Document	t Title:	L_(019)_254-2072
"10-MWe	Solar Thermal Central Receiver Pilot Plant:	: Conceptual Design Review"
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DOE OFFICE OF PATENT COUNSEL (OPC)

DOE Form RA-426 (10/80)

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U.S. DEPARTMENT OF ENERGY

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OMB NO. 038-R0190

DOE AND MAJOR CONTRACTOR RECOMMENDATIONS FOR ANNOUNCEMENT AND DISTRIBUTION OF DOCUMENTS

See Instructions on Reverse Side

1.	DOE Report No. DOE/ET/21007-1	L (STMPO-	-283)	2. Contract DE-AC	№. 03-78E1	r21007	3.	Subject Cate UC-62	gory No.	<u></u>
4.	Title "10 MWe SOLAR	THERMAL	CENTRAL	RECEIVER	PILOT	PLANT:	CONCEPTIAL	DESIGN	REVIEW	 10/16/78'
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	b. Copies being tra	nsmitted for sp	ecial distribu	tion per attach	ed comple	te address lis	st.			
	C. Two completely	legible, reprod	ucible copies	being transmi	tted to DO	E-TIC. (Cla	ssified documents,	see instructio	ns)	
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	Has an invention disclo	sure been subm	itted to DOE	covering any	aspect of th	his informat	ion product? 🛛 N	lo 🗌 Yes		
	If so, identify the D	OE (or other) of	disclosure nur	nber and to wi	hom the di	sclosure was	submitted.			
	Are there any patent-re	lated objection	s to the releas	se of this infor	mation pro	oduct?	No 🗌 Yes If so	o, state these	objections.	
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Orga	nization Post 0	ffice Box	366, Da	aggett, C	A_ 923	27	(619) 254-2	672		
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): Roger S. Ga	aither, Asst. Chief for Prosecution	Subcondact No.
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P.O. Box 80	08, L-376	Report No.
Livermore,	California 94550 \sim	DOE/FT/21007-1 (STMP0-283)
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n an the second s		Technical Representative
		S. D. Elliott. Jr.
1. Document	Title:	(619) 254-2672
"10-Mile !	Solar Thermal Central Receiver Pilot Plar	nt: Conceptual Design Review"
2. Type of Do	Conv of Oral Presentation Other (nlease	\Box Journal Article, \Box Adstract of Summary, specify):
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