

**10 MWe Solar Thermal  
Central Receiver Pilot Plant  
Solar Facilities Design Integration**

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**PSS FINAL DESIGN CALCULATIONS  
BOOK 15 OF 26--MAIN STEAM PIPING  
CONSTRUCTION PACKAGE 9 (RADL 7-8)**

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**September 1980**

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**STEARNS-ROGER ENGINEERING CORP  
4500 CHERRY DRIVE  
P.O. BOX 5888  
DENVER, CO 80217**

**PREPARED FOR THE  
U.S. DEPARTMENT OF ENERGY  
SOLAR ENERGY  
UNDER CONTRACT DE-AC-03-79SF10499**

## PREFACE

This document is provided by McDonnell Douglas Astronautics Company (MDAC) in accordance with Department of Energy Contract Number DE-AC03-79SF10499, Reports and Deliverables List (RADL Item 7-8). The report was prepared by Stearns-Roger Engineering Corporation under MDAC Subcontract Number 78012035.

The Plant Support Subsystem Final Design Calculations (RADL Item 7-8) are arranged in a twenty-six book volume as shown on the master Table of Contents.

Book 15 of this document is provided in support of the Mechanical Equipment Installation, Construction Package No. 9 and includes isometric drawings, weight, seismic, thermal and stress analysis for the main steam piping design.

Questions concerning this report should be directed to R.J. Perkins at (714) 896-3073.

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BOOK 4 - RECEIVER TOWER  
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Note: This document includes design calculation for the receiver tower steel (Construction Package 5A) which was previously submitted by MDAC letter A3-228-EP-RJP-46, dated 16 January 1980, and therefore, is not included in this submittal. Please transfer your copy to your RADL ITEM 7-8 file, marking it as BOOK 4 of 25.

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PAGE

BOOK 25 - COLLECTOR FIELD  
ELECTRICAL, CONSTRUCTION  
PACKAGE 11A

Note: This document was pre-  
viously submitted by MDAC  
Letter A3-228-EP-RJP-262,  
dated 7 March 1980 and  
therefore is not included  
in this submittal. Please  
transfer your copy to your  
RADL ITEM 7-8 file, mark-  
ing it as BOOK 25 of 25.

BOOK-26-MDAC GENERAL ANALYSIS AND  
BACKGROUND DATA

- 1 Plant Process and Preliminary Component Requirements
- 2 Receiver Subsystem Calculations
- 3 Thermal Storage Subsystems Calculations
- 4 Analysis of Plant Cost Reduction Options
- 5 Collector Field Design and Plant Power Calculations
- 6 Miscellaneous Plant Calculations

LINE MS-3 MAIN STEAM - DESUPER HEATER BRANCH

Stearns-Roger

# PIPE STRESS ANALYSIS REPORT

FOR

IOMWe SOLAR PILOT PLANT

C-21700

MAIN STEAM DESUPERHEATER BRANCH  
PIPING SYSTEM  
6"-MS-3-QEB

CONDITIONS ANALYZED / ANAL. I.D. NO.

THERMAL T-21700-MS-3-A-3  
DEAD WEIGHT W-21700-MS-3-A-3  
PRESSURE P-21700-MS-3-A-0  
SEISMIC X-21700-MS-3-A-1

CODE REFERENCE: ANSI B31.1-1973  
W/ APPENDA C.  
(SUMMER 1979)

ANALYST G.H. MAY DATE 7-25-80



DIVISION USAGE					
MM	P	PP	SH	FI	SP

**Stearns-Roger**  
INCORPORATED  
ENGINEERING STANDARD

STANDARD NUMBER  
EE16.01.2

**APPROVALS**  
Des. Sect. \_\_\_\_\_  
Sect. Supv. \_\_\_\_\_  
Div. \_\_\_\_\_

PIPING ANALYSIS RESULTS

PAGE 1 OF 1  
ISSUED 8-31-73  
REVISED 4-15-74

Date FEB. 28, 1980

TO: T.E. OLSON  
FROM: G. H. MAY, Piping Engineering Group  
Client DOE Project SOLAR I Job No. C-21700  
Pipe Line Analyzed MAIN STEAM DESUPERHEATER BRANCH / 6"-MS-3-QEB  
Reference Dwg. PB-2A

This piping has been analyzed for the THERMAL, DEAD WEIGHT & PRESSURE & OPERATIONAL BASIS EARTHQUAKE loading condition and is found to be:

- Satisfactory, as is.
- Satisfactory, with comments noted.
- Unsatisfactory - See Comments.

COMMENT (1) ALL PIPE STRESSES ARE SATISFACTORY.  
(2) ALL FORCES AND MOMENTS ARE REASONABLE AND SHOULD BE ACCEPTED.

RECOMMENDATION (1) USE THE REFERENCED ROUTING.  
(2) SUBMIT THE FORCES AND MOMENTS ON THE DESUPERHEATER TO ROCKETDYNE FOR THEIR APPROVAL.

ATTACHS: PIPE STRESS SUMMARY REPORT, SUMMARY OF FORCES & MOMENTS

G. H. May  
Signature

Copies to: Sender  
Analysis Folder - Job File  
Piping Engrg. Supervisor TCT  
K.E. NOBLE

DIVISION USAGE						<b>Stearns-Roger</b> INCORPORATED ENGINEERING STANDARD	STANDARD NUMBER
MM	P	PP	SH	FI	SP		EE16.01.8
APPROVALS						PIPE STRESS SUMMARY REPORT  ANSI B31.1 - 1973 EDITION	PAGE <u>1</u> OF <u>1</u>
Des. Sect. <u>SV 2nd</u>							ISSUED 5/27/75 REVISED
Sect. Supv. <u>J. H. May</u>							
Div. _____							

10MWE SOLAR PILOT PLANT - DOE  
Job Name

MAIN STEAM DESUPERHEATER BRANCH 16" MS-3-QEB  
System/Pipe Line Name

C-21700  
Job No.

MS-3  
Analysis No.

1. Loading Conditions Analyzed

Pressure X  
Weight X  
Thermal Expansion X  
Sustained Mech. Loads \_\_\_\_\_  
Occasional Loads X

Analysis Ident. Code

P-MS-3-A-0  
W-MS-3-A-3  
T-MS-3-A-3  
X-MS-3-A-1

2. Stress Evaluation (Code Equations)

Eq. (11)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} \leq 1.0 S_h$  3357 psi  $\leq$  10360 psi

Material: - ASTM A335 P22

PRESS = 1565 PSIG  $\Rightarrow$  SLP = 1885 PSI  
Temp = 960°F @ Pt. 370

Eq. (12)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} + \frac{0.75 iM_B}{z} \leq K S_h$  9374 psi  $\leq$  12430 psi

Material: - ASTM A335 P22

Temp = 960°F @ Pt. 400

Eq. (13)  $S_E = \frac{iM_C}{z} \leq S_A$  7550 psi  $\leq$  19200 psi

Material: - ASTM A335 P22

Temp = 960°F @ Pt. 360M

Eq. (14)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} + \frac{iM_C}{z} \leq (S_h + S_A)$  9560 psi  $\leq$  29560 psi

Material: - ASTM A335 P22

Temp = 960°F @ Pt. 360M

3. Stress Evaluation (Local or Special)

<u>Loading</u>	<u>Analysis Ident. Code</u>	<u>Calculated Stress (psi)</u>	<u>Allowable Stress (psi)</u>
_____	_____	_____	_____
_____	_____	_____	_____

REMARKS:

- (1) ALL PIPE STRESSES ARE LESS THAN THE ALLOWABLES.
- (2) EQ. 12 INCLUDES STRESSES FROM AN OPERATIONAL BASIS EARTHQUAKE BASED ON GROUND RESPONSE SPECTRUM.

J. H. May  
Prepared By

2-28-80  
Date

J. H. M  
Approved By

2-28-80  
Date



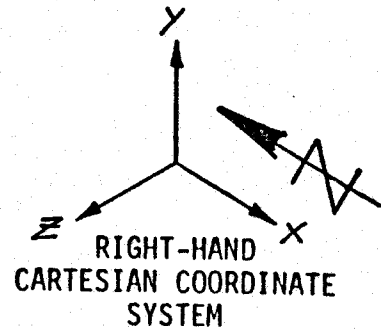
CUSTOMER: DOE  
 PROJECT: 10 MWE SOLAR PILOT PLANT  
 JOB NO: C-21700  
 BY: GHM DATE: FEB. 28, 1980  
 REF. DWGS: P13-2 A  
 ANALYSIS CODE: T/W/X-MS-3-A-3/3/1

SUMMARY OF FORCES & MOMENTS  
 ON SYSTEM TERMINAL EQUIPMENT  
 ( SYSTEM )

MAIN STEAM DESUPERHEATER BRANCH  
 6"-MS-3-QEB

DIVISION USAGE				
MM	P	PP	SH	FI
X				
APPROVALS				
Des. Sect.	[Signature]			
Spec. Superv.	[Signature]			
Div. [Signature]				

THE REPORTED REACTIONS BASED ON A THERMAL EXPANSION ANALYSIS FROM ...70°F TO 960°F USING E<sub>c</sub>, THE COLD MOD. OF ELASTICITY, AND .....0...% COLD SPRING.



**Stearns-Roger**  
 INCORPORATED  
 ENGINEERING STANDARD

SUMMARY OF FORCES & MOMENTS  
 PER. ANSI B31.1-1973 PWR. PIPING CODE

EQUIPMENT CONNECTIONS	LOC. NO.	FORCES (LBS)				MOMENTS (FT.-LBS)			
		X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
DESUPERHEATER (THERMAL)	400	-745	-399	-542	1004	1456	-6319	7248	9726
DS-301 STEAM INLET (WEIGHT)	400	-4	321	-3	321	-1148	-47	-119	1156
10" EAST 1I (SEISMIC)	400	901	345	1243	1574	6493	10402	7713	14487

STANDARD NUMBER  
 EE 16.01.7  
 PAGE 1 OF 1  
 ISSUED 2/28/74  
 REVISED 4/18/78

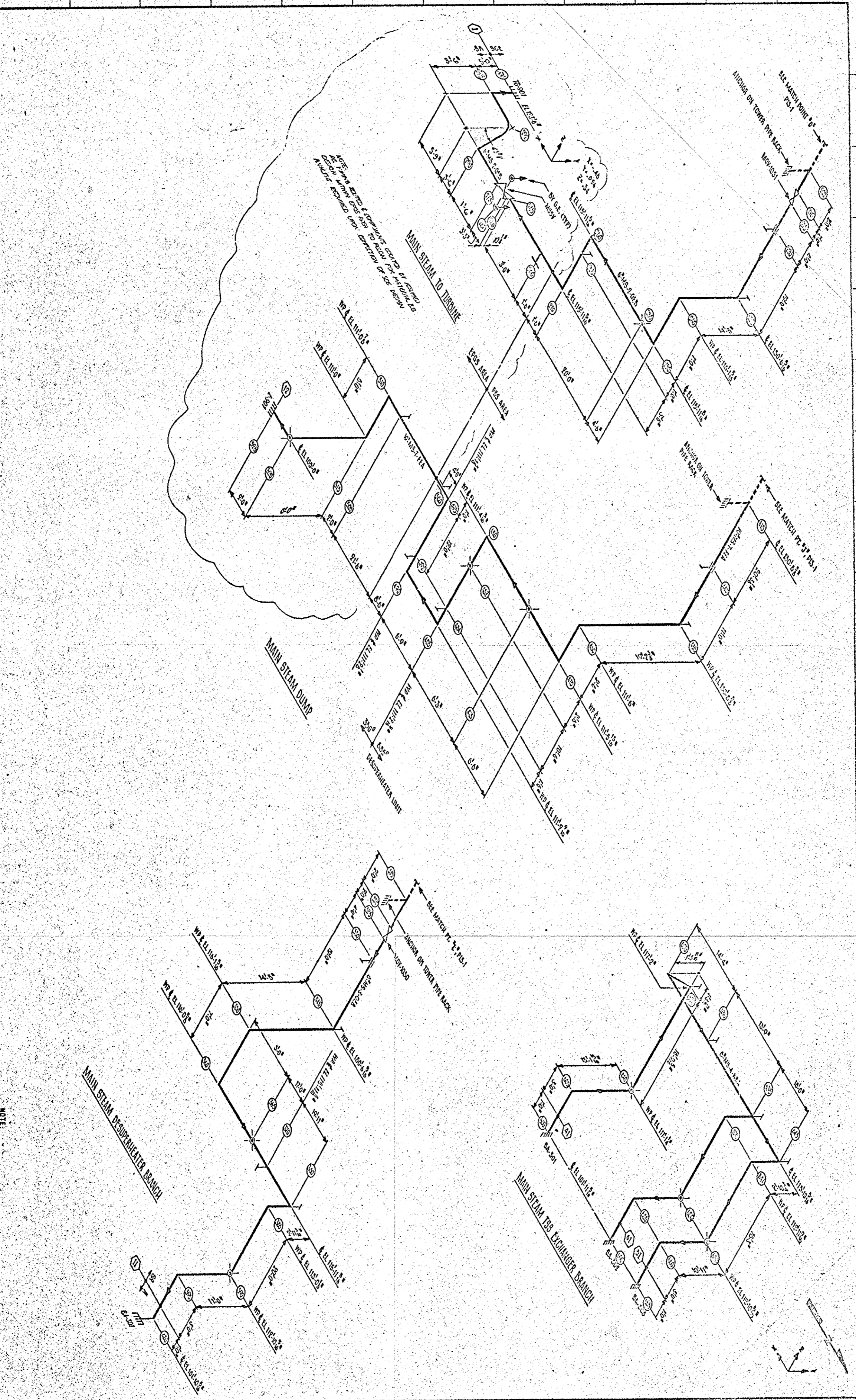


REVISIONS		DATE		BY		CHECKED		APPROVED		TITLES	
1	ISSUED FOR PERMIT	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00
2	REVISED PER CLIENT ANALYSIS DOCUMENTS	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00

SOLAR FACILITIES DESIGN INTEGRATION		DATE		BY		CHECKED		APPROVED		TITLES	
1	ISSUED FOR PERMIT	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00
2	REVISED PER CLIENT ANALYSIS DOCUMENTS	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00

1	ISSUED FOR PERMIT	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00
2	REVISED PER CLIENT ANALYSIS DOCUMENTS	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00

SOLAR FACILITIES DESIGN INTEGRATION		DATE		BY		CHECKED		APPROVED		TITLES	
1	ISSUED FOR PERMIT	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00	1/23/00
2	REVISED PER CLIENT ANALYSIS DOCUMENTS	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00	7/14/00







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FOREWORD

THE FOLLOWING REPORT WAS GENERATED BY THE DYNAPLEX PIPING ANALYSIS PROGRAM. THE PROGRAM ACCOMMODATES THE WIDEST RANGE OF PRACTICAL SITUATIONS ENCOUNTERED IN MODERN PIPING SYSTEMS INCLUDING DYNAMIC AS WELL AS THERMAL, WEIGHT AND PRESSURE EFFECTS.

FOR FURTHER INFORMATION OR PROMPT ASSISTANCE IN USING THE DYNAPLEX PROGRAM, PLEASE CONTACT YOUR LOCAL COMPUTER CENTER REPRESENTATIVE OR:

ENGINEERING SERVICES MANAGER  
AUTUM COMPUTING CORPORATION  
1 METRO PLAZA  
505 THORNTON STREET  
EDISON, N.J. 08817

THE STRESS FORMULATION OF  
ANSI B31.1 - 1977  
INCLUDING THE LATEST MANDATORY UPDATES  
HAS BEEN APPLIED IN THIS ANALYSIS

THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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SIFARNS - ROGER CORP

100% SOLAR PILOT PLANT - C-21700

MS VALVE TO THER STOR (T/W-MS-3-A-373)

2/25/80

DYNAPLEX (MOD204F-UCC)  
LAST UPDATE - 12/25/79

DYNAFLEX

STEARNS - ROGER CORP

PAGE 1

PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
JOB - PS VALVE TO INER STOR (T/W-PS-3-A-3/3)

DATE 2/25/80

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	PRE	ID	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
1	PROP								STEARNS - ROGER CORP, 2/25/80, PROJ. 10MW SO LAR PILOT PLANT - C-2 1700, JOBS VALVE TO I NER STOR (T/W-PS-3-A- 3/3)	1
2	GEN								APPLY B31.1-1973, EXA	2
3			515	320			-2-0		MAT=LCH, UC=6.625, WT=.864, UNIF=15.2, TEMP=960.	3
4				325			-2-0		RIGID, WEIGHT=1315.	4
5				330			-4-0			5
6				345			-19-0	L		6
7				350			-14-5	L		7
8				360			-7-0	L		8
9				365	-3-0					9
10				375	-17-0					10
11				380	-19-11			L		11
12				385			-2-11-3/16	L		12
13				390			-20-0	L		13
14				395			-11-0	L		14
15				400			-4-0			15
16	AFC	315								16
17	AFC	400								17
18	RAD	330			RIGID					18
19	RAD	330			RIGID					19
20	RAD	345			RIGID					20
21	RAD	370			RIGID					21
22	RAD	380			RIGID					22
23	PRG								LUC(365,390N), USE TABLE 1	23
24	CCC	400			-39-11		-24-4-3/16	-58-0		24

PROJECT - 10% SOLAR PILOT PLANT - C-21700  
 JOB - MS VALVE TO THER STOR (1/2-MS-3-A-3/3)

DATE 2/25/80

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*
*   EDITED PIPING SYSTEM DESCRIPTION
*   -----
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EXPLANATORY NOTES

IN THE EDITED PIPING SYSTEM DESCRIPTION:

- (1) POINT LOCATION NUMBER SUFFIXES N, M AND F IDENTIFY THE NEAR POINT, THE MID POINT AND THE FAR POINT RESPECTIVELY OF A PIPING ELEMENT IN THE DIRECTION IN WHICH THE LINE IS TRAVERSED. FRACTIONAL NODE NUMBERS APPEARING IN THE FOLLOWING REPORTS IDENTIFY THOSE NODES AT WHICH THE PROGRAM HAS INTRODUCED EVENLY SPACED LOOP POINTS AND ARE DETERMINED FROM THE MAXIMUM LOOP SPACING CRITERIA ENTERED BY THE USER. IF FRACTIONAL NODE NUMBERS ARE ASSIGNED TO MORE THAN ONE ELEMENT ORIGINATING AT THE SAME BRANCH POINT, A LETTER WILL PRECEDE THE FRACTIONAL PORTION OF THE NUMBER TO UNIQUELY IDENTIFY EACH POINT (E.G. 2.A1). LETTERS ARE ASSIGNED TO ELEMENTS ALPHABETICALLY IN THE ORDER OF THEIR APPEARANCE IN THE INPUT DATA.
- (2) AMBIENT TEMPERATURE IS ASSUMED TO BE 70 DEGREES FAHRENHEIT.
- (3) THE MAXIMUM ALLOWABLE DIFFERENCE IN LOOP CLOSURE CALCULATIONS IS ASSUMED TO BE .25 INCHES.
- (4) THE MAXIMUM ALLOWABLE DIFFERENCE BETWEEN THE CALCULATED AND INPUT COORDINATES OF USER DESIGNATED CHECK POINTS IS ASSUMED TO BE .25 INCHES.
- (8) IN ORDER TO MAKE IT POSSIBLE IN CERTAIN SITUATIONS TO PERFORM INDEPENDENT WEIGHT AND THERMAL ANALYSES IN THE SAME RUN AND APPLY THE CODE STRESS CALCULATIONS CORRECTLY, THE PROGRAM MAKES CERTAIN ASSUMPTIONS. IF THE INPUT DATA SPECIFIES IMPOSED MOVEMENTS ON ANCHORS (ANC) OR DISPLACEMENTS OR ROTATIONS ON RIGID RESTRAINTS (RAD,RAR) THESE ARE ASSUMED BY THE PROGRAM TO BE THERMAL EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE THERMAL. ALTERNATELY, IF THE INPUT DATA SPECIFIES FORCES AND MOMENTS (FOR,MOM) OR IMPOSED DISPLACEMENTS OR ROTATIONS ON FLEXIBLE RESTRAINTS (RAD,RAR) THESE ARE ASSIGNED TO BE WEIGHT EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE WEIGHT. IF THESE ASSUMPTIONS ARE INVALID FOR A PARTICULAR SITUATION, THE USER CAN ACCURATELY MODEL THE MOST GENERAL SITUATION BY USING THE AUXILIARY SUPPORT MOVEMENTS (ASM) AND THE AUXILIARY CONCENTRATED LOADS (ACL) FEATURE OF THE PROGRAM AS AN ALTERNATE METHOD OF SPECIFYING THESE EFFECTS.
- (9) A POINT IDENTIFIED AS A TENTATIVE HANGER LOCATION IS ASSUMED TO BE FREE TO MOVE IN THE VERTICAL DIRECTION FOR THE INITIAL THERMAL LOADING CASE AND FULLY RESTRAINED IN THE VERTICAL DIRECTION FOR THE INITIAL WEIGHT LOADING CASE.
- (15) THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.



GENERAL DATA

COORDINATES OF ALL POINTS OF THE SYSTEM ARE WITH RESPECT TO POINT 315, WHICH IS THE ORIGIN

HANGER DESIGN DATA:

THE MAXIMUM RIGID SUPPORT DISPLACEMENT CRITERION = .10 INCHES.  
 VARIABLE SPRING HANGERS SELECTED FROM TABLE C10

MATERIALS USED (TEMPERATURE DEPENDENT PROPERTIES ARE SHOWN AT POINT OF TEMPERATURE CHANGE)

LCM - LOW CARBON MILD STEEL  
 YOUNG'S MODULUS AT AMBIENT TEMPERATURE = 29,900,000. PSI  
 POISSON'S RATIO = .30  
 DENSITY = 500. POUNDS PER CUBIC FOOT

LOADING CONDITIONS ANALYZED

%FREE THERMAL  
 %RESTRAINED WEIGHT

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) FARR	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	WT. UNIF
ANCHR		--315	0.00	0.00	0.00								
	STRAIGHT	(					2.00		6.625	.864	LCM: 960.		54.3: 15.2:
	( * THERMAL EXPANSION OF LCM BETWEEN AMBIENT AND 960. DEGREES FARRERHEIT IS 3.46 INCHES PER 100 FEET												
	( * A UNIFORM DISTRIBUTED LOAD OF 15.20 PLF IS ACTING IN THE -Y DIRECTION.												
		>-320	0.00	0.00	-2.00								
	RIGID	(					2.00		N/A:	N/A:		N/A:	N/A:
	( * WEIGHT OF ELEMENT = 1315. POUNDS												
		>-325	0.00	0.00	-4.00								
	STRAIGHT	(					4.00						
		>-330	0.00	0.00	-6.00								

\* NOTES PERTAINING TO POINT 330 APPEAR ON THE FOLLOWING PAGE

ELIATED PIPELINE SYSTEM DESCRIPTION (CONTINUED)

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (CEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	WT. UNIF
		>330	0.00	0.00	-8.00				6.625	.864	LCM: 960.	54.3	15.2
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 330											
		( IN DIRECTION X											
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 330											
		( IN DIRECTION Y											
		(											
	STRAIGHT	(					16.25						
		(											
		>345N	0.00	0.00	-26.25								
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 345N											
		( IN DIRECTION Y											
		(											
		(											
	BEND	( 345N	0.00	-0.22	-26.78	.75	1.18	90.000					
		(											
		>345F	0.00	-0.75	-27.00								
	STRAIGHT	(					12.92						
		(											
		>350N	0.00	-13.67	-27.00								
15	BEND	( 350N	0.00	-14.20	-27.22	.75	1.18	90.000					
		(											
		>350F	0.00	-14.42	-27.75								
	STRAIGHT	(					5.50						
		(											
		>360N	0.00	-14.42	-33.25								
	BEND	( 360N	-0.22	-14.42	-33.78	.75	1.18	90.000					
		(											
		>360F	-0.75	-14.42	-34.00								
	STRAIGHT	(					2.25						
		(											
		>365	-3.00	-14.42	-34.00								
		(											
		( * TENTATIVE HANGER AT POINT 365 - SEE NOTE (9) ABOVE.											
		( HANGER LENGTH NOT SPECIFIED.											
		(											
	STRAIGHT	(					17.00						
		(											
		>370	-20.00	-14.42	-34.00								

\* NOTES PERTAINING TO POINT 370 APPEAR ON THE FOLLOWING PAGE

DYNAPLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 5

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
		>-370	-20.00	-14.42	-34.00				6.625	.664	LCM: 960.	54.3	15.2
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 370											
		( IN DIRECTION Y											
		(											
	STRAIGHT	- (					19.17						
		(											
		>-380N	-39.17	-14.42	-34.00								
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 380N											
		( IN DIRECTION Y											
		(											
		(											
	BEND	- (	380N	-39.70	-14.64	-34.00	.75	1.18	90.000				
		(											
		>-380F	-39.92	-15.17	-34.00								
	STRAIGHT	- (					1.43						
		(											
		>-385N	-39.92	-16.60	-34.00								
		(											
	BEND	- (	385N	-39.92	-17.13	-34.22	.75	1.18	90.000				
		(											
16		>-385F	-39.92	-17.35	-34.75								
	STRAIGHT	- (					18.50						
		(											
		>-390N	-39.92	-17.35	-53.25								
		(											
		( * TENTATIVE HANGER AT POINT 390N - SEE NOTE (9) ABOVE.											
		( HANGER LENGTH NOT SPECIFIED.											
		(											
		(											
	BEND	- (	390N	-39.92	-17.57	-53.78	.75	1.18	90.000				
		(											
		>-390F	-39.92	-18.10	-54.00								
	STRAIGHT	- (					9.50						
		(											
		>-395N	-39.92	-27.60	-54.00								
		(											
	BEND	- (	395N	-39.92	-28.13	-54.22	.75	1.18	90.000				
		(											
		>-395F	-39.92	-28.35	-54.75								
	STRAIGHT	- (					3.25						
		(											
	ANCHR	- (	400	-39.92	-28.35	-58.00							

\* NOTES PERTAINING TO POINT 400 APPEAR ON THE FOLLOWING PAGE

DYNAFLEX

EDITED PIPE SYSTEM DESCRIPTION (CONTINUED)

PAGE 6

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (FAHR)	DIST. LOADS (LBS PER FT)		
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF	
ANCHP		--400	-39.92	-28.35	-58.00				6.625	.864	LCM	960	54.3	15.2
* CONTROL COORDINATES OF POINT 400, IN FEET :														
X = -39.92, Y = -28.35, Z = -58.00														
RESULTANT DIFFERENCE IS ZERO														

DYNAFLEX

LOADING - FREEZE THERMAL

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\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM  
 ON THE ANCHORS AND RESTRAINTS.

LIC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
315	-1693.	-123.	542.	1782.	-307.	4240.	2693.	5033.
400	-745.	-399.	-542.	1004.	1456.	-6319.	7248.	9726.
<b>RESTRAINTS:</b>								
330	2439.	0.	0.	2439.	0.	0.	0.	0.
330	0.	258.	0.	258.	0.	0.	0.	0.
18 345	0.	568.	0.	568.	0.	0.	0.	0.
370	0.	-868.	0.	868.	0.	0.	0.	0.
380	0.	565.	0.	565.	0.	0.	0.	0.

\*\*\*\*\*  
 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 7550. OCCURS AT POINT 360M

EXPLANATORY NOTES:  
-----

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ANSI B31.1 - 1977  
PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (15) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE MODULUS IS USED FOR THE BRANCH LEG AT REDUCED OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK. THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNALLEX

LOADING - FREEZE THERMAL

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## INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	MEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	EXPANSION STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP		-- 315	542.	1698.	2693.	307.	4240.	1.00	1.00	20.02	3016.
	STR - (	>- 320	542.	1698.	2693.	62.	854.	1.00	1.00	20.02	1694.
	PIGID - (	>- 325	542.	1698.	2693.	183.	2533.	1.00	1.00	20.02	2219.
	STR - (	>- 330	542.	1698.	2693.	674.	9307.	1.00	1.00	20.02	5820.
	STR - (	>- 345N	542.	750.	2693.	1797.	4296.	1.00	1.00	20.02	3224.
	BEND - (	345N	880.	754.	5106.	2051.	1528.	1.00	1.00	20.02	3422.
	STR - (	>- 345F	703.	922.	4855.	1918.	2134.	1.00	1.00	20.02	3379.
	STR - (	>- 350N	703.	922.	4855.	7493.	5085.	1.00	1.00	20.02	6157.
	BEND - (	350N	880.	754.	2030.	5218.	9127.	1.00	1.00	20.02	6416.
	STR - (	>- 350F	542.	1025.	8052.	4965.	5410.	1.00	1.00	20.02	6532.
	STR - (	>- 360N	542.	1025.	8052.	1098.	9513.	1.00	1.00	20.02	7498. *
20	BEND - (	360N	910.	717.	5072.	9790.	6098.	1.00	1.00	20.02	7550. **
	STR - (	>- 360F	745.	888.	571.	9666.	7525.	1.00	1.00	20.02	7349. *
	STR - (	>- 365	745.	888.	571.	5944.	8446.	1.00	1.00	20.02	6198.
	STR - (	>- 370	745.	567.	571.	6007.	771.	1.00	1.00	20.02	3645.
	STR - (	>- 380N	745.	567.	571.	2837.	11163.	1.00	1.00	20.02	6911. *
	BEND - (	380N	809.	595.	7609.	2885.	8585.	1.00	1.00	20.02	7088. **
	STR - (	>- 380F	399.	922.	11569.	2578.	978.	1.00	1.00	20.02	7127. **
	STR - (	>- 385N	399.	922.	11569.	1510.	1754.	1.00	1.00	20.02	7070. **
	BEND - (	385N	666.	752.	7277.	1954.	8853.	1.00	1.00	20.02	6967. **
	STR - (	>- 385F	542.	246.	951.	1862.	11010.	1.00	1.00	20.02	6716.
	STR - (	>- 390N	542.	246.	951.	5524.	2779.	1.00	1.00	20.02	3749.

DYNALOX

LOADING - FREE THERMAL

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	EXPANSION STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
		>- 390N	542.	846.	951.	5524.	2779.	1.00	1.00	20.02	3749.
	BEND - (	390N	666.	752.	2801.	5617.	1688.	1.00	1.00	20.02	3895.
		>- 390F	399.	922.	3338.	392.	5417.	1.00	1.00	20.02	3820.
	STR - (	395N	399.	922.	3338.	6689.	266.	1.00	1.00	20.02	4482.
		>- 395N	399.	922.	3338.	6689.	266.	1.00	1.00	20.02	4482.
	BEND - (	395N	666.	752.	2533.	66.	7485.	1.00	1.00	20.02	4736.
		>- 395F	542.	846.	7248.	159.	3897.	1.00	1.00	20.02	4932.
	STR - (	400	542.	846.	7248.	1456.	6319.	1.00	1.00	20.02	5828.
	ANCHR	-- 400	542.	846.	7248.	1456.	6319.	1.00	1.00	20.02	5828.



\*\*\*\*\*  
 \* INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

EXPLANATORY NOTES

THE FORCES AND MOMENTS AT ANY POINT SHOWN BELOW REPRESENT THE ACTION OF THE FAR END OF THE BRANCH ACTING ON THE NEAR END OF THE BRANCH.

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
ANCHOR		-- 315	-1693.	-123.	542.	-307.	4240.	2693.
	STRAIGHT - (	>- 320	-1693.	-123.	542.	-62.	854.	2693.
	RIGID - (	>- 325	-1693.	-123.	542.	183.	-2533.	2693.
	STRAIGHT - (	>- 330	-1693.	-123.	542.	670.	-9307.	2693.
	STRAIGHT - (	>- 345H	745.	135.	542.	-1797.	4296.	2693.
	BEND - (	345H	745.	703.	542.	-2051.	4691.	2530.
	(	>- 345F	745.	703.	542.	-1918.	4855.	2130.
22	STRAIGHT - (	>- 3500	745.	703.	542.	5085.	4855.	-7493.
	BEND - (	3500	745.	703.	542.	5216.	5019.	-7889.
	(	>- 350F	745.	703.	542.	4965.	5414.	-8052.
	STRAIGHT - (	>- 3600	745.	703.	542.	1096.	9513.	-8052.
	BEND - (	3600	745.	703.	542.	726.	9790.	-7898.
	(	>- 360F	745.	703.	542.	571.	9666.	-7525.
	STRAIGHT - (	>- 365	745.	703.	542.	571.	8446.	-5944.
	STRAIGHT - (	>- 370	745.	-165.	542.	571.	-771.	6007.
	STRAIGHT - (	>- 370H	745.	-165.	542.	571.	-11163.	2837.
	(	370H	745.	399.	542.	690.	-11450.	2885.
	BEND - (	370H	745.	399.	542.	978.	-11569.	2578.
	(	>- 370F	745.	399.	542.			

LOADING - ZEPPEL THERMAL

INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES (CONTINUED)

NODE	ELEMENT	LOC.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
DESTINATION	OPERATION	NO.						
>=	380F		745.	399.	542.	978.	-11569.	2578.
STRAIGHT - (								
>=	3850		745.	399.	542.	1759.	-11569.	1510.
MEMO - (								
>=	3850		745.	399.	542.	1959.	-11406.	1115.
STRAIGHT - (								
>=	385F		745.	399.	542.	1862.	-11010.	951.
STRAIGHT - (								
>=	3900		745.	399.	542.	-5524.	2779.	951.
MEMO - (								
>=	3904		745.	399.	542.	-5617.	3174.	788.
STRAIGHT - (								
>=	390F		745.	399.	542.	-5417.	3338.	392.
STRAIGHT - (								
>=	3954		745.	399.	542.	-266.	3338.	-6689.
MEMO - (								
>=	395F		745.	399.	542.	-66.	3502.	-7084.
STRAIGHT - (								
>=	400		745.	399.	542.	-159.	3897.	-7248.
MEMO - (								
>=	400		745.	399.	542.	-1456.	6319.	-7248.

DYKAFLEX

LOADING - ZEROED THERMAL

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\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
 \*-----\*  
 \*  
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POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 315	-.000	-.000	.000	-.000	.000	.000
	STRAIGHT -(	>- 320	-.006	-.000	-.169	-.002	.021	.029
	RIGID -(	>- 325	-.015	-.001	-.338	-.002	.021	.029
	STRAIGHT -(	>- 330	.000	.000	-.677	.006	-.077	.088
	STRAIGHT -(	>- 345D	.990	.000	-2.220	-.037	-.268	.353
	BEAD -(	345A	1.035	-.023	-2.263	-.045	-.250	.361
	STRAIGHT -(	>- 345E	1.087	-.070	-2.276	-.054	-.235	.369
	STRAIGHT -(	>- 350H	2.007	-1.163	-2.159	.031	.104	.225
	BEAD -(	350B	2.025	-1.205	-2.182	.053	.124	.191
24	STRAIGHT -(	>- 350E	2.018	-1.217	-2.230	.075	.148	.162
	STRAIGHT -(	>- 360N	1.759	-1.081	-2.695	.145	.319	-.077
	BEAD -(	360B	1.703	-1.060	-2.724	.151	.361	-.105
	STRAIGHT -(	>- 360E	1.641	-1.039	-2.700	.156	.403	-.137
	STRAIGHT -(	>- 365	1.450	-.959	-2.489	.163	.488	-.200
	STRAIGHT -(	>- 370	.013	-.000	-.075	.216	.759	-.197
	STRAIGHT -(	>- 380H	-1.608	.000	2.295	.275	.283	.155
	BEAD -(	380B	-1.645	-.056	2.311	.281	.237	.168
	STRAIGHT -(	>- 380E	-1.645	-.089	2.290	.289	.197	.180
	STRAIGHT -(	>- 385H	-1.589	-.210	2.262	.297	.108	.192
	BEAD -(	385B	-1.571	-.241	2.150	.305	.068	.201

DYNALFX

LOADING - FREEZE THERMAL

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	BEND - (	3850	-1.571	-.241	2.150	.305	.068	.201
	(							
	>- 385F		-1.566	-.226	2.091	.313	.022	.209
	STRAIGHT - (							
	(							
	>- 3900		-.692	.899	.527	.172	-.295	.304
	BEND - (	3900	-.645	.899	.475	.148	-.283	.306
	(							
	>- 390F		-.598	.860	.441	.124	-.272	.307
	STRAIGHT - (							
	(							
	>- 3950		-.064	.057	.340	.012	-.101	.183
	BEND - (	3950	-.041	.012	.320	.011	-.087	.153
	(							
	>- 395F		-.026	-.005	.275	.011	-.069	.127
	STRAIGHT - (							
	(							
ANCHOR	--	400	-.000	-.000	-.000	.000	-.000	.000



\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
315	-12.	-886.	3.	886.	-845.	29.	29.	846.
400	-4.	321.	-3.	321.	-1148.	-47.	-119.	1156.
<b>RESTRAINTS:</b>								
330	16.	0.	0.	16.	0.	0.	0.	0.
330	0.	-1557.	0.	1557.	0.	0.	0.	0.
345	0.	-1835.	0.	1835.	0.	0.	0.	0.
365	0.	-938.	0.	938.	0.	0.	0.	0.
570	0.	-1432.	0.	1432.	0.	0.	0.	0.
580	0.	-1381.	0.	1381.	0.	0.	0.	0.
390	0.	-2117.	0.	2117.	0.	0.	0.	0.

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DYNAPLEX

LOADING - RESTRAINED WEIGHT

(+UNIF)

PAGE 16

\*\*\*\*\*  
 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 1472. OCCURS AT POINT 370

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ANSI B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (7) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP	STR - (	-- 315	3.	886.	29.	845.	29.	1.00	1.00	20.02	507.
	>-	320	3.	747.	29.	788.	6.	1.00	1.00	20.02	472.
RIGID - (	STR - (	>- 325	3.	599.	29.	936.	17.	1.00	1.00	20.02	561.
	>-	330	3.	681.	29.	2014.	63.	1.00	1.00	20.02	1208.
	STR - (	>- 345N	3.	588.	29.	1169.	16.	1.00	1.00	20.02	701.
	BEND - (	345F	851.	855.	33.	517.	7.	1.00	1.00	20.02	311.
	STR - (	>- 345F	1165.	5.	19.	26.	254.	1.00	1.00	20.02	153.
	>-	350N	267.	5.	19.	31.	219.	1.00	1.00	20.02	133.
	BEND - (	350N	158.	162.	9.	165.	38.	1.00	1.00	20.02	102.
	STR - (	>- 350F	3.	185.	34.	56.	23.	1.00	1.00	20.02	41.
	>-	360N	3.	197.	34.	46.	87.	1.00	1.00	20.02	63.
29	BEND - (	360N	5.	238.	132.	48.	154.	1.00	1.00	20.02	125.
	STR - (	>- 360F	4.	279.	257.	153.	48.	1.00	1.00	20.02	182.
	STR - (	>- 365	4.	503.	257.	956.	42.	1.00	1.00	20.02	594.
	STR - (	>- 370	4.	754.	257.	2403.	4.	1.00	1.00	20.02	1472. **
	STR - (	>- 380F	4.	803.	257.	760.	55.	1.00	1.00	20.02	482.
	BEND - (	380N	546.	542.	221.	343.	142.	1.00	1.00	20.02	259.
	STR - (	>- 380F	721.	5.	57.	176.	255.	1.00	1.00	20.02	189.
	>-	385N	622.	5.	57.	251.	170.	1.00	1.00	20.02	185.
	BEND - (	385N	409.	412.	79.	119.	158.	1.00	1.00	20.02	128.
	STR - (	>- 385F	3.	540.	167.	178.	54.	1.00	1.00	20.02	150.
	>-	390N	3.	746.	167.	1731.	27.	1.00	1.00	20.02	1042.



DYKAFLEX

LOADING - PRESTRAINED WEIGHT (+UNIF)

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELFM TYPE	LRC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL:	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		STRESS PSI	
		>- 390B	3.	746.	167.	1731.	27.	1.00	1.00	20.02	:	1042.
	BEND	-( 390M	939.	942.	138.	1013.	97.	1.00	1.00	20.02	:	615.
		>- 390F	1279.	5.	30.	164.	723.	1.00	1.00	20.02	:	444.
	STR	-( 395N	629.	5.	30.	697.	122.	1.00	1.00	20.02	:	425.
	BEND	-( 395R	414.	418.	107.	564.	63.	1.00	1.00	20.02	:	346.
		>- 395F	3.	547.	119.	263.	33.	1.00	1.00	20.02	:	174.
	STR	-( 400	3.	321.	119.	1148.	47.	1.00	1.00	20.02	:	693.
	ANCHP										:	

\*\*\*\*\*  
 \* INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES \*  
 \*\*\*\*\*

EXPLANATORY NOTES

THE FORCES AND MOMENTS AT ANY POINT SHOWN BELOW REPRESENT THE ACTION OF THE FAR END OF THE BRANCH ACTING ON THE NEAR END OF THE BRANCH.

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
31	ANCHOR	-- 315	-12.	-886.	3.	-845.	29.	29.
	STRAIGHT - (	>- 320	-12.	-747.	3.	788.	6.	29.
	RIGID - (	>- 325	-12.	599.	3.	936.	-17.	29.
	STRAIGHT - (	>- 330	4.	-680.	3.	-2014.	-63.	29.
	STRAIGHT - (	>- 345H	4.	588.	3.	-1169.	16.	29.
	BEND - (	345H	4.	-1206.	3.	-517.	18.	28.
	STRAIGHT - (	>- 345F	4.	-1165.	3.	-254.	19.	26.
	STRAIGHT - (	>- 350H	4.	-267.	3.	-219.	19.	-31.
	BEND - (	350H	4.	-226.	3.	-165.	20.	-33.
	STRAIGHT - (	>- 350F	4.	-185.	3.	-56.	23.	-34.
	STRAIGHT - (	>- 340H	4.	197.	3.	-87.	46.	-34.
	BEND - (	360H	4.	238.	3.	-202.	48.	15.
	STRAIGHT - (	>- 360F	4.	279.	3.	-257.	48.	153.
	STRAIGHT - (	>- 36H	4.	-503.	3.	-257.	42.	956.
	STRAIGHT - (	>- 370	4.	-754.	3.	-257.	-4.	2443.
STRAIGHT - (	>- 380H	4.	-803.	3.	-257.	-55.	760.	
BEND - (	380H	4.	-762.	3.	-257.	-56.	343.	
STRAIGHT - (	>- 380F	4.	-721.	3.	-255.	-57.	176.	

DYNAFLEX

LOADING = RESTRAINED WEIGHT (+UNIF)

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INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
	STRAIGHT - (	>- 380F	4.	-721.	3.	-255.	-57.	176.
		>- 385H	4.	-622.	3.	-251.	-57.	170.
	BEND - (	385H	4.	-581.	3.	-119.	-56.	168.
	STRAIGHT - (	>- 385F	4.	-540.	3.	178.	-54.	167.
		>- 390N	4.	746.	3.	-1731.	27.	167.
	BEND - (	390N	4.	-1330.	3.	-1013.	29.	166.
	STRAIGHT - (	>- 390F	4.	-1289.	3.	-723.	30.	164.
		>- 395H	4.	-629.	3.	-697.	30.	122.
	BEND - (	395H	4.	-588.	3.	-564.	31.	120.
	STRAIGHT - (	>- 395F	4.	-547.	3.	-263.	33.	119.
ANCHOR		-- 400	4.	-321.	3.	1148.	47.	119.

LEADING - RESTRAINED WEIGHT (+UNIF)

\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
 \* \*\*\*\*\*

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR	STRAIGHT - (	315	.000	.000	.000	.000	.000	.000
	>-	320	.000	.000	.000	.000	.000	.000
	>-	325	.000	.000	.000	.000	.000	.000
	STRAIGHT - (	330	.000	.000	.000	.007	-.001	.001
	STRAIGHT - (	3450	.007	.000	.000	.018	-.002	.004
	MEMB - (	3450	.008	.002	-.001	.015	-.002	.004
	>-	345F	.008	.002	-.002	.013	-.002	.004
	>-	3500	.020	.002	-.020	.000	-.001	.004
	MEMB - (	3500	.020	.002	-.020	.000	-.001	.004
33	>-	350F	.020	.002	-.020	.001	-.001	.004
	STRAIGHT - (	3600	.021	.002	-.020	.001	.000	.003
	MEMB - (	3600	.021	.002	-.020	.001	.000	.003
	>-	360F	.021	.002	-.020	.000	.001	.003
	STRAIGHT - (	365	.021	.000	-.020	.003	.001	.008
	STRAIGHT - (	370	.021	.000	-.013	.027	.002	.010
	STRAIGHT - (	3800	.021	.000	-.007	.053	.000	-.032
	MEMB - (	3800	.019	.003	-.005	.054	.000	-.030
	>-	380F	.016	.005	.001	.055	.000	-.029
	STRAIGHT - (	3850	.008	.005	.018	.057	-.001	-.028
	MEMB - (	3850	.005	.002	.024	.058	-.001	-.027

LOADING - RESIDUAL WEIGHT (LUNIF)  
 DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT IDENTIFICATION	ELEMENT NO.	LOC.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
MEMO - (	395F	(	.005	.002	.024	-.058	-.001	-.027
STRAIGHT - (	395F	(	.003	-.004	.027	-.058	-.001	-.027
	396J	>	.012	-.000	.027	.035	-.002	-.010
MEMO - (	390F	(	.012	.003	.026	.029	-.002	-.009
STRAIGHT - (	390F	>	.011	.005	.023	.026	-.002	-.009
	395F	(	.000	.005	-.001	-.002	-.001	-.003
MEMO - (	395G	(	-.000	.004	-.000	-.005	-.001	-.002
STRAIGHT - (	395F	>	-.000	.004	-.000	-.007	-.001	-.002
ANCHOR	400	--	-.000	.000	-.000	-.000	-.000	-.000

HANGER DESIGN DATAEXPLANATORY NOTES:

- (1) ALL MOVEMENTS SHOWN BELOW REFER TO THE INITIAL EXPANSION LOADING CASE WHEREIN TENTATIVE HANGER LOCATIONS HAVE NOT BEEN VERTICALLY RESTRAINED AND ALL WEIGHT EFFECTS HAVE BEEN SUPPRESSED.
- (2) THE WEIGHT REACTIONS SHOWN BELOW REFER TO THE WEIGHT LOADING CASE WHEREIN TENTATIVE HANGER LOCATIONS HAVE BEEN RIGIDLY RESTRAINED IN THE VERTICAL DIRECTION AND ALL THERMAL EFFECTS INCLUDING IMPOSED ANCHOR MOVEMENTS HAVE BEEN SUPPRESSED.
- (3) IN THE FOLLOWING TABLE VSH DENOTES A VARIABLE SPRING HANGER AND CSH DENOTES A CONSTANT SUPPORT HANGER.
- (4) THE SUPPORT LOADS IN THE INSTALLED CONDITION HAVE BEEN CALCULATED TO PRODUCE THE CORRECT WEIGHT BALANCE IN THE OPERATING CONDITION.
- (5) IF THE FREE VERTICAL MOVEMENT EXCEEDS THE MAXIMUM RIGID SUPPORT DISPLACEMENT CRITERION OF .10 IN., A VARIABLE SPRING IS SELECTED. IF A VARIABLE SPRING HANGER CANNOT BE FOUND WHICH SATISFIES BOTH THE LOAD VARIATION CRITERION SELECTED BY THE USER AND THE WORKING RANGE OF THE SPRINGS LISTED IN SPRING TABLE (1) OF THE USER'S MANUAL, A CONSTANT SUPPORT HANGER IS RECOMMENDED.
- (6) THE SO-CALLED "THEORETICAL" SPRING INSTALLATION LOAD SHOWN BELOW PRESUPPOSES THAT THE HANGER LOCATION IS RESTRAINED AGAINST VERTICAL MOVEMENT WHILE THE SPRING IS SET TO THE COLD LOAD.

DYNAFLEX

HANGER DESIGN DATA TABLE

HANGER LOC. NO.	SUPPORT LOAD (POUNDS)	FREE VERTICAL MOVEMENT (INCHES)	HORIZONTAL MOVEMENT (INCHES)		TYPE	PRELIMINARY HANGER SELECTION SPECIFICATION	SWING	
			X	Z			LENGTH (FEET)	ANGLE (DEG)
365	938.	-0.96	1.45	-2.49	VSH	1 - MIDDLE RANGE SPRING. SPRING RATE = 200 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 746 LBS SPRING LOAD IN THE OPERATING CONDITION = 938 LBS	N/A	N/A
300N	2117.	.90	-0.69	.53	VSH	1 - SHORT RANGE SPRING. SPRING RATE = 1200 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 3196 LBS SPRING LOAD IN THE OPERATING CONDITION = 2117 LBS	N/A	N/A

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DYNAFLEX

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 \*  
 \* STATIC SOLUTION ACCURACY CHECKS \*  
 \*  
 \*\*\*\*\*

EACH BASIC STATIC SOLUTION COMPRISING EACH LOAD OR LOADING COMBINATION  
 SPECIFIED IN THIS RUN HAS BEEN SUBJECTED TO EQUILIBRIUM AND  
 COMPATIBILITY CHECKS FOR ALL POINTS IN THE SYSTEM:

LOADING - CENTER THERMAL

-----  
 STATIC EQUILIBRIUM AND COMPATIBILITY HAVE BEEN SATISFIED FOR THIS  
 BASIC STATIC SOLUTION AT ALL POINTS IN THE SYSTEM.

LOADING - RESTRAINED WEIGHT (+UNIF)

-----  
 STATIC EQUILIBRIUM AND COMPATIBILITY HAVE BEEN SATISFIED FOR THIS  
 BASIC STATIC SOLUTION AT ALL POINTS IN THE SYSTEM.



DYNAPLEX

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DYNAFLEX

STEARNS - ROGER CORP

PAGE 1

PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
JOB - HS VALVE TO THER STOR (X-RS-3-A-1)

DATE 2/26/80

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	FRM	TO	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
1	PHD								STEARNS - ROGER CORP, 2/26/80, PROJ. 10MW SO LAR PILOT PLANT - C-2 1700, JOMMS VALVE TO T HER STOR (X-RS-3-A-1)	1
2	GLD								APPLY 531.1-1973, EXA	2
3			315	320			-2-0		MAT=LCH, DU=6.625, WT=.864, UNIF=15.2, TEMP=950.	3
4				325			-2-0		RIGID, HEIGHT=1315.	4
5				330			-4-0			5
6				335			-19-0	L		6
7				350		-14-5		L		7
8				360			-7-0	L		8
9				365	-4-0					9
10				370	-17-0					10
11				380	-19-11			L		11
12				385		-2-11-3/16		L		12
13				390			-20-0	L		13
14				395		-11-0		L		14
15				400			-4-0			15
16	ACC	415								16
17	ACC	400								17
18	ACC	330			RIGID					18
19	ACC	330			RIGID					19
20	ACC	345			RIGID					20
21	ACC	370			RIGID					21
22	ACC	380			RIGID					22
23	ACC	400			-4-0	-2-11-3/16	-58-0			23
24	BYL								SPEC 1(FRE/G, .17.013, .25/.CR2, 2.57.638, 9.7.531, 33.7.125), SPEC 2(FRE/G, .17.009, .25/.CR5, 2.57.638, 9.7.531, 33.7.125), SHOCK 1(X/1.0SPEC 1), SHOCK 2(Z/1.0SPEC 1), SHOCK 3(Y/1.0SPEC 2),	24

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FOREWORD

THE FOLLOWING REPORT WAS GENERATED BY THE DYNAFLEX PIPING ANALYSIS PROGRAM. THE PROGRAM ACCOMMODATES THE WIDEST RANGE OF PRACTICAL SITUATIONS ENCOUNTERED IN MODERN PIPING SYSTEMS INCLUDING DYNAMIC AS WELL AS THERMAL, WEIGHT AND PRESSURE EFFECTS.

FOR FURTHER INFORMATION OR PROMPT ASSISTANCE IN USING THE DYNAFLEX PROGRAM, PLEASE CONTACT YOUR LOCAL COMPUTER CENTER REPRESENTATIVE OR:

ENGINEERING SERVICES MANAGER  
 AUBURN COMPUTING CORPORATION  
 1 METRO PLAZA  
 505 THORNBALL STREET  
 EDISON, N.J. 08917

THE STRESS FORMULATION OF  
 ANSI B31.1 - 1977  
 INCLUDING THE LATEST MANDATORY UPDATES  
 HAS BEEN APPLIED IN THIS ANALYSIS

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THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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*****
*
*
*
*          STEARNS - ROGER CORP
*          -----
*
*          TOWN SOLAR PLOT PLANT - C-21700
*
*          BS VALVE TO THER STOR (X-RS-3-A-1)
*
*
*
*
*
*
*          2/26/80
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DYNAFLEX (MOD204F-000)

LAST UPDATE - 12/25/79

INPUT DATA - (CARD IMAGES)

1	:	0	:	7	:	10	:	13	:	25	:	37	:	49	:	60	:								
LINE NUMBER	:	TYPE	:	LOC	:	FREQ	:	DELTA	:	X	:	DELTA	:	Y	:	DELTA	:	Z	:	RADIUS	:	ADDITIONAL-DATA	:	LINE NUMBER	
25	:	060	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:	:
	:		:		:		:		:		:		:		:		:		:		:		:		:
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DYNALLEX

STEARNS - ROGER CORP

PAGE 3

PROJECT - TCRW SOLAR PILOT PLANT - C-21700  
JOB - MS VALVE THER STOR (X-MS-3-A-1)

DATE 2/26/80

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\*  
\* EDITED PIPING SYSTEM DESCRIPTION \*  
\*-----\*  
\*  
\*\*\*\*\*

EXPLANATORY NOTESIN THE EDITED PIPING SYSTEM DESCRIPTION:

- (1) POINT LOCATION NUMBER SUFFIXES N, M AND F IDENTIFY THE NEAR POINT, THE MID POINT AND THE FAR POINT RESPECTIVELY OF A PIPING ELEMENT IN THE DIRECTION IN WHICH THE LINE IS TRAVERSED. FRACTIONAL NODE NUMBERS APPEARING IN THE FOLLOWING REPORTS IDENTIFY THOSE NODES AT WHICH THE PROGRAM HAS INTRODUCED EVENLY SPACED LUMP POINTS AND ARE DETERMINED FROM THE MAXIMUM LUMP SPACING CRITERIA ENTERED BY THE USER. IF FRACTIONAL NODE NUMBERS ARE ASSIGNED TO MORE THAN ONE ELEMENT ORIGINATING AT THE SAME BRANCH POINT, A LETTER WILL PRECEDE THE FRACTIONAL PORTION OF THE NUMBER TO UNIQUELY IDENTIFY EACH POINT (E.G. 2.A1). LETTERS ARE ASSIGNED TO ELEMENTS ALPHABETICALLY IN THE ORDER OF THEIR APPEARANCE IN THE INPUT DATA.
- (2) AMBIENT TEMPERATURE IS ASSUMED TO BE 70 DEGREES FAHRENHEIT.
- (3) THE MAXIMUM ALLOWABLE DIFFERENCE IN LUMP CLOSURE CALCULATIONS IS ASSUMED TO BE .25 INCHES.
- (4) THE MAXIMUM ALLOWABLE DIFFERENCE BETWEEN THE CALCULATED AND INPUT COORDINATES OF USER DESIGNATED CHECK POINTS IS ASSUMED TO BE .25 INCHES.
- (5) IN ORDER TO MAKE IT POSSIBLE IN CERTAIN SITUATIONS TO PERFORM INDEPENDENT WEIGHT AND THERMAL ANALYSES IN THE SAME RUN AND APPLY THE CODE STRESS CALCULATIONS CORRECTLY, THE PROGRAM MAKES CERTAIN ASSUMPTIONS. IF THE INPUT DATA SPECIFIES IMPOSED MOVEMENTS ON ACCORDS (ASC) OR DISPLACEMENTS OR ROTATIONS ON RIGID RESTRAINTS (RAD, RAR) THESE ARE ASSUMED BY THE PROGRAM TO BE THERMAL EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE THERMAL. ALTERNATELY, IF THE INPUT DATA SPECIFIES FORCES AND MOMENTS (FOR, MOM) OR IMPOSED DISPLACEMENTS OR ROTATIONS ON FLEXIBLE RESTRAINTS (RAD, RAR) THESE ARE ASSUMED TO BE WEIGHT EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE WEIGHT. IF THESE ASSUMPTIONS ARE INVALID FOR A PARTICULAR SITUATION, THE USER CAN ACCURATELY MODEL THE MOST GENERAL SITUATION BY USING THE AUXILIARY SUPPORT MOVEMENTS (ASM) AND THE AUXILIARY CONCENTRATED LOADS (ACL) FEATURE OF THE PROGRAM AS AN ALTERNATE METHOD OF SPECIFYING THESE EFFECTS.
- (6) THE TABLE OF RESULTS APPEARS AT THE END OF THIS REPORT.

GENERAL DATA

COORDINATES OF ALL POINTS OF THE SYSTEM ARE WITH RESPECT TO POINT 315, WHICH IS THE ORIGIN

DYNAFLEX

EDITED INPUT SYSTEM DESCRIPTION (CONTINUED)

PAGE 4

MATERIALS USED (TEMPERATURE DEPENDENT PROPERTIES ARE SHOWN AT POINT OF TEMPERATURE CHANGE)

LCR - LOW CARBON MILD STEEL  
 YOUNG'S MODULUS AT AMBIENT TEMPERATURE = 29,900,000. PSI  
 POISSON'S RATIO = .30  
 DENSITY = 500. POUNDS PER CUBIC FOOT

SHOCK AND SPECTRUM DATA

DOME CUTOFF SET AT 20

FREQUENCY CUTOFF SET AT 33.00 CPS

SPECTRUM 1		SPECTRUM 2	
(LOGARITHMIC INTERPOLATION)		(LOGARITHMIC INTERPOLATION)	
FREQUENCY	G	FREQUENCY	G
---CPS---	-----	---CPS---	-----
.10	.01	.10	.01
.25	.08	.25	.06
2.50	.64	2.50	.64
9.00	.53	9.00	.53
33.00	.13	33.00	.13

SHOCK SPECIFICATIONS

43

SHOCK 1 CONSISTS OF :  
 100.00 PERCENT OF SPECTRUM 1 IN THE X DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING  
 THE CLOSELY SPACED FREQUENCY (CSF) METHOD.  
 CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND  
 THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

SHOCK 2 CONSISTS OF :  
 100.00 PERCENT OF SPECTRUM 1 IN THE Z DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING  
 THE CLOSELY SPACED FREQUENCY (CSF) METHOD.  
 CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND  
 THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

SHOCK 3 CONSISTS OF :

100.00 PERCENT OF SPECTRUM 2 IN THE Y DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

SHOCK 4 CONSISTS OF :

100.00 PERCENT OF SPECTRUM 1 IN THE X DIRECTION AND  
 100.00 PERCENT OF SPECTRUM 2 IN THE Y DIRECTION AND  
 100.00 PERCENT OF SPECTRUM 1 IN THE Z DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

LOADING CONDITIONS ANALYZED

- SHOCK 1
- SHOCK 2
- SHOCK 3
- SHOCK 4

44

DIST. LOADS (LBS PER FT)

POINT TYPE	FLUENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE	WALL	TEMP	PRESS PIPE	
			DIAMETER (IN)	THICK (IN)	MATL (FAHR)				(PSI)	WT. UNIF			
ANCHP		--315	0.00	0.00	0.00								
	STRAIGHT	(					2.00	6.625	.864	LCM: 960.	54.3	15.2	
		(											
		(											
		(											
		(											
		(											
		(											
		(											
		>-320	0.00	0.00	-2.00								

\* NOTES PERTAINING TO POINT 320 APPEAR ON THE FOLLOWING PAGE



DYNAMIC

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 6

POINT TYPE	ELEMENT DESIGNATION	POINT NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	MATERIAL	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z								PRESS (PSI)	PIPE WT. UNIF
:	:	:	:	:	:	:	:	6.625	.864	LCM	960	:	54.3	15.2
:	>-320	:	0.00	0.00	-2.00	:	:	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
	RIGID	-	:	:	:	2.00	:	N/A	N/A	:	:	:	K/A	N/A
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(* HEIGHT OF ELEMENT = 1315. POUNDS	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
:	>-325	:	0.00	0.00	-4.00	:	4.00	:	:	:	:	:	:	:
	STRAIGHT	-	:	:	:	:	:	:	:	:	:	:	:	:
:	>-330	:	0.00	0.00	-8.00	:	:	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 330	:	:	:	:	:	:	:	:	:	:	:	:	:
:	( IN DIRECTION X	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 330	:	:	:	:	:	:	:	:	:	:	:	:	:
:	( IN DIRECTION Y	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
	STRAIGHT	-	:	:	:	:	4.56	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
	STRAIGHT	-	>-330.1	0.00	0.00	-12.56	4.56	:	:	:	:	:	:	:
	STRAIGHT	-	>-330.2	0.00	0.00	-17.13	4.56	:	:	:	:	:	:	:
	STRAIGHT	-	>-330.3	0.00	0.00	-21.69	4.56	:	:	:	:	:	:	:
45	STRAIGHT	-	>-345K	0.00	0.00	-26.25	4.56	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 345N	:	:	:	:	:	:	:	:	:	:	:	:	:
:	( IN DIRECTION Y	:	:	:	:	:	:	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
	BEND	-	345K	0.00	-2.22	-26.76	.75	1.18	90.000	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
	STRAIGHT	-	>-345F	0.00	-2.75	-27.00	4.31	:	:	:	:	:	:	:
	STRAIGHT	-	>-345F.1	0.00	-5.06	-27.00	4.31	:	:	:	:	:	:	:
	STRAIGHT	-	>-345F.2	0.00	-9.36	-27.00	4.31	:	:	:	:	:	:	:
	STRAIGHT	-	>-350K	0.00	-13.67	-27.00	4.31	:	:	:	:	:	:	:
:	(	:	:	:	:	:	:	:	:	:	:	:	:	:
	BEND	-	350K	0.00	-14.20	-27.22	.75	1.18	90.000	:	:	:	:	:



POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)		
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF	
									6.625	.864	LCY	960.	54.3	15.2
	STRAIGHT	>-380F	-49.92	-15.17	-34.00		1.43							
		>-385F	-49.92	-16.60	-34.00									
	BEND	( 385)	-39.92	-17.13	-34.22	.75	1.18	90.000						
	STRAIGHT	>-385F	-39.92	-17.35	-34.75		4.63							
	STRAIGHT	>-385F.1	-39.92	-17.35	-39.33		4.63							
	STRAIGHT	>-385F.2	-39.92	-17.35	-44.00		4.63							
	STRAIGHT	>-385F.3	-39.92	-17.35	-48.63		4.63							
		>-390F	-39.92	-17.35	-53.25									
	BEND	( 390)	-39.92	-17.57	-53.73	.75	1.18	90.000						
	STRAIGHT	>-390F	-39.92	-18.16	-54.00		4.75							
	STRAIGHT	>-390F.1	-39.92	-22.85	-54.00		4.75							
47		>-395F	-39.92	-27.60	-54.00									
	BEND	( 395)	-39.92	-28.13	-54.22	.75	1.18	90.000						
	STRAIGHT	>-395F	-39.92	-28.35	-54.75		3.25							
ANCHP		--400	-39.92	-28.35	-55.00									
* CONTROL COORDINATES OF POINT 400, IN FEET :														
X = -39.92, Y = -28.35, Z = -55.00														
RESULTANT DIFFERENCE IS ZERO														

DYNAFLEX

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\*\*\*\*\*  
 \*  
 \* LUMPED DYNAMIC MODEL \*  
 \* ----- \*  
 \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

- (1) THE MASS OF THE SYSTEM IS LUMPED AT THE POINTS SHOWN BELOW. FRACTIONAL LOCATION NUMBERS, IF ANY, REFER FROM ADDITIONAL MASS POINTS LOCATED BY THE PROGRAM BASED ON THE MAXIMUM LUMP SPACING CRITERION GIVEN IN THE INPUT DATA.
- (2) THE MASS AT A POINT IS A COMBINATION OF DISTRIBUTED MASS LUMPED IN A LINEAR FASHION PLUS ANY ADDITIONAL CONCENTRATED MASS INPUT BY THE USER (SHOWN IN PARENTHESES).
- (3) ROTATIONAL DEGREES OF FREEDOM ARE NOT INCLUDED IN THE MODEL UNLESS SPECIFICALLY ADDED BY THE USER. ALL THREE DISPLACEMENT DEGREES OF FREEDOM ARE AUTOMATICALLY DELETED AT ANCHORS. THE DEGREE OF FREEDOM ASSOCIATED WITH A RIGID RESTRAINT OR BORDER ORIENTED IN THE X, Y OR Z DIRECTION IS ALSO DELETED.

DISPLACEMENT FREEDOM

ROTATIONAL FREEDOM

-----  
 MOMENT OF INERTIA (LB - IN X IN)  
 -----

LOC. NO.	HEIGHT (FEET)	DIRECTION			MOMENT OF INERTIA (LB - IN X IN)		
		X	Y	Z	X	Y	Z
315	89.				0.	0.	0.
48	320	X	Y	Z	0.	0.	0.
	325	X	Y	Z	0.	0.	0.
	330			Z	0.	0.	0.
	330.1	X	Y	Z	0.	0.	0.
	330.2	X	Y	Z	0.	0.	0.
	330.3	X	Y	Z	0.	0.	0.
	345H	X		Z	0.	0.	0.
	345F	X	Y	Z	0.	0.	0.
	345F.1	X	Y	Z	0.	0.	0.
	345F.2	X	Y	Z	0.	0.	0.
	350H	X	Y	Z	0.	0.	0.
	350F	X	Y	Z	0.	0.	0.

DYNAFLEX

LINKED DYNAMIC MODEL (CONTINUED)

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## DISPLACEMENT FREEDOM

## ROTATIONAL FREEDOM

## MOMENT OF INERTIA (LB - IN X IN)

LOC. ID.	WEIGHT (POUNDS)	DIRECTION			MOMENT OF INERTIA (LB - IN X IN)		
		X	Y	Z	X	Y	Z
350F.1	191.	X	Y	Z	0.	0.	0.
360.1	156.	Y	Y	Z	0.	0.	0.
370F	119.	X	Y	Z	0.	0.	0.
365	226.	X	Y	Z	0.	0.	0.
365.1	295.	X	Y	Z	0.	0.	0.
365.2	295.	X	Y	Z	0.	0.	0.
365.3	295.	X	Y	Z	0.	0.	0.
370	319.	X		Z	0.	0.	0.
370.1	333.	X	Y	Z	0.	0.	0.
370.2	333.	X	Y	Z	0.	0.	0.
49 370.3	333.	X	Y	Z	0.	0.	0.
3800	207.	X		Z	0.	0.	0.
380F	91.	Y	Y	Z	0.	0.	0.
3850	91.	X	Y	Z	0.	0.	0.
385F	202.	X	Y	Z	0.	0.	0.
385F.1	321.	X	Y	Z	0.	0.	0.
385F.2	321.	X	Y	Z	0.	0.	0.
385F.3	321.	X	Y	Z	0.	0.	0.
3900	202.	X	Y	Z	0.	0.	0.
390F	206.	X	Y	Z	0.	0.	0.
390F.1	330.	X	Y	Z	0.	0.	0.
3950	206.	X	Y	Z	0.	0.	0.
395F	154.	X	Y	Z	0.	0.	0.

DYNAPLEX

LOGO DYNAMIC MODEL (CONTINUED)

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LOG. NO.	DISPLACEMENT FREEDOM		ROTATIONAL FREEDOM	
	OF LOG	DIRECTION	MOMENT OF INERTIA (LB-IN X 10)	
---	(POUNDS)			
407	113.			
		X	0.	0.
		Y	0.	0.
		Z	0.	0.

NUMBER OF DYNAMIC DEGREES OF FREEDOM = 100, TOTAL MASS OF SYSTEM = 9825. POUNDS

\*\*\*\*\*  
 \* FREQUENCIES, PERIODS OF VIBRATION AND MASS PARTICIPATION FACTORS \*  
 \*-----\*  
 \*-----\*

MODE NO.	FREQUENCY (CPS)	PERIOD (SECONDS)	MASS PARTICIPATION FACTORS		
			X	Y	Z
1	.80053	1.13585	5.81331626	.04724832	.64842915
2	1.87980	.53197	.98141578	.45430736	3.10306712
3	3.09022	.32702	.26667501	.04580247	1.70408241
4	3.91919	.25515	.98239791	.13011141	.42862710
5	6.16749	.23995	.75514853	.62167820	1.03757545
6	5.28776	.18912	1.03627413	.34984765	.15507050
7	7.78927	.12904	.03756246	1.41655161	.84478677
8	9.12371	.10960	.14267544	.41555116	.04905854
9	11.81558	.08465	.00269364	.25176260	.26694942
10	12.40673	.08060	.02949952	1.83648268	.41712082
11	14.99943	.06667	.20970328	.91435421	.18149968
12	16.32961	.06124	.17221260	.19809667	.04984137
13	17.77357	.05626	.06591277	.23382666	.02805239
14	18.19506	.05496	.27405085	.70991063	.00892138
15	20.65028	.04743	.22952225	1.21738988	.18950453
16	25.65096	.03898	.01869728	.22351457	.30338071
17	26.36611	.03768	.00476015	1.39896417	.07143881
18	31.88191	.03137	.40124551	.78789004	.37900367
19	32.58816	.03069	.49242111	.04006710	.25567220

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NOTE:

SECTION 5.7.2 OF THE U.S. NUCLEAR REGULATORY COMMISSION STANDARD REVIEW PLAN DATED JUNE, 1975 STATES THAT AN ADEQUATE NUMBER OF DEGREES OF FREEDOM IN DYNAMIC MODELING MAY BE TAKEN EQUAL TO FIVE THE NUMBER OF MODES WITH FREQUENCIES LESS THAN 33 CPS.

FOR THIS ANALYSIS THE RATIO OF DYNAMIC DEGREES OF FREEDOM TO FREQUENCIES LESS THAN 33 CPS = 5.26

THE COLD ELASTIC MODULUS WAS USED IN CALCULATING THE NATURAL FREQUENCIES OF THE SYSTEM IN THIS ANALYSIS.

DYNALLEX

MODE ORTHOGONALITY CHECK  
-----

ON DIAGONAL TERMS: MAX. = 1.00000 MIN. = 1.00000

OFF DIAGONAL TERMS: MAX. = 1.04160 X 10<sup>-13</sup> MIN. = -1.43525 X 10<sup>-13</sup>



GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE DETAILED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
515	2295.	374.	410.	2273.	937.	5518.	3339.	6517.
400	786.	70.	348.	862.	2057.	10061.	6867.	12353.
<b>RESTRAINTS:</b>								
53	2934.	0.	0.	2934.	0.	0.	0.	0.
530	0.	788.	0.	788.	0.	0.	0.	0.
345	0.	612.	0.	612.	0.	0.	0.	0.
370	0.	466.	0.	466.	0.	0.	0.	0.

LOADING - SHOCK LOADING NO. 1

FORCES AND MOMENTS IN ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
	X	Y	Z	X	Y	Z
---	---	---	---	---	---	---
RESTRAINTS (CONTINUED) -	9.	159.	0.	0.	0.	0.
---	---	---	---	---	---	---

\*\*\*\*\*  
\*  
\* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
\*-----\*  
\*  
\*\*\*\*\*

THE MAXIMUM STRESS OF 7624. OCCURS AT POINT 330

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ASTM B31.1 - 1977  
PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
(AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (15) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
BUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP	STR	-- 315	410.	2236.	3339.	937.	5518.	1.00	1.00	20.02	3906.
		>- 321	410.	2236.	3339.	188.	1109.	1.00	1.00	20.02	2112.
RIGID	STR	-- 325	410.	2234.	3339.	560.	3298.	1.00	1.00	20.02	2833.
		>- 349	410.	846.	3339.	2053.	12103.	1.00	1.00	20.02	7624. **
		>- 330.1	410.	835.	3339.	167.	8778.	1.00	1.00	20.02	5629.
		>- 330.2	410.	808.	3339.	1717.	5529.	1.00	1.00	20.02	4005.
		>- 330.3	410.	772.	3339.	3569.	2515.	1.00	1.00	20.02	3294.
		>- 345H	410.	691.	3339.	5401.	1252.	1.00	1.00	20.02	3879.
BEND	STR	-- 345H	303.	743.	2869.	5231.	1937.	1.00	1.00	20.02	3759.
		>- 345F	289.	733.	1538.	2885.	4988.	1.00	1.00	20.02	3574.
		>- 345F.1	289.	697.	1538.	528.	3329.	1.00	1.00	20.02	2220.
55	STR	-- 345F.2	289.	697.	1538.	2257.	1939.	1.00	1.00	20.02	2007.
		>- 350H	289.	640.	1538.	4491.	1424.	1.00	1.00	20.02	2970.
BEND	STR	-- 350H	260.	597.	2804.	1407.	4060.	1.00	1.00	20.02	3075.
		>- 350F	331.	563.	4849.	1309.	1699.	1.00	1.00	20.02	3177.
		>- 350F.1	309.	537.	4849.	858.	2552.	1.00	1.00	20.02	3324.
		>- 360H	281.	500.	4849.	980.	3529.	1.00	1.00	20.02	3642.
BEND	STR	-- 360H	253.	490.	3951.	3739.	2907.	1.00	1.00	20.02	3696.
		>- 365F	345.	369.	1103.	4642.	3881.	1.00	1.00	20.02	3686.
		>- 365	345.	369.	1103.	4032.	4187.	1.00	1.00	20.02	3545.
		>- 365.1	277.	347.	1103.	2919.	4837.	1.00	1.00	20.02	3449.
		>- 365.2	276.	320.	1103.	1896.	5400.	1.00	1.00	20.02	3493.

DYNAFLEX

LOADING - SHOCK LOADING NO. 1

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## INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEX TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESIDUAL SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 365.2	215.	320.	1103.	1896.	5400.	1.00	1.00	20.02	3493.
	STR - (	>- 365.3	147.	295.	1103.	1228.	5791.	1.00	1.00	20.02	3608.
	STR - (	>- 370	178.	210.	1103.	1558.	5981.	1.00	1.00	20.02	3762.
	STR - (	>- 370.1	178.	210.	1103.	1205.	5984.	1.00	1.00	20.02	3717.
	STR - (	>- 370.2	244.	246.	1103.	1420.	5855.	1.00	1.00	20.02	3671.
	STR - (	>- 370.3	324.	271.	1103.	2029.	5720.	1.00	1.00	20.02	3697.
	STR - (	>- 380F	411.	280.	1103.	2792.	5671.	1.00	1.00	20.02	3845.
	BEND - (	380B	360.	374.	3546.	2714.	4559.	1.00	1.00	20.02	3824.
	STR - (	>- 390F	194.	509.	5677.	2496.	1017.	1.00	1.00	20.02	3766.
	STR - (	>- 395B	103.	531.	5677.	1877.	898.	1.00	1.00	20.02	3623.
	BEND - (	395B	181.	532.	2957.	868.	4990.	1.00	1.00	20.02	3515.
57	STR - (	>- 395F	201.	525.	1565.	849.	5300.	1.00	1.00	20.02	3351.
	STR - (	>- 395F.1	204.	576.	1565.	910.	2789.	1.00	1.00	20.02	1993.
	STR - (	>- 395F.2	242.	675.	1565.	1114.	1016.	1.00	1.00	20.02	1302.
	STR - (	>- 395F.3	276.	710.	1565.	1327.	3463.	1.00	1.00	20.02	2412.
	STR - (	>- 390B	276.	710.	1565.	1520.	6599.	1.00	1.00	20.02	4165.
	BEND - (	390B	204.	752.	5810.	1477.	4085.	1.00	1.00	20.02	4351.
	STR - (	>- 390F	70.	822.	7124.	1037.	1329.	1.00	1.00	20.02	4387.
	STR - (	>- 390F.1	70.	855.	7124.	2587.	259.	1.00	1.00	20.02	4544.
	STR - (	>- 395B	70.	855.	7124.	6280.	1871.	1.00	1.00	20.02	5800.
	BEND - (	395B	269.	910.	1800.	2059.	9724.	1.00	1.00	20.02	6052.
	STR - (	>- 395F	348.	789.	6867.	2114.	7662.	1.00	1.00	20.02	6294.

DYNAFLEX

LOADING - SHOCK LOADING NO. 1

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	MEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		:	:
		>- 395F	548.	789.	6867.	2114.	7662.	1.00	1.00	20.02	:	6294.
	STR = (										:	
ANCHP		-- 400	548.	789.	6867.	2057.	10061.	1.00	1.00	20.02	:	7403. *

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\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR	STRAIGHT - (	315	.000 (.0)	.000 (.0)	.000 (.0)	.000	.000	.000
	>-	320	.008 (1.6)	.001 (1.1)	.000 (.0)	.005	.028	.036
	STRAIGHT - (	325	.019 (3.9)	.003 (2.6)	.000 (.0)	.005	.028	.036
	>-	330	.000 (.0)	.000 (.0)	.000 (.1)	.017	.101	.109
	STRAIGHT - (	330.1	.196 (34.0)	.029 (18.6)	.000 (.2)	.038	.299	.191
	>-	330.2	.552 (78.8)	.062 (29.9)	.000 (.3)	.023	.434	.273
	STRAIGHT - (	330.3	1.007 (123.3)	.063 (21.4)	.000 (.3)	.027	.509	.356
	>-	3450	1.506 (171.7)	.000 (.0)	.000 (.4)	.112	.525	.438
	BEND - (	3450	1.583 (174.8)	.014 (3.1)	.006 (1.6)	.135	.523	.449
	>-	345F	1.653 (176.6)	.020 (4.5)	.022 (5.6)	.157	.520	.461
	STRAIGHT - (	345F.1	2.058 (119.0)	.020 (4.5)	.200 (33.9)	.231	.496	.490
	>-	345F.2	2.477 (105.8)	.020 (4.6)	.431 (56.5)	.276	.471	.473
	STRAIGHT - (	3500	2.865 (111.4)	.020 (4.6)	.690 (44.8)	.294	.449	.414
BEND - (	3500	2.928 (112.4)	.038 (5.2)	.722 (88.8)	.295	.444	.395	
>-	350F	2.994 (112.7)	.067 (6.3)	.736 (90.4)	.296	.438	.379	
STRAIGHT - (	350F.1	3.238	.236	.736	.294	.418	.310	

DYNAFLEX

LOADING - SPOCK LOADING NO. 1

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DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>- 560F.1		5.238	.236	.736	.294	.416	.310
	STRAIGHT - (		(108.3)	(29.1)	(90.4)			
	>- 560F		3.466	.403	.736	.286	.384	.241
	(		(109.0)	(51.1)	(90.4)			
	HEAD - (	560F	3.507	.425	.748	.283	.389	.227
	(		(110.5)	(54.0)	(89.2)			
	>- 560F		3.524	.416	.776	.280	.353	.209
	STRAIGHT - (		(111.2)	(54.2)	(86.4)			
	>- 565		3.524	.333	.896	.267	.316	.173
	STRAIGHT - (		(111.2)	(48.5)	(78.4)			
	>- 565.1		3.524	.212	1.161	.242	.238	.122
	STRAIGHT - (		(111.2)	(39.6)	(81.2)			
	>- 565.2		3.524	.121	1.247	.216	.151	.090
	STRAIGHT - (		(111.2)	(29.3)	(94.1)			
	>- 565.3		3.524	.051	1.317	.191	.070	.068
	STRAIGHT - (		(111.2)	(14.9)	(101.6)			
	>- 570		3.524	.000	1.301	.166	.043	.048
	STRAIGHT - (		(111.2)	(.0)	(97.6)			
	>- 570.1		3.524	.038	1.175	.134	.185	.030
	STRAIGHT - (		(111.2)	(14.7)	(80.5)			
	>- 570.2		3.524	.056	.933	.110	.300	.012
	STRAIGHT - (		(111.2)	(19.7)	(56.9)			
60	>- 570.3		3.524	.048	.581	.083	.410	.027
	STRAIGHT - (		(111.2)	(13.2)	(39.0)			
	>- 580		3.524	.000	.153	.057	.515	.072
	(		(111.2)	(.0)	(50.2)			
	HEAD - (	580	3.520	.009	.122	.052	.536	.083
	(		(111.2)	(2.4)	(53.4)			
	>- 580F		3.510	.013	.119	.048	.554	.094
	STRAIGHT - (		(111.1)	(3.6)	(56.4)			
	>- 585		3.461	.013	.131	.043	.595	.106
	(		(111.6)	(3.6)	(62.1)			
	HEAD - (	585	3.441	.014	.135	.040	.613	.115
	(		(110.7)	(4.0)	(64.3)			
	>- 585F		3.366	.017	.137	.038	.635	.122
	STRAIGHT - (		(107.8)	(5.5)	(65.2)			
	>- 585F.1		2.716	.045	.137	.024	.706	.159
	STRAIGHT - (		(66.8)	(21.6)	(65.2)			
	>- 585F.2		2.622	.058	.137	.009	.727	.196
	STRAIGHT - (		(109.3)	(39.3)	(65.2)			
	>- 585F.3		1.345	.052	.137	.019	.692	.235
	STRAIGHT - (		(132.1)	(26.9)	(65.1)			
	>- 590		.719	.023	.137	.044	.596	.273
	(		(153.1)	(11.1)	(65.1)			
	HEAD - (	590	.666	.017	.134	.050	.568	.276



DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT	LOC.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X AXIS	Y AXIS	Z AXIS	X AXIS	Y AXIS	Z AXIS
MEMO - (	5990		.134	.017	(64.1)	.059	.568	.276
STRAIGHT - (	5991		.129	.015	(61.3)	.056	.545	.278
STRAIGHT - (	5992		.065	.015	(31.3)	.046	.362	.263
MEMO - (	5993		.066	.015	(3.1)	.037	.179	.175
STRAIGHT - (	5994		.062	.013	(6.5)	.028	.120	.121
ANCHOR	400		.000	.000	.000	.000	.000	.000

DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

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 \*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	Y	Z	RESULTANT	X	Y	Z	RESULTANT	
<b>ANCHORS:</b>								
315	745.	869.	1160.	1629.	2174.	1862.	951.	3016.
400	417.	264.	1177.	1277.	6038.	2598.	3334.	7371.
<b>RESTRAINTS:</b>								
29	330	1035.	0.	0.	1035.	0.	0.	0.
	531	0.	1829.	0.	1829.	0.	0.	0.
	545H	0.	872.	0.	872.	0.	0.	0.
	370	0.	705.	0.	705.	0.	0.	0.

LOADING - SHOCK LOADING NO. 2

FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
	X	Y	Z	X	Y	Z
RESTRAINTS (CONTINUED)	0.	430.	0.	0.	0.	0.
ANCHORS	0.	430.	0.	0.	0.	0.

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 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \* ----- \*  
 \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 7531. OCCURS AT POINT 345H

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ASME B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (15) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP		-- 315	1160.	1144.	951.	2174.	1862.	1.00	1.00	20.02	1807.
	STR -(										
	>-	320	1160.	1144.	951.	436.	373.	1.00	1.00	20.02	666.
	RIGID -(										
	>-	325	1160.	1140.	951.	1300.	1114.	1.00	1.00	20.02	1174.
	STR -(										
	>-	330	1160.	1140.	951.	4763.	4080.	1.00	1.00	20.02	3801.
	STR -(										
	>-	330.1	1160.	998.	951.	369.	2762.	1.00	1.00	20.02	1764.
	STR -(										
	>-	330.2	1160.	975.	951.	3987.	1548.	1.00	1.00	20.02	2626.
	STR -(										
	>-	330.3	1160.	975.	951.	8261.	908.	1.00	1.00	20.02	5013.
	STR -(										
	>-	345N	1160.	326.	951.	12453.	1408.	1.00	1.00	20.02	7531. **
	BEND -(	345N	906.	793.	1272.	12243.	1123.	1.00	1.00	20.02	7407. *
	STR -(										
	>-	345F	250.	1178.	1530.	11648.	929.	1.00	1.00	20.02	7062. *
	STR -(										
	>-	345F.1	254.	1116.	1530.	1096.	6708.	1.00	1.00	20.02	4175.
	STR -(										
	>-	345F.2	257.	1001.	1530.	1581.	2156.	1.00	1.00	20.02	1846.
	STR -(										
	>-	350N	257.	1001.	1530.	2105.	2580.	1.00	1.00	20.02	2196.
	BEND -(	350N	729.	572.	1156.	2973.	2298.	1.00	1.00	20.02	2356.
	STR -(										
	>-	350F	880.	291.	2187.	3060.	1539.	1.00	1.00	20.02	2435.
	STR -(										
	>-	350F.1	800.	289.	2187.	2596.	1553.	1.00	1.00	20.02	2237.
	STR -(										
	>-	360N	692.	285.	2187.	2168.	1620.	1.00	1.00	20.02	2085.
	BEND -(	360N	467.	504.	1050.	1536.	2794.	1.00	1.00	20.02	2011.
	STR -(										
	>-	360F	122.	669.	2044.	1312.	2630.	1.00	1.00	20.02	1897.
	STR -(										
	>-	365	113.	618.	2044.	1541.	1021.	1.00	1.00	20.02	1652.
	STR -(										
	>-	365.1	97.	500.	2044.	836.	2430.	1.00	1.00	20.02	1968.
	STR -(										
	>-	365.2	82.	470.	2044.	1592.	3612.	1.00	1.00	20.02	2664.

DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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## INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 365.2	82.	479.	2044.	1592.	3612.	1.00	1.00	20.02	2664.
	STR - (	>- 365.3	40.	435.	2044.	3078.	4218.	1.00	1.00	20.02	3360.
	STR - (	>- 370	83.	434.	2044.	4747.	4214.	1.00	1.00	20.02	3996.
	STR - (	>- 370.1	105.	394.	2044.	3520.	3527.	1.00	1.00	20.02	3151.
	STR - (	>- 370.2	127.	481.	2044.	1885.	2408.	1.00	1.00	20.02	2204.
	STR - (	>- 370.3	177.	675.	2044.	625.	2196.	1.00	1.00	20.02	1837.
	STR - (	>- 380D	193.	735.	2044.	1435.	4102.	1.00	1.00	20.02	2877.
	BEND - (	380E	352.	671.	4355.	1463.	2023.	1.00	1.00	20.02	3008.
	STR - (	>- 380F	379.	677.	4503.	1410.	1780.	1.00	1.00	20.02	3022.
	STR - (	>- 385M	379.	677.	4503.	1174.	1594.	1.00	1.00	20.02	2948.
	BEND - (	385N	712.	354.	2435.	1613.	3890.	1.00	1.00	20.02	2914.
99	STR - (	>- 385F	668.	432.	1056.	1527.	4363.	1.00	1.00	20.02	2842.
	STR - (	>- 385F.1	718.	434.	1056.	1434.	3452.	1.00	1.00	20.02	2328.
	STR - (	>- 385F.2	805.	437.	1056.	2703.	2442.	1.00	1.00	20.02	2273.
	STR - (	>- 385F.3	899.	436.	1056.	4090.	1414.	1.00	1.00	20.02	2669.
	STR - (	>- 390I	997.	444.	1056.	5392.	1197.	1.00	1.00	20.02	3370.
	BEND - (	390J	945.	465.	1359.	5306.	870.	1.00	1.00	20.02	3323.
	STR - (	>- 390F	290.	1119.	1333.	4805.	792.	1.00	1.00	20.02	3026.
	STR - (	>- 390F.1	275.	1205.	1333.	1067.	575.	1.00	1.00	20.02	1080.
	STR - (	>- 390D	267.	1249.	1333.	6110.	3023.	1.00	1.00	20.02	4163.
	BEND - (	390E	1014.	781.	1951.	6681.	2965.	1.00	1.00	20.02	4526.
	STR - (	>- 395F	1177.	493.	3534.	6809.	1511.	1.00	1.00	20.02	4633.

DYNAFLEX

LOADING - SPOCK LOADING NO. 2

PAGE 2P

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT ELEM TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
		>- 395F	1177.	493.	3334.	6409.	1511.	1.00	1.00	20.02	4633.
	STR -(										
	ANCHR	-- 400	1177.	493.	3334.	6038.	2598.	1.00	1.00	20.02	4417.

DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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 \*  
 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )  
 DISPLACEMENTS (INCHES) ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 315	.000	.000	.000	.000	.000	.000
	STRAIGHT -(		(.0)	(.0)	(.0)			
	>- 325	.003	.003	.000	.011	.009	.010	
	RIGID -(		(.6)	(.7)	(.0)			
	>- 325	.006	.006	.000	.011	.009	.010	
	STRAIGHT -(		(1.6)	(1.7)	(.0)			
	>- 330	.000	.000	.000	.040	.034	.031	
	STRAIGHT -(		(.0)	(.0)	(.1)			
	>- 330.1	.066	.066	.000	.088	.099	.054	
	STRAIGHT -(		(15.3)	(13.6)	(.2)			
	>- 330.2	.121	.143	.000	.054	.139	.078	
	STRAIGHT -(		(41.2)	(26.2)	(.2)			
	>- 330.3	.324	.147	.001	.062	.157	.101	
	STRAIGHT -(		(72.1)	(25.6)	(.3)			
89	>- 345F	.474	.000	.001	.259	.156	.125	
	(		(104.9)	(.0)	(.0)			
	BEND -(		.490	.032	.014	.312	.154	.128
	(		(106.7)	(5.5)	(2.6)			
	>- 345F	.494	.047	.052	.364	.152	.132	
	STRAIGHT -(		(163.5)	(8.2)	(9.3)			
	>- 345F.1	.489	.047	.462	.528	.143	.145	
	STRAIGHT -(		(60.1)	(8.2)	(79.0)			
	>- 345F.2	.511	.047	.980	.606	.140	.155	
	STRAIGHT -(		(72.2)	(8.2)	(160.5)			
	>- 350K	.545	.047	1.532	.604	.147	.165	
	(		(64.2)	(8.2)	(243.2)			
	BEND -(		.552	.074	1.598	.594	.149	.167
	(		(83.6)	(11.9)	(253.2)			
>- 350F	.559	.139	1.625	.582	.151	.170		
STRAIGHT -(		(79.5)	(27.6)	(257.3)				
>- 350F.1	.566	.066	1.625	.552	.158	.184		



DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESTINATION	ELEMENT DESTINATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>- 350F.1		.586	.466	1.625	.552	.158	.184
	STRAIGHT - (		(51.0)	(85.8)	(257.3)			
	>- 360F		.620	.775	1.625	.525	.163	.203
	(		(34.4)	(152.3)	(257.3)			
	BEND - (	360F	.627	.824	1.624	.516	.165	.208
	(		(26.1)	(163.7)	(255.7)			
	>- 360F		.630	.824	1.620	.509	.167	.215
	STRAIGHT - (		(37.2)	(166.1)	(252.1)			
	>- 365		.630	.722	1.607	.484	.170	.226
	STRAIGHT - (		(37.2)	(153.9)	(238.6)			
	>- 365.1		.630	.517	1.585	.437	.169	.234
	STRAIGHT - (		(37.2)	(126.7)	(223.1)			
	>- 365.2		.630	.313	1.541	.390	.170	.220
	STRAIGHT - (		(37.2)	(90.2)	(218.7)			
	>- 365.3		.630	.133	1.459	.344	.185	.181
	STRAIGHT - (		(37.2)	(45.1)	(218.7)			
	>- 370		.630	.000	1.328	.297	.217	.112
	STRAIGHT - (		(37.2)	(.0)	(214.8)			
	>- 370.1		.630	.070	1.123	.245	.259	.033
	STRAIGHT - (		(37.2)	(32.2)	(198.9)			
	>- 370.2		.630	.074	.875	.193	.238	.020
	STRAIGHT - (		(37.2)	(50.8)	(172.8)			
69	>- 370.3		.630	.041	.617	.142	.285	.042
	STRAIGHT - (		(37.2)	(33.6)	(105.0)			
	>- 380F		.630	.000	.408	.094	.241	.036
	(		(37.2)	(.0)	(130.3)			
	BEND - (	380F	.630	.004	.395	.089	.227	.033
	(		(37.1)	(5.3)	(130.5)			
	>- 380F		.629	.005	.396	.085	.216	.031
	STRAIGHT - (		(37.6)	(7.6)	(131.6)			
	>- 385F		.626	.005	.416	.079	.191	.030
	(		(42.4)	(7.6)	(135.9)			
	BEND - (	385F	.622	.008	.423	.079	.180	.030
	(		(45.2)	(7.9)	(138.0)			
	>- 385F		.615	.016	.427	.079	.169	.032
	STRAIGHT - (		(48.3)	(9.2)	(139.0)			
	>- 385F.1		.562	.089	.426	.069	.136	.047
	STRAIGHT - (		(61.0)	(43.2)	(139.0)			
	>- 385F.2		.472	.139	.426	.034	.129	.070
	STRAIGHT - (		(77.0)	(60.3)	(138.9)			
	>- 385F.3		.379	.141	.426	.034	.128	.095
	STRAIGHT - (		(86.0)	(52.9)	(138.9)			
	>- 390F		.291	.068	.426	.123	.115	.120
	(		(92.5)	(25.0)	(138.9)			
	BEND - (	390F	.276	.053	.420	.146	.110	.123

LOADING - SHOCK LEADING NO. 2

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
BEAD - (	3950	3950F	.276	.055	.420	.146	.110	.123
			(41.3)	(14.0)	(136.8)			
			.259	.046	.402	.168	.106	.127
STRAIGHT - (	3950	3950F	.132	.046	.205	.209	.072	.124
			(40.7)	(15.7)	(67.3)			
			.026	.046	.020	.144	.039	.083
BEAD - (	3950	3950F	.016	.040	.005	.116	.034	.070
			(8.6)	(13.8)	(1.7)			
			.010	.029	.000	.087	.027	.059
STRAIGHT - (	400	400	.000	.000	.000	.000	.000	.000
			(6.0)	(10.2)	(.0)			
			.000	.000	.000	.000	.000	.000

GENERAL NOTES FOR THIS TABLE:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE POTTED PILING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

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 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \* \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
315	126.	138.	291.	338.	343.	313.	252.	529.
400	146.	211.	196.	323.	1215.	489.	1107.	1715.
<b>RESTRAINTS:</b>								
71 330	182.	0.	0.	182.	0.	0.	0.	0.
330	0.	290.	0.	290.	0.	0.	0.	0.
345 <sup>H</sup>	0.	312.	0.	312.	0.	0.	0.	0.
370	0.	262.	0.	262.	0.	0.	0.	0.

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

FORCES (POUNDS)

MOMENTS (FOOT-POUNDS)

LOC. NO.

X Y Z RESULTANT

X Y Z RESULTANT

RESTRAINTS (CONTINUED)

3A05

0. 199. 0. 199.

0. 0. 0. 0.

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*
* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS
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THE MAXIMUM STRESS OF 1139. OCCURS AT POINT 350H

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ANSI B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP	STR - (	>- 315	281.	187.	252.	343.	313.	1.00	1.00	20.02	317.
	STR - (	>- 320	281.	187.	252.	67.	61.	1.00	1.00	20.02	161.
	RIGID - (	>- 325	281.	181.	252.	208.	189.	1.00	1.00	20.02	226.
	STR - (	>- 330	281.	181.	252.	740.	680.	1.00	1.00	20.02	621.
	STR - (	>- 330.1	281.	155.	252.	84.	446.	1.00	1.00	20.02	311.
	STR - (	>- 340.2	281.	143.	252.	636.	291.	1.00	1.00	20.02	445.
	STR - (	>- 350.3	281.	143.	252.	1223.	257.	1.00	1.00	20.02	764.
	STR - (	>- 345N	281.	146.	252.	1802.	571.	1.00	1.00	20.02	1113. **
	BEND - (	345N	398.	133.	341.	1808.	233.	1.00	1.00	20.02	1111. **
	STR - (	>- 345F	309.	284.	399.	1712.	237.	1.00	1.00	20.02	1063. **
	STR - (	>- 345F.1	315.	246.	399.	206.	1031.	1.00	1.00	20.02	674.
74	STR - (	>- 345F.2	319.	196.	399.	322.	1267.	1.00	1.00	20.02	819.
	STR - (	>- 350N	319.	196.	399.	429.	1406.	1.00	1.00	20.02	1138. **
	BEND - (	350N	313.	195.	289.	1800.	434.	1.00	1.00	20.02	1123. **
	STR - (	>- 350F	160.	333.	451.	1659.	367.	1.00	1.00	20.02	1054. **
	STR - (	>- 350F.1	128.	309.	451.	817.	344.	1.00	1.00	20.02	596.
	STR - (	>- 360N	128.	309.	451.	299.	584.	1.00	1.00	20.02	477.
	BEND - (	360N	140.	254.	292.	625.	461.	1.00	1.00	20.02	497.
	STR - (	>- 360F	121.	261.	348.	612.	436.	1.00	1.00	20.02	496.
	STR - (	>- 365	100.	158.	348.	611.	502.	1.00	1.00	20.02	518.
	STR - (	>- 365.1	100.	158.	348.	909.	588.	1.00	1.00	20.02	682.
	STR - (	>- 365.2	83.	129.	348.	840.	834.	1.00	1.00	20.02	739.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEMENT TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULES ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 365.2	85.	129.	348.	840.	634.	1.00	1.00	20.02	739.
	STR - (	>- 365.3	69.	167.	348.	657.	1009.	1.00	1.00	20.02	751.
	STR - (	>- 370.1	60.	206.	348.	1057.	1026.	1.00	1.00	20.02	907.
	STR - (	>- 370.2	65.	189.	348.	960.	857.	1.00	1.00	20.02	799.
	STR - (	>- 370.3	77.	142.	348.	1087.	590.	1.00	1.00	20.02	770.
	STR - (	>- 380M	114.	193.	348.	804.	450.	1.00	1.00	20.02	590.
	BEND - (	>- 380M	127.	148.	348.	316.	696.	1.00	1.00	20.02	503.
	BEND - (	>- 380M	126.	140.	704.	305.	406.	1.00	1.00	20.02	520.
	STR - (	>- 385F	93.	175.	760.	284.	307.	1.00	1.00	20.02	520.
	BEND - (	>- 385F	93.	175.	760.	302.	291.	1.00	1.00	20.02	520.
	BEND - (	>- 385F	87.	174.	518.	318.	628.	1.00	1.00	20.02	524.
75	STR - (	>- 385F	117.	156.	349.	358.	726.	1.00	1.00	20.02	528.
	STR - (	>- 385F.1	112.	139.	349.	716.	687.	1.00	1.00	20.02	630.
	STR - (	>- 385F.2	126.	109.	349.	809.	670.	1.00	1.00	20.02	681.
	STR - (	>- 385F.3	126.	109.	349.	741.	485.	1.00	1.00	20.02	571.
	BEND - (	>- 390M	147.	168.	349.	871.	281.	1.00	1.00	20.02	587.
	BEND - (	>- 390M	142.	147.	260.	881.	330.	1.00	1.00	20.02	585.
	STR - (	>- 390F	151.	189.	264.	817.	283.	1.00	1.00	20.02	542.
	STR - (	>- 390F.1	183.	242.	260.	339.	214.	1.00	1.00	20.02	288.
	BEND - (	>- 395M	200.	244.	264.	995.	999.	1.00	1.00	20.02	860.
	BEND - (	>- 395M	228.	219.	650.	1085.	692.	1.00	1.00	20.02	927.
	BEND - (	>- 395F	195.	257.	1107.	1106.	220.	1.00	1.00	20.02	947.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT FLEX TYPE	LUC NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)		SIF		SECTION : LONGITUDINAL:	
		AXIAL	RESISTANT SHEAR	TORSION	BENDING	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN-PLANE PLANE	OUT-OF-PLANE PLANE	MODULUS ( IN (3) )	STRESS PSI
	>- 595F	196.	257.	1107.	1106.	220.	1.00	1.00	20.02	20.02	947.
ANCHR	-- 400	196.	257.	1107.	1215.	489.	1.00	1.00	20.02	20.02	1028.



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 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 315	.000 (.0)	.000 (.0)	.000 (.0)	.000	.000	.000
	STRAIGHT - (	>- 320	.000 (1.4)	.000 (1.9)	.000 (.1)	.002	.002	.003
	RIGID - (	>- 325	.001 (3.4)	.001 (4.7)	.000 (.1)	.002	.002	.003
	STRAIGHT - (	>- 330	.000 (.0)	.000 (.0)	.000 (.2)	.006	.006	.008
	STRAIGHT - (	>- 330.1	.011 (25.5)	.011 (25.5)	.000 (.3)	.013	.016	.014
	STRAIGHT - (	>- 330.2	.030 (42.4)	.022 (58.9)	.000 (.4)	.008	.022	.021
	STRAIGHT - (	>- 330.3	.053 (35.2)	.022 (43.3)	.000 (.5)	.010	.026	.027
77		>- 345H	.077 (48.6)	.069 (.6)	.000 (.6)	.038	.027	.033
	RIGID - (	345H	.019 (50.9)	.005 (5.7)	.002 (2.9)	.046	.027	.034
		>- 345F	.077 (47.9)	.007 (8.2)	.008 (9.1)	.053	.027	.035
	STRAIGHT - (	>- 345F.1	.056 (36.5)	.007 (8.2)	.067 (62.0)	.076	.028	.038
	STRAIGHT - (	>- 345F.2	.056 (49.6)	.007 (8.2)	.140 (90.2)	.084	.031	.037
		>- 350H	.060 (45.7)	.007 (8.2)	.214 (75.6)	.087	.036	.035
	RIGID - (	350H	.059 (44.7)	.010 (8.1)	.223 (71.1)	.088	.037	.034
		>- 350F	.057 (46.6)	.020 (11.2)	.227 (69.0)	.090	.038	.034
	STRAIGHT - (	>- 350F.1	.048 (.48)	.071 (.71)	.227 (.227)	.093	.040	.033

DYNALOX

LOADING - SHOCK LOADING NO. 3

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 350F.1	.048	.071	.227	.093	.040	.033
	STRAIGHT -(	(30.6)	(63.7)	(69.0)				
		>- 3600	.045	.124	.227	.092	.041	.033
	(	(21.6)	(164.2)	(69.0)				
	BEAD -(	3600	.047	.135	.226	.091	.041	.033
	(	(26.7)	(184.6)	(68.1)				
		>- 360F	.047	.135	.225	.090	.040	.034
	STRAIGHT -(	(26.6)	(186.3)	(65.9)				
		>- 365	.047	.124	.223	.086	.040	.034
	STRAIGHT -(	(26.6)	(177.0)	(57.7)				
		>- 365.1	.047	.099	.220	.078	.036	.036
	STRAIGHT -(	(24.7)	(163.6)	(50.6)				
		>- 365.2	.047	.069	.217	.071	.029	.039
	STRAIGHT -(	(24.7)	(138.9)	(63.8)				
		>- 365.3	.047	.033	.208	.064	.024	.040
	STRAIGHT -(	(26.7)	(80.6)	(76.1)				
		>- 370	.047	.000	.190	.056	.029	.033
	STRAIGHT -(	(24.7)	(.0)	(73.3)				
		>- 370.1	.047	.026	.157	.049	.040	.019
	STRAIGHT -(	(26.7)	(95.9)	(60.5)				
		>- 370.2	.047	.035	.114	.041	.050	.004
	STRAIGHT -(	(24.6)	(149.1)	(59.4)				
78		>- 370.3	.047	.025	.073	.035	.052	.019
	STRAIGHT -(	(24.6)	(112.9)	(63.7)				
		>- 3800	.047	.000	.058	.030	.047	.027
	(	(28.5)	(.0)	(73.5)				
	BEAD -(	3800	.047	.003	.060	.030	.045	.028
	(	(23.9)	(14.8)	(71.2)				
		>- 380F	.047	.004	.062	.029	.044	.028
	STRAIGHT -(	(17.5)	(21.1)	(62.6)				
		>- 3850	.049	.004	.068	.026	.040	.028
	(	(43.0)	(21.0)	(45.2)				
	BEAD -(	3850	.049	.005	.070	.027	.039	.028
	(	(57.6)	(13.2)	(43.7)				
		>- 385F	.049	.006	.071	.026	.038	.028
	STRAIGHT -(	(64.4)	(13.4)	(44.1)				
		>- 385F.1	.059	.025	.071	.016	.031	.027
	STRAIGHT -(	(62.4)	(25.6)	(44.1)				
		>- 385F.2	.072	.034	.071	.005	.026	.029
	STRAIGHT -(	(63.1)	(110.0)	(44.1)				
		>- 385F.3	.041	.029	.071	.013	.022	.034
	STRAIGHT -(	(61.2)	(75.7)	(44.1)				
		>- 3900	.048	.012	.071	.024	.018	.040
	(	(64.7)	(95.5)	(44.0)				
	BEAD -(	3900	.067	.009	.070	.027	.017	.041

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	Y AXIS	Z AXIS	X AXIS
	BEND - (	390F	.027	.009	.070	.027	.017	.041
	(		(28.4)	(47.4)	(43.3)			
	>-	390F	.022	.008	.066	.029	.017	.042
	STRAIGHT - (		(25.5)	(48.7)	(41.6)			
	>-	390F.1	.041	.008	.034	.034	.010	.040
	STRAIGHT - (		(43.9)	(46.5)	(39.5)			
	>-	395H	.006	.008	.003	.024	.005	.027
	(		(12.6)	(46.1)	(10.1)			
	BEND - (	395H	.003	.007	.001	.020	.005	.023
	(		(10.5)	(45.2)	(3.3)			
	>-	395H	.002	.005	.000	.015	.004	.019
	STRAIGHT - (		(8.1)	(37.0)	(.0)			
ANCHOR	--	400	.000	.000	.000	.000	.000	.000

GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
315	2339.	956.	1262.	2817.	2392.	5832.	3481.	7201.
400	901.	345.	1243.	1574.	6493.	10462.	7713.	14487.
<b>RESTRAINTS:</b>								
08	3116.	0.	0.	3116.	0.	0.	0.	0.
330	0.	2012.	0.	2012.	0.	0.	0.	0.
3450	0.	1110.	0.	1110.	0.	0.	0.	0.
370	0.	885.	0.	885.	0.	0.	0.	0.

DYNAMIFLEX Loading - SHOCK LOADING NO. 4

FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOG. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
5806	0.	503.	0.	503.	0.	0.	0.	0.

RESTRAINTS (CONTINUED)

DYNAFLEX

LOADING - SHOCK LOADING NO. 4

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*****
*
* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS
* -----
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*****
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THE MAXIMUM STRESS OF 8681. OCCURS AT POINT 400

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ACSI 031.1 - 1977  
PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
(AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (14) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		:	:
ANCHR		-- 315	1262.	2519.	3481.	2392.	5832.	1.00	1.00	20.02	:	4315.
	STR - (	>- 320	1262.	2519.	3481.	479.	1172.	1.00	1.00	20.02	:	2220.
	RIGID - (	>- 325	1262.	2513.	3481.	1431.	3486.	1.00	1.00	20.02	:	3075.
	STR - (	>- 330	1262.	1327.	3481.	5239.	12790.	1.00	1.00	20.02	:	8541. *
	STR - (	>- 330.1	1262.	1310.	3481.	414.	9213.	1.00	1.00	20.02	:	5907.
	STR - (	>- 330.2	1262.	1274.	3481.	4587.	5749.	1.00	1.00	20.02	:	4810.
	STR - (	>- 330.3	1262.	1231.	3481.	9082.	2687.	1.00	1.00	20.02	:	6047.
	STR - (	>- 345F	1262.	426.	3481.	13693.	1920.	1.00	1.00	20.02	:	8544. *
	BEND - (	345F	1035.	1095.	3157.	13435.	2252.	1.00	1.00	20.02	:	8380. *
		>- 345F	494.	1411.	2206.	3040.	12786.	1.00	1.00	20.02	:	7986. *
	STR - (	>- 345F.1	498.	1339.	2206.	1234.	7559.	1.00	1.00	20.02	:	4776.
	STR - (	>- 345F.2	501.	1204.	2206.	2774.	3165.	1.00	1.00	20.02	:	2848.
	STR - (	>- 350N	501.	1204.	2206.	4978.	3456.	1.00	1.00	20.02	:	3865.
	BEND - (	350N	836.	850.	3047.	3750.	4686.	1.00	1.00	20.02	:	4033.
		>- 350F	954.	710.	5338.	3719.	2321.	1.00	1.00	20.02	:	4139.
	STR - (	>- 350F.1	870.	697.	5338.	2853.	3007.	1.00	1.00	20.02	:	4050.
	STR - (	>- 360N	758.	654.	5338.	2598.	3927.	1.00	1.00	20.02	:	4223.
	BEND - (	360N	550.	748.	4098.	4090.	4058.	1.00	1.00	20.02	:	4237.
		>- 361F	410.	814.	2349.	4142.	5085.	1.00	1.00	20.02	:	4175.
	STR - (	>- 365	381.	753.	2349.	4360.	4338.	1.00	1.00	20.02	:	3945.
	STR - (	>- 365.1	319.	662.	2349.	3170.	5445.	1.00	1.00	20.02	:	4029.
	STR - (	>- 365.2	245.	583.	2349.	2614.	6550.	1.00	1.00	20.02	:	4454.

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 365.2	245.	563.	2349.	2614.	6550.	1.00	1.00	20.02	4454.
	STR - (	>- 365.3	195.	559.	2349.	3578.	7235.	1.00	1.00	20.02	4988.
	STR - (	>- 370	182.	564.	2349.	5107.	7388.	1.00	1.00	20.02	5563.
	STR - (	>- 370.1	217.	479.	2349.	3660.	6999.	1.00	1.00	20.02	4938.
	STR - (	>- 370.2	286.	558.	2349.	2598.	6359.	1.00	1.00	20.02	4350.
	STR - (	>- 370.3	370.	659.	2349.	2270.	6144.	1.00	1.00	20.02	4170.
	STR - (	>- 380N	461.	754.	2349.	3154.	7033.	1.00	1.00	20.02	4829.
	BEND - (	380M	513.	780.	5660.	3098.	5004.	1.00	1.00	20.02	4893.
	STR - (	>- 380F	403.	848.	7286.	2881.	2073.	1.00	1.00	20.02	4857.
	STR - (	>- 385M	474.	878.	7286.	2235.	1852.	1.00	1.00	20.02	4700.
	BEND - (	385M	740.	663.	3864.	1859.	6359.	1.00	1.00	20.02	4596.
84	STR - (	>- 385F	707.	698.	1920.	1783.	6904.	1.00	1.00	20.02	4425.
	STR - (	>- 385F.1	755.	737.	1920.	1844.	4491.	1.00	1.00	20.02	3128.
	STR - (	>- 385F.2	842.	774.	1920.	3045.	2728.	1.00	1.00	20.02	2706.
	STR - (	>- 385F.3	1045.	854.	1920.	4363.	3772.	1.00	1.00	20.02	3643.
	STR - (	>- 390N	1045.	854.	1920.	5669.	6713.	1.00	1.00	20.02	5390.
	BEND - (	390M	995.	1014.	5980.	5577.	4190.	1.00	1.00	20.02	5506.
	STR - (	>- 390F	333.	1460.	7253.	1335.	5052.	1.00	1.00	20.02	5357.
	STR - (	>- 390F.1	336.	1529.	7253.	2818.	666.	1.00	1.00	20.02	4680.
	STR - (	>- 395N	336.	1529.	7253.	7041.	6468.	1.00	1.00	20.02	7191.
	BEND - (	395M	1070.	1152.	2755.	7072.	10190.	1.00	1.00	20.02	7614.
	STR - (	>- 395F	1243.	965.	7713.	7215.	7812.	1.00	1.00	20.02	7873. *



LOADING - SHOCK LOADING NO. 4

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT FLEM TYPE	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)		SIF		SECTION : LONGITUDINAL :	
	AXIAL	SHEAR	TORSION	BENDING	PLANE	OUT-OF-PLANE	MOMENTS	STRESS
					( IN (3) )			PSI
>= 355F	1243.	365.	7715.	7215.	1.00	1.00	20.02	7873. **
ANCHR	1243.	365.	7715.	10402.	1.00	1.00	20.02	8681. **

\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 315	.000	.000	.000	.000	.000	.000
	STRAIGHT -(		(.0)	(.0)	(.0)			
	>- 320	.008	.003	.000	.012	.029	.038	
	RIGID -(		(2.2)	(2.3)	(.1)			
	>- 325	.020	.008	.000	.012	.029	.038	
	STRAIGHT -(		(5.4)	(5.6)	(.1)			
	>- 330	.000	.000	.000	.044	.106	.113	
	STRAIGHT -(		(.0)	(.0)	(.2)			
	>- 330.1	.207	.075	.000	.097	.315	.199	
	STRAIGHT -(		(45.2)	(42.1)	(.4)			
	>- 330.2	.582	.158	.000	.059	.456	.285	
	STRAIGHT -(		(92.5)	(71.1)	(.5)			
	>- 330.3	1.059	.161	.001	.069	.533	.371	
	STRAIGHT -(		(147.1)	(54.7)	(.7)			
98	>- 345H	1.580	.000	.001	.225	.548	.457	
	(	(207.0)	(.0)	(.8)				
	BEND -(		1.659	.035	.016	.343	.546	.468
	(	(211.1)	(8.6)	(4.2)				
	>- 345I	1.727	.051	.057	.400	.543	.481	
	STRAIGHT -(		(205.2)	(12.5)	(13.9)			
	>- 345F.1	2.116	.051	.508	.581	.517	.512	
	STRAIGHT -(		(141.9)	(12.5)	(106.3)			
	>- 345F.2	2.530	.051	1.080	.671	.493	.499	
	STRAIGHT -(		(145.7)	(12.5)	(192.6)			
	>- 350H	2.917	.051	1.693	.678	.473	.447	
	(	(169.3)	(12.4)	(268.4)				
	BEND -(		2.980	.042	1.768	.669	.470	.431
	(	(169.1)	(14.4)	(277.6)				
>- 350F	3.046	.156	1.798	.659	.465	.417		
STRAIGHT -(		(162.6)	(26.6)	(221.3)				
>- 350F.1	3.291	.527	1.798	.632	.448	.362		

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESTINATION	ELEMENT DESTINATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>> 359F.1		3.291	.527	1.798	.632	.448	.362
	STRAIGHT - (		(124.6)	(123.4)	(281.3)			
	>> 360N		3.521	.883	1.799	.605	.420	.317
	(		(116.3)	(233.3)	(281.3)			
	BEAD - (	360N	3.563	.937	1.802	.595	.466	.310
	(		(119.3)	(252.6)	(279.3)			
	>> 360F		3.580	.935	1.810	.588	.393	.302
	STRAIGHT - (		(120.7)	(256.9)	(274.5)			
	>> 365		3.580	.805	1.854	.559	.361	.286
	STRAIGHT - (		(120.7)	(257.5)	(257.7)			
	>> 365.1		3.580	.568	1.942	.505	.294	.266
	STRAIGHT - (		(120.7)	(210.7)	(242.7)			
	>> 365.2		3.580	.443	1.995	.452	.229	.241
	STRAIGHT - (		(120.7)	(168.2)	(246.4)			
	>> 365.3		3.580	.146	1.977	.398	.199	.197
	STRAIGHT - (		(120.7)	(94.6)	(252.9)			
	>> 370		3.580	.000	1.869	.345	.234	.126
	STRAIGHT - (		(120.7)	(.0)	(247.0)			
	>> 370.1		3.580	.084	1.633	.285	.323	.048
	STRAIGHT - (		(120.7)	(104.6)	(222.9)			
	>> 370.2		3.580	.099	1.284	.224	.419	.024
	STRAIGHT - (		(120.7)	(160.0)	(191.4)			
87	>> 370.3		3.580	.028	.850	.168	.502	.053
	STRAIGHT - (		(120.7)	(129.4)	(163.1)			
	>> 380N		3.540	.000	.439	.114	.570	.085
	(		(120.6)	(.0)	(157.8)			
	BEAD - (	380N	3.576	.010	.418	.108	.584	.094
	(		(119.6)	(15.9)	(157.9)			
	>> 380F		3.567	.014	.418	.101	.597	.103
	STRAIGHT - (		(118.6)	(22.7)	(156.3)			
	>> 385N		3.537	.014	.441	.095	.626	.114
	(		(126.8)	(22.7)	(156.1)			
	BEAD - (	385N	3.497	.017	.450	.092	.640	.122
	(		(132.7)	(20.3)	(158.4)			
	>> 385F		3.722	.025	.453	.091	.658	.129
	STRAIGHT - (		(133.9)	(17.2)	(159.7)			
	>> 385F.1		2.772	.162	.453	.075	.720	.168
	STRAIGHT - (		(141.1)	(94.3)	(159.7)			
	>> 385F.2		2.077	.155	.453	.036	.739	.210
	STRAIGHT - (		(157.4)	(129.1)	(159.7)			
	>> 385F.3		1.740	.153	.453	.041	.704	.255
	STRAIGHT - (		(149.2)	(96.2)	(159.6)			
	>> 390N		.760	.072	.453	.133	.607	.301
	(		(197.9)	(52.1)	(159.6)			
	BEAD - (	390N	.766	.056	.446	.157	.579	.305

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	ALLO - (	590B	.766 (198.1)	.056 (51.5)	.446 (157.2)	.157	.579	.305
	>= 590F		.604	.049	.427	.179	.555	.309
	STRAIGHT - (	390F.1	.342 (97.5)	.049 (51.5)	.218 (84.1)	.222	.349	.293
	>= 395B		.093 (23.9)	.049 (51.2)	.021 (12.4)	.153	.143	.196
	bFEU - (	395B	.066 (18.5)	.043 (47.7)	.005 (3.8)	.123	.156	.164
	>= 395F		.040 (13.9)	.031 (58.7)	.000 (.1)	.092	.123	.136
ANCHOR		400	.000	.000	.000	.000	.000	.000

STEAKS - RUGER CORP

PROJECT 100% SOLAR PILET PLANT - C-21700  
JOB 03 VALVE TO THER STOR (A-MS-5-A-1)  
DATE 2/26/80

\* \* \* \* \*  
\* \* \* \* \*  
\* \* \* \* \*  
\* \* \* \* \*  
\* \* \* \* \*  
\* \* \* \* \*

STRESS SUMMARY

EXPLANATORY NOTES:

- (1) EQUATION NUMBERS REFER TO ARTICLE 104.6 OF THE ANSI 551.1-1977 PIPING CODE.
- (7) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE MODULUS IS USED FOR THE BRANCH LEG AT REDUCED OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK. THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.
- (9) THE COLD MODULUS WAS USED IN ANALYZING THE OCCASIONAL LOADS.

DYNAFLEX

STRESS SUMMARY (CONTINUED)

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LOC. NO.	STRESSES ARE IN PSI			
	ED (12)	ED (12)	ED (12)	ED (12)
	SHOCK 1	SHOCK 2	SHOCK 3	SHOCK 4
315	3906	1807	317	4315
320	2112	656	161	2220
325	2833	1170	226	3075
330	7624	3801	621	8541
330.1	5629	1760	311	5907
330.2	4005	2626	405	4810
330.3	3294	5013	764	6047
345N	3879	7531	1113	8544
345M	3759	7407	1111	8380
345F 06	3574	7062	1063	7986
345F.1	2220	4175	670	4776
345F.2	2007	1846	819	2848
350N	2970	2196	1136	3865
350M	3075	2356	1123	4033
350F	3177	2435	1054	4139
350F.1	3324	2237	596	4050
360N	3642	2025	477	4223
360M	3696	2011	497	4237
360F	3686	1897	496	4175
365	3545	1652	512	3945
365.1	3449	1968	602	4029

STRESS SUMMARY (CONTINUED)

LOC.	STRESSED ABL ID. PSI			
	EU (12)	FU (12)	EU (12)	FU (12)
	SHEET 1	SHEET 2	SHEET 3	SHEET 4
365.2	3493	2664	739	4454
365.3	3608	3360	751	4988
370	3762	3996	907	5563
370.1	3717	3151	799	4938
370.2	3671	2200	770	4350
370.3	3697	1837	590	4170
380N	3845	2877	503	4629
380M	3524	3009	520	4893
380F	3766	3022	520	4857
385N	3623	2944	520	4700
385N 10	3515	2914	524	4596
385F	3351	2842	520	4425
385F.1	1993	2328	630	3128
385F.2	1302	2273	681	2706
385F.3	2012	2669	571	3643
390M	4165	3370	587	5390
390M	4351	3323	585	5506
390F	4387	3026	542	5357
390F.1	4540	1680	286	4680
395N	5800	4163	460	7191
395M	6052	4526	927	7614

STRESS SUMMARY (CONTINUED)

DYNALLEX

STRESSES ARE IN PSI

EU (12) EU (12) EU (12) EU (12)

SHOCK 1 SHOCK 2 SHOCK 3 SHOCK 4

LOC. NO.

395F 6290 4633 947 7873

900 7493 4417 1928 8681



DYNALLEX

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LINE MS-4 MAIN STEAM - INTERNAL ANCHOR TO CONDENSER

PIPE STRESS ANALYSIS REPORT

FOR

IOMWe SOLAR PILOT PLANT

C-21700

MAIN STEAM - INTERNAL ANCHOR TO CONDENSER

PIPING SYSTEM

10"-MS-7 -FEA

95

CONDITIONS ANALYZED / ANAL. I.D. NO.

THERMAL T-21700-MS-4-A-5

DEAD WEIGHT W-21700-MS-4-A-3

PRESSURE P-21700-MS-4-A-0

SEISMIC X-21700-MS-4-A-1

CODE REFERENCE: ANSI B31.1-1973  
W/ APPENDA C.  
(SUMMER 1979)

ANALYST G.H.MAY

DATE 7-24-80



DIVISION USAGE					
MM	P	PP	SH	FI	SP

**Stearns-Roger**  
INCORPORATED  
**ENGINEERING STANDARD**

STANDARD NUMBER  
EE16.01.2

APPROVALS  
Des. Sect. \_\_\_\_\_  
Sect. Supv. \_\_\_\_\_  
Div. \_\_\_\_\_

PIPING ANALYSIS RESULTS

PAGE 1 OF 1  
ISSUED 8-31-73  
REVISED 4-15-74

Date FEB. 28, 1980

TO: T.E. OLSON  
FROM: G.H. MAY, Piping Engineering Group  
Client DOE Project SOLAR I Job No. C-21700  
Pipe Line Analyzed MAIN STEAM CONDENSER DUMP / 10"-MS-7-PEA  
Reference Dwg. P13-2 A

This piping has been analyzed for the THERMAL, DEAD WEIGHT, PRESSURE & OPERATIONAL BASIS EARTHQUAKE loading condition and is found to be:

- Satisfactory, as is.
- Satisfactory, with comments noted.
- Unsatisfactory - See Comments.

COMMENT (1) ALL PIPE STRESSES ARE SATISFACTORY.  
(2) ALL FORCES AND MOMENTS ARE REASONABLE AND SHOULD BE ACCEPTED.  
(3) ROUTING IN THE EPGS AREA IS "BEST ESTIMATE."

RECOMMENDATION (1) USE THE REFERENCED ROUTING.  
(2) SUBMIT THE CONDENSER FORCES AND MOMENTS TO SCE FOR APPROVAL.

ATTACHS: PIPE STRESS SUMMARY REPORT, SUMMARY OF FORCES & MOMENTS

G.H. MAY  
Signature

Copies to: Sender  
Analysis Folder - Job File  
Piping Engrg. Supervisor TCT  
K.E. NOBLE

<b>Stearns-Roger</b> <small>INCORPORATED</small>						<b>STANDARD NUMBER</b>
<b>ENGINEERING STANDARD</b>						EE16.01.8
<b>PIPE STRESS SUMMARY REPORT</b>						PAGE <u>1</u> OF <u>1</u>
<b>ANSI B31.1 - 1973 EDITION</b>						ISSUED 5/27/75
<b>APPROVALS</b>						REVISED
Des. Sect. <u>HY 2nd</u>						
Sect. Supv. <u>[Signature]</u>						
Div. _____						

10MW SOLAR PILOT PLANT - DOE  
Job Name

MAIN STEAM CONDENSER DUMP / 10" MS-7-FEA  
System/Pipe Line Name

C-21700  
Job No.

MS-4  
Analysis No.

**1. Loading Conditions Analyzed**

Pressure X

Weight X

Thermal Expansion X

Sustained Mech. Loads \_\_\_\_\_

Occasional Loads X

Analysis Ident. Code

D-MS-4-A-0

W-MS-4-A-3

T-MS-4-A-5

X-MS-4-A-1

**2. Stress Evaluation (Code Equations)**

Eq. (11)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} \leq 1.0 S_h$  1756 psi  $\leq$  13500 psi

Material: - ASTM A335 P11 Temp = 885°F @ Pt. 455N

Eq. (12)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} + \frac{0.75 iM_B}{z} \leq K S_h$  9396 psi  $\leq$  16200 psi

Material: - ASTM A335 P11 Temp = 885°F @ Pt. 455N

Eq. (13)  $S_E = \frac{iM_C}{z} \leq S_A$  14355 psi  $\leq$  19910 psi

Material: - ASTM A335 P11 Temp = 885°F @ Pt. 460F

Eq. (14)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} + \frac{iM_C}{z} \leq (S_h + S_A)$  15174 psi  $\leq$  33410 psi

Material: - ASTM A335 P11 Temp = 885°F @ Pt. 460F

**3. Stress Evaluation (Local or Special)**

Loading	Analysis Ident. Code	Calculated Stress (psi)	Allowable Stress (psi)
_____	_____	_____	_____
_____	_____	_____	_____

**REMARKS:**

- (1) ALL PIPE STRESSES ARE LESS THAN THE ALLOWABLES.
- (2) EQ. 12 INCLUDES STRESSES DUE TO AN OPERATIONAL BASIS EARTHQUAKE BASED ON GROUND RESPONSE SPECTRUM.

J. H. Miller 2-28-80  
Prepared By Date

[Signature] 2-28-80  
Approved By Date

CUSTOMER: DOE  
 PROJECT: 10MW SOLAR PILOT PLANT  
 JOB NO: C-21700  
 BY: GHM DATE: FEB. 28, 1980  
 REF. DWGS: P13-2 A  
 ANALYSIS CODE: T/W/X-MS-4-A-5/3/1

SUMMARY OF FORCES & MOMENTS  
 ON SYSTEM TERMINAL EQUIPMENT  
 ( SYSTEM )

MAIN STEAM CONDENSER PUMP LINE  
 10"-MS-7-FEA

DIVISION USAGE					
MM	P	PP	SH	FI	SP
X					

APPROVALS  
 Des. Sec'y *[Signature]*  
 Sect. Supv *[Signature]*  
 DWG *[Signature]*

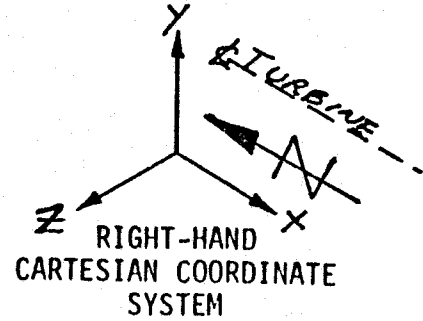
**Stearns-Roger**  
 INCORPORATED  
 ENGINEERING STANDARD

STANDARD NUMBER  
 EE 16.01.7

PAGE 1 OF 1

SUMMARY OF FORCES & MOMENTS  
 PER. ANSI B31.1-1973 PWR. PIPING CODE

ISSUED 2/28/74  
 REVISED 4/18/78



THE REPORTED REACTIONS BASED ON A THERMAL EXPANSION ANALYSIS FROM ..70..°F TO 885/350 F USING E<sub>c</sub>, THE COLD MOD. OF ELASTICITY, AND .....% COLD SPRING.

EQUIPMENT CONNECTIONS	LOC. NO.	FORCES (LBS)				MOMENTS (FT.-LBS)			
		X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
CONDENSER THERMAL	525	769	-733	408	1138	4100	3230	-1130	5402
CONN. INTERFACE WEIGHT	525	2	487	-3	487	-243	68	-508	567
12 SEISMIC	525	1641	246	440	1717	1541	5735	5710	7832





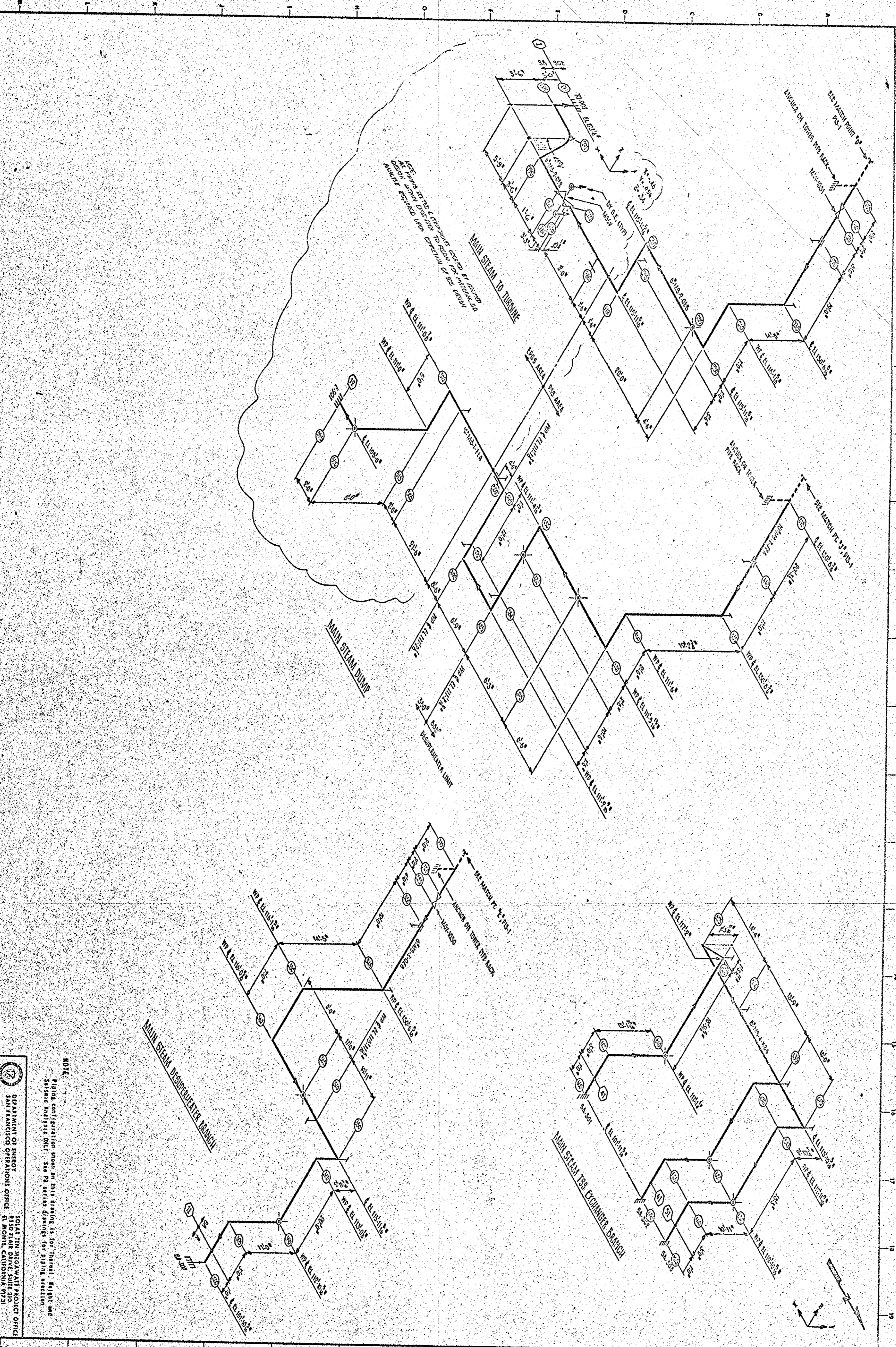
REVISIONS		DATE		BY		CHECKED	
1	ISSUED FOR PERMIT	11/11/11	11/11/11	...	...	...	...
2	...	...	...	...	...	...	...
3	...	...	...	...	...	...	...
4	...	...	...	...	...	...	...
5	...	...	...	...	...	...	...

EFFECTIVE DRAWINGS		DATE		BY		CHECKED	
1	...	...	...	...	...	...	...
2	...	...	...	...	...	...	...
3	...	...	...	...	...	...	...
4	...	...	...	...	...	...	...
5	...	...	...	...	...	...	...

MAIN RECORD		DATE		BY		CHECKED	
1	...	...	...	...	...	...	...
2	...	...	...	...	...	...	...
3	...	...	...	...	...	...	...
4	...	...	...	...	...	...	...
5	...	...	...	...	...	...	...



**NOTE:**  
 Piping configuration shown on this drawing is for Thermal, Nuclear and  
 System Analysis only. See P&ID sheets for piping configuration.

**SOLAR FACILITIES DESIGN INTEGRATOR**  
 101 ANALYSIS ISOMETRIC  
 MAIN STEAM - P&ID - ISS. A. E. P. S.  
 ANALYSIS ISOMETRIC (P&ID - 2)  
 407-2005-11-1971

**SOLAR THERMOPOWER PROJECT OFFICE**  
 101 ANALYSIS ISOMETRIC  
 MAIN STEAM - P&ID - ISS. A. E. P. S.  
 ANALYSIS ISOMETRIC (P&ID - 2)  
 407-2005-11-1971

BRANCH CONNECTION SYMBOLS	PIPE SUPPORT SYMBOLS	ANCHOR OR RESTRAINT NO.	DISPLACEMENT (COLD TO HOT) IN.			ANCHOR &/OR TERM. POINTS	COORDINATE		
			X	Y	Z		X	Y	Z
UNREINFORCED PIPE—PIPE INTERSECTION	CONSTANT SUPPORT	205	0.0	0.0	0.0	2053 MS-2	0'-0"	0'-0"	0'-0"
REINFORCED PIPE—PIPE INTERSECTION	VARIABLE SPRING SUPPORT	300	-4.0	0.54	.35	3000	48'-0 1/2"	-6'-10 3/16"	-38'-6"
ASA TEE	RIGID ROD OR STRUT	315	0.0	0.0	0.0	3153 MS-3	0'-0"	0'-0"	0'-0"
WELDOLET	HYDRAULIC SNUBBER	400				4000	39'-11"	-38'-4 3/16"	-58'-0"
SWEEPOLET	RIGID GUIDE	415				4153 MS-4	0'-0"	0'-0"	0'-0"
	ANCHOR	500				5000	33'-6"	-23'-2 3/8"	-33'-0 1/8"
		600				6000	0'-0"	0'-0"	0'-0"
		650				6503 MS-5	40'-1"	0'-0 1/2"	-0'-2"
		690				6900	-23'-1"	0'-0 1/2"	-0'-2"

MODELING SYMBOLS	PIPING SYMBOLS
POINT OF INTEREST	ELBOW
DATA POINT NO.	BEND
MASS POINT OR CENTER OF GRAVITY	VALVES (ASSUMED RIGID)
INTERFACE POINT	REDUCER OR INCREASER
	EQUIPMENT NOZZLE

ANALYZED CONDITIONS			VALVES & SPECIAL COMPONENTS			
OPERATING TEMP (°F)	DESIGN PRESSURE (PSIG)	CASE	VALVE OR SPEC. COMP. NO.	WGT. (LBS.)	STATUS EST. OR FINAL	REFERENCE & COMMENTS
960	1565	MS-2	MSSV	3000	EST	GE Prelim Turbine info.
960	1565	MS-3	MOV-1031	1315	EST	Powell 125023 WE
885	100	MS-4/PT 915-935	MOV-1030	1315	EST	" " "
350	100	MS-4/PT 935-525				
650	1335	MS-5				

NOTES:

- 1) DIMENSIONS ARE IN FEET AND INCHES UNLESS NOTED OTHERWISE.
- 2) ALL ELBOWS ARE ASA LONG RADIUS UNLESS NOTED OTHERWISE.
- 3) PIPE ANALYZED FROM COLD TO HOT.
- 4) THERMAL ANALYSIS TEMPERATURE RANGE IS FROM 70° F. TO OPERATING TEMPERATURE UNLESS NOTED OTHERWISE.
- 5) SCALE: NONE
- 6) THIS DATA AND ATTACHED DRAWINGS USED FOR ANALYSIS PURPOSES ONLY.

DATA POINTS		PIPE MATERIAL	LINE SPEC.	PIPE SIZE (IN.)		WEIGHT LBS./LINEAR FT.			INS. THK.
FROM	TO			O.D.	W.T.	PIPE	FLUID	INSULATION	
205 / 315	300 / 400	A335 P22	QEB	6.625	.864	54.3	3.1	15.2	5
415	484	↓ P11	FEH	10.75	.365	41.3	34.1	20.6	5
600	645 / 690	A106 Gr. B	KBA	8.625	.500	37.4	19.8	16.1	3 1/2
484	525	A335 P11	FEH	10.75	.365	41.3	34.1	8.7	3 1/2

NO.	REFERENCES	NO.	ANALYSIS CODES	NO.	REVISIONS	DATE	BY	CH'D	APP'D
			T/MS-MS-2-A-4/5/3	△					
			T/MS-MS-3-A-3/3/1	△					
			T/MS-MS-4-A-5/5/1	△					
			T/MS-MS-5-A-3/3/1	△					

**PIPING ANALYSIS DATA**

**Stearns-Roger** (PIB-2)  
INCORPORATED

10 MAINE SOLAR PILOT PLANT - DAGGET, CALIF.  
MAIN STEAM - PSS, TSS & EPG5  
FOR ISOMETRIC PIB 2

DRAWN: KEN	DATE: 2-12-80	ANAL. BY G.H.M.	DATE: 2-28-80
CHECKED: KEN	DATE: 2-3-80	APP'D. BY [Signature]	DATE: 3-19-80
JOB NO. C-21700	SYSTEM NO. MS-2, 3, 4, 5	103 SHEET 1 OF 1	



DYNAFLEX

STEARNS - ROGER CORP

PAGE 1

PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
 JOB - MS VALVE TO TURBINE DUMP(17W-MS-4-A-5/3)

DATE 2/25/80

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	PER	TO	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
1	PLD								STEARNS - ROGER CORP, 2/25/80, PROJ. 10MW SO LAR PILOT PLANT - C-2 1700, JOBS VALVE TO T URBINE DUMP(17W-MS-4- A-5/3)	1
2	VER								APPLY 831.1-1973, FEA	2
3			415	440			-20-3-1/8		FAT=LCP, UU=10.75, WT=.365, UNIF=20.62, TEMP=285.	3
4				455			-11-0	L		4
5				460		-19-2-3/8		L		5
6				470			-2-6	L		6
7				475:6-6				L		7
8				470:6-3				L		8
9				462			-2-6			9
10				460			-10-0			10
11				455			-1-6	L		11
12				460:6-9				L	TEMP=350., UNIF=8.67	12
13				460			1-6			13
14				465			12-0	L		14
15				500:4-0						15
16				505:26-0						16
17				517:2-0				L		17
18				515			-5-0	L		18
19				520		-6-0		L		19
20				525:25-0						20
21	ANC	415								21
22	ANC	525								22
23	FAU	400		RIGID						23
24	FAU	440			RIGID					24
25	FAU	455			RIGID					25
26	FAU	460			RIGID					26
27	FAU	462			RIGID					27
28	FAU	500			RIGID					28
29	FAU	500				RIGID				29
30	FAU	505			RIGID					30
31	PLD								LUC(475,482,520N)	31

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FOREWORD

THE FOLLOWING REPORT WAS GENERATED BY THE DYNAFLEX PIPING ANALYSIS PROGRAM. THE PROGRAM ACCOMMODATES THE WIDEST RANGE OF PRACTICAL SITUATIONS ENCOUNTERED IN MODERN PIPING SYSTEMS INCLUDING DYNAMIC AS WELL AS THERMAL, WEIGHT AND PRESSURE EFFECTS.

FOR FURTHER INFORMATION OR PROMPT ASSISTANCE IN USING THE DYNAFLEX PROGRAM, PLEASE CONTACT YOUR LOCAL COMPUTER CENTER REPRESENTATIVE OR:

ENGINEERING SERVICES MANAGER  
 ADION COMPUTING CORPORATION  
 1 METRO PLAZA  
 505 THORNALL STREET  
 EDISON, N.J. 08817

THE STRESS FORMULATION OF  
 ANSI B31.1 - 1977  
 INCLUDING THE LATEST MANDATORY UPDATES  
 HAS BEEN APPLIED IN THIS ANALYSIS

THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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*****
*
*
*
* STEARNS - ROGER CORP
*
*
* 100W SOLAR PILOT PLANT - C-21700
*
* 15 VALVE TO TURBINE PUMP (17, -6S-1-A-5/3)
*
*
* 2/25/80
*
*
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DYNAFLEX (R00204F-000)  
 LAST UPDATE - 12/25/79

105



DYNAFLEX

STEARNS - ROGER CORP

PAGE 3

PROJECT - 100W SOLAR PILET PLANT - C-21700  
JOB - MS VALVE IN TURBINE DUMP (1/W-MS-4-A-5/3)

DATE 2/25/80

\*\*\*\*\*  
\*  
\* EDITED PIPING SYSTEM DESCRIPTION \*  
\*-----\*  
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EXPLANATORY NOTESIN THE EDITED PIPING SYSTEM DESCRIPTION:

- (1) POINT LOCATION NUMBER SUFFIXES N, M AND F IDENTIFY THE NEAR POINT, THE MID POINT AND THE FAR POINT RESPECTIVELY OF A PIPING ELEMENT IN THE DIRECTION IN WHICH THE LINE IS TRAVERSED. FRACTIONAL NODE NUMBERS APPEARING IN THE FOLLOWING REPORTS IDENTIFY THOSE NODES AT WHICH THE PROGRAM HAS INTRODUCED EVENLY SPACED LUMP POINTS AND ARE DETERMINED FROM THE MAXIMUM LUMP SPACING CRITERIA ENTERED BY THE USER. IF FRACTIONAL NODE NUMBERS ARE ASSIGNED TO MORE THAN ONE ELEMENT ORIGINATING AT THE SAME BRANCH POINT, A LETTER WILL PRECEDE THE FRACTIONAL PORTION OF THE NUMBER TO UNIQUELY IDENTIFY EACH POINT (E.G. 2.A1). LETTERS ARE ASSIGNED TO ELEMENTS ALPHABETICALLY IN THE ORDER OF THEIR APPEARANCE IN THE INPUT DATA.
- (2) AMBIENT TEMPERATURE IS ASSUMED TO BE 70 DEGREES FAHRENHEIT.
- (3) THE MAXIMUM ALLOWABLE DIFFERENCE IN LOOP CLOSURE CALCULATIONS IS ASSUMED TO BE .25 INCHES.
- (4) THE MAXIMUM ALLOWABLE DIFFERENCE BETWEEN THE CALCULATED AND INPUT COORDINATES OF USER DESIGNATED CHECK POINTS IS ASSUMED TO BE .25 INCHES.
- (5) IN ORDER TO MAKE IT POSSIBLE IN CERTAIN SITUATIONS TO PERFORM INDEPENDENT WEIGHT AND THERMAL ANALYSES IN THE SAME RUN AND APPLY THE CODE STRESS CALCULATIONS CORRECTLY, THE PROGRAM MAKES CERTAIN ASSUMPTIONS. IF THE INPUT DATA SPECIFIES IMPOSED MOVEMENTS OR ANCHORS (ANC) OR DISPLACEMENTS OR ROTATIONS ON RIGID RESTRAINTS (RAD,RAR) THESE ARE ASSUMED BY THE PROGRAM TO BE THERMAL EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE THERMAL. ALTERNATELY, IF THE INPUT DATA SPECIFIES FORCES AND MOMENTS (FOR,MOM) OR IMPOSED DISPLACEMENTS OR ROTATIONS ON FLEXIBLE RESTRAINTS (RAD,RAR) THESE ARE ASSUMED TO BE WEIGHT EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE WEIGHT. IF THESE ASSUMPTIONS ARE INVALID FOR A PARTICULAR SITUATION, THE USER CAN ACCURATELY MODEL THE MOST GENERAL SITUATION BY USING THE AUXILIARY SUPPORT MOVEMENTS (ASM) AND THE AUXILIARY CONCENTRATED LOADS (ACL) FEATURE OF THE PROGRAM AS AN ALTERNATE METHOD OF SPECIFYING THESE EFFECTS.
- (6) A POINT IDENTIFIED AS A TENTATIVE HANGER LOCATION IS ASSUMED TO BE FREE TO MOVE IN THE VERTICAL DIRECTION FOR THE INITIAL THERMAL LOADING CASE AND FULLY RESTRAINED IN THE VERTICAL DIRECTION FOR THE INITIAL WEIGHT LOADING CASE.
- (7) THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.





ELATED PIPING SYSTEM DESCRIPTION (CONTINUED)

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
		>440	0.00	0.00	-20.25				10.750	.365	LOC: 085.	41.3	20.6
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 440											
		( IN DIRECTION X											
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 440											
		( IN DIRECTION Y											
		(											
	STRAIGHT	(				9.75							
		(											
		>455N	0.00	0.00	-30.01								
		(											
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 455N											
		( IN DIRECTION Y											
		(											
		(											
	BEND	( 455E	0.00	-.37	-30.89	1.25	1.96	90.000					
		(											
		>455E	0.00	-1.25	-31.26								
	STRAIGHT	(				16.70							
		(											
		>460N	0.00	-17.95	-31.26								
		(											
110	BEND	( 460E	0.00	-18.83	-31.63	1.25	1.96	90.000					
		(											
		>460E	0.00	-19.20	-32.51								
		(											
		( * POINT 460E IS EQUIVALENT TO POINT 470N											
		(											
		(											
		(											
	BEND	( 470N	.37	-19.20	-33.39	1.25	1.96	90.000					
		(											
		>470E	1.25	-19.20	-33.76								
	STRAIGHT	(				5.25							
		(											
		>475	6.50	-19.20	-33.76								

\* POINTS PERTAINING TO POINT 475 APPEAR ON THE FOLLOWING PAGE



DNVAFLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 7

POINT TYPE	ELEMENT DESIGNATION	POINT NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	MATERIAL	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z								PRESS (PSI)	PIPE WT. UNIF
		>4900	18.25	-19.20	-48.25				10.750	.365	LCM	350.	41.3	8.7
	BEND	( 4900	19.13	-19.20	-47.89	1.25	1.96	90.000						
		(												
	STRAIGHT	(	>490F	19.50	-19.20	-47.01								
		(	>492	19.50	-19.20	-46.75								
		(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 492											
		(	IN DIRECTION Y											
	STRAIGHT	(					10.75							
		(	>4950	19.50	-19.20	-36.01								
	BEND	( 4950	19.27	-19.20	-35.13	1.25	1.96	90.000						
		(	>495F	20.75	-19.20	-34.76								
	STRAIGHT	(					2.75							
		(	>500	23.50	-19.20	-34.76								
		(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 500											
		(	IN DIRECTION Z											
		(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 500											
		(	IN DIRECTION Y											
	STRAIGHT	(					26.00							
		(	>505	49.50	-19.20	-34.76								
		(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 505											
		(	IN DIRECTION Y											
	STRAIGHT	(					.75							
		(	>510F	50.25	-19.20	-34.76								
	BEND	( 5100	51.13	-19.20	-35.13	1.25	1.96	90.000						
		(	>510F	51.50	-19.20	-36.61								

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POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
								:10.750:	:.365:	LCM:	350.:	: 41.3:	8.7:
	STRAIGHT	( >-S10F	51.50	-19.20	-36.01		2.50						
		( >-S15F	51.50	-19.20	-36.51								
	BEND	( S15F	51.50	-19.56	-39.39	1.25	1.96	90.000					
		( >-S15F	51.50	-20.05	-39.76		3.50						
		( >-S20F	51.50	-23.95	-39.76								
		( * TENTATIVE HANGER AT POINT S20F - SEE NOTE (9) ABOVE.											
		( HANGER LENGTH NOT SPECIFIED.											
		(											
		(											
	BEND	( S20F	51.87	-24.83	-39.76	1.25	1.96	90.000					
		( >-S20F	52.75	-25.20	-39.76		.75						
	STRAIGHT	(											
	ANCHP	(--S25	53.50	-25.20	-39.76								
		( * CONTROL COORDINATES OF POINT S25, IN FEET :											
		( X = 53.50, Y = -25.20, Z = -39.76											
		( RESULTANT DIFFERENCE IS ZERO											

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 \*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
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EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)				
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT	
<b>ANCHORS:</b>									
415	277.	-86.	461.	909.	-579.	-1869.	-4803.	5186.	
525	769.	-733.	408.	1138.	4180.	3230.	-1130.	5402.	
<b>RESTRAINTS:</b>									
400	-1648.	0.	0.	1648.	0.	0.	0.	0.	
406	0.	848.	0.	848.	0.	0.	0.	0.	
455	0.	310.	0.	310.	0.	0.	0.	0.	
484	0.	-1816.	0.	1816.	0.	0.	0.	0.	
492	0.	1545.	0.	1545.	0.	0.	0.	0.	
500	0.	0.	-1269.	1269.	0.	0.	0.	0.	
503	0.	-852.	0.	852.	0.	0.	0.	0.	
505	0.	784.	0.	784.	0.	0.	0.	0.	

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\* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
\*-----\*  
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THE MAXIMUM STRESS OF 14,355. OCCURS AT POINT 460F

EXPLANATORY NOTES:  
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- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ANSI B31.1 - 1977  
PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
(AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

DYNAFLEX

LOADING - VREF% THERMAL

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT ELEMENT TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	EXPANSION : STRESS PSI
		AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF-PLANE		
ANCHOR	-- 415	861.	292.	4605.	579.	1869.	1.00	1.00	29.90	2081.
STR - (	>- 466	861.	1082.	4605.	1176.	3779.	1.00	1.00	29.90	2497.
STR - (	>- 455D	861.	1319.	4805.	6255.	3718.	2.61	2.61	29.90	9114.
BEND - (	455E	1366.	783.	6306.	6887.	88.	2.61	2.61	29.90	9762.
STR - (	>- 455F	1072.	1154.	4679.	6519.	3842.	2.61	2.61	29.90	9300.
STR - (	>- 460E	1072.	1154.	4679.	7853.	8997.	2.61	2.61	29.90	13408. *
BEND - (	460F	1366.	783.	3535.	8222.	10349.	2.61	2.61	29.90	14251. *
STR - (	>- 466F	861.	1319.	9958.	7590.	5640.	2.61	2.61	29.90	14355. **
BEND - (	470E	1152.	1074.	2067.	6004.	11461.	2.61	2.61	29.90	13697. *
STR - (	>- 470F	769.	1374.	6250.	5525.	8618.	2.61	2.61	29.90	12539.
STR - (	>- 475	769.	1374.	6250.	2992.	1006.	1.00	1.00	29.90	2810.
STR - (	>- 480E	769.	1374.	6250.	3297.	2366.	2.61	2.61	29.90	7790.
BEND - (	480F	1152.	1074.	6485.	3777.	1800.	2.61	2.61	29.90	8067.
STR - (	>- 480F	861.	1319.	3705.	3412.	4911.	2.61	2.61	29.90	7354.
STR - (	>- 482	861.	1319.	3705.	3571.	2451.	1.00	1.00	29.90	2287.
STR - (	>- 484	861.	1319.	3705.	7620.	5622.	1.00	1.00	29.90	4699.
STR - (	>- 485D	861.	1076.	3705.	5610.	7494.	2.61	2.61	29.90	10645.
BEND - (	485E	1152.	747.	2407.	6178.	7262.	2.61	2.61	29.90	10280.
STR - (	>- 485F	769.	1138.	6564.	5699.	2775.	2.61	2.61	29.90	9540.
STR - (	>- 490D	769.	1138.	6564.	2041.	587.	2.61	2.61	29.90	7198.
BEND - (	490E	861.	1374.	4096.	999.	5575.	2.61	2.61	29.90	7505.
STR - (	>- 490F	861.	1070.	1517.	4.	7494.	2.61	2.61	29.90	7954.

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DYNAFLEX

LOADING - CREEP THERMAL

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	MEM TYPE	CUL. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	EXPANSION STRESS ( PSI )
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 490F	861.	1070.	1517.	4.	7494.	2.61	2.61	29.90	7954.
	STR - (	>- 492	861.	1110.	1517.	7680.	188.	1.00	1.00	29.90	3128.
	STR - (	>- 495N	861.	1110.	1517.	8453.	931.	2.61	2.61	29.90	8996.
	BEND - (	>- 495B	65.	1403.	1683.	9408.	456.	2.61	2.61	29.90	10081.
	STR - (	>- 495F	769.	1176.	1933.	10490.	315.	2.61	2.61	29.90	11156.
	STR - (	>- 500	769.	411.	1933.	1888.	12857.	1.00	1.00	29.90	5272.
	STR - (	>- 505	769.	839.	1933.	552.	2246.	1.00	1.00	29.90	1210.
	STR - (	>- 510B	769.	839.	1933.	1940.	1101.	2.61	2.61	29.90	3086.
	BEND - (	>- 510N	255.	1109.	2414.	1298.	60.	2.61	2.61	29.90	2866.
	STR - (	>- 510F	408.	1062.	2017.	469.	1016.	2.61	2.61	29.90	2412.
	STR - (	>- 515N	408.	1062.	2017.	816.	1453.	2.61	2.61	29.90	2735.
117	BEND - (	>- 515B	250.	1114.	2735.	1613.	280.	2.61	2.61	29.90	3332.
	STR - (	>- 515F	733.	870.	2414.	2242.	1056.	2.61	2.61	29.90	3617.
	STR - (	>- 520B	733.	870.	2414.	1635.	3670.	2.61	2.61	29.90	4900.
	BEND - (	>- 520F	1062.	409.	1038.	2046.	4663.	2.61	2.61	29.90	5432.
	STR - (	>- 520F	769.	839.	4180.	1680.	2924.	2.61	2.61	29.90	5615.
	ANCHR	-- 525	769.	839.	4180.	1139.	3230.	1.00	1.00	29.90	2168.



DYNAFLEX

LOADING - ZERO THERMAL

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 \*  
 \* INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES \*  
 \*-----\*  
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EXPLANATORY NOTES

THE FORCES AND MOMENTS AT ANY POINT SHOWN BELOW REPRESENT THE ACTION OF THE FAR END OF THE BRANCH ACTING ON THE NEAR END OF THE BRANCH.

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
ANCHOR		415	279.	-86.	861.	-579.	-1869.	-4803.
	STRAIGHT - (	> 440	-769.	762.	861.	1170.	3779.	-4803.
	STRAIGHT - (	> 455	-769.	1072.	861.	-6255.	-3718.	-4803.
	BEND - (	455	-769.	1072.	861.	-6887.	-4397.	-4521.
	STRAIGHT - (	> 459	-769.	1072.	861.	-6519.	-4679.	-3842.
	BEND - (	460	-769.	1072.	861.	7853.	-4679.	8997.
118	BEND - (	460	-769.	1072.	861.	8222.	-4960.	9676.
	STRAIGHT - (	> 469	-769.	1072.	861.	7590.	-5640.	9958.
	BEND - (	470	-769.	1072.	861.	6643.	-6004.	9565.
	STRAIGHT - (	> 479	-769.	1072.	861.	6250.	-5525.	8618.
	STRAIGHT - (	> 475	-769.	1072.	861.	6250.	-1006.	2992.
	BEND - (	480	-769.	1072.	861.	6250.	3297.	-2366.
	STRAIGHT - (	> 470	-769.	1072.	861.	5858.	3777.	-3313.
	STRAIGHT - (	> 476	-769.	1072.	861.	4911.	3412.	-3705.
	STRAIGHT - (	> 482	-769.	1072.	861.	3571.	2451.	-3705.
	STRAIGHT - (	> 474	-769.	1072.	861.	-7660.	-5622.	-3705.
	BEND - (	485	-769.	-744.	861.	-7494.	-5814.	-3705.
	BEND - (	485	-769.	-744.	861.	-6837.	-6178.	-3433.

INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES (CONTINUED)

POINT DESIGNATION	ELEMENT NO.	LOC.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
BEND	4950	(	-769.	-700.	861.	-6637.	-6178.	-3433.
STRAIGHT	4951	)	-769.	-700.	861.	-6564.	-5699.	-2775.
	4990	(	-769.	-700.	861.	-6564.	-2041.	387.
BEND	4900	(	-769.	-700.	861.	-6637.	-999.	1044.
STRAIGHT	4901	)	-769.	-700.	861.	-7094.	-4.	1317.
STRAIGHT	492	(	-769.	801.	861.	-7680.	188.	1317.
BEND	4950	(	-769.	801.	861.	931.	6453.	1317.
STRAIGHT	4951	)	-769.	801.	861.	1639.	9408.	1023.
STRAIGHT	4951	(	-769.	801.	861.	1933.	10490.	315.
STRAIGHT	500	(	-769.	-51.	-408.	1933.	12857.	-1888.
STRAIGHT	505	(	-769.	733.	-408.	1933.	2206.	-552.
STRAIGHT	5100	(	-769.	733.	-408.	1933.	1900.	-1101.
BEND	5100	(	-769.	733.	-408.	1664.	1298.	-1749.
STRAIGHT	5101	)	-769.	733.	-408.	1016.	469.	-2017.
STRAIGHT	5151	(	-769.	733.	-408.	-816.	-1453.	-2017.
BEND	5150	(	-769.	733.	-408.	-1613.	-2133.	-1736.
STRAIGHT	5151	)	-769.	733.	-408.	-2202.	-2414.	-1056.
BEND	5200	(	-769.	733.	-408.	-3670.	-2014.	1635.
STRAIGHT	5201	)	-769.	733.	-408.	-4031.	-2563.	2046.
STRAIGHT	5201	(	-769.	733.	-408.	-4180.	-2928.	1680.
ANCHOR	525	(	-769.	733.	-408.	-4180.	-3230.	1130.

DYNALLEX

LOADING - CIRCULAR THERMAL

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 \* DISPLACEMENTS AND ROTATIONS \*  
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DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 415	.000	-.000	.000	-.000	-.000	-.000
	STRAIGHT - (	>- 411	-.000	.000	-1.547	.010	.033	-.217
	STRAIGHT - (	>- 455A	-.091	.000	-2.291	-.032	.034	-.322
	BEND - (	455B	-.119	-.042	-2.352	-.123	.004	-.326
	STRAIGHT - (	>- 455F	-.182	-.121	-2.348	-.216	.000	-.358
	BEND - (	460A	-1.415	-1.396	-1.502	-.197	-.174	-.284
	STRAIGHT - (	>- 460F	-1.441	-1.473	-1.503	-.066	-.229	-.158
	BEND - (	470A	-1.303	-1.496	-1.606	.142	-.408	-.030
120	STRAIGHT - (	>- 470F	-1.202	-1.475	-1.550	.203	-.488	.099
	STRAIGHT - (	>- 475	-.801	-1.332	-.993	.277	-.517	.151
	STRAIGHT - (	>- 480A	-.420	-1.168	-.453	.346	-.507	.154
	BEND - (	480B	-.316	-1.113	-.391	.361	-.458	.143
	STRAIGHT - (	>- 480F	-.208	-1.031	-.425	.411	-.408	.149
	STRAIGHT - (	>- 482	-.102	-.922	-.520	.420	-.401	.138
	STRAIGHT - (	>- 484	.784	-.000	-1.322	.383	-.430	.051
	STRAIGHT - (	>- 485	.806	.020	-1.341	.380	-.432	.049
	BEND - (	485A	.921	.083	-1.371	.283	-.515	.008
	STRAIGHT - (	>- 485F	1.030	.099	-1.296	.243	-.598	-.052
	STRAIGHT - (	>- 490A	1.126	.046	-.749	.181	-.626	-.061

DYNAFLEX

LOADING - 0EREG THERMAL

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 490H	1.126	.046	-.749	.181	-.626	-.061
		(						
	BEND	(- 490H	1.097	.024	-.622	.149	-.647	-.028
		(						
	STRAIGHT	>- 490F	.985	.003	-.552	.061	-.653	.003
		(						
	STRAIGHT	>- 492	.950	.000	-.547	.058	-.653	.003
		(						
		>- 495H	-.461	-.029	-.303	-.004	-.574	.035
		(						
	BEND	(- 495H	-.549	-.027	-.245	.006	-.452	.035
		(						
	STRAIGHT	>- 495F	-.560	-.021	-.166	.011	-.315	.036
		(						
	STRAIGHT	>- 500	-.498	-.000	-.000	.023	-.260	.032
		(						
	STRAIGHT	>- 505	.049	.000	-.279	.135	.077	-.022
		(						
		>- 510H	.106	-.003	.267	.136	.080	-.023
		(						
	BEND	(- 510H	.119	.002	.241	.140	.103	-.033
		(						
121	STRAIGHT	>- 510F	.107	.026	.213	.149	.115	-.036
		(						
		>- 515H	.046	.105	.156	.149	.113	-.047
		(						
	BEND	(- 515H	.023	.123	.126	.133	.099	-.048
		(						
	STRAIGHT	>- 515F	.006	.113	.095	.106	.095	-.056
		(						
		>- 520H	-.036	.034	.023	.088	.076	-.054
		(						
	BEND	(- 520H	-.035	.011	.007	.034	.051	-.028
		(						
	STRAIGHT	>- 520F	-.017	.000	.000	.007	.004	-.002
		(						
ANCHOR		-- 525	.000	-.000	.000	.000	.000	-.000



DYNAFLEX

LOADING - RESTRAINED WEIGHT

(+UNIF)

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 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
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EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)				
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT	
<b>ANCHORS:</b>									
415	11.	-687.	-11.	687.	-2519.	-73.	55.	2520.	
525	2.	487.	-3.	487.	-243.	68.	-568.	567.	
<b>RESTRAINTS:</b>									
440	-15.	0.	0.	15.	0.	0.	0.	0.	
440	0.	-863.	0.	863.	0.	0.	0.	0.	
123 4554	0.	-1640.	0.	1640.	0.	0.	0.	0.	
475	0.	-770.	0.	770.	0.	0.	0.	0.	
482	0.	-421.	0.	421.	0.	0.	0.	0.	
484	0.	-551.	0.	551.	0.	0.	0.	0.	
492	0.	-585.	0.	585.	0.	0.	0.	0.	
500	0.	0.	14.	14.	0.	0.	0.	0.	
500	0.	-1146.	0.	1146.	0.	0.	0.	0.	
505	0.	-657.	0.	657.	0.	0.	0.	0.	

DYNAFLEX

LOADING = UNRESTRAINED WEIGHT (+UNIF)

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FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

FORCES (POUNDS)

MOMENTS (FOOT-POUNDS)

LOC.  
NO.

X Y Z RESULTANT

X Y Z RESULTANT

RESTRAINTS (CONTINUED)

LOC. NO.	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
5200	0.	-916.	0.	916.	0.	0.	0.	0.

DYNAPLEY

LOADING - RESTRAINED WEIGHT

(\*UNIF)

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 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*-----\*  
 \*  
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THE MAXIMUM STRESS OF 1094. OCCURS AT POINT 455N

EXPLANATORY NOTES:  
-----

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ANSI B31.1 - 1977  
PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
(AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (15) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLEX

LOADING - PRESRAINED WEIGHT

(+CONT)

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP		-- 415	11.	687.	55.	2519.	73.	1.00	1.00	29.90	1011. **
	STR - (	> 440	11.	568.	55.	1313.	147.	1.00	1.00	29.90	531. :
	STR - (	> 455N	11.	1330.	55.	1388.	128.	2.61	2.61	29.90	1094. **:
	BEND - (	455M	905.	899.	129.	242.	50.	2.61	2.61	29.90	219. :
		> 455F	1208.	11.	126.	205.	58.	2.61	2.61	29.90	194. :
	STR - (	> 460N	173.	11.	126.	22.	90.	2.61	2.61	29.90	123. :
	BEND - (	460M	87.	72.	154.	61.	24.	2.61	2.61	29.90	131. :
		> 460F	11.	52.	93.	128.	124.	2.61	2.61	29.90	157. :
	BEND - (	470M	6.	13.	36.	118.	173.	2.61	2.61	29.90	167. :
		> 470F	2.	71.	137.	107.	60.	2.61	2.61	29.90	145. :
	STR - (	> 475	2.	375.	137.	1161.	50.	1.00	1.00	29.90	470. :
126	STR - (	> 480N	2.	66.	137.	5.	61.	2.61	2.61	29.90	118. :
	BEND - (	480M	6.	10.	124.	15.	83.	2.61	2.61	29.90	118. :
		> 480F	11.	57.	36.	21.	122.	2.61	2.61	29.90	101. :
	STR - (	> 482	11.	287.	36.	2.	24.	1.00	1.00	29.90	17. :
	STR - (	> 484	11.	167.	36.	404.	44.	1.00	1.00	29.90	164. :
	STR - (	> 485N	11.	172.	36.	44.	359.	2.61	2.61	29.90	285. :
	BEND - (	485M	6.	111.	173.	50.	155.	2.61	2.61	29.90	186. :
		> 485F	2.	51.	199.	61.	83.	2.61	2.61	29.90	175. :
	STR - (	> 490N	2.	163.	199.	107.	156.	2.61	2.61	29.90	215. :
	BEND - (	490M	4.	212.	316.	116.	136.	2.61	2.61	29.90	285. :
		> 490F	11.	261.	403.	116.	82.	2.61	2.61	29.90	336. :

DYNAFLEX

LOADING - RESTRAINED WEIGHT (+UNIF)

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR	>- 490F	11.	261.	403.	118.	82.	2.61	2.61	29.90	336.
	STR	>- 492	11.	312.	403.	149.	118.	1.00	1.00	29.90	179.
	STR	>- 495N	11.	226.	403.	97.	315.	2.61	2.61	29.90	407.
	BEND	>- 495N	9.	275.	417.	99.	286.	2.61	2.61	29.90	404.
	STR	>- 495F	2.	324.	14.	108.	764.	2.61	2.61	29.90	605.
	STR	>- 500	2.	462.	14.	1844.	138.	1.00	1.00	29.90	742.
	STR	>- 503	2.	241.	14.	959.	69.	1.00	1.00	29.90	386.
	STR	>- 510N	2.	203.	14.	67.	793.	2.61	2.61	29.90	624.
	BEND	>- 510N	3.	154.	502.	65.	594.	2.61	2.61	29.90	503.
	STR	>- 510F	3.	105.	583.	66.	190.	2.61	2.61	29.90	484.
	STR	>- 515N	3.	20.	583.	297.	71.	2.61	2.61	29.90	516.
127	BEND	>- 515N	51.	47.	463.	260.	360.	2.61	2.61	29.90	503.
	STR	>- 515F	114.	3.	73.	231.	581.	2.61	2.61	29.90	493.
	STR	>- 520N	623.	5.	73.	574.	240.	2.61	2.61	29.90	491.
	BEND	>- 520N	464.	467.	223.	356.	120.	2.61	2.61	29.90	342.
	STR	>- 520F	2.	524.	243.	129.	70.	2.61	2.61	29.90	223.
	ANCHR	-- 525	2.	487.	243.	508.	68.	1.00	1.00	29.90	228.

DYNAFLEX

LOADING - RESTRAINED WEIGHT

(+UNIF)

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 \*  
 \* INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES \*  
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EXPLANATORY NOTES

THE FORCES AND MOMENTS AT ANY POINT SHOWN BELOW REPRESENT THE ACTION OF THE FAR END OF THE BRANCH ACTING ON THE NEAR END OF THE BRANCH.

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
ANCHOR		-- 415	11.	-647.	-11.	-2519.	-73.	55.
	STRAIGHT -(	> 440	11.	568.	-11.	-1313.	147.	55.
	STRAIGHT -(	> 4550	-2.	-1330.	-11.	-1388.	128.	55.
	END -(	4550	-2.	-1269.	-11.	-242.	127.	56.
	STRAIGHT -(	> 4551	-2.	-1208.	-11.	205.	126.	58.
	END -(	4600	-2.	-173.	-11.	22.	126.	90.
128	END -(	4600	-2.	-113.	-11.	61.	125.	92.
	STRAIGHT -(	> 4601	-2.	-52.	-11.	128.	124.	93.
	END -(	4700	-2.	9.	-11.	148.	118.	97.
	STRAIGHT -(	> 4701	-2.	70.	-11.	137.	107.	60.
	STRAIGHT -(	> 475	-2.	-375.	-11.	137.	50.	-1161.
	STRAIGHT -(	> 4800	-2.	-65.	-11.	137.	-5.	-61.
	END -(	4850	-2.	-4.	-11.	146.	-15.	-29.
	STRAIGHT -(	> 4801	-2.	57.	-11.	122.	-21.	-36.
	STRAIGHT -(	> 482	-2.	-287.	-11.	2.	-24.	-36.
	STRAIGHT -(	> 484	-2.	-147.	-11.	-464.	-74.	-36.
	STRAIGHT -(	> 4850	-2.	-172.	-11.	-359.	-44.	-36.
	END -(	4851	-2.	-111.	-11.	-232.	-50.	13.

INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES (CONTINUED)

POINT DESTINATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
	BEND - (	4850	-2.	-111.	-11.	-252.	-50.	13.
	>-	485F	-2.	-50.	-11.	-199.	-61.	83.
	STRAIGHT - (	4900	-2.	162.	-11.	-199.	-107.	-156.
	>-	490F	-2.	211.	-11.	-128.	-116.	-320.
	BEND - (	4960	-2.	261.	-11.	82.	-118.	-403.
	>-	496F	-2.	-312.	-11.	149.	-118.	-403.
	STRAIGHT - (	4950	-2.	226.	-11.	-313.	-97.	-403.
	>-	495F	-2.	275.	-11.	-93.	-99.	-498.
	BEND - (	4950	-2.	324.	-11.	14.	-108.	-764.
	>-	500	-2.	462.	-11.	14.	-138.	-1844.
	STRAIGHT - (	505	-2.	-241.	3.	14.	-69.	-959.
	>-	5100	-2.	-203.	3.	14.	-67.	-793.
129	BEND - (	5100	-2.	-154.	3.	77.	-65.	-634.
	>-	510F	-2.	-105.	3.	190.	-66.	-583.
	STRAIGHT - (	5150	-2.	20.	3.	297.	-71.	-583.
	>-	5150	-2.	69.	3.	260.	-73.	-582.
	BEND - (	5150	-2.	118.	3.	231.	-73.	-581.
	>-	5200	-2.	-623.	3.	240.	-73.	-574.
	STRAIGHT - (	5200	-2.	-574.	3.	242.	-72.	-356.
	>-	520F	-2.	-524.	3.	243.	-70.	129.
	ANCHOR	525	-2.	-567.	3.	243.	-68.	508.

DYNAFLEX

LOADING - RESTRAINED WEIGHT

(+UNIF)

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 \* DISPLACEMENTS AND ROTATIONS \*  
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POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 415	.000	-.000	-.000	-.000	-.000	.000
	STRAIGHT -(	>- 440	-.000	-.000	-.000	.007	.001	.002
	STRAIGHT -(	>- 4550	-.005	-.000	-.000	-.007	.004	.004
	BEND -(	4550	-.006	-.003	.001	-.018	.005	.003
	STRAIGHT -(	>- 4550F	-.005	-.004	.005	-.017	.005	.004
		>- 4600	.011	-.005	.058	-.014	.010	.006
	BEND -(	4600	.011	-.006	.060	-.013	.010	.006
		>- 4600F	.010	-.008	.061	-.012	.011	.006
	BEND -(	4700	.008	-.010	.060	-.010	.013	.007
130		>- 4700F	.007	-.009	.058	-.009	.014	.009
	STRAIGHT -(	>- 475	.007	-.000	.041	-.008	.015	.005
	STRAIGHT -(	>- 4800	.007	.002	.025	-.006	.015	.001
	BEND -(	4800	.005	.002	.023	-.006	.015	.001
		>- 4800F	.003	.001	.021	-.004	.015	.001
	STRAIGHT -(	>- 480	-.001	-.000	.021	-.004	.015	.001
	STRAIGHT -(	>- 484	-.033	-.000	.021	.003	.014	.000
		>- 4850	-.054	.000	.021	.003	.014	.000
	BEND -(	4850	-.057	.000	.020	-.001	.014	-.001
		>- 4850F	-.050	-.000	.018	-.002	.013	-.001
	STRAIGHT -(	>- 4900	-.058	-.000	.007	-.004	.012	-.000

DYNAFLEX

LOADING - RESTRAINED WEIGHT

(+UNIF)

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 4900	-.032	-.000	.007	-.004	.012	-.000
		(						
	BEND	- ( 4900	-.037	-.000	.005	-.003	.011	-.003
		(						
	STRAIGHT	- ( >- 490F	-.035	-.000	.004	-.002	.009	-.004
		(						
	STRAIGHT	- ( >- 492	-.035	-.000	.004	-.002	.009	-.004
		(						
		>- 4950	-.017	.015	.004	-.013	.007	-.014
		(						
	BEND	- ( 4950	-.015	.016	.003	-.013	.006	-.015
		(						
	STRAIGHT	- ( >- 495F	-.015	.014	.002	-.011	.004	-.022
		(						
	STRAIGHT	- ( >- 500	-.015	-.000	.000	-.011	.004	-.028
		(						
	STRAIGHT	- ( >- 505	-.015	-.000	-.006	-.011	-.001	.035
		(						
		>- 5100	-.015	.005	-.006	-.011	-.001	.034
		(						
	BEND	- ( 5100	-.015	.010	-.006	-.013	-.002	.026
		(						
131	STRAIGHT	- ( >- 510F	-.014	.009	-.006	-.013	-.003	.024
		(						
		>- 5150	-.013	.002	-.006	-.012	-.003	.020
		(						
	BEND	- ( 5150	-.011	.001	-.005	-.008	-.002	.018
		(						
	STRAIGHT	- ( >- 515F	-.008	-.000	-.004	-.005	.000	.012
		(						
		>- 5200	-.001	-.000	-.000	-.004	-.001	.008
		(						
	BEND	- ( 5200	.000	.000	-.000	-.001	.000	.001
		(						
	STRAIGHT	- ( >- 520F	.000	.000	.000	-.000	.000	-.000
		(						
ANCHOR		-- 525	.000	.000	-.000	-.000	.000	-.000

DYNAFLEX

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## HANGER DESIGN DATA

## EXPLANATORY NOTES:

- (1) ALL MOVEMENTS SHOWN BELOW REFER TO THE INITIAL EXPANSION LOADING CASE WHEREIN TENTATIVE HANGER LOCATIONS HAVE NOT BEEN VERTICALLY RESTRAINED AND ALL WEIGHT EFFECTS HAVE BEEN SUPPRESSED.
- (2) THE WEIGHT REACTIONS SHOWN BELOW REFER TO THE WEIGHT LOADING CASE WHEREIN TENTATIVE HANGER LOCATIONS HAVE BEEN FIRMLY RESTRAINED IN THE VERTICAL DIRECTION AND ALL THERMAL EFFECTS INCLUDING IMPOSED ANCHOR MOVEMENTS HAVE BEEN SUPPRESSED.
- (3) IN THE FOLLOWING TABLE VSH DENOTES A VARIABLE SPRING HANGER AND CSH DENOTES A CONSTANT SUPPORT HANGER.
- (4) THE SUPPORT LOADS IN THE INSTALLED CONDITION HAVE BEEN CALCULATED TO PRODUCE THE CORRECT WEIGHT BALANCE IN THE OPERATING CONDITION.
- (5) IF THE FREE VERTICAL MOVEMENT EXCEEDS THE MAXIMUM RIGID SUPPORT DISPLACEMENT CRITERION OF .10 IN., A VARIABLE SPRING IS SELECTED. IF A VARIABLE SPRING HANGER CANNOT BE FOUND WHICH SATISFIES BOTH THE LOAD VARIATION CRITERION SELECTED BY THE USER AND THE WORKING RANGE OF THE SPRINGS LISTED IN SPRING TABLE (1) OF THE USER'S MANUAL, A CONSTANT SUPPORT HANGER IS RECOMMENDED.
- (6) THE NOMINAL ROD DIAMETER FOR RIGID HANGERS IS TAKEN FROM ANSI B31.1 - 1977 TABLE 121.2.2(A), (CARRYING CAPACITIES OF THREADED ASTM A 107 HOT ROLLED CARBON STEEL) ASSUMING A ROD TEMPERATURE OF 650 DEGREES FAHRENHEIT. A MINIMUM 1/2 IN. DIAMETER ROD WILL BE SELECTED FOR ALL PIPE WHICH IS NOMINALLY 2-1/2 IN. OR LARGER.
- (8) THE SO-CALLED THEORETICAL SPRING INSTALLATION LOAD SHOWN BELOW PRESUPPOSES THAT THE HANGER LOCATION IS RESTRAINED AGAINST VERTICAL MOVEMENT WHILE THE SPRING IS SET TO THE COLD LOAD.

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HANGER DESIGN DATA TABLE

HANGER LOC. NO.	SUPPORT LOAD (POUNDS)	FREE VERTICAL MOVEMENT (INCHES)	HORIZONTAL MOVEMENT (INCHES)		TYPE	PRELIMINARY HANGER SELECTION		SWING	
						SPECIFICATION		LENGTH (FEET)	ANGLE (DEG)
475	770.	-1.33	-0.00	-0.99	VSH	1 - MIDDLE RANGE SPRING. SPRING RATE = 150 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 570 LBS SPRING LOAD IN THE OPERATING CONDITION = 770 LBS		N/A	N/A
482	421.	-0.92	-0.10	-0.52	VSH	1 - MIDDLE RANGE SPRING. SPRING RATE = 84 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 343 LBS SPRING LOAD IN THE OPERATING CONDITION = 421 LBS		N/A	N/A
520N	916.	.03	-0.00	.02	RIGID	REQUIRED MINIMUM CROSS SECTIONAL ROGT AREA = .13 SQ IN BASED ON AN ALLOWABLE STRESS OF 9000 PSI		N/A	N/A



DYNAFLEX

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*
*   STATIC SOLUTION ACCURACY CHECKS   *
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EACH BASIC STATIC SOLUTION COMPRISING EACH LOAD OR LOADING COMBINATION SPECIFIED IN THIS RUN HAS BEEN SUBJECTED TO EQUILIBRIUM AND COMPATIBILITY CHECKS FOR ALL POINTS IN THE SYSTEM:

LOADING - 2PIEPC INERTIAL

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 STATIC EQUILIBRIUM AND COMPATIBILITY HAVE BEEN SATISFIED FOR THIS BASIC STATIC SOLUTION AT ALL POINTS IN THE SYSTEM.

LOADING - CONSTRAINED WEIGHT (+UNIF)

-----  
 STATIC EQUILIBRIUM AND COMPATIBILITY HAVE BEEN SATISFIED FOR THIS BASIC STATIC SOLUTION AT ALL POINTS IN THE SYSTEM.

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FOREWORD  
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THE FOLLOWING REPORT WAS GENERATED BY THE DYNAFLEX PIPING ANALYSIS PROGRAM. THE PROGRAM ACCOMMODATES THE WIDEST RANGE OF PRACTICAL SITUATIONS ENCOUNTERED IN MODERN PIPING SYSTEMS INCLUDING DYNAMIC AS WELL AS THERMAL, HEIGHT AND PRESSURE EFFECTS.

FOR FURTHER INFORMATION OR PROMPT ASSISTANCE IN USING THE DYNAFLEX PROGRAM, PLEASE CONTACT YOUR LOCAL COMPUTER CENTER REPRESENTATIVE OR:

ENGINEERING SERVICES MANAGER  
AUBURN COMPUTING CORPORATION  
1 METRO PLAZA  
505 THORNALL STREET  
EDISON, N.J. 08817

THE STRESS FORMULATION OF  
ANSI B31.1 - 1977  
INCLUDING THE LATEST MANDATORY UPDATES  
HAS BEEN APPLIED IN THIS ANALYSIS

THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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*
* STEARNS - ROGER CORP, 2/26/80, PROJ. 10MW
* SOLAR PILOT PLANT - C-21700, JONES VALVE
* ID 11700, JONES VALVE TO TURBINE DUMP (X=M
* S-1-A-1)
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DYNAFLEX (MOD204F-UCC)  
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LAST UPDATE - 12/25/79

NY:AFLEX

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STEARNS - ROGER CORP, 27  
 26/80, PROJ. 10MW SOLAR PI  
 LOT PLANT - C-21700, JUMP  
 S VALVE TO 1700, JUMP S V  
 TURBINE DUMP (X-MS  
 S-4-A-1)

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYP	LOC	FRE	TO	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
1	GEN								STEARNS - ROGER CORP, 27/26/80, PROJ. 10MW SO EAR PILOT PLANT - C-2 1700, JUMPS VALVE TO T 1700, JUMPS VALVE TO T TURBINE DUMP (X-MS-4-A- (1)	1
2	GEN								APPLY R31.1-1973, EXA	2
3			415	440			-20-3-178		RAT=LCP, CO=10.75, WT=.365, UNIF=20.62, TEMP=885.	3
4				455			-11-0	L		4
5				460			-19-2-378	L		5
6				470			-2-6	L		6
7				475	6-6					7
8				480	6-5			L		8
9				482			-2-8			9
10				484			-10-6			10
11				485			-1-6	L		11
12				490	6-9			L	TEMP=350., UNIF=8.67	12
13				492			1-6			13
14				495			12-0	L		14
15				500	4-0					15
16				505	26-0					16
17				515	2-0			L		17
18				515			-5-0	L		18
19				520			-6-0	L		19
20				525	2-1					20
21		400	415							21
22		400	525							22
23		400	400		RIGID					23
24		400	400			RIGID				24
25		400	455			RIGID				25
26		400	484			RIGID				26
27		400	482			RIGID				27
28		400	500			RIGID				28
29		400	500				RIGID			29
30		400	505			RIGID				30

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INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC NO.	FREQ	THU	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
31	CCC	525			53-6	-25-2-3/8	-39-9-1/8			31
32	CYC								SPEC 1(FREQ, .17.013, .257.082, 2.57.638, 9.7.531, 33.7.125), SPEC 2(FREQ, .17.009, .257.055, 2.57.638, 9.7.531, 33.7.125), SHOCK 1(X/1.0SPEC 1), SHOCK 2(Z/1.0 SPEC 1), SHOCK 3(Y/1.SPEC2), SHOCK 4(X/1.SPEC1, Y/1.SPEC2, Z/1.SPEC1), LUG, 1.92	32
33	DPG								CURP POINTS(AUTOMATIC), MAX SPACING=5-0, EXCLLDE ALL M NODES, MODE CUTOFF=20, FREQUENCY CUTOFF=33	33
34	POT								SHOCK 1, SHOCK 2, SHOCK 3, SHOCK 4	34

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STEARNS - ROGER CORP, 27  
267RD, PROJ. 101W SULLY PI  
LOT PLANT - C-21700, JOBM  
S VALVE TO ~~11700, JOBM~~  
~~AT THE TURBINE DRIVE (X-M~~  
S-4-A-1)

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\*  
\* EDITED PIPING SYSTEM DESCRIPTION \*  
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EXPLANATORY NOTES

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IN THE EDITED PIPING SYSTEM DESCRIPTION:

- (1) POINT LOCATION NUMBER SUFFIXES N, M AND F IDENTIFY THE NEAR POINT, THE MID POINT AND THE FAR POINT RESPECTIVELY OF A PIPING ELEMENT IN THE DIRECTION IN WHICH THE LINE IS TRAVERSED. FRACTIONAL NODE NUMBERS APPEARING IN THE FOLLOWING REPORTS IDENTIFY THOSE NODES AT WHICH THE PROGRAM HAS INTRODUCED EVENLY SPACED LUMP POINTS AND ARE DETERMINED FROM THE MAXIMUM LUMP SPACING CRITERIA ENTERED BY THE USER. IF FRACTIONAL NODE NUMBERS ARE ASSIGNED TO MORE THAN ONE ELEMENT ORIGINATING AT THE SAME BRANCH POINT, A LETTER WILL PRECEDE THE FRACTIONAL PORTION OF THE NUMBER TO UNIQUELY IDENTIFY EACH POINT (E.G. 2.A1). LETTERS ARE ASSIGNED TO ELEMENTS ALPHABETICALLY IN THE ORDER OF THEIR APPEARANCE IN THE INPUT DATA.
- (2) AMBIENT TEMPERATURE IS ASSUMED TO BE 70 DEGREES FAHRENHEIT.
- (3) THE MAXIMUM ALLOWABLE DIFFERENCE IN LUMP CLOSURE CALCULATIONS IS ASSUMED TO BE .25 INCHES.
- (4) THE MAXIMUM ALLOWABLE DIFFERENCE BETWEEN THE CALCULATED AND INPUT COORDINATES OF USER DESIGNATED CHECK POINTS IS ASSUMED TO BE .25 INCHES.
- (5) IN ORDER TO MAKE IT POSSIBLE IN CERTAIN SITUATIONS TO PERFORM INDEPENDENT WEIGHT AND THERMAL ANALYSES IN THE SAME RUN AND APPLY THE CODE STRESS CALCULATIONS CORRECTLY, THE PROGRAM MAKES CERTAIN ASSUMPTIONS. IF THE INPUT DATA SPECIFIES IMPOSED MOVEMENTS ON ANCHORS (ANC) OR DISPLACEMENTS OR ROTATIONS ON RIGID RESTRAINTS (RAD, RAR) THESE ARE ASSUMED BY THE PROGRAM TO BE THERMAL EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE THERMAL. ALTERNATELY, IF THE INPUT DATA SPECIFIES FORCES AND MOMENTS (FOR, MOM) OR IMPOSED DISPLACEMENTS OR ROTATIONS ON FLEXIBLE RESTRAINTS (RAD, RAR) THESE ARE ASSUMED TO BE WEIGHT EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE WEIGHT. IF THESE ASSUMPTIONS ARE INVALID FOR A PARTICULAR SITUATION, THE USER CAN ACCURATELY MODEL THE MOST GENERAL SITUATION BY USING THE AUXILIARY SUPPORT MOVEMENTS (ASM) AND THE AUXILIARY CONCENTRATED LOADS (ACL) FEATURE OF THE PROGRAM AS AN ALTERNATE METHOD OF SPECIFYING THESE EFFECTS.
- (15) THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

GENERAL DATA

-----  
COORDINATES OF ALL POINTS OF THE SYSTEM ARE WITH RESPECT TO POINT #15, WHICH IS THE ORIGIN

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DYNAFLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 4

MATERIALS USED (TEMPERATURE DEPENDENT PROPERTIES ARE SHOWN AT POINT OF TEMPERATURE CHANGE)

LCR - LOW CHROME MOLY STEEL  
 YOUNG'S MODULUS AT AMBIENT TEMPERATURE = 29,900,000. PSI  
 POISSON'S RATIO = .30  
 DENSITY = 500. POUNDS PER CUBIC FOOT

SHOCK AND SPECTRUM DATA

MODE CUTOFF SET AT 20

FREQUENCY CUTOFF SET AT 33.00 CPS

SPECTRUM 1		SPECTRUM 2	
(LOGARITHMIC INTERPOLATION)		(LOGARITHMIC INTERPOLATION)	
FREQUENCY	G	FREQUENCY	G
---CPS	-----	---CPS	-----
.10	.01	.10	.01
.25	.08	.25	.06
2.50	.64	2.50	.64
9.00	.53	9.00	.53
33.00	.13	33.00	.13

SHOCK SPECIFICATIONS

SHOCK 1 CONSISTS OF :  
 100.00 PERCENT OF SPECTRUM 1 IN THE X DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

SHOCK 2 CONSISTS OF :  
 100.00 PERCENT OF SPECTRUM 1 IN THE Z DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.



DYNAFLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 5

SHOCK 3 CONSISTS OF :

100.00 PERCENT OF SPECTRUM 2 IN THE Y DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD.

CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

SHOCK 4 CONSISTS OF :

100.00 PERCENT OF SPECTRUM 1 IN THE X DIRECTION AND

100.00 PERCENT OF SPECTRUM 2 IN THE Y DIRECTION AND

100.00 PERCENT OF SPECTRUM 1 IN THE Z DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD.

CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

LOADING CONDITIONS ANALYZED

- SHOCK 1
- SHOCK 2
- SHOCK 3
- SHOCK 4

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POINT Type	ELEMENT DESIGNATION	POINT LOC. ID.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) MATERIAL (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							WT.	UNIF
ANCHOR		--415	0.00	0.00	0.00								
STRAIGHT	-(					4.05		10.750	.365	LCM: 885.		41.3	20.6
	*)												
	*)												
	*)												
	*)												
	*)												
	*)												
	*)												
STRAIGHT	-(	>-415.1	0.00	0.00	-4.05								
STRAIGHT	-(	>-415.2	0.00	0.00	-4.10		4.05						
STRAIGHT	-(	>-415.3	0.00	0.00	-12.15		4.05						

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
								10.750	.365	LCM	885.	41.3	20.6
STRAIGHT	(	>-415.3	0.00	0.00	-12.16		4.05						
STRAIGHT	(	>-415.4	0.00	0.00	-16.21		4.05						
	(	>-440	0.00	0.00	-20.25								
	(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 440											
	(	IN DIRECTION X											
	(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 440											
	(	IN DIRECTION Y											
STRAIGHT	(						4.88						
STRAIGHT	(	>-440.1	0.00	0.00	-25.10		4.88						
	(	>-455W	0.00	0.00	-30.01								
	(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 455W											
	(	IN DIRECTION Y											
	(												
	(												
BEAD	(	455E	0.00	-1.37	-30.89	1.25	1.96	90.000					
STRAIGHT	(	>-455E	0.00	-1.25	-31.26		4.17						
STRAIGHT	(	>-455E.1	0.00	-5.42	-31.26		4.17						
STRAIGHT	(	>-455E.2	0.00	-9.60	-31.26		4.17						
STRAIGHT	(	>-455E.3	0.00	-13.77	-31.26		4.17						
	(	>-460F	0.00	-17.95	-31.26								
BEAD	(	460E	0.00	-10.83	-31.63	1.25	1.96	90.000					
	(	>-460E	0.00	-19.20	-32.51								

\* NOTES PERTAINING TO POINT 460E APPEAR ON THE FOLLOWING PAGE

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DYNAFLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

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DIST. LOADS  
(LBS PER FT)

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	MATERIAL	TEMP (FAHR)	DISTRIBUTED LOADS (LBS PER FT)	
			X	Y	Z								PRESS (PSI)	WT. UNIF
		>460F	0.00	-19.20	-32.51				10.750	.365	LCM	885	41.3	20.6
		(												
		(* POINT 460F IS EQUIVALENT TO POINT 470F												
		(												
		(												
		HEAD - ( 470F	.37	-19.20	-33.49	1.25	1.96	90.000						
		(												
		>470F	1.25	-19.20	-33.76									
	STRAIGHT - (						2.63							
		>470F.1	3.88	-19.20	-33.76									
	STRAIGHT - (						2.63							
		>475	6.50	-19.20	-33.76									
	STRAIGHT - (						2.50							
		>475.1	9.00	-19.20	-33.76									
	STRAIGHT - (						2.50							
		>480F	11.50	-19.20	-33.76									
		(												
		HEAD - ( 480F	12.34	-19.20	-34.14	1.25	1.96	90.000						
		(												
144		>480F	12.75	-19.20	-35.01									
	STRAIGHT - (						1.25							
		>482	12.75	-19.20	-36.26									
	STRAIGHT - (						3.50							
		>482.1	12.75	-19.20	-39.76									
	STRAIGHT - (						3.50							
		>482.2	12.75	-19.20	-43.26									
	STRAIGHT - (						3.50							
		>484	12.75	-19.20	-46.76									
		(												
		(* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 484												
		( IN DIRECTION Y												
		(												
		STRAIGHT - (					.25							
		(												
		>485F	12.75	-19.20	-47.01									
		(												
		HEAD - ( 485F	13.12	-19.20	-47.39	1.25	1.96	90.000						
		(												
		>485F	14.00	-19.20	-48.26									

\* POINTS 480F THROUGH TO POINT 485F APPEAR ON THE FOLLOWING PAGE



POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) MATERIAL (FAHR)	DIST. LOADS (LBS PER FT)		
			X	Y	Z							PRESS (PSI)	PIPE UNIF	
		>-500	23.50	-19.20	-34.76				10.750	.365	LCM	350	41.3	8.7
		(												
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 500												
		( IN DIRECTION Y												
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 500												
		( IN DIRECTION Z												
	STRAIGHT	-					4.33							
		(												
	STRAIGHT	>-500.1	27.83	-19.20	-34.76		4.33							
		(												
	STRAIGHT	>-500.2	32.17	-19.20	-34.76		4.33							
		(												
	STRAIGHT	>-500.3	36.50	-19.20	-34.76		4.33							
		(												
	STRAIGHT	>-500.4	40.83	-19.20	-34.76		4.33							
		(												
	STRAIGHT	>-500.5	45.17	-19.20	-34.76		4.33							
		(												
146		>-505	49.50	-19.20	-34.76									
		(												
		( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 505												
		( IN DIRECTION Y												
	STRAIGHT	-					.75							
		(												
		>-510	50.25	-19.20	-34.76									
		(												
	BEND	-	51.13	-19.20	-35.13	1.25	1.96	90.000						
		(												
	STRAIGHT	>-510F	51.50	-19.20	-36.01		2.50							
		(												
		>-515	51.50	-19.20	-36.51									
		(												
	BEND	-	51.50	-19.56	-39.33	1.25	1.96	90.000						
		(												
	STRAIGHT	>-515F	51.50	-20.05	-39.76		3.50							
		(												
		>-520	51.50	-24.95	-39.76									
		(												
	BEND	-	51.57	-24.95	-39.76	1.25	1.96	90.000						



DYNAFLEX

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\*\*\*\*\*  
 \*  
 \* LUMPED DYNAMIC MODEL \*  
 \*-----\*  
 \*  
 \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

- (1) THE MASS OF THE SYSTEM IS LUMPED AT THE POINTS SHOWN BELOW. FRACTIONAL LOCATION NUMBERS, IF ANY, RESULT FROM ADDITIONAL MASS POINTS LOCATED BY THE PROGRAM BASED ON THE MAXIMUM LUMP SPACING CRITERION GIVEN IN THE INPUT DATA.
- (2) THE MASS AT A POINT IS A COMBINATION OF DISTRIBUTED MASS LUMPED IN A LINEAR FASHION PLUS ANY ADDITIONAL CONCENTRATED MASS INPUT BY THE USER (SHOWN IN PARENTHESES).
- (3) ROTATIONAL DEGREES OF FREEDOM ARE NOT INCLUDED IN THE MODEL UNLESS SPECIFICALLY ADDED BY THE USER. ALL THREE DISPLACEMENT DEGREES OF FREEDOM ARE AUTOMATICALLY DELETED AT ANCHORS. THE DEGREE OF FREEDOM ASSOCIATED WITH A RIGID RESTRAINT OR SHOCKER ORIENTED IN THE X, Y OR Z DIRECTION IS ALSO DELETED.

DISPLACEMENT FREEDOM

ROTATIONAL FREEDOM

LOC. NO.	WEIGHT (POUNDS)	DIRECTION			MOMENT OF INERTIA (LB - IN <sup>2</sup> IN)			
					X	Y	Z	
415	125.				0.	0.	0.	
148	415.1	251.	X	Y	Z	0.	0.	0.
	415.2	251.	X	Y	Z	0.	0.	0.
	415.3	251.	X	Y	Z	0.	0.	0.
	415.4	251.	X	Y	Z	0.	0.	0.
440	277.			Z	0.	0.	0.	
440.1	302.	X	Y	Z	0.	0.	0.	
455B	212.			Z	0.	0.	0.	
455F	190.	X	Y	Z	0.	0.	0.	
455F.1	259.	X	Y	Z	0.	0.	0.	
455F.2	259.	X	Y	Z	0.	0.	0.	
455F.3	259.	X	Y	Z	0.	0.	0.	
460H	190.	X	Y	Z	0.	0.	0.	

LUMPED DYNAMIC MODEL (CONTINUED)

LOC. NO.	HEIGHT (INCHES)	DISPLACEMENT FREEDOM			ROTATIONAL FREEDOM			
		X	Y	Z	MOMENT OF INERTIA (LB-IN <sup>2</sup> )	X	Y	Z
460F	122.	X	Y	Z	0.	0.	0.	0.
470F	142.	X	Y	Z	0.	0.	0.	0.
470F.1	165.	X	Y	Z	0.	0.	0.	0.
475	159.	X	Y	Z	0.	0.	0.	0.
475.1	155.	X	Y	Z	0.	0.	0.	0.
480F	135.	X	Y	Z	0.	0.	0.	0.
480F	100.	X	Y	Z	0.	0.	0.	0.
482	147.	X	Y	Z	0.	0.	0.	0.
482.1	217.	X	Y	Z	0.	0.	0.	0.
482.2	217.	X	Y	Z	0.	0.	0.	0.
484	115.	X	Y	Z	0.	0.	0.	0.
485F	69.	X	Y	Z	0.	0.	0.	0.
485F	167.	X	Y	Z	0.	0.	0.	0.
490F	155.	X	Y	Z	0.	0.	0.	0.
490F	55.	X	Y	Z	0.	0.	0.	0.
492	96.	X	Y	Z	0.	0.	0.	0.
492.1	176.	X	Y	Z	0.	0.	0.	0.
492.2	173.	X	Y	Z	0.	0.	0.	0.
495F	130.	X	Y	Z	0.	0.	0.	0.
495F	115.	X	Y	Z	0.	0.	0.	0.
500	177.	X	Y	Z	0.	0.	0.	0.
500.1	217.	X	Y	Z	0.	0.	0.	0.
500.2	217.	X	Y	Z	0.	0.	0.	0.



DYNAFLEX

LOOPEO DYNAMIC MODEL (CONTINUED)

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LIC. NO.	DISPLACEMENT FREEDOM			ROTATIONAL FREEDOM			
	WEIGHT (POUNDS)	DIRECTION			MOMENT OF INERTIA (LB - IN X IN)		
		X	Y	Z	X	Y	Z
500.3	217.	X	Y	Z	0.	0.	0.
500.4	217.	X	Y	Z	0.	0.	0.
500.5	217.	X	Y	Z	0.	0.	0.
505	127.	X		Z	0.	0.	0.
5100	68.	X	Y	Z	0.	0.	0.
510F	112.	X	Y	Z	0.	0.	0.
5150	112.	X	Y	Z	0.	0.	0.
515F	137.	X	Y	Z	0.	0.	0.
5200	137.	X	Y	Z	0.	0.	0.
520F	68.	X	Y	Z	0.	0.	0.
525	19.				0.	0.	0.

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NUMBER OF DYNAMIC DEGREES OF FREEDOM = 127, TOTAL MASS OF SYSTEM = 7949. POUNDS

\*\*\*\*\*  
 \* FREQUENCIES, PERIODS OF VIBRATION AND MASS PARTICIPATION FACTORS \*  
 \*\*\*\*\*

## MASS PARTICIPATION FACTORS

MODE NO.	FREQUENCY (CPS)	PERIOD (SECONDS)	MASS PARTICIPATION FACTORS		
			X	Y	Z
1	1.90055	.52616	3.18847055	.19734389	.51729546
2	2.29081	.43577	.39934866	.05364096	2.14600820
3	5.36132	.18652	1.84092416	.03236287	1.03682192
4	5.90181	.16944	.82598151	1.22215502	.01870259
5	6.69972	.14926	.20199215	.83437125	.25487094
6	8.80444	.11358	.53352223	.62929805	.13437585
7	9.23293	.10831	.57422589	.29945055	.29759387
8	11.00991	.09083	.42321912	.24158472	1.85664116
9	12.21764	.08185	.32656904	1.25248901	.32021996
10	19.05302	.05249	.08555753	.09100971	1.22503314
11	23.49000	.04274	.05532389	.12709661	.71966985
12	24.39459	.04099	.38504365	.08513607	.22510560
13	27.36154	.03655	.17309114	.03512859	.93335955
14	29.38517	.03403	.01501677	1.57166913	.05681710
15	36.26057	.02735	1.54785242	.02300938	.02642861

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## NOTE:

SECTION 3.7.2 OF THE U.S. NUCLEAR REGULATORY COMMISSION STANDARD REVIEW PLAN DATED JUNE, 1975 STATES THAT AN ADEQUATE NUMBER OF DEGREES OF FREEDOM IN DYNAMIC MODELING MAY BE TAKEN EQUAL TO TWICE THE NUMBER OF MODES WITH FREQUENCIES LESS THAN 33 CPS.

FOR THIS ANALYSIS THE RATIO OF DYNAMIC DEGREES OF FREEDOM TO FREQUENCIES LESS THAN 33 CPS = 8.47

THE COIL ELASTIC MODULUS WAS USED IN CALCULATING THE NATURAL FREQUENCIES OF THE SYSTEM IN THIS ANALYSIS.

DYNAPLEX

MINI ORTHOGONALITY CHECK  
-----

ON DIAGONAL TERMS: MAX. = 1.00000 MIN. = 1.00000

OFF DIAGONAL TERMS: MAX. = 1.61338 X 10<sup>-13</sup> MIN. = -1.30390 X 10<sup>-13</sup>

DYNAFLEX

LOADING - SHOCK LOADING NO. 1

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GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
415	445.	34.	287.	530.	229.	2950.	4445.	5340.
525	1554.	153.	298.	1589.	1124.	4924.	5440.	7423.
<b>RESTRAINTS:</b>								
153	440	1098.	0.	0.	1098.	0.	0.	0.
	440	0.	332.	0.	332.	0.	0.	0.
	455N	0.	206.	0.	206.	0.	0.	0.
	484	0.	129.	0.	129.	0.	0.	0.

LOADING - SHOCK LOADING NO. 1

FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM OF THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
	X	Y	Z	X	Y	Z
492	0.	144.	0.	144.	0.	0.
500	0.	343.	0.	343.	0.	0.
509	0.	0.	568.	568.	0.	0.
505	0.	216.	0.	216.	0.	0.

RESTRAINTS (CONTINUED)

```

*****
*
* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS
*
*****

```

THE MAXIMUM STRESS OF 6426. OCCURS AT POINT 485F

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
ASME B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLE MODEL HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLX

LOADING - SHEAR LOADING NO. 1

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		:	:
ANCHR	STR - (	-- 415	287.	446.	4445.	229.	2950.	1.00	1.00	29.90	:	2143.
		>- 415.1	287.	442.	4445.	90.	1153.	1.00	1.00	29.90	:	1843.
	STR - (	>- 415.2	287.	442.	4445.	48.	642.	1.00	1.00	29.90	:	1802.
		>- 415.3	287.	429.	4445.	186.	2390.	1.00	1.00	29.90	:	2027.
	STR - (	>- 415.4	287.	429.	4445.	322.	4111.	1.00	1.00	29.90	:	2433.
		>- 440	286.	739.	4445.	458.	5817.	1.00	1.00	29.90	:	2944.
	STR - (	>- 440.1	286.	722.	4445.	999.	2542.	1.00	1.00	29.90	:	2094.
		>- 455E	286.	684.	4445.	2452.	801.	2.61	2.61	29.90	:	4030.
	BEND - (	455H	370.	642.	3674.	2524.	2128.	2.61	2.61	29.90	:	3990.
		>- 455F	262.	693.	1537.	2348.	3679.	2.61	2.61	29.90	:	3628.
	STR - (	>- 455F.1	264.	611.	1537.	1288.	1183.	1.00	1.00	29.90	:	934.
156		>- 455F.2	264.	611.	1537.	1289.	347.	1.00	1.00	29.90	:	817.
	STR - (	>- 455F.3	265.	527.	1537.	3092.	1134.	1.00	1.00	29.90	:	1458.
		>- 460N	266.	321.	1537.	2061.	4494.	2.61	2.61	29.90	:	4059.
	BEND - (	460M	314.	275.	2309.	2158.	4399.	2.61	2.61	29.90	:	4247.
		>- 460F	205.	363.	4786.	2046.	1796.	2.61	2.61	29.90	:	4317.
	BEND - (	470E	256.	277.	2221.	1865.	4541.	2.61	2.61	29.90	:	4224.
		>- 470F	185.	330.	1801.	1775.	4483.	2.61	2.61	29.90	:	4035.
	STR - (	>- 470F.1	126.	418.	1801.	3855.	1405.	1.00	1.00	29.90	:	1798.
		>- 475	185.	312.	1801.	3235.	1148.	1.00	1.00	29.90	:	1555.
	STR - (	>- 475.1	284.	411.	1801.	2643.	1052.	1.00	1.00	29.90	:	1351.
		>- 480N	341.	319.	1801.	1116.	2039.	2.61	2.61	29.90	:	2305.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS ( PSI )
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF-PLANE		
		>- 480N	341.	319.	1801.	1116.	2039.	2.61	2.61	29.90	2305.
	HEND	( 480P	257.	390.	682.	1098.	2392.	2.61	2.61	29.90	2132.
		>- 480F	142.	445.	1721.	592.	1471.	2.61	2.61	29.90	1938.
	STR	(									
		>- 482	161.	574.	1721.	1130.	969.	1.00	1.00	29.90	913.
	STR	(									
		>- 482.1	193.	681.	1721.	443.	2122.	1.00	1.00	29.90	1111.
	STR	(									
		>- 482.2	235.	780.	1721.	1328.	4102.	1.00	1.00	29.90	1863.
	STR	(									
		>- 483	235.	780.	1721.	2537.	6509.	1.00	1.00	29.90	2887.
	STR	(									
		>- 485N	275.	838.	1721.	6695.	2611.	2.61	2.61	29.90	5794. *
	HEND	( 485M	655.	589.	3158.	7336.	940.	2.61	2.61	29.90	6305. **
		>- 485F	781.	406.	2982.	7514.	1350.	2.61	2.61	29.90	6426. ***
	STR	(									
		>- 490N	944.	442.	2982.	6978.	290.	2.61	2.61	29.90	5954. *
157	HEND	( 490M	609.	446.	2176.	6537.	1912.	2.61	2.61	29.90	5606. :
		>- 490F	326.	990.	433.	5876.	2616.	2.61	2.61	29.90	4912. :
	STR	(									
		>- 492	330.	1015.	433.	2543.	5436.	1.00	1.00	29.90	2415. :
	STR	(									
		>- 492.1	349.	1095.	433.	1925.	1915.	1.00	1.00	29.90	1104. :
	STR	(									
		>- 492.2	363.	1138.	433.	1316.	2251.	1.00	1.00	29.90	1061. :
	STR	(									
		>- 495N	373.	1158.	433.	6173.	750.	2.61	2.61	29.90	4887. :
	HEND	( 495M	759.	951.	779.	7234.	132.	2.61	2.61	29.90	5705. :
		>- 495F	1144.	414.	543.	7804.	579.	2.61	2.61	29.90	6153. *
	STR	(									
		>- 500	1185.	313.	593.	1007.	8371.	1.00	1.00	29.90	3392. :
	STR	(									
		>- 500.1	1224.	309.	593.	467.	7559.	1.00	1.00	29.90	3048. :
	STR	(									
		>- 500.2	1224.	309.	593.	897.	6761.	1.00	1.00	29.90	2747. :



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LOADING - SHOCK LOADING NO. 1

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS ( PSI )
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 500.2	1224.	309.	593.	897.	6761.	1.00	1.00	29.90	2747.
	STR - (	>- 500.3	1267.	304.	593.	1594.	5950.	1.00	1.00	29.90	2484.
	STR - (	>- 500.4	1315.	303.	593.	2287.	5108.	1.00	1.00	29.90	2258.
	STR - (	>- 500.5	1364.	309.	593.	2949.	4247.	1.00	1.00	29.90	2089.
	STR - (	>- 505	1425.	320.	593.	3603.	3431.	1.00	1.00	29.90	2011.
	STR - (	>- 510N	1479.	324.	593.	3305.	3644.	2.61	2.61	29.90	3885.
	BEND - (	510N	931.	1194.	3014.	2627.	2213.	2.61	2.61	29.90	3583.
	STR - (	>- 510F	283.	1488.	3719.	1316.	526.	2.61	2.61	29.90	3121.
	STR - (	>- 515N	292.	1539.	3719.	599.	2595.	2.61	2.61	29.90	3582.
	BEND - (	515N	274.	1545.	4997.	579.	615.	2.61	2.61	29.90	3974.
	STR - (	>- 515F	159.	1559.	4477.	394.	1806.	2.61	2.61	29.90	3798.
158	STR - (	>- 520N	153.	1582.	4477.	3622.	763.	2.61	2.61	29.90	4555.
	BEND - (	520N	1148.	1099.	2630.	4971.	3857.	2.61	2.61	29.90	5347.
	STR - (	>- 520F	1554.	335.	1124.	5485.	4752.	2.61	2.61	29.90	5758.
ANCHR		-- 525	1554.	335.	1124.	5440.	4924.	1.00	1.00	29.90	2979.

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LOADING - SHOCK LOADING NO. 1

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\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )  
 DISPLACEMENTS (INCHES)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 415	.000	.000	.000	.000	.000	.000
	STRAIGHT - (		(.0)	(.0)	(.0)			
	>- 415.1		.007	.001	.000	.001	.014	.040
	STRAIGHT - (		(23.5)	(.4)	(.0)			
	>- 415.2		.021	.002	.000	.001	.016	.080
	STRAIGHT - (		(58.0)	(1.1)	(.0)			
	>- 415.3		.032	.002	.000	.000	.006	.121
	STRAIGHT - (		(66.5)	(1.3)	(.1)			
	>- 415.4		.026	.002	.000	.001	.017	.161
	STRAIGHT - (		(40.8)	(.8)	(.1)			
	>- 440		.000	.000	.000	.004	.052	.201
	STRAIGHT - (		(.0)	(.0)	(.1)			
	>- 440.1		.073	.004	.000	.002	.086	.249
	STRAIGHT - (		(51.8)	(1.0)	(.1)			
159		>- 4550	.168	.000	.000	.013	.094	.298
	(		(117.6)	(.0)	(.2)			
	BEND - (	4550	.208	.005	.003	.047	.099	.312
	(		(124.5)	(1.2)	(.7)			
	>- 455F		.275	.010	.015	.081	.107	.353
	STRAIGHT - (		(130.0)	(2.2)	(3.5)			
	>- 455F.1		.588	.010	.092	.093	.094	.371
	STRAIGHT - (		(127.1)	(2.2)	(21.1)			
	>- 455F.2		.912	.010	.176	.098	.081	.371
	STRAIGHT - (		(148.4)	(2.2)	(39.6)			
	>- 455F.3		1.230	.010	.261	.094	.069	.356
	STRAIGHT - (		(144.9)	(2.2)	(57.8)			
	>- 4600		1.530	.010	.339	.084	.058	.330
	(		(226.0)	(2.2)	(74.8)			
BEND - (	4600	1.588	.015	.352	.060	.041	.273	
(		(234.4)	(3.3)	(78.0)				
>- 460F		1.609	.023	.356	.043	.036	.247	

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LOADING - SHOCK LOADING NO. 1

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 460F	1.609	.023	.356	.043	.036	.247
	(		(236.9)	(8.2)	(79.1)			
	BEND -	( 470F	1.604	.038	.356	.049	.052	.221
	(		(236.1)	(8.3)	(78.2)			
		>- 470F	1.601	.065	.359	.066	.074	.165
	STRAIGHT -	(	(235.7)	(12.2)	(76.2)			
		>- 470F.1	1.601	.106	.376	.074	.080	.147
	STRAIGHT -	(	(235.7)	(40.9)	(73.1)			
		>- 475	1.601	.220	.400	.082	.085	.132
	STRAIGHT -	(	(235.7)	(73.0)	(75.9)			
		>- 475.1	1.601	.285	.427	.090	.089	.119
	STRAIGHT -	(	(235.7)	(103.4)	(82.8)			
		>- 480F	1.601	.344	.458	.099	.092	.110
	(		(235.7)	(132.8)	(92.9)			
	BEND -	( 480F	1.595	.354	.470	.113	.100	.082
	(		(234.5)	(139.1)	(97.5)			
		>- 480F	1.579	.336	.475	.135	.107	.069
	STRAIGHT -	(	(231.0)	(133.8)	(99.9)			
		>- 482	1.557	.300	.475	.138	.107	.064
	STRAIGHT -	(	(225.9)	(120.4)	(99.9)			
		>- 482.1	1.500	.198	.475	.140	.103	.051
	STRAIGHT -	(	(215.3)	(80.8)	(99.8)			
160		>- 482.2	1.453	.096	.475	.136	.091	.037
	STRAIGHT -	(	(212.6)	(39.8)	(99.8)			
		>- 484	1.427	.000	.475	.124	.074	.024
	STRAIGHT -	(	(215.9)	(.0)	(99.8)			
		>- 485F	1.426	.006	.475	.123	.073	.023
	(		(216.3)	(2.7)	(99.8)			
	BEND -	( 485F	1.431	.025	.471	.098	.099	.026
	(		(218.0)	(10.3)	(99.2)			
		>- 485F	1.440	.028	.446	.091	.186	.019
	STRAIGHT -	(	(219.0)	(11.4)	(93.2)			
		>- 490F	1.439	.014	.260	.062	.235	.014
	(		(219.0)	(5.8)	(50.4)			
	BEND -	( 490F	1.417	.007	.209	.050	.324	.006
	(		(216.5)	(3.5)	(42.2)			
		>- 490F	1.350	.001	.132	.019	.405	.007
	STRAIGHT -	(	(208.4)	(.6)	(39.1)			
		>- 492	1.329	.000	.182	.018	.408	.007
	STRAIGHT -	(	(205.9)	(.0)	(39.1)			
		>- 492.1	1.016	.008	.182	.006	.429	.008
	STRAIGHT -	(	(174.6)	(7.4)	(39.1)			
		>- 492.2	.750	.009	.182	.008	.427	.010
	STRAIGHT -	(	(157.1)	(13.0)	(39.0)			
		>- 495F	.402	.010	.182	.014	.402	.013



DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
415	80.	109.	687.	700.	727.	529.	2052.	2240.
525	517.	182.	303.	626.	969.	1409.	1703.	2414.
<b>RESTRAINTS:</b>								
162	246.	0.	0.	246.	0.	0.	0.	0.
440	0.	1052.	0.	1052.	0.	0.	0.	0.
455N	0.	809.	0.	809.	0.	0.	0.	0.
484	0.	486.	0.	486.	0.	0.	0.	0.

FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

FORCES (POUNDS)

MOMENTS (FOOT-POUNDS)

LDC.  
NO.

X Y Z RESULTANT

X Y Z RESULTANT

RESTRAINTS (CONTINUED) -

LDC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
492	0.	284.	0.	284.	0.	0.	0.	0.
500	0.	131.	0.	131.	0.	0.	0.	0.
503	0.	0.	718.	718.	0.	0.	0.	0.
505	0.	214.	0.	214.	0.	0.	0.	0.

\*\*\*\*\*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 6469, OCCURS AT POINT 455N

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ANSI B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A ( \* ) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE GILD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 BOLT CUT CORRECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP		-- 415	687.	135.	2052.	727.	529.	1.00	1.00	29.90	899.
	STR -(	>- 415.1	687.	134.	2052.	286.	205.	1.00	1.00	29.90	835.
	STR -(	>- 415.2	687.	134.	2052.	154.	117.	1.00	1.00	29.90	827.
	STR -(	>- 415.3	687.	130.	2052.	590.	428.	1.00	1.00	29.90	874.
	STR -(	>- 415.4	687.	130.	2052.	1022.	727.	1.00	1.00	29.90	965.
	STR -(	>- 440	686.	969.	2052.	1450.	1017.	1.00	1.00	29.90	1088.
	STR -(	>- 440.1	686.	966.	2052.	3164.	1119.	1.00	1.00	29.90	1578.
	STR -(	>- 455N	686.	250.	2052.	7767.	1883.	2.61	2.61	29.90	6469. **
	BEND -(	455N	583.	440.	2820.	7637.	377.	2.61	2.61	29.90	6390. **
	STR -(	>- 455F	146.	716.	2113.	7082.	1804.	2.61	2.61	29.90	5964. **
	STR -(	>- 455F.1	149.	710.	2113.	1002.	4245.	1.00	1.00	29.90	1945.
165	STR -(	>- 455F.2	153.	666.	2113.	416.	1584.	1.00	1.00	29.90	1073.
	STR -(	>- 455F.3	157.	582.	2113.	551.	622.	1.00	1.00	29.90	936.
	STR -(	>- 460N	164.	348.	2113.	2682.	847.	2.61	2.61	29.90	2758.
	BEND -(	460N	355.	149.	1280.	2924.	1898.	2.61	2.61	29.90	2911.
	STR -(	>- 460F	342.	176.	692.	2909.	2117.	2.61	2.61	29.90	2906.
	BEND -(	470N	222.	239.	2293.	2182.	1790.	2.61	2.61	29.90	2851.
	STR -(	>- 470F	79.	317.	2705.	2398.	951.	2.61	2.61	29.90	2931.
	STR -(	>- 470F.1	109.	257.	2705.	1212.	2858.	1.00	1.00	29.90	1652.
	STR -(	>- 475	152.	210.	2705.	1576.	3101.	1.00	1.00	29.90	1768.
	STR -(	>- 475.1	193.	186.	2705.	1955.	3158.	1.00	1.00	29.90	1844.
	STR -(	>- 480N	285.	203.	2705.	3076.	2328.	2.61	2.61	29.90	3694.



DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
		>- 480N	285.	203.	2705.	3076.	2328.	2.61	2.61	29.90	3694.
	BEND	- ( 480M	205.	284.	3584.	2919.	428.	2.61	2.61	29.90	3640.
		>- 480F	134.	323.	2502.	2665.	2519.	2.61	2.61	29.90	3481.
	STR	- (									
		>- 482	168.	342.	2502.	2345.	2321.	1.00	1.00	29.90	1662.
	STR	- (									
		>- 482.1	220.	364.	2502.	1907.	1400.	1.00	1.00	29.90	1382.
	STR	- (									
		>- 482.2	377.	340.	2502.	1529.	1139.	1.00	1.00	29.90	1262.
	STR	- (									
		>- 484	420.	482.	2502.	1214.	1805.	1.00	1.00	29.90	1330.
	STR	- (									
		>- 485N	445.	473.	2502.	1863.	1307.	2.61	2.61	29.90	2652.
	BEND	- ( 485M	277.	587.	2017.	2064.	573.	2.61	2.61	29.90	2775.
		>- 485F	281.	586.	1775.	2209.	2027.	2.61	2.61	29.90	2732.
	STR	- (									
		>- 490N	224.	630.	1775.	3378.	448.	2.61	2.61	29.90	3013.
166	BEND	- ( 490M	454.	490.	1117.	3666.	1216.	2.61	2.61	29.90	3152.
		>- 490F	502.	441.	211.	3659.	1509.	2.61	2.61	29.90	3052.
	STR	- (									
		>- 492	501.	241.	211.	1217.	3617.	1.00	1.00	29.90	1534.
	STR	- (									
		>- 492.1	501.	244.	211.	862.	3122.	1.00	1.00	29.90	1302.
	STR	- (									
		>- 492.2	502.	247.	211.	515.	2857.	1.00	1.00	29.90	1168.
	STR	- (									
		>- 495N	504.	245.	211.	2854.	235.	2.61	2.61	29.90	2251.
	BEND	- ( 495M	441.	346.	297.	3024.	71.	2.61	2.61	29.90	2383.
		>- 495F	221.	515.	219.	3383.	254.	2.61	2.61	29.90	2665.
	STR	- (									
		>- 500	229.	244.	219.	478.	4574.	1.00	1.00	29.90	1847.
	STR	- (									
		>- 500.1	250.	227.	219.	493.	3852.	1.00	1.00	29.90	1560.
	STR	- (									
		>- 500.2	250.	227.	219.	661.	3216.	1.00	1.00	29.90	1321.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 500.2	250.	227.	219.	661.	3218.	1.00	1.00	29.90	1321.
	STR - (	>- 500.3	261.	203.	219.	817.	2625.	1.00	1.00	29.90	1107.
	STR - (	>- 500.4	319.	195.	219.	902.	2003.	1.00	1.00	29.90	886.
	STR - (	>- 500.5	362.	215.	219.	950.	1369.	1.00	1.00	29.90	675.
	STR - (	>- 505	408.	246.	219.	1030.	1077.	1.00	1.00	29.90	604.
	STR - (	>- 510F	451.	317.	219.	1114.	1077.	2.61	2.61	29.90	1227.
	BEND - (	510M	401.	379.	945.	1034.	689.	2.61	2.61	29.90	1223.
	STR - (	>- 510F	256.	408.	1189.	724.	210.	2.61	2.61	29.90	1104.
	STR - (	>- 515A	282.	529.	1189.	604.	694.	2.61	2.61	29.90	1179.
	BEND - (	515M	326.	504.	1450.	660.	296.	2.61	2.61	29.90	1270.
	STR - (	>- 515F	186.	570.	1254.	482.	571.	2.61	2.61	29.90	1145.
167	STR - (	>- 520M	182.	599.	1254.	1233.	596.	2.61	2.61	29.90	1456.
	BEND - (	520M	450.	435.	949.	1653.	1196.	2.61	2.61	29.90	1764.
	STR - (	>- 520F	517.	353.	969.	1762.	1321.	2.61	2.61	29.90	1686.
	ANCHR	-- 525	517.	353.	969.	1703.	1409.	1.00	1.00	29.90	969.

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 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )  
 DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. ID.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 415	.000	.000	.000	.000	.000	.000
	STRAIGHT -(		(.6)	(.0)	(.0)			
	>- 415.1	.001	.002	.000	.004	.003	.019	
	STRAIGHT -(		(2.9)	(1.5)	(.1)			
	>- 415.2	.004	.005	.000	.004	.003	.037	
	STRAIGHT -(		(8.1)	(4.0)	(.2)			
	>- 415.3	.006	.008	.000	.001	.001	.056	
	STRAIGHT -(		(11.0)	(4.9)	(.3)			
	>- 415.4	.005	.007	.000	.004	.003	.074	
	STRAIGHT -(		(8.8)	(5.4)	(.5)			
	>- 440	.000	.000	.000	.013	.009	.093	
	STRAIGHT -(		(.6)	(.0)	(.6)			
	>- 440.1	.013	.014	.001	.006	.016	.115	
	STRAIGHT -(		(19.5)	(4.0)	(.7)			
>- 455F	.033	.000	.001	.040	.025	.137		
168		(	(43.8)	(.0)	(.8)			
	BEND -(	455F	.037	.017	.009	.146	.039	.139
	(		(46.6)	(4.5)	(2.9)			
	>- 455F	.047	.031	.046	.247	.041	.154	
	STRAIGHT -(		(44.4)	(8.0)	(12.1)			
	>- 455F.1	.170	.031	.282	.288	.059	.164	
	STRAIGHT -(		(43.3)	(8.0)	(66.6)			
	>- 455F.2	.313	.031	.545	.309	.077	.168	
	STRAIGHT -(		(41.8)	(6.0)	(120.4)			
	>- 455F.3	.459	.031	.818	.311	.096	.168	
	STRAIGHT -(		(100.2)	(8.0)	(172.9)			
	>- 460F	.604	.031	1.046	.299	.116	.165	
	(		(131.4)	(7.9)	(227.9)			
	BEND -(	460F	.625	.052	1.138	.261	.126	.150
(		(135.1)	(10.9)	(239.2)				
>- 460F	.612	.096	1.157	.221	.151	.140		

DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

OUTPUT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 460F	.612	.096	1.157	.221	.151	.140
		(	(130.6)	(21.1)	(243.4)			
	BEAD	-( 470F	.585	.123	1.144	.189	.180	.147
		(	(123.1)	(28.8)	(240.6)			
		>- 470F	.571	.110	1.168	.180	.212	.149
	STRAIGHT	-(	(119.8)	(27.5)	(233.0)			
		>- 470F.1	.571	.039	.989	.184	.223	.145
	STRAIGHT	-(	(119.8)	(16.9)	(209.0)			
		>- 475	.571	.059	.864	.149	.237	.139
	STRAIGHT	-(	(119.8)	(20.9)	(185.9)			
		>- 475.1	.571	.125	.738	.134	.250	.131
	STRAIGHT	-(	(119.8)	(33.9)	(165.8)			
		>- 4800	.571	.191	.605	.119	.263	.122
		(	(119.8)	(48.7)	(149.3)			
	BEAD	-( 4801	.551	.203	.554	.117	.304	.104
		(	(114.6)	(51.7)	(144.9)			
		>- 480F	.498	.191	.530	.097	.342	.103
	STRAIGHT	-(	(101.1)	(49.1)	(143.5)			
		>- 482	.421	.166	.530	.092	.347	.096
	STRAIGHT	-(	(81.7)	(43.4)	(145.5)			
		>- 482.1	.256	.103	.530	.080	.358	.076
	STRAIGHT	-(	(50.2)	(28.0)	(143.5)			
169		>- 482.2	.300	.048	.530	.070	.362	.057
	STRAIGHT	-(	(87.2)	(13.5)	(143.4)			
		>- 484	.507	.000	.530	.062	.360	.037
	STRAIGHT	-(	(134.3)	(.0)	(143.3)			
		>- 485F	.524	.003	.530	.061	.360	.036
		(	(137.6)	(.9)	(143.3)			
	BEAD	-( 485H	.583	.011	.503	.053	.346	.031
		(	(146.1)	(3.4)	(137.6)			
		>- 485F	.608	.011	.441	.054	.330	.012
	STRAIGHT	-(	(147.7)	(3.7)	(118.9)			
		>- 4900	.808	.006	.155	.037	.314	.003
		(	(147.7)	(2.3)	(53.3)			
	BEAD	-( 490F	.506	.004	.102	.029	.272	.007
		(	(146.7)	(1.8)	(57.4)			
		>- 490F	.541	.001	.082	.012	.230	.012
	STRAIGHT	-(	(146.8)	(.5)	(58.3)			
		>- 492	.530	.000	.082	.012	.228	.012
	STRAIGHT	-(	(136.4)	(.0)	(58.3)			
		>- 492.1	.365	.006	.082	.006	.211	.012
	STRAIGHT	-(	(109.9)	(7.4)	(58.4)			
		>- 492.2	.225	.009	.082	.003	.194	.013
	STRAIGHT	-(	(83.0)	(14.9)	(58.4)			
		>- 495F	.112	.010	.082	.004	.177	.013



GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
415	206.	68.	81.	232.	327.	1347.	301.	1419.
525	114.	63.	115.	174.	416.	370.	316.	640.
<b>RESTRAINTS:</b>								
171	437.	0.	0.	437.	0.	0.	0.	0.
470	0.	101.	0.	101.	0.	0.	0.	0.
455B	0.	220.	0.	220.	0.	0.	0.	0.
484	0.	79.	0.	79.	0.	0.	0.	0.

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>RESTRAINTS (CONTINUED)</b>								
492	0.	37.	0.	37.	0.	0.	0.	0.
500	0.	135.	0.	135.	0.	0.	0.	0.
500	0.	0.	116.	116.	0.	0.	0.	0.
505	0.	160.	0.	160.	0.	0.	0.	0.

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 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
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THE MAXIMUM STRESS OF 1575. OCCURS AT POINT 480M

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ANSI B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, FITTERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 GATE CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		:	:
ANCHR		-- 415	81.	217.	301.	327.	1347.	1.00	1.00	29.90	:	569.
	STR - (	>- 415.1	81.	209.	301.	52.	513.	1.00	1.00	29.90	:	239.
	STR - (	>- 415.2	81.	209.	301.	162.	308.	1.00	1.00	29.90	:	184.
	STR - (	>- 415.3	81.	193.	301.	214.	1086.	1.00	1.00	29.90	:	460.
	STR - (	>- 415.4	81.	180.	301.	103.	1803.	1.00	1.00	29.90	:	735.
	STR - (	>- 440	81.	243.	301.	194.	2468.	1.00	1.00	29.90	:	1001.
	STR - (	>- 440.1	81.	243.	301.	243.	1156.	1.00	1.00	29.90	:	489.
	STR - (	>- 455H	81.	263.	301.	527.	323.	2.61	2.61	29.90	:	539.
	BEND - (	455H	154.	228.	346.	495.	337.	2.61	2.61	29.90	:	543.
	STR - (	>- 455F	201.	189.	397.	446.	399.	2.61	2.61	29.90	:	563.
	STR - (	>- 455F.1	200.	138.	397.	764.	197.	1.00	1.00	29.90	:	354.
174	STR - (	>- 455F.2	200.	92.	397.	940.	243.	1.00	1.00	29.90	:	421.
	STR - (	>- 455F.3	199.	75.	397.	987.	327.	1.00	1.00	29.90	:	446.
	STR - (	>- 460F	198.	101.	397.	277.	1016.	2.61	2.61	29.90	:	882.
	BEND - (	460H	142.	172.	724.	304.	826.	2.61	2.61	29.90	:	894.
	STR - (	>- 460F	192.	210.	1034.	370.	449.	2.61	2.61	29.90	:	930.
	BEND - (	470H	104.	209.	1068.	339.	405.	2.61	2.61	29.90	:	934.
	STR - (	>- 470F	67.	224.	681.	280.	801.	2.61	2.61	29.90	:	853.
	STR - (	>- 470F.1	75.	224.	681.	346.	296.	1.00	1.00	29.90	:	336.
	STR - (	>- 475	84.	202.	681.	397.	620.	1.00	1.00	29.90	:	401.
	STR - (	>- 475.1	94.	181.	681.	635.	980.	1.00	1.00	29.90	:	542.
	STR - (	>- 480F	102.	124.	681.	1323.	801.	2.61	2.61	29.90	:	1325. *

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEX TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESIDUAL SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
		>- 480M	102.	124.	681.	1323.	601.	2.61	2.61	29.90	1325. *
	BEND	-( 480M	155.	43.	197.	1395.	1040.	2.61	2.61	29.90	1373. **
		>- 480F	121.	106.	811.	1359.	690.	2.61	2.61	29.90	1354. *
	STR	-(									
		>- 482	107.	117.	811.	644.	1238.	1.00	1.00	29.90	648. :
	STR	-(									
		>- 482.1	89.	154.	811.	313.	875.	1.00	1.00	29.90	495. :
	STR	-(									
		>- 482.2	75.	221.	811.	295.	573.	1.00	1.00	29.90	416. :
	STR	-(									
		>- 484	75.	221.	811.	942.	473.	1.00	1.00	29.90	534. :
	STR	-(									
		>- 485M	91.	158.	811.	476.	973.	2.61	2.61	29.90	1061. :
		(									
	BEND	-( 485M	78.	164.	1304.	480.	240.	2.61	2.61	29.90	1105. :
		(									
		>- 485F	98.	154.	1126.	463.	661.	2.61	2.61	29.90	1086. :
	STR	-(									
		>- 490M	86.	166.	1126.	552.	253.	2.61	2.61	29.90	1003. :
		(									
175	BEND	-( 490M	133.	131.	744.	589.	821.	2.61	2.61	29.90	984. :
		(									
		>- 490F	110.	145.	229.	546.	984.	2.61	2.61	29.90	900. :
	STR	-(									
		>- 492	125.	123.	229.	956.	524.	1.00	1.00	29.90	447. :
	STR	-(									
		>- 492.1	134.	130.	229.	674.	226.	1.00	1.00	29.90	300. :
	STR	-(									
		>- 492.2	143.	133.	229.	389.	269.	1.00	1.00	29.90	211. :
	STR	-(									
		>- 495M	151.	134.	229.	606.	97.	2.61	2.61	29.90	514. :
		(									
	BEND	-( 495M	161.	122.	149.	661.	163.	2.61	2.61	29.90	546. :
		(									
		>- 495F	100.	175.	49.	635.	209.	2.61	2.61	29.90	525. :
	STR	-(									
		>- 500	95.	147.	49.	326.	617.	1.00	1.00	29.90	281. :
	STR	-(									
		>- 500.1	94.	108.	49.	540.	623.	1.00	1.00	29.90	331. :
	STR	-(									
		>- 500.2	93.	108.	49.	897.	695.	1.00	1.00	29.90	456. :

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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## INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL:	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		STRESS PSI	
	STR - (	>- 500.2	93.	108.	49.	697.	695.	1.00	1.00	29.90	456.	:
	STR - (	>- 500.3	92.	44.	49.	1003.	720.	1.00	1.00	29.90	496.	:
	STR - (	>- 500.4	90.	57.	49.	799.	630.	1.00	1.00	29.90	409.	:
	STR - (	>- 500.5	97.	124.	49.	351.	422.	1.00	1.00	29.90	221.	:
	STR - (	>- 505	101.	166.	49.	385.	212.	1.00	1.00	29.90	177.	:
	BEND - (	>- 510M	106.	113.	49.	215.	386.	2.61	2.61	29.90	349.	:
	BEND - (	>- 510M	115.	104.	292.	209.	264.	2.61	2.61	29.90	349.	:
	STR - (	>- 510F	93.	124.	397.	149.	64.	2.61	2.61	29.90	336.	:
	BEND - (	>- 515M	107.	128.	397.	215.	185.	2.61	2.61	29.90	382.	:
	BEND - (	>- 515M	118.	118.	435.	234.	121.	2.61	2.61	29.90	399.	:
	STR - (	>- 515F	63.	154.	312.	174.	273.	2.61	2.61	29.90	353.	:
176	BEND - (	>- 520M	63.	162.	312.	227.	279.	2.61	2.61	29.90	373.	:
	BEND - (	>- 520M	103.	140.	359.	300.	329.	2.61	2.61	29.90	448.	:
	STR - (	>- 520F	114.	131.	416.	323.	331.	2.61	2.61	29.90	487.	:
	ANCHR	-- 525	114.	131.	416.	316.	370.	1.00	1.00	29.90	257.	:

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 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 415	.000	.000	.000	.000	.000	.000
	STRAIGHT -(		(.0)	(.0)	(.0)			
	>-	415.1	.003	.001	.000	.001	.006	.003
	STRAIGHT -(		(5.9)	(24.9)	(.0)			
	>-	415.2	.010	.002	.000	.001	.007	.005
	STRAIGHT -(		(17.3)	(61.2)	(.1)			
	>-	415.3	.014	.002	.000	.000	.002	.008
	STRAIGHT -(		(25.3)	(69.5)	(.1)			
	>-	415.4	.012	.001	.000	.001	.008	.011
	STRAIGHT -(		(21.9)	(40.7)	(.2)			
	>-	440	.000	.000	.000	.001	.023	.014
	STRAIGHT -(		(.0)	(.0)	(.2)			
	>-	440.1	.032	.001	.000	.000	.038	.017
	STRAIGHT -(		(55.4)	(14.8)	(.2)			
177	>-	455F	.074	.000	.000	.003	.043	.020
	(		(126.3)	(.0)	(.3)			
	BEAD -(	455H	.081	.001	.001	.010	.042	.021
	(		(137.4)	(3.5)	(1.5)			
	>-	455F	.083	.002	.003	.016	.041	.025
	STRAIGHT -(		(137.7)	(4.9)	(5.3)			
	>-	455F.1	.077	.002	.018	.018	.040	.028
	STRAIGHT -(		(111.1)	(5.0)	(28.2)			
	>-	455F.2	.075	.002	.034	.018	.039	.032
	STRAIGHT -(		(79.5)	(5.0)	(51.7)			
	>-	455F.3	.079	.002	.049	.017	.038	.036
	STRAIGHT -(		(45.7)	(5.1)	(72.9)			
	>-	460F	.091	.002	.063	.016	.038	.039
	(		(22.6)	(5.2)	(91.5)			
	BEAD -(	460F	.095	.003	.065	.015	.037	.046
	(		(24.2)	(6.6)	(95.0)			
	>-	460F	.096	.005	.066	.016	.036	.049

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		> 460F	.096	.005	.066	.016	.036	.049
	(	(26.2)		(11.0)	(96.3)			
	BEND - (	470B	.096	.006	.064	.018	.039	.050
	(	(32.0)		(14.2)	(91.8)			
		> 470F	.097	.009	.057	.018	.041	.055
	STRAIGHT - (	(35.6)		(17.8)	(79.9)			
		> 470F.1	.097	.035	.040	.022	.041	.056
	STRAIGHT - (	(35.6)		(54.7)	(43.2)			
		> 475	.097	.068	.031	.025	.039	.056
	STRAIGHT - (	(35.6)		(98.1)	(15.2)			
		> 475.1	.097	.097	.034	.028	.036	.054
	STRAIGHT - (	(35.6)		(139.5)	(32.5)			
		> 480B	.097	.125	.045	.032	.031	.051
	(	(35.6)		(179.4)	(60.2)			
	BEND - (	480C	.096	.130	.048	.037	.015	.039
	(	(37.5)		(147.9)	(67.3)			
		> 480F	.095	.125	.049	.048	.014	.033
	STRAIGHT - (	(37.8)		(100.8)	(68.5)			
		> 482	.094	.113	.049	.049	.015	.031
	STRAIGHT - (	(34.5)		(162.6)	(68.5)			
		> 482.1	.089	.075	.049	.052	.020	.025
	STRAIGHT - (	(28.0)		(109.2)	(68.5)			
178		> 482.2	.087	.037	.049	.052	.023	.018
	STRAIGHT - (	(40.1)		(53.7)	(68.5)			
		> 484	.089	.000	.049	.048	.024	.012
	STRAIGHT - (	(40.4)		(.0)	(68.5)			
		> 485A	.090	.003	.049	.048	.024	.012
	(	(42.0)		(3.7)	(68.5)			
	BEND - (	485B	.091	.010	.047	.039	.025	.012
	(	(47.0)		(15.9)	(66.0)			
		> 485F	.093	.011	.043	.037	.026	.008
	STRAIGHT - (	(48.6)		(15.3)	(59.1)			
		> 490A	.093	.005	.021	.026	.026	.005
	(	(48.6)		(8.5)	(27.5)			
	BEND - (	490B	.091	.003	.016	.021	.027	.003
	(	(47.0)		(5.7)	(23.6)			
		> 490F	.088	.000	.014	.010	.029	.006
	STRAIGHT - (	(61.9)		(1.4)	(23.1)			
		> 492	.064	.000	.014	.009	.029	.006
	STRAIGHT - (	(40.2)		(.0)	(23.1)			
		> 492.1	.063	.005	.014	.005	.030	.007
	STRAIGHT - (	(35.9)		(16.9)	(23.1)			
		> 492.2	.042	.008	.014	.004	.029	.009
	STRAIGHT - (	(15.0)		(56.9)	(23.1)			
		> 495B	.023	.010	.014	.004	.027	.010

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESTINATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 4950	.023	.010	.014	.004	.027	.010
	(	( 4951	(19.5)	(53.6)	(23.0)			
	BEND -	( 4951	.026	.010	.013	.003	.021	.011
	(	( 4951	(22.1)	(53.2)	(22.3)			
		>- 495F	.019	.008	.009	.003	.017	.013
	STRAIGHT -	( 495F	(22.2)	(42.3)	(18.1)			
		>- 500	.019	.000	.000	.003	.015	.014
	STRAIGHT -	( 500	(22.2)	(.0)	(.0)			
		>- 500.1	.019	.012	.012	.003	.012	.012
	STRAIGHT -	( 500.1	(22.3)	(66.9)	(27.8)			
		>- 500.2	.019	.020	.021	.003	.007	.007
	STRAIGHT -	( 500.2	(22.4)	(116.2)	(49.7)			
		>- 500.3	.019	.023	.025	.003	.002	.001
	STRAIGHT -	( 500.3	(22.3)	(132.7)	(61.5)			
		>- 500.4	.019	.020	.024	.003	.003	.007
	STRAIGHT -	( 500.4	(22.2)	(111.9)	(61.1)			
		>- 500.5	.019	.011	.019	.004	.007	.011
	STRAIGHT -	( 500.5	(22.2)	(61.6)	(49.8)			
		>- 505	.019	.000	.012	.004	.009	.012
	STRAIGHT -	( 505	(22.2)	(.0)	(32.7)			
		>- 510F	.019	.002	.010	.004	.009	.011
	(	( 510F	(22.2)	(10.0)	(29.9)			
	BEND -	( 510F	.018	.003	.009	.004	.010	.009
	(	( 510F	(21.0)	(18.5)	(27.4)			
		>- 510F	.016	.003	.008	.005	.010	.008
	STRAIGHT -	( 510F	(17.9)	(17.7)	(27.0)			
		>- 515F	.012	.001	.006	.005	.010	.008
	(	( 515F	(12.1)	(6.4)	(26.9)			
	BEND -	( 515F	.010	.001	.008	.005	.009	.008
	(	( 515F	(11.2)	(2.7)	(25.6)			
		>- 515F	.008	.001	.007	.007	.008	.009
	STRAIGHT -	( 515F	(9.0)	(1.3)	(21.5)			
		>- 520F	.002	.001	.002	.006	.006	.008
	(	( 520F	(2.0)	(1.3)	(4.5)			
	BEND -	( 520F	.006	.000	.001	.003	.005	.005
	(	( 520F	(.4)	(.7)	(1.5)			
		>- 520F	.000	.000	.000	.001	.000	.000
	STRAIGHT -	( 520F	(.0)	(.1)	(.1)			
		-- 525	.000	.000	.000	.000	.000	.000

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ANCHOR

GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PILING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING LOCAL QUANTITIES BY SRSS.

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 \*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
415	495.	133.	748.	908.	829.	3286.	4905.	5962.
525	1641.	246.	440.	1717.	1541.	5135.	5710.	7832.
<b>RESTRAINTS:</b>								
180	1207.	0.	0.	1207.	0.	0.	0.	0.
440	0.	1108.	0.	1108.	0.	0.	0.	0.
455	0.	863.	0.	863.	0.	0.	0.	0.
484	0.	509.	0.	509.	0.	0.	0.	0.

DYNAFLEX

LOADING - SHOCK LOADING NO. 4

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FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>RESTRAINTS (CONTINUED) -</b>								
492	0.	321.	0.	321.	0.	0.	0.	0.
500	0.	391.	0.	391.	0.	0.	0.	0.
500	0.	0.	923.	923.	0.	0.	0.	0.
505	0.	344.	0.	344.	0.	0.	0.	0.

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 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 7640. OCCURS AT POINT 455N

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ACSJ 531.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (15) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLEX

LOADING - SHOCK LOADING NO. 4

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	MEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESIDUAL SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHR		-- 415	748.	514.	4905.	829.	5286.	1.00	1.00	29.90	2392.
	STR - (	>- 415.1	748.	507.	4905.	304.	1278.	1.00	1.00	29.90	2038.
	STR - (	>- 415.2	748.	507.	4905.	228.	722.	1.00	1.00	29.90	1992.
	STR - (	>- 415.3	748.	483.	4905.	654.	2659.	1.00	1.00	29.90	2254.
	STR - (	>- 415.4	748.	483.	4905.	1076.	4548.	1.00	1.00	29.90	2719.
	STR - (	>- 440	748.	1251.	4905.	1533.	6400.	1.00	1.00	29.90	3294.
	STR - (	>- 440.1	748.	1230.	4905.	3326.	3009.	1.00	1.00	29.90	2667.
	STR - (	>- 455R	748.	774.	4905.	8162.	2072.	2.61	2.61	29.90	7640. **
	BEND - (	455M	708.	811.	4804.	8058.	2187.	2.61	2.61	29.90	7553. *
	STR - (	>- 455F	361.	1014.	2642.	7474.	4117.	2.61	2.61	29.90	7004. *
	STR - (	>- 455F.1	363.	985.	2642.	1802.	4411.	1.00	1.00	29.90	2186.
183	STR - (	>- 455F.2	364.	909.	2642.	1649.	1640.	1.00	1.00	29.90	1412.
	STR - (	>- 455F.3	366.	789.	2642.	3292.	1438.	1.00	1.00	29.90	1790.
	STR - (	>- 460R	370.	484.	2642.	3593.	4605.	2.61	2.61	29.90	4986.
	BEND - (	460M	494.	357.	2737.	3647.	4862.	2.61	2.61	29.90	5226.
	STR - (	>- 460F	405.	455.	4977.	3584.	2801.	2.61	2.61	29.90	5286.
	BEND - (	470M	356.	421.	3366.	2891.	4898.	2.61	2.61	29.90	5181.
	STR - (	>- 470F	212.	509.	3320.	2996.	4652.	2.61	2.61	29.90	5059.
	STR - (	>- 470F.1	182.	476.	3320.	4059.	3199.	1.00	1.00	29.90	2465.
	STR - (	>- 475	212.	444.	3320.	3619.	3365.	1.00	1.00	29.90	2389.
	STR - (	>- 475.1	213.	416.	3320.	3346.	3470.	1.00	1.00	29.90	2349.
	STR - (	>- 480R	456.	394.	3320.	3530.	3197.	2.61	2.61	29.90	4551.

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEX TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS (PSI)	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF-PLANE		:	:
		>- 480N	456.	398.	3320.	3530.	3197.	2.61	2.61	29.90	:	4551.
	BEND	- ( 480N	563.	480.	3650.	3416.	2643.	2.61	2.61	29.90	:	4436.
		>- 480F	230.	560.	3143.	3152.	2998.	2.61	2.61	29.90	:	4208.
	STR	- (									:	
		>- 482	247.	613.	3143.	2682.	2803.	1.00	1.00	29.90	:	2004.
	STR	- (									:	
		>- 482.1	563.	799.	3143.	1983.	2689.	1.00	1.00	29.90	:	1841.
	STR	- (									:	
		>- 482.2	451.	881.	3143.	2047.	4296.	1.00	1.00	29.90	:	2288.
	STR	- (									:	
		>- 484	451.	881.	3143.	2967.	6771.	1.00	1.00	29.90	:	3224.
	STR	- (									:	
		>- 485H	531.	975.	3143.	6966.	3078.	2.61	2.61	29.90	:	6460.
	BEND	- ( 485H	716.	848.	4426.	7636.	1127.	2.61	2.61	29.90	:	6977. **
		>- 485F	836.	730.	3648.	7846.	2524.	2.61	2.61	29.90	:	7067. **
	STR	- (									:	
		>- 490N	974.	787.	3648.	7772.	590.	2.61	2.61	29.90	:	6747.
184	BEND	- ( 490N	771.	986.	2557.	7518.	2410.	2.61	2.61	29.90	:	6506.
		>- 490F	610.	1093.	533.	6775.	3087.	2.61	2.61	29.90	:	5852.
	STR	- (									:	
		>- 492	617.	1066.	533.	2977.	6550.	1.00	1.00	29.90	:	2895.
	STR	- (									:	
		>- 492.1	625.	1129.	533.	2215.	3670.	1.00	1.00	29.90	:	1733.
	STR	- (									:	
		>- 492.2	636.	1172.	533.	1466.	3648.	1.00	1.00	29.90	:	1592.
	STR	- (									:	
		>- 495H	605.	1191.	533.	6828.	792.	2.61	2.61	29.90	:	5405.
	BEND	- ( 495H	893.	1019.	847.	7868.	222.	2.61	2.61	29.90	:	6207.
		>- 495F	1159.	888.	634.	8529.	666.	2.61	2.61	29.90	:	6726.
	STR	- (									:	
		>- 500	1211.	423.	634.	1161.	9559.	1.00	1.00	29.90	:	3872.
	STR	- (									:	
		>- 500.1	1252.	393.	634.	862.	8506.	1.00	1.00	29.90	:	3440.
	STR	- (									:	
		>- 500.2	1252.	393.	634.	1439.	7520.	1.00	1.00	29.90	:	3082.

DYNAFLX

LOADING - SHOCK LOADING NO. 4

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELFN TYPE	LOC. ID.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 500.2	1252.	398.	634.	1430.	7520.	1.00	1.00	29.90	3082.
	STR - (	>- 500.3	1301.	366.	634.	2056.	6543.	1.00	1.00	29.90	2764.
	STR - (	>- 500.4	1397.	365.	634.	2585.	5525.	1.00	1.00	29.90	2460.
	STR - (	>- 500.5	1419.	396.	634.	3116.	4483.	1.00	1.00	29.90	2206.
	STR - (	>- 505	1486.	436.	634.	3767.	3603.	1.00	1.00	29.90	2107.
	STR - (	>- 510F	1550.	467.	634.	3494.	3819.	2.61	2.61	29.90	4089.
	BEND - (	510F	1021.	1257.	3172.	2831.	2333.	2.61	2.61	29.90	3802.
	STR - (	>- 510F	393.	1571.	3925.	1509.	570.	2.61	2.61	29.90	3327.
	STR - (	>- 515H	426.	1633.	3925.	877.	2683.	2.61	2.61	29.90	3790.
	BEND - (	515H	442.	1627.	5221.	908.	694.	2.61	2.61	29.90	4191.
	STR - (	>- 515F	253.	1667.	4660.	646.	1914.	2.61	2.61	29.90	3982.
185	STR - (	>- 520H	246.	1699.	4660.	3833.	1008.	2.61	2.61	29.90	4796.
	BEND - (	520H	1238.	1190.	2819.	5247.	4052.	2.61	2.61	29.90	5648.
	STR - (	>- 520F	1641.	504.	1541.	5770.	4944.	2.61	2.61	29.90	6079.
ANCHR	STR - (	-- 525	1641.	504.	1541.	5710.	5135.	1.00	1.00	29.90	3143.

DYNAFILEY

LOADING - SHOCK LOADING NO. 4

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\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	EFFECT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT			
			X	Y	Z	X AXIS	Y AXIS	Z AXIS	
ANCHOR		-- 415	.000 (.0)	.000 (.0)	.000 (.0)	.000	.000	.000	
	STRAIGHT - (	>- 415.1	.003 (24.4)	.002 (24.9)	.000 (.1)	.004	.016	.044	
	STRAIGHT - (	>- 415.2	.024 (61.9)	.006 (61.3)	.000 (.2)	.004	.018	.089	
	STRAIGHT - (	>- 415.3	.035 (72.0)	.008 (69.7)	.000 (.4)	.002	.006	.133	
	STRAIGHT - (	>- 415.4	.031 (47.1)	.007 (40.9)	.000 (.5)	.005	.019	.177	
	STRAIGHT - (	>- 440	.000 (.0)	.000 (.0)	.001 (.6)	.014	.057	.222	
	STRAIGHT - (	>- 440.1	.081 (78.3)	.014 (15.4)	.001 (.8)	.006	.096	.275	
186		>- 455H	.187 (178.0)	.000 (.0)	.001 (.9)	.042	.106	.329	
		BEAD - (	455H	.226 (193.8)	.018 (5.8)	.010 (3.4)	.154	.114	.343
			>- 455F	.291 (194.5)	.032 (9.6)	.049 (13.7)	.261	.121	.386
		STRAIGHT - (	>- 455F.1	.617 (174.3)	.032 (9.7)	.297 (75.3)	.303	.117	.406
		STRAIGHT - (	>- 455F.2	.967 (181.9)	.032 (9.7)	.574 (136.9)	.324	.119	.408
		STRAIGHT - (	>- 455F.3	1.315 (215.2)	.032 (9.7)	.860 (196.3)	.326	.125	.395
			>- 460K	1.647 (262.4)	.032 (9.7)	1.139 (256.7)	.311	.135	.371
		BEAD - (	460K	1.709 (271.6)	.054 (13.2)	1.193 (268.9)	.268	.137	.315
			>- 460F	1.720 (.099)	.099	1.212	.225	.160	.288

DYNAFLEX

LOADING - SHOCK LOADING NO. 4

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (MICRONS)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 460F	1.724	.099	1.212	.225	.160	.268
		(	(271.7)	(24.5)	(273.5)			
	BEND	- ( 470B	1.710	.129	1.200	.197	.192	.270
		(	(266.2)	(33.2)	(269.1)			
		>- 470F	1.703	.129	1.166	.192	.228	.229
	STRAIGHT	- (	(266.4)	(35.0)	(257.8)			
		>- 470F.1	1.703	.155	1.059	.181	.241	.214
	STRAIGHT	- (	(266.4)	(29.3)	(225.6)			
		>- 475	1.703	.238	.952	.172	.255	.199
	STRAIGHT	- (	(266.4)	(124.0)	(201.4)			
		>- 475.1	1.703	.326	.853	.164	.268	.186
	STRAIGHT	- (	(246.8)	(177.0)	(188.1)			
		>- 480A	1.703	.413	.760	.158	.281	.172
		(	(266.4)	(228.4)	(185.9)			
	BEND	- ( 480B	1.690	.428	.728	.166	.320	.138
		(	(263.7)	(239.4)	(187.2)			
		>- 480F	1.658	.406	.714	.173	.358	.128
	STRAIGHT	- (	(255.6)	(230.2)	(187.8)			
		>- 482	1.616	.361	.714	.173	.364	.119
	STRAIGHT	- (	(242.7)	(206.9)	(187.8)			
		>- 482.1	1.524	.236	.714	.170	.373	.095
	STRAIGHT	- (	(223.8)	(138.7)	(187.7)			
187		>- 482.2	1.487	.114	.714	.161	.374	.070
	STRAIGHT	- (	(233.3)	(68.3)	(187.7)			
		>- 484	1.517	.000	.714	.147	.368	.046
	STRAIGHT	- (	(261.3)	(.0)	(187.6)			
		>- 485A	1.522	.068	.714	.146	.368	.044
		(	(263.7)	(4.6)	(187.6)			
	BEND	- ( 485B	1.548	.029	.691	.118	.361	.042
		(	(270.9)	(17.6)	(182.0)			
		>- 485F	1.565	.032	.629	.112	.379	.024
	STRAIGHT	- (	(273.0)	(19.4)	(162.2)			
		>- 490A	1.565	.016	.304	.077	.393	.016
		(	(273.0)	(16.4)	(78.3)			
	BEND	- ( 490B	1.536	.009	.233	.062	.424	.010
		(	(270.0)	(6.9)	(75.1)			
		>- 490F	1.457	.001	.201	.025	.467	.015
	STRAIGHT	- (	(259.0)	(1.6)	(75.9)			
		>- 492	1.433	.000	.201	.024	.468	.016
	STRAIGHT	- (	(255.3)	(.0)	(73.9)			
		>- 492.1	1.083	.012	.201	.010	.479	.017
	STRAIGHT	- (	(206.0)	(21.6)	(73.9)			
		>- 492.2	.736	.015	.201	.009	.470	.019
	STRAIGHT	- (	(178.3)	(41.9)	(73.9)			
		>- 495A	.418	.016	.201	.015	.480	.021

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT	LOC.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
>- 4956			.416	.016	.201	.015	.440	.021
(			(196.6)	(61.3)	(73.8)			
>- 4957			.756	.012	.172	.021	.340	.022
(			(192.2)	(60.8)	(68.0)			
>- 4958			.537	.015	.120	.022	.226	.024
(			(195.9)	(40.5)	(52.0)			
>- 500			.337	.000	.000	.026	.186	.026
(			(195.9)	(.0)	(.0)			
>- 500.1			.337	.024	.159	.032	.121	.026
(			(195.9)	(76.6)	(75.9)			
>- 500.2			.537	.045	.221	.038	.064	.019
(			(195.9)	(133.5)	(136.8)			
>- 500.3			.537	.057	.255	.044	.019	.009
(			(195.8)	(153.4)	(171.6)			
>- 500.4			.336	.056	.244	.050	.036	.011
(			(195.7)	(131.0)	(174.0)			
>- 500.5			.536	.038	.198	.056	.070	.030
(			(195.6)	(73.9)	(148.8)			
>- 505			.336	.000	.123	.062	.097	.054
(			(195.5)	(.0)	(115.2)			
>- 5100			.536	.009	.108	.063	.101	.059
(			(195.5)	(12.8)	(110.9)			
>- 5100			.327	.019	.086	.052	.138	.100
(			(196.9)	(24.5)	(106.0)			
>- 5101			.245	.019	.075	.044	.164	.114
(			(196.4)	(23.3)	(104.0)			
>- 5150			.216	.005	.075	.046	.160	.136
(			(196.9)	(12.0)	(103.9)			
>- 5150			.177	.013	.071	.050	.134	.139
(			(125.4)	(12.3)	(101.2)			
>- 5151			.140	.017	.062	.055	.126	.153
(			(105.5)	(12.4)	(87.6)			
>- 5200			.027	.017	.022	.053	.090	.147
(			(20.0)	(12.5)	(20.4)			
>- 5201			.005	.009	.008	.024	.066	.084
(			(3.4)	(6.5)	(6.0)			
>- 5201			.000	.001	.001	.003	.006	.007
(			(.0)	(.4)	(.5)			
>- 525			.000	.000	.000	.000	.000	.000
ANCHOR								

DYNAFLEX

STRAKES - ROGER COMP

PAGE 52

PROJECT 1000 SOLAR FILGT PLANT - C-21700  
 JOB MS VALVE TO 11700  
 DATE 2/26/80

\*\*\*\*\*  
 \*  
 \* STRESS SUMMARY \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

- (1) EQUATION NUMBERS REFER TO ARTICLE 104.8 OF THE ANSI B31.1-1977 PIPING CODE.
- (7) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE MODULUS IS USED FOR THE BRANCH LEG AT REDUCED BUILT CORRECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK. THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.
- (9) THE COLD MODULUS WAS USED IN ANALYZING THE OCCASIONAL LOADS.





STRESS SUMMARY (CONTINUED)

LOC. NO.	STRESS	AREA	IN.	PSI
480N	2305	3694	1325	4851
480M	2132	3740	1373	4434
480F	1934	5081	1350	4204
482	913	1662	644	2004
482.1	1111	1302	495	1841
482.2	1863	1262	416	2288
484	2887	1350	534	3224
485N	5794	2652	1061	6460
485M	6305	2775	1195	6977
485F	6426	2742	1066	7051
490N	5954	5613	1603	6787
490M	5606	5152	944	5508
490F	4912	5052	909	5352
492	2415	1534	447	2895
492.1	1104	1302	300	1733
492.2	1061	1164	211	1542
495N	4887	2251	514	5405
495M	5705	2483	546	6207
495F	6153	2665	525	6724
500	3392	1847	281	3872
500.1	3044	1560	331	3101

Eq (12) in (12) F1 (12) F2 (12)  
 SHEET 1 SHEET 2 SHEET 3 SHEET 4

STRESS SUMMARY (CONTINUED)

STRESSES ARE IN PSI

EQ (12) EQ (12) EQ (12) EQ (12)

SHOCK 1 SHOCK 2 SHOCK 3 SHOCK 4

LOC.  
IN.

500.2	2747	1321	456	3082
500.3	2094	1107	496	2764
500.4	2252	826	409	2460
500.5	2049	675	221	2206
505	2011	604	177	2107
510N	3895	1227	549	4089
510M	3583	1223	309	3802
510F	3121	1104	336	3327
515M	3582	1179	382	3790
515M	3974	1270	599	4191
515F	3748	1145	553	3982
520N	4555	1456	573	4796
520M	5347	1764	448	5646
520F	5758	1866	487	6079
525	2979	969	257	3143

DYNAFLEX

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**Stearns-Roger**

**PIPE STRESS ANALYSIS REPORT**

FOR

**IOMWe SOLAR PILOT PLANT**

**C-21700**

MAIN STEAM TSS EXCHANGER BRANCH

PIPING SYSTEM

8" MS-4-KBA

CONDITIONS ANALYZED / ANAL. I.D. NO.

THERMAL T-21700-MS-5-A-2

DEAD WEIGHT W-21700-MS-5-A-2

PRESSURE P-21700-MS-5-A-0

SEISMIC X-21700-MS-5-A-1

CODE REFERENCE: **ANSI B31.1-1973**

**W/ ADDENDA C.**

**(SUMMER 1979)**

ANALYST G.H. MAY

DATE 7-25-80



DIVISION USAGE					
MM	P	PP	SH	FI	SP

**Stearns-Roger**  
INCORPORATED  
ENGINEERING STANDARD

STANDARD NUMBER  
EE16.01.2

APPROVALS  
Des. Sect: \_\_\_\_\_  
Sect. Supv. \_\_\_\_\_  
Div. \_\_\_\_\_

PIPING ANALYSIS RESULTS

PAGE 1 OF 1  
ISSUED 8-31-73  
REVISED 4-15-74

Date FEB. 28, 1980

TO: T.E. OLSON  
FROM: G. H. MAY, Piping Engineering Group  
Client DOE Project SOLAR I Job No. C-21700  
Pipe Line Analyzed MAIN STEAM TSS EXCHANGER BRANCH 18"-MS-4-KBA  
Reference Dwg. PI3-2 A

This piping has been analyzed for the THERMAL, DEAD WEIGHT, PRESSURE & OPERATIONAL BASIS EARTHQUAKE loading condition and is found to be:

- Satisfactory, as is.
- Satisfactory, with comments noted.
- Unsatisfactory - See Comments.

COMMENT (1) ALL PIPE STRESSES ARE SATISFACTORY.  
(2) ALL ANCHOR FORCES AND MOMENTS ARE REASONABLE AND SHOULD BE ACCEPTED.

RECOMMENDATION (1) SUBMIT THE FORCES AND MOMENTS TO ROCKWELL FOR APPROVAL.  
(2) USE THE REFERENCED ROUTING.

ATTACHS: PIPE STRESS SUMMARY REPORT, SUMMARY OF FORCES & MOMENTS

G. H. May  
Signature

Copies to: Sender  
Analysis Folder - Job File  
Piping Engrg. Supervisor TCT  
H.E. NOBLE



DIVISION USAGE						<b>Stearns-Roger</b> INCORPORATED ENGINEERING STANDARD	STANDARD NUMBER
MM	P	PP	SH	FI	SP		EE16.01.8
APPROVALS						PIPE STRESS SUMMARY REPORT  ANSI B31.1 - 1973 EDITION	PAGE <u>1</u> OF <u>1</u>
Des. Sect. <u>MS-5</u>							ISSUED 5/27/75 REVISED
Sect. Supt. <u>[Signature]</u>							
Div. _____							

10 MWE SOLAR PILOT PLANT - DOE  
Job Name

MAIN STEAM TSS EXCHANGER BRANCH 18" MS-4KBA  
System/Pipe Line Name

C-21700  
Job No.

MS-5  
Analysis No.

1. Loading Conditions Analyzed

Pressure X  
 Weight X  
 Thermal Expansion X  
 Sustained Mech. Loads \_\_\_\_\_  
 Occasional Loads X

Analysis Ident. Code

P-MS-5-A-0  
W-MS-5-A-2  
T-MS-5-A-2  
X-MS-5-A-1

2. Stress Evaluation (Code Equations)

Eq. (11)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} \leq 1.0 S_h$       5649 psi  $\leq$  15000 psi

Material: - ASTM A106 GR. B

PRESS = 1385 PSI  $\Rightarrow$  SLP = 4955 PSI  
 Temp = 650°F @ PT. 655N

Eq. (12)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} + \frac{0.75 iM_B}{z} \leq K S_h$       1733 psi  $\leq$  18000 psi

Material: - ASTM A106 GR. B

Temp = 650°F @ PT. 605N

Eq. (13)  $S_E = \frac{iM_C}{z} \leq S_A$       4446 psi  $\leq$  22500 psi

Material: - ASTM A106 GR. B

Temp = 650°F @ PT. 605N

Eq. (14)  $\frac{Pd^2}{Do^2-d^2} + \frac{0.75 iM_A}{z} + \frac{iM_C}{z} \leq (S_h + S_A)$       9462 psi  $\leq$  37500 psi

Material: - ASTM A106 GR. B

Temp = 650°F @ PT. 605N

3. Stress Evaluation (Local or Special)

Loading	Analysis Ident. Code	Calculated Stress (psi)	Allowable Stress (psi)
_____	_____	_____	_____
_____	_____	_____	_____

REMARKS:

1) ALL PIPE STRESSES ARE SATISFACTORY.

2) EQ. 12 INCLUDES AN OPERATIONAL BASIS EARTHQUAKE BASED ON A GROUND RESPONSE SPECTRUM.

[Signature]  
Prepared By

7-28-80  
Date

[Signature]  
Approved By

7-28-80  
Date

CUSTOMER: *DOE*  
 PROJECT: *10MWB SOLAR PILOT PLANT*  
 JOB NO: *C-21700*  
 BY: *GMM* DATE: *FEB. 28, 1980*  
 REF. DWGS: *P13-2 A*  
 ANALYSIS CODE: *T/WIX-MS-5-A-212/1*

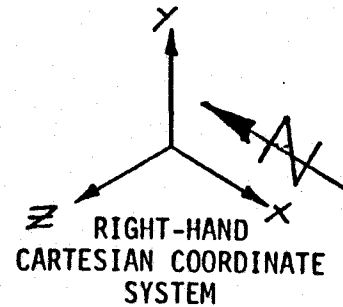
SUMMARY OF FORCES & MOMENTS  
 ON SYSTEM TERMINAL EQUIPMENT  
 ( SYSTEM )

*MAIN STEAM TSS EXCHANGER BRANCH*  
*8" - MS - 4 - HBA*

DIVISION USAGE				
MM	P	PP	SH	FI
				SP
X				

APPROVALS  
 Des. Sect. *[Signature]*  
 Sect. Superv. *[Signature]*  
 Div. *[Signature]*

THE REPORTED REACTIONS BASED ON A THERMAL EXPANSION ANALYSIS FROM ...*70*...°F TO *650*...°F USING  $E_c$ , THE COLD MOD. OF ELASTICITY, AND .....*0*...% COLD SPRING.



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EQUIPMENT CONNECTIONS	LOC. NO.	FORCES (LBS)				MOMENTS (FT.-LBS)			
		X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<i>DESUPERHEATER DS-30</i> (THERMAL)	<i>600</i>	<i>346</i>	<i>-212</i>	<i>-120</i>	<i>423</i>	<i>2171</i>	<i>3970</i>	<i>-3934</i>	<i>5996</i>
<i>STEAM OUTLET, 10" EAST</i> (WEIGHT)	<i>600</i>	<i>3</i>	<i>-136</i>	<i>-3</i>	<i>136</i>	<i>94</i>	<i>22</i>	<i>63</i>	<i>115</i>
<i>HI, (SA-301)</i> (SEISMIC)	<i>600</i>	<i>813</i>	<i>627</i>	<i>1658</i>	<i>1950</i>	<i>9931</i>	<i>6524</i>	<i>8235</i>	<i>14457</i>
<i>TSS COND. E-302</i> (THERMAL)	<i>665</i>	<i>-317</i>	<i>-173</i>	<i>-173</i>	<i>400</i>	<i>1769</i>	<i>-3886</i>	<i>2804</i>	<i>5108</i>
<i>STEAM INLET, 10" EAST</i> (WEIGHT)	<i>665</i>	<i>-4</i>	<i>58</i>	<i>-3</i>	<i>58</i>	<i>-484</i>	<i>-57</i>	<i>-11</i>	<i>488</i>
<i>SI, (SA-303)</i> (SEISMIC)	<i>665</i>	<i>792</i>	<i>366</i>	<i>1119</i>	<i>1419</i>	<i>4851</i>	<i>8038</i>	<i>6679</i>	<i>11522</i>
<i>TSS Cond. E-301</i> (THERMAL)	<i>690</i>	<i>-29</i>	<i>24</i>	<i>293</i>	<i>296</i>	<i>3577</i>	<i>-398</i>	<i>249</i>	<i>3608</i>
<i>STEAM INLET, 10" EAST</i> (WEIGHT)	<i>690</i>	<i>1</i>	<i>83</i>	<i>6</i>	<i>83</i>	<i>-532</i>	<i>12</i>	<i>29</i>	<i>533</i>
<i>GI (SA-302)</i> (SEISMIC)	<i>690</i>	<i>959</i>	<i>431</i>	<i>1388</i>	<i>1741</i>	<i>6085</i>	<i>8731</i>	<i>7742</i>	<i>13160</i>

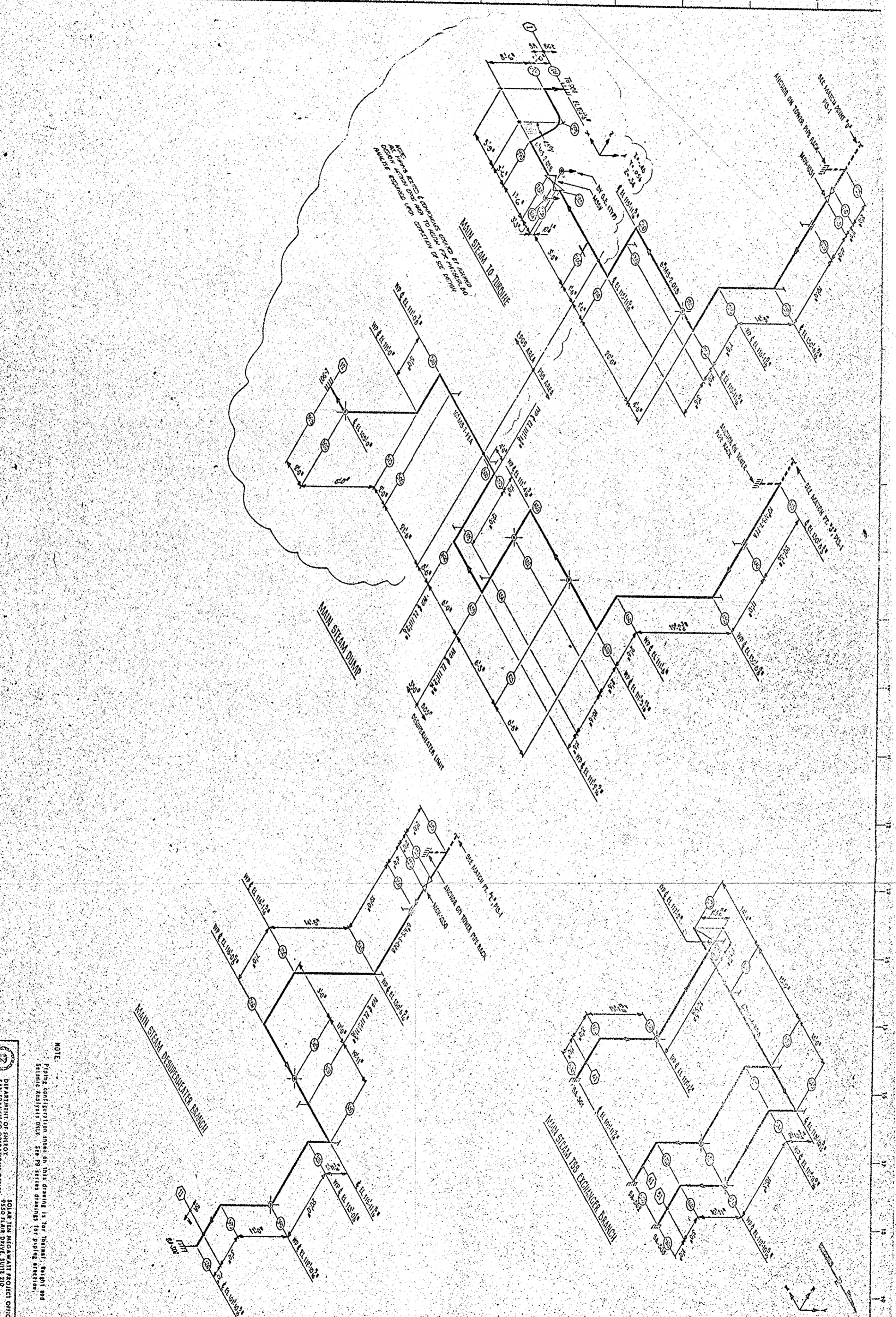
SUMMARY OF FORCES & MOMENTS  
 PER. ANSI B31.1-1973 PWR. PIPING CODE

**Stearns-Roger**  
 INCORPORATED  
 ENGINEERING STANDARD

STANDARD NUMBER  
 EE 16.01.7  
 PAGE 1 OF 1  
 ISSUED 2/28/74  
 REVISED 4/18/78



REVISION		DATE		BY		CHECKED	
1	ISSUED FOR CONSTRUCTION	12/15/11	12/15/11	...	...	...	...
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90	...	...	...	...	...	...	...
91	...	...	...	...	...	...	...
92	...	...	...	...	...	...	...
93	...	...	...	...	...	...	...
94	...	...	...	...	...	...	...
95	...	...	...	...	...	...	...
96	...	...	...	...	...	...	...
97	...	...	...	...	...	...	...
98	...	...	...	...	...	...	...
99	...	...	...	...	...	...	...
100	...	...	...	...	...	...	...



NOTE:  
 Piping configuration shown on this drawing is for thermal, safety and  
 control purposes only. All other piping drawings for piping installation.

SOLAR FACILITIES DESIGN INTEGRATOR  
 ANALYSIS: ISOMETRIC  
 MAIN STEAM - P35, T35 & EPG  
 ANALYSIS ISOMETRIC (P35-T35)  
 4072005191921

DEPARTMENT OF ENERGY  
 SAN FRANCISCO OPERATIONS OFFICE  
 1000 CALIFORNIA STREET, SUITE 100  
 SAN FRANCISCO, CALIFORNIA 94133-0001  
 SOLAR THE MCGRAW HILL GROUP  
 9550 FLAM DRIVE, SUITE 210  
 SAN FRANCISCO, CALIFORNIA 94133-0001

BRANCH CONNECTION SYMBOLS	PIPE SUPPORT SYMBOLS	ANCHOR OR RESTRAINT NO.	DISPLACEMENT (COLD TO HOT) IN.			ANCHOR &/OR TERM. POINTS	COORDINATE		
			X	Y	Z		X	Y	Z
UNREINFORCED PIPE-PIPE INTERSECTION	CONSTANT SUPPORT	205	0.0	0.0	0.0	205 MS-2	0'-0"	0'-0"	0'-0"
REINFORCED PIPE-PIPE INTERSECTION	VARIABLE SPRING SUPPORT	300	-4.6	0.54	.35	300	43'-0 1/2"	-6'-10 5/16"	-38'-6"
ASA TEE	RIGID ROD OR STRUT	215	0.0	0.0	0.0	215 MS-3	0'-0"	0'-0"	0'-0"
WELDOLET	HYDRAULIC SNUBBER	400				400	-39'-11"	-28'-4 3/16"	-58'-0"
SWEEPOLET	RIGID GUIDE	415				415 MS-4	0'-0"	0'-0"	0'-0"
	ANCHOR	525				525	53'-6"	-25'-2 3/8"	-39'-9 1/8"
		600				600	0'-0"	0'-0"	0'-0"
		665				665 MS-5	-46'-1"	0'-0 1/2"	-0'-2"
		690	↓	↓	↓	690	-28'-1"	0'-0 1/2"	-0'-2"

MODELING SYMBOLS	PIPING SYMBOLS
POINT OF INTEREST	ELBOW
DATA POINT NO.	BEND
MASS POINT OR CENTER OF GRAVITY	VALVES (ASSUMED RIGID)
INTERFACE POINT	REDUCER OR INCREASER
	EQUIPMENT NOZZLE

ANALYZED CONDITIONS				VALVES & SPECIAL COMPONENTS			
OPERATING TEMP (°F)	DESIGN PRESSURE (PSIG)	CASE	VALVE OR SPEC. COMP. NO.	WGT. (LBS.)	STATUS EST. OR FINAL	REFERENCE & COMMENTS	
960	1565	MS-2	MSSV	3000	EST	GE Prelim Turbine Info.	
960	1565	MS-3	MOV-1031	1315	EST	Powell 125023 WG	
885	100	MS-4/PT 415-485	MOV-1030	1315	EST		
350	100	MS-4/PT 415-525					
650	1385	MS-5					

NOTES:

- 1) DIMENSIONS ARE IN FEET AND INCHES UNLESS NOTED OTHERWISE.
- 2) ALL ELBOWS ARE ASA LONG RADIUS UNLESS NOTED OTHERWISE.
- 3) PIPE ANALYZED FROM COLD TO HOT.
- 4) THERMAL ANALYSIS TEMPERATURE RANGE IS FROM 70° F. TO OPERATING TEMPERATURE UNLESS NOTED OTHERWISE.
- 5) SCALE: NONE
- 6) THIS DATA AND ATTACHED DRAWINGS USED FOR ANALYSIS PURPOSES ONLY.

DATA POINTS		PIPE MATERIAL	LINE SPEC.	PIPE SIZE (IN.)		WEIGHT LBS./LINEAR FT.			INS. THK.
FROM	TO			O.D.	W.T.	PIPE	FLUID	INSULATION	
205 / 215	300 / 400	A335 P22	QEB	6.625	.8104	54.2	3.1	15.2	5
415	484	↓ P11	FEA	10.75	.365	41.3	34.1	20.6	5
600	665 / 690	A106 Gr. B	KBH	8.625	.500	43.4	19.8	16.1	3 1/2
484	525	A335 P11	FEA	10.75	.365	41.3	34.1	8.7	3 1/2

NO.	REFERENCES	NO.	ANALYSIS CODES	NO.	REVISIONS	DATE	BY	CH'D	APP'D
			T/W/K-MS-2-A-4/5/3	▲					
			T/W/K-MS-3-A-3/3/1	▲					
			T/W/K-MS-4-A-5/2/1	▲					
			T/W/K-MS-5-A-2/2/1	▲					

**PIPING ANALYSIS DATA**

**Stearns-Roger**  
INCORPORATED (PIB-2)

10 MINE SOLAR PILOT PLANT - DAGGET, CALIF.  
**MAIN STEAM - PSS TSS & EPGs**  
FOR ISOMETRIC PIB-2

DRAWN: KEN	DATE: 2-15-80	ANAL. BY G.H.M.	DATE: 2-28-80
CHECKED: KEN	DATE: 3-3-80	APP'D. BY: [Signature]	DATE: 3-19-80
JOB NO. C-21700	SYSTEM NO. MS-2,3,4&5	SHEET 1 OF 1	



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***** ** ** ** ***** ***** ** ***** **
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FOREWORD

THE FOLLOWING REPORT WAS GENERATED BY THE DYNAPLEX PIPING ANALYSIS PROGRAM. THE PROGRAM ACCOMMODATES THE WIDEST RANGE OF PRACTICAL SITUATIONS ENCOUNTERED IN MODERN PIPING SYSTEMS INCLUDING DYNAMIC AS WELL AS THERMAL, WEIGHT AND PRESSURE EFFECTS.

FOR FURTHER INFORMATION OR PROMPT ASSISTANCE IN USING THE DYNAPLEX PROGRAM, PLEASE CONTACT YOUR LOCAL COMPUTER CENTER REPRESENTATIVE OR:

ENGINEERING SERVICES MANAGER  
AUTON COMPUTING CORPORATION  
1 BETRO PLAZA  
505 THORNALL STREET  
EDISON, N.J. 08817

THE STRESS FORMULATION OF  
ANSI B31.1 - 1977  
INCLUDING THE LATEST MANDATORY UPDATES  
HAS BEEN APPLIED IN THIS ANALYSIS

THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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*
* STEARNS - ROGER CORP
* -----
*
* IOWA SOLAR PILOT PLANT - C-21700
*
* AS DESUP TO TR STOR HTR(T/W-MS-C5-A-272)
*
*
*
* 2/25/80
*
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DYNAPLEX (MOD204F-UCC)  
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LAST UPDATE - 12/25/79

DYNAFLEX

STEARNS - ROGER COMP

PAGE 1

PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
JOB - MS DESUP TO TH STOR HTR(T/M-MS-05-A-2/2)

DATE 2/25/80

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	FR	TO	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
1	MEM								STEARNS - ROGER CORP, 2/25/80, PROJ. 10MW SO LAR PILOT PLANT - C-2 1700, JOHNS DESUP TO T H STOR HTR(T/M-MS-05- A-2/2)	1
2	GEN								APPLY R31.1-1973, EXA	2
3		600	605				4-0	L	MAT=LCS, OD=8.625, WT=.500, UNIF=11.1, TEMP=650.	3
4				610		15-1	15/16	L		4
5				620			16-9-5/16	L		5
6				625		-1-3	11/16	L		6
7				630	-14-4					7
8				635	-13-9				WLT	8
9				665	-18-0			L		9
10				667		-2-10	3/4	L		10
11				655			-18-3	L		11
12				660		-10-11		L		12
13				665			-4-0			13
14				635	670		-2-10-3/4	L		14
15				680			-18-3	L		15
16				685		-10-11		L		16
17				690			-4-0			17
18		600								18
19		665								19
20		690								20
21		620	N			RIGID				21
22		630	N			RIGID				22
23		635	N			RIGID				23
24		645	N			RIGID				24
25		610	F	655	680				LOC(610F, 655N, 680N), USE TABLE 1	25
26		665			-4-1	0-0-1/2	-0-2			26
27		690			-2-1	0-0-1/2	-0-2			27

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PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
JOB - HS DESUP TO TH STOP HTR(T/W-MS-05-A-2/2)

DATE 2/25/80

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\*  
\* EDITED PIPING SYSTEM DESCRIPTION \*  
\*-----\*  
\*  
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EXPLANATORY NOTES

IN THE EDITED PIPING SYSTEM DESCRIPTION:

- (1) POINT LOCATION NUMBER SUFFIXES N, M AND F IDENTIFY THE NEAR POINT, THE MID POINT AND THE FAR POINT RESPECTIVELY OF A PIPING ELEMENT IN THE DIRECTION IN WHICH THE LINE IS TRAVERSED. FRACTIONAL NODE NUMBERS APPEARING IN THE FOLLOWING REPORTS IDENTIFY THOSE NODES AT WHICH THE PROGRAM HAS INTRODUCED EVENLY SPACED LUMP POINTS AND ARE DETERMINED FROM THE MAXIMUM LOOP SPACING CRITERIA ENTERED BY THE USER. IF FRACTIONAL NODE NUMBERS ARE ASSIGNED TO MORE THAN ONE ELEMENT ORIGINATING AT THE SAME BRANCH POINT, A LETTER WILL PRECEDE THE FRACTIONAL PORTION OF THE NUMBER TO UNIQUELY IDENTIFY EACH POINT (E.G. 2.A1). LETTERS ARE ASSIGNED TO ELEMENTS ALPHABETICALLY IN THE ORDER OF THEIR APPEARANCE IN THE INPUT DATA.
- (2) AMBIENT TEMPERATURE IS ASSUMED TO BE 70 DEGREES FAHRENHEIT.
- (3) THE MAXIMUM ALLOWABLE DIFFERENCE IN LOOP CLOSURE CALCULATIONS IS ASSUMED TO BE .25 INCHES.
- (4) THE MAXIMUM ALLOWABLE DIFFERENCE BETWEEN THE CALCULATED AND INPUT COORDINATES OF USER DESIGNATED CHECK POINTS IS ASSUMED TO BE .25 INCHES.
- (8) IN ORDER TO MAKE IT POSSIBLE IN CERTAIN SITUATIONS TO PERFORM INDEPENDENT WEIGHT AND THERMAL ANALYSES IN THE SAME RUN AND APPLY THE CODE STRESS CALCULATIONS CORRECTLY, THE PROGRAM MAKES CERTAIN ASSUMPTIONS. IF THE INPUT DATA SPECIFIES IMPOSED MOVEMENTS ON ANCHORS (ANC) OR DISPLACEMENTS OR ROTATIONS ON RIGID RESTRAINTS (RAD, RAR) THESE ARE ASSUMED BY THE PROGRAM TO BE THERMAL EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE THERMAL. ALTERNATELY, IF THE INPUT DATA SPECIFIES FORCES AND MOMENTS (FOR, MOM) OR IMPOSED DISPLACEMENTS OR ROTATIONS ON FLEXIBLE RESTRAINTS (RAD, RAR) THESE ARE ASSUMED TO BE WEIGHT EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE WEIGHT. IF THESE ASSUMPTIONS ARE INVALID FOR A PARTICULAR SITUATION, THE USER CAN ACCURATELY MODEL THE MOST GENERAL SITUATION BY USING THE AUXILIARY SUPPORT MOVEMENTS (ASH) AND THE AUXILIARY CONCENTRATED LOADS (ACL) FEATURE OF THE PROGRAM AS AN ALTERNATE METHOD OF SPECIFYING THESE EFFECTS.
- (9) A POINT IDENTIFIED AS A TENTATIVE HANGER LOCATION IS ASSUMED TO BE FREE TO MOVE IN THE VERTICAL DIRECTION FOR THE INITIAL THERMAL LOADING CASE AND FULLY RESTRAINED IN THE VERTICAL DIRECTION FOR THE INITIAL WEIGHT LOADING CASE.
- (15) THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)		
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF	
	BEND - (	610B	0.00	14.87	4.29	1.00	1.57	90.000	8.625	.500	LCS	650	43.4	11.1
	(	S-610F	0.00	15.16	5.00									
	(	* TENTATIVE HANGER AT POINT 610F - SEE NOTE (9) ABOVE.												
	(	HANGER LENGTH NOT SPECIFIED.												
	(													
	STRAIGHT - (						15.36							
	(	S-620N	0.00	15.16	20.36									
	(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 620N												
	(	IN DIRECTION Y												
	(													
	BEND - (	620H	0.00	15.09	20.74	1.00	.79	45.000						
	(	S-620F	0.00	14.87	21.07									
209	STRAIGHT - (						.43							
	(	S-625N	0.00	14.56	21.38									
	(													
	BEND - (	625H	-0.29	14.06	21.88	1.00	1.57	90.000						
	(	S-625F	-1.00	13.85	22.08									
	STRAIGHT - (						13.33							
	(	S-630	-14.33	13.85	22.08									
	(	* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 630												
	(	IN DIRECTION Y												
	(													
	STRAIGHT - (						13.75							
	(													
BR PT	(	S-635	-22.08	13.85	22.08									

\* NOTES PERTAINING TO POINT 635 APPEAR ON THE FOLLOWING PAGE



POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
								8.625	.500	LCS	650.	43.4	11.1
	STRAIGHT - (	>-660F	-46.08	.04	2.83		3.00						
ANCHR		--665	-46.08	.04	-1.17								
* CONTROL COORDINATES OF POINT 665, IN FEET :													
X = -46.08, Y = .04, Z = -1.17													
RESULTANT DIFFERENCE IS ZERO													
HR PT		--635	-28.08	13.85	22.08								
( * WELDING TEE AT POINT 635													
( * RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 635													
( IN DIRECTION Y													
	STRAIGHT - (						1.90						
211		>-6700	-28.08	11.96	22.08								
		( BEND - (	6700	-28.08	11.25	21.79	1.00	1.57	90.000				
		>-6700F	-28.08	10.96	21.08		16.25						
	STRAIGHT - (												
		>-6800	-28.08	10.96	4.83								
( * TENTATIVE HANGER AT POINT 6800 - SEE NOTE (9) ABOVE.													
( HANGER LENGTH NOT SPECIFIED.													
	BEND - (	6800	-28.08	10.67	4.13	1.00	1.57	90.000					
		>-6800F	-28.08	9.96	3.83		8.92						
	STRAIGHT - (												
		>-6850	-28.08	1.04	3.83								
	BEND - (	6850	-28.08	.33	3.54	1.00	1.57	90.000					
		>-6850F	-28.08	.04	2.83								
	STRAIGHT - (						3.00						
ANCHR		--690	-28.08	.04	-1.17								

\* NOTES PERTAINING TO POINT 690 APPEAR ON THE FOLLOWING PAGE

DYNAFLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 7

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	MATERIAL	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z								PRESS (PSI)	PIPE WT. UNIF
ANCHP		--690	-28.08	.00	-.17				8.625	.500	LCS	650.	43.4	11.1
* CONTROL COORDINATES OF POINT 690, IN FEET : X = -28.08, Y = .00, Z = -.17 RESULTANT DIFFERENCE IS ZERO														

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DYNAFLEX

LOADING - 2FREED THERMAL

PAGE 8

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)				
	X	Y	Z	RESULTANT	Y	Z	RESULTANT		
<b>ANCHORS:</b>									
600	346.	-212.	-120.	423.	2171.	3970.	-3934.	5996.	
665	-317.	-173.	-173.	400.	1769.	-3886.	2804.	5108.	
690	-27.	24.	293.	296.	3577.	-398.	249.	3608.	
<b>RESTRAINTS:</b>									
620	0.	285.	0.	285.	0.	0.	0.	0.	
213 630	0.	-262.	0.	262.	0.	0.	0.	0.	
635	0.	299.	0.	299.	0.	0.	0.	0.	
645	0.	40.	0.	40.	0.	0.	0.	0.	

\*\*\*\*\*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 4445. OCCURS AT POINT 605N

EXPLANATORY NOTES:  
 -----

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ANSI B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A ( \*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (14) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 INLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLEX

LOADING - YFLEW THERMAL

PAGE 10

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	EXPANSION STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHR		-- 600	120.	406.	3934.	2171.	3970.	1.00	1.00	24.51	2935.
	STR - (	>- 605F	120.	405.	3934.	1534.	2933.	1.77	1.77	24.51	4446. **
	BEND - (	605M	235.	352.	809.	1419.	4611.	1.77	1.77	24.51	4231. *
		>- 605F	212.	366.	2587.	1442.	3588.	1.77	1.77	24.51	4024. *
	STR - (	>- 610M	212.	366.	2587.	3024.	965.	1.77	1.77	24.51	3541.
	BEND - (	610M	235.	352.	2613.	3046.	902.	1.77	1.77	24.51	3558.
		>- 610F	120.	406.	1311.	2931.	2241.	1.77	1.77	24.51	3387.
	STR - (	>- 620M	120.	353.	1311.	331.	3073.	1.77	1.77	24.51	2904.
	BEND - (	620M	139.	347.	2414.	312.	2470.	1.77	1.77	24.51	2999.
		>- 620F	136.	341.	3202.	315.	1491.	1.77	1.77	24.51	3067.
	STR - (	>- 625M	136.	346.	3202.	1641.	329.	1.77	1.77	24.51	3125.
215	BEND - (	625M	341.	152.	2507.	1846.	2007.	1.77	1.77	24.51	3204.
		>- 625F	346.	140.	363.	1851.	3168.	1.77	1.77	24.51	3189.
	STR - (	>- 630	346.	225.	363.	1896.	1947.	1.00	1.00	24.51	1342.
BR PT		-- 635	346.	225.	363.	715.	294.	1.35	1.35	24.51	567.
HQ PT		-- 635	317.	219.	308.	870.	42.	1.35	1.35	24.51	612.
	STR - (	>- 645M	317.	245.	308.	1397.	2987.	1.77	1.77	24.51	2865.
	BEND - (	645M	346.	261.	2580.	1427.	2017.	1.77	1.77	24.51	2968.
		>- 645F	173.	361.	3160.	1254.	134.	1.77	1.77	24.51	2943.
	STR - (	>- 647M	173.	361.	3160.	21.	970.	1.77	1.77	24.51	2859.
	BEND - (	647M	245.	317.	1641.	93.	2697.	1.77	1.77	24.51	2732.
		>- 647F	173.	361.	654.	21.	2844.	1.77	1.77	24.51	2524.

DYNAFLY

LOADING - FREE THERMAL

PAGE 10

INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLER TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	EXPANSION STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHR		-- 600	120.	406.	3934.	2171.	3970.	1.00	1.00	24.51	2935.
	STR - (	>- 605H	120.	405.	3934.	1534.	2933.	1.77	1.77	24.51	4446. **
	HEND - (	605M	235.	352.	809.	1419.	4611.	1.77	1.77	24.51	4231. *
		>- 605F	212.	366.	2587.	1442.	3588.	1.77	1.77	24.51	4024. *
	STR - (	>- 610H	212.	366.	2587.	3024.	965.	1.77	1.77	24.51	3541.
	HEND - (	610M	235.	352.	2613.	3046.	902.	1.77	1.77	24.51	3558.
		>- 610F	120.	406.	1311.	2931.	2241.	1.77	1.77	24.51	3387.
	STR - (	>- 620H	120.	353.	1311.	331.	3073.	1.77	1.77	24.51	2904.
	HEND - (	620M	139.	347.	2414.	312.	2470.	1.77	1.77	24.51	2999.
		>- 620F	136.	348.	3202.	315.	1491.	1.77	1.77	24.51	3067.
	STR - (	>- 625H	136.	348.	3202.	1641.	329.	1.77	1.77	24.51	3125.
216	HEND - (	625M	341.	152.	2507.	1846.	2007.	1.77	1.77	24.51	3204.
		>- 625F	346.	140.	363.	1851.	3168.	1.77	1.77	24.51	3189.
	STR - (	>- 630	346.	225.	363.	1896.	1947.	1.00	1.00	24.51	1342.
BR PT		-- 635	346.	225.	363.	715.	294.	1.35	1.35	24.51	567.
BR PT		-- 645	317.	219.	308.	870.	42.	1.35	1.35	24.51	612.
	STR - (	>- 645H	317.	245.	308.	1397.	2987.	1.77	1.77	24.51	2865.
	HEND - (	645M	346.	201.	2380.	1427.	2017.	1.77	1.77	24.51	2968.
		>- 645F	173.	361.	3160.	1254.	134.	1.77	1.77	24.51	2943.
	STR - (	>- 647H	173.	361.	3160.	21.	970.	1.77	1.77	24.51	2859.
	HEND - (	647M	245.	317.	1641.	93.	2697.	1.77	1.77	24.51	2732.
		>- 647F	173.	361.	654.	21.	2844.	1.77	1.77	24.51	2524.



DYNAFLEX

LOADING - FREE THERMAL

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\*\*\*\*\*  
 \* INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES \*  
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EXPLANATORY NOTES

THE FORCES AND MOMENTS AT ANY POINT SHOWN BELOW REPRESENT THE ACTION OF THE FAR END OF THE BRANCH ACTING ON THE NEAR END OF THE BRANCH.

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
ANCHOR	STRAIGHT	600	346.	-212.	-120.	2171.	3970.	-3934.
		> 6050	346.	-212.	-120.	1534.	2933.	-3934.
	BEND	6050	346.	-212.	-120.	1419.	2688.	-3832.
		> 605F	346.	-212.	-120.	1442.	2587.	-3588.
		> 6100	346.	-212.	-120.	3024.	2587.	965.
	BEND	6100	346.	-212.	-120.	3046.	2485.	1210.
		> 610F	346.	-212.	-120.	2931.	2241.	1311.
	STRAIGHT	6200	346.	72.	-120.	-331.	-3073.	1311.
		> 6200	346.	72.	-120.	-312.	-3206.	1285.
	BEND	6200	346.	72.	-120.	-315.	-3310.	1210.
		> 620F	346.	72.	-120.	-329.	-3424.	1104.
		> 6250	346.	72.	-120.	-353.	-3562.	952.
	> 625F	346.	72.	-120.	-363.	-3549.	931.	
	> 630	346.	-190.	-120.	-363.	-1947.	1896.	
BRANCH PT.		635	346.	-190.	-120.	-363.	-294.	-715.
BRANCH PT.	STRAIGHT	635	317.	133.	173.	-308.	-42.	-870.
	> 6450	317.	173.	173.	-306.	-2987.	1397.	
	BEND	6450	317.	173.	173.	-257.	-3110.	1427.

INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
	BEND - (	6450	317.	173.	173.	-257.	-3110.	1427.
		)						
	STRAIGHT - (	>- 645F	317.	173.	173.	-134.	-3160.	1254.
		)						
	BEND - (	>- 6470	317.	173.	173.	21.	-3160.	970.
		)						
	BEND - (	647F	317.	173.	173.	93.	-3068.	746.
		)						
	STRAIGHT - (	>- 647F	317.	173.	173.	21.	-2844.	654.
		)						
		>- 6551	317.	173.	173.	-2794.	2303.	654.
		)						
	BEND - (	6550	317.	173.	173.	-2866.	2527.	561.
		)						
	STRAIGHT - (	>- 655F	317.	173.	173.	-2794.	2620.	337.
		)						
		>- 6600	317.	173.	173.	-1249.	2620.	-2487.
		)						
	BEND - (	6660	317.	173.	173.	-1177.	2712.	-2711.
		)						
	STRAIGHT - (	>- 666F	317.	173.	173.	-1249.	2936.	-2804.
		)						
ANCHOR		-- 665	317.	173.	173.	-1769.	3886.	-2804.
BRANCH PT.	STRAIGHT - (	635	29.	-24.	-293.	-56.	-253.	155.
		)						
219		>- 6700	29.	-24.	-293.	-612.	-253.	99.
		)						
	BEND - (	6700	29.	-24.	-293.	-612.	-244.	78.
		)						
	STRAIGHT - (	>- 670F	29.	-24.	-293.	-881.	-224.	70.
		)						
		>- 6800	29.	-24.	-293.	-493.	252.	70.
		)						
	BEND - (	6800	29.	-24.	-293.	-563.	272.	61.
		)						
	STRAIGHT - (	>- 680F	29.	-24.	-293.	-763.	281.	41.
		)						
		>- 6850	29.	-24.	-293.	-3379.	281.	-220.
		)						
	BEND - (	6850	29.	-24.	-293.	-3580.	289.	-241.
		)						
	STRAIGHT - (	>- 685F	29.	-24.	-293.	-3649.	310.	-249.
		)						
ANCHOR		-- 690	29.	-24.	-293.	-3577.	398.	-249.

DYNALÉX

LOADING - FREE THERMAL

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 \* DISPLACEMENTS AND ROTATIONS \*  
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POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 600	.000	-.000	-.000	.000	.000	-.000
	STRAIGHT - (	>- 605	.010	-.005	.153	.016	.029	-.043
		( 605	.020	.006	.191	.030	.062	-.065
	STRAIGHT - (	>- 605F	.037	.040	.211	.040	.082	-.103
		>- 610	.428	.713	.433	.127	.205	-.152
	BEND - (	610	.463	.740	.469	.157	.213	-.148
		>- 610F	.505	.729	.515	.187	.231	-.148
	STRAIGHT - (	>- 620	1.282	.000	1.300	.243	.213	-.075
		( 620	1.297	-.023	1.316	.241	.199	-.075
220		>- 620F	1.307	-.051	1.322	.240	.188	-.077
	STRAIGHT - (	>- 625	1.314	-.082	1.322	.239	.183	-.075
		( 625	1.309	-.128	1.333	.243	.162	-.078
	STRAIGHT - (	>- 625F	1.276	-.138	1.354	.249	.130	-.073
		>- 630	.595	-.000	1.560	.232	.028	-.020
BRANCH PT.		-- 635	-.107	.000	1.562	.213	-.016	.003
BRANCH PT.		-- 635	-.107	.000	1.562	.213	-.016	.003
	STRAIGHT - (	>- 645	-.976	.000	1.420	.194	-.088	.015
		( 645	-1.010	-.018	1.393	.200	-.113	.030
	STRAIGHT - (	>- 645F	-1.020	-.056	1.356	.206	-.126	.043
		>- 647	-1.011	-.102	1.317	.205	-.136	.046

DYNAPLEX

LOADING - FREE THERMAL

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DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 6475	-1.011	-.102	1.317	.205	-.136	.046
	BEND - (	6475	-.995	-.126	1.272	.206	-.151	.060
		>- 6476	-.966	-.110	1.223	.207	-.178	.070
	STRAIGHT - (	6551	-.271	.523	.393	.144	-.191	.109
	BEND - (	6550	-.238	.527	.349	.115	-.171	.106
		>- 6556	-.212	.498	.319	.087	-.161	.104
	STRAIGHT - (	6609	-.032	.042	.210	.037	-.076	.077
	BEND - (	6601	-.019	.008	.191	.025	-.059	.048
		>- 6607	-.010	-.004	.153	.013	-.029	.031
	STRAIGHT - (	665	-.000	-.000	-.000	.000	-.000	.000
ANCHOR		-- 635	-.107	.000	1.562	.213	-.016	.003
	STRAIGHT - (	6700	-.106	-.097	1.477	.212	-.017	.003
221		>- 6700	-.104	-.120	1.431	.204	-.019	.005
	BEND - (	6700	-.101	-.106	1.383	.196	-.021	.006
		>- 6800	-.025	.503	.552	.165	-.020	.010
	BEND - (	6800	-.022	.512	.506	.159	-.018	.009
		>- 6806	-.019	.485	.468	.153	-.017	.009
	STRAIGHT - (	6850	-.003	.030	.220	.101	-.008	.007
	BEND - (	6850	-.002	-.002	.193	.067	-.006	.004
		>- 6851	-.001	-.009	.153	.030	-.003	.003
	STRAIGHT - (	690	-.000	.000	.000	.000	-.000	.000
ANCHOR		-- 690	-.000	.000	.000	.000	-.000	.000





DYNAFLEX

LOADING - RESTRAINED WEIGHT (+UNIF)

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 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
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EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)				
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT	
<b>ANCHORS:</b>									
609	3.	-136.	-3.	136.	94.	22.	63.	115.	
665	-4.	58.	-3.	58.	-484.	-57.	-11.	488.	
690	1.	83.	6.	83.	-532.	12.	29.	535.	
<b>RESTRAINTS:</b>									
223 610F	0.	-1364.	0.	1364.	0.	0.	0.	0.	
620F	0.	-817.	0.	817.	0.	0.	0.	0.	
630	0.	-809.	0.	809.	0.	0.	0.	0.	
635	0.	-1446.	0.	1448.	0.	0.	0.	0.	
645F	0.	-1000.	0.	1000.	0.	0.	0.	0.	
655F	0.	-1409.	0.	1409.	0.	0.	0.	0.	
680F	0.	-1428.	0.	1428.	0.	0.	0.	0.	

DYNAPLEX

LOADING - RESTRAINTS WEIGHT

(+UNIF)

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 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
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THE MAXIMUM STRESS OF 694. OCCURS AT POINT 655N

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ASME B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 PORTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNALOX

LOADING - RESTRAINED WEIGHT (\*UNIF)

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHP		-- 602	3.	136.	65.	94.	22.	1.00	1.00	24.51	56.
	STR	>- 605N	3.	28.	63.	68.	14.	1.77	1.77	24.51	61.
	BEND	-( 605B	40.	52.	53.	33.	37.	1.77	1.77	24.51	47.
		>- 605F	110.	4.	11.	6.	66.	1.77	1.77	24.51	43.
	STR	>- 610N	431.	4.	11.	29.	103.	1.77	1.77	24.51	70.
	BEND	-( 610B	616.	620.	82.	283.	67.	1.77	1.77	24.51	196.
		>- 610F	3.	917.	106.	917.	8.	1.77	1.77	24.51	599.
	STR	>- 620N	3.	427.	106.	482.	35.	1.77	1.77	24.51	321.
	BEND	-( 620B	153.	375.	112.	323.	7.	1.77	1.77	24.51	222.
		>- 620F	269.	273.	101.	194.	48.	1.77	1.77	24.51	145.
225	STR	>- 625N	253.	256.	101.	47.	79.	1.77	1.77	24.51	88.
	BEND	-( 625B	155.	277.	88.	24.	43.	1.77	1.77	24.51	65.
		>- 625F	3.	274.	156.	170.	117.	1.77	1.77	24.51	168.
	STR	>- 630	3.	453.	156.	986.	2.	1.00	1.00	24.51	489.
BR PT		-- 635	3.	394.	156.	1245.	35.	1.35	1.35	24.51	624.
BR PT		-- 645	4.	503.	49.	1295.	14.	1.35	1.35	24.51	645.
	STR	>- 645N	4.	576.	49.	615.	36.	1.77	1.77	24.51	401.
	BEND	-( 645B	374.	380.	61.	221.	7.	1.77	1.77	24.51	149.
		>- 645F	499.	5.	39.	66.	46.	1.77	1.77	24.51	58.
	STR	>- 647N	442.	5.	39.	43.	62.	1.77	1.77	24.51	55.
	BEND	-( 647B	200.	204.	15.	80.	66.	1.77	1.77	24.51	69.
		>- 647F	3.	356.	58.	347.	34.	1.77	1.77	24.51	229.

DYNAFLEX

LOADING - RESTRAINED WEIGHT

(+DRIFF)

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEX TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 677F	3.	356.	58.	347.	34.	1.77	1.77	24.51	229.
	STR - (	>- 655D	3.	879.	58.	1068.	36.	1.77	1.77	24.51	694. **
	BEND - (	655D	589.	593.	68.	460.	13.	1.77	1.77	24.51	302.
	STR - (	>- 655F	790.	5.	40.	217.	54.	1.77	1.77	24.51	148.
	STR - (	>- 660B	317.	5.	40.	191.	15.	1.77	1.77	24.51	127.
	BEND - (	660B	185.	189.	38.	107.	21.	1.77	1.77	24.51	75.
	STR - (	>- 660F	3.	222.	11.	65.	44.	1.77	1.77	24.51	51.
	ANCHP	-- 665	3.	58.	11.	484.	57.	1.00	1.00	24.51	239.
	BR PT	-- 635	551.	6.	21.	50.	107.	1.35	1.35	24.51	60.
	STR - (	>- 670B	448.	6.	21.	118.	47.	1.77	1.77	24.51	83.
226	BEND - (	670B	290.	282.	18.	1.	47.	1.77	1.77	24.51	33.
	STR - (	>- 670F	6.	362.	45.	270.	19.	1.77	1.77	24.51	178.
	STR - (	>- 680B	6.	904.	45.	1049.	5.	1.77	1.77	24.51	681. *
	BEND - (	680B	613.	605.	36.	426.	28.	1.77	1.77	24.51	278.
	STR - (	>- 680F	418.	5.	6.	182.	44.	1.77	1.77	24.51	122.
	STR - (	>- 685B	352.	6.	6.	232.	31.	1.77	1.77	24.51	152.
	BEND - (	685B	208.	200.	26.	147.	16.	1.77	1.77	24.51	97.
	STR - (	>- 685F	6.	246.	29.	39.	8.	1.77	1.77	24.51	32.
	ANCHP	-- 690	6.	83.	29.	532.	12.	1.00	1.00	24.51	261.

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 \* INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES \*  
 \*\*\*\*\*

EXPLANATORY NOTES

THE FORCES AND MOMENTS AT ANY POINT SHOWN BELOW REPRESENT THE ACTION OF THE FAR END OF THE BRANCH ACTING ON THE NEAR END OF THE BRANCH.

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
ANCHOR	STRAIGHT - (	600	3.	-136.	-3.	94.	22.	63.
		>= 6050	3.	28.	-3.	-68.	14.	63.
	BEND - (	6050	3.	71.	-3.	-33.	12.	64.
		>= 605F	3.	114.	-3.	-6.	11.	66.
	STRAIGHT - (	6100	3.	831.	-3.	29.	11.	103.
		>= 610F	3.	874.	-3.	283.	10.	105.
	STRAIGHT - (	610F	3.	917.	-3.	917.	8.	106.
		>= 6200	3.	-427.	-3.	482.	-35.	106.
	BEND - (	6200	3.	-405.	-3.	323.	-37.	106.
		>= 620F	3.	-333.	-3.	194.	-37.	105.
	STRAIGHT - (	6250	3.	-360.	-3.	79.	-38.	104.
		>= 625F	3.	-317.	-3.	-92.	-39.	6.
STRAIGHT - (	625F	3.	-274.	-3.	-156.	-38.	-203.	
	>= 630	3.	453.	-3.	-156.	-2.	986.	
BRANCH PT.	STRAIGHT - (	635	3.	394.	-3.	-156.	35.	1245.
BRANCH PT.	STRAIGHT - (	635	4.	-503.	3.	-49.	14.	1295.
		>= 6450	4.	-576.	3.	-49.	-36.	615.
	BEND - (	6450	4.	-533.	3.	-48.	-38.	221.

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DYWIDAG

LOADING - RESTRAINED FLIGHT

(+DNIF)

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INTERNAL FORCES AND MOMENTS ORIENTED TO X Y AND Z AXES (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
			X	Y	Z	X	Y	Z
	LEAD - (	6450	4.	-533.	3.	-48.	-38.	221.
	)							
	>- 645F		4.	-490.	3.	-46.	-39.	66.
	STRAIGHT - (							
	)		4.	-402.	3.	-43.	-39.	62.
	(							
	BEAD - (	6470	4.	-399.	3.	80.	-38.	59.
	)							
	>- 647F		4.	-356.	3.	347.	-34.	58.
	STRAIGHT - (							
	)		4.	-879.	3.	-1068.	36.	58.
	(							
	BEAD - (	6550	4.	-836.	3.	-460.	39.	57.
	)							
	>- 655F		4.	-794.	3.	-217.	40.	54.
	STRAIGHT - (							
	)		4.	-307.	3.	-191.	40.	15.
	(							
	BEAD - (	6600	4.	-264.	3.	-107.	41.	12.
	)							
	>- 660F		4.	-222.	3.	65.	44.	11.
	STRAIGHT - (							
228	ANCHOR	-- 665	4.	-58.	3.	484.	57.	11.
	BRANCH PT.	-- 675	-1.	-551.	-6.	-107.	21.	-50.
	STRAIGHT - (							
	)		-1.	-448.	-6.	-118.	21.	-47.
	(							
	BEAD - (	6700	-1.	-405.	-6.	1.	20.	-46.
	)							
	>- 670F		-1.	-362.	-6.	270.	19.	-45.
	STRAIGHT - (							
	)		-1.	-904.	-6.	-1049.	-5.	-45.
	(							
	BEAD - (	6800	-1.	-861.	-6.	-426.	-6.	-45.
	)							
	>- 680F		-1.	-818.	-6.	-182.	-6.	-44.
	STRAIGHT - (							
	)		-1.	-332.	-6.	-232.	-6.	-31.
	(							
	BEAD - (	6850	-1.	-289.	-6.	-147.	-7.	-30.
	)							
	>- 685F		-1.	-246.	-6.	39.	-8.	-29.
	STRAIGHT - (							
	)		-1.	-83.	-6.	532.	-12.	-29.
	ANCHOR	-- 690						

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 \* DISPLACEMENTS AND ROTATIONS \*  
 \*\*\*\*\*

POINT DESIGNATION	ELEMENT DESIGNATION	LIM. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT			
			X	Y	Z	X AXIS	Y AXIS	Z AXIS	
ANCHOR	STRAIGHT - (	>> 600	.000	-.000	-.000	.000	.000	.000	
		>> 605	.000	-.000	-.000	-.000	.000	.001	
	BEND - (	>> 605	.000	.000	-.000	-.001	.000	.001	
		>> 605F	-.000	.000	-.000	-.001	.000	.001	
	STRAIGHT - (	>> 610	-.008	.000	-.002	-.001	.001	.005	
		BEND - (	>> 610	-.009	.000	-.003	.001	.000	.005
	STRAIGHT - (	>> 610F	-.009	-.000	-.002	.006	.000	.006	
		>> 620	-.009	-.000	-.002	-.010	-.000	.012	
	229	BEND - (	>> 620	-.009	.001	-.002	-.008	-.001	.012
			>> 620F	-.008	.001	-.002	-.006	-.000	.012
		STRAIGHT - (	>> 625	-.008	.001	-.002	-.006	-.001	.012
			BEND - (	>> 625	-.006	.001	-.001	-.006	-.001
STRAIGHT - (		>> 625F	-.006	-.000	-.001	-.007	-.001	.011	
		>> 630	-.006	-.000	-.004	-.014	-.001	-.004	
BRANCH PT.		STRAIGHT - (	>> 635	-.006	-.000	-.008	-.022	-.001	.006
			BRANCH PT.	>> 635	-.006	-.000	-.008	-.022	-.001
STRAIGHT - (		>> 645	-.006	-.000	-.011	-.025	-.001	-.011	
		BEND - (	>> 645	-.006	.001	-.010	-.025	-.002	-.007
STRAIGHT - (		>> 645F	-.007	.002	-.006	-.025	-.002	-.006	
		>> 647	-.003	.002	-.001	-.026	-.002	-.006	

DYNAFLEX

LOADING - RESTRAINED WEIGHT

(+UNIF)

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DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
		>- 647	-.004	.002	-.001	-.026	-.002	-.006
	BEAD - (	647	-.009	.000	.003	-.026	-.002	-.005
	STRAIGHT - (	647	-.009	-.004	.004	-.023	-.002	-.005
		>- 655	.000	-.000	.004	.015	-.002	-.001
	BEAD - (	655	.001	.001	.004	.007	-.002	-.001
	STRAIGHT - (	655	.001	.002	.003	.004	-.002	-.001
		>- 660	-.000	.002	-.000	-.001	-.001	-.000
	BEAD - (	660	-.000	.002	-.000	-.002	-.001	-.000
	STRAIGHT - (	660	-.000	.001	-.000	-.003	-.000	-.000
ANCHOR		-- 665	-.000	.000	-.000	-.000	-.000	-.000
BRANCH PT.	STRAIGHT - (	635	-.006	-.000	-.008	-.022	-.001	.006
230		>- 670	-.004	-.000	.001	-.023	-.001	.006
	BEAD - (	670	-.003	-.001	.004	-.023	-.000	.005
	STRAIGHT - (	670	-.002	-.005	.006	-.022	-.000	.005
		>- 680	-.002	-.000	.006	.015	.000	.002
	BEAD - (	680	-.002	.002	.005	.008	.000	.002
	STRAIGHT - (	680	-.002	.002	.004	.005	.000	.002
		>- 685	-.000	.002	-.000	-.000	.000	.001
	BEAD - (	685	.000	.002	-.000	-.002	.000	.000
	STRAIGHT - (	685	.000	.001	.000	-.003	.000	.000
ANCHOR		-- 675	.000	.000	.000	-.000	.000	.000



HANGER DESIGN DATA  
-----EXPLANATORY NOTES:  
-----

- (1) ALL MOVEMENTS SHOWN BELOW REFER TO THE INITIAL EXPANSION LOADING CASE WHEREIN TENTATIVE HANGER LOCATIONS HAVE NOT BEEN VERTICALLY RESTRAINED AND ALL WEIGHT EFFECTS HAVE BEEN SUPPRESSED.
- (2) THE WEIGHT REACTIONS SHOWN BELOW REFER TO THE WEIGHT LOADING CASE WHEREIN TENTATIVE HANGER LOCATIONS HAVE BEEN RIGIDLY RESTRAINED IN THE VERTICAL DIRECTION AND ALL THERMAL EFFECTS INCLUDING IMPOSED ANCHOR MOVEMENTS HAVE BEEN SUPPRESSED.
- (3) IN THE FOLLOWING TABLE VSH DENOTES A VARIABLE SPRING HANGER AND CSH DENOTES A CONSTANT SUPPORT HANGER.
- (4) THE SUPPORT LOADS IN THE INSTALLED CONDITION HAVE BEEN CALCULATED TO PRODUCE THE CORRECT WEIGHT BALANCE IN THE OPERATING CONDITION.
- (5) IF THE FREE VERTICAL MOVEMENT EXCEEDS THE MAXIMUM RIGID SUPPORT DISPLACEMENT CRITERION OF .10 IN., A VARIABLE SPRING IS SELECTED. IF A VARIABLE SPRING HANGER CANNOT BE FOUND WHICH SATISFIES BOTH THE LOAD VARIATION CRITERION SELECTED BY THE USER AND THE WORKING RANGE OF THE SPRINGS LISTED IN SPRING TABLE (1) OF THE USER'S MANUAL, A CONSTANT SUPPORT HANGER IS RECOMMENDED.
- (6) THE NOMINAL ROD DIAMETER FOR RIGID HANGERS IS TAKEN FROM ANSI H31.1 - 1977 TABLE 121.2.2(A), (CARRYING CAPACITIES OF THREADED ASTM A 107 HOT ROLLED CARBON STEEL) ASSUMING A ROD TEMPERATURE OF 650 DEGREES FAHRENHEIT. A MINIMUM 1/2 IN. DIAMETER ROD WILL BE SELECTED FOR ALL PIPE WHICH IS NOMINALLY 2-1/2 IN. OR LARGER.
- (8) THE SO-CALLED THEORETICAL SPRING INSTALLATION LOAD SHOWN BELOW PRESUPPOSES THAT THE HANGER LOCATION IS RESTRAINED AGAINST VERTICAL MOVEMENT WHILE THE SPRING IS SET TO THE COLD LOAD.

DYNAFLEX

HANGER DESIGN DATA TABLE

HANGER LOC. NO.	SUPPORT LOAD (POUNDS)	FREE VERTICAL MOVEMENT (INCHES)	HORIZONTAL MOVEMENT (INCHES)		TYPE	PRELIMINARY HANGER SELECTION SPECIFICATION	SWING	
			X	Z			LENGTH (FEET)	ANGLE (DEG)
610F	1364.	.73	.50	.52	VSH	1 - SHORT RANGE SPRING. SPRING RATE = 680 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 1860 LBS SPRING LOAD IN THE OPERATING CONDITION = 1364 LBS	N/A	N/A
655N	1409.	.52	-.27	.39	VSH	1 - SHORT RANGE SPRING. SPRING RATE = 680 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 1765 LBS SPRING LOAD IN THE OPERATING CONDITION = 1409 LBS	N/A	N/A
680N	1428.	.50	-.05	.55	VSH	1 - SHORT RANGE SPRING. SPRING RATE = 680 LBS/IN THEORETICAL SPRING INSTALLATION LOAD = 1770 LBS SPRING LOAD IN THE OPERATING CONDITION = 1428 LBS	N/A	N/A

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DYNAPLEX

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*
*   STATIC SOLUTION ACCURACY CHECKS   *
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*****

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EACH BASIC STATIC SOLUTION COMPRISING EACH LOAD OR LOADING COMBINATION SPECIFIED IN THIS RUN HAS BEEN SUBJECTED TO EQUILIBRIUM AND COMPATIBILITY CHECKS FOR ALL POINTS IN THE SYSTEM:

LOADING = ZEROER THERMAL

STATIC EQUILIBRIUM AND COMPATIBILITY HAVE BEEN SATISFIED FOR THIS BASIC STATIC SOLUTION AT ALL POINTS IN THE SYSTEM.

LOADING = RESTRAINED FLIGHT (FOR IF)

STATIC EQUILIBRIUM AND COMPATIBILITY HAVE BEEN SATISFIED FOR THIS BASIC STATIC SOLUTION AT ALL POINTS IN THE SYSTEM.

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DYNAFLEX

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FOREWORD

THE FOLLOWING REPORT WAS GENERATED BY THE DYNAFLEX PIPING ANALYSIS PROGRAM. THE PROGRAM ACCOMMODATES THE WIDEST RANGE OF PRACTICAL SITUATIONS ENCOUNTERED IN MODERN PIPING SYSTEMS INCLUDING DYNAMIC AS WELL AS THERMAL, WEIGHT AND PRESSURE EFFECTS.

FOR FURTHER INFORMATION OR PROMPT ASSISTANCE IN USING THE DYNAFLEX PROGRAM, PLEASE CONTACT YOUR LOCAL COMPUTER CENTER REPRESENTATIVE OR:

ENGINEERING SERVICES MANAGER  
AUTOM COMPUTING CORPORATION  
1 METRO PLAZA  
505 THORNALL STREET  
EDISON, N.J. 08817

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THE STRESS FORMULATION OF  
ANSI B31.1 - 1977  
INCLUDING THE LATEST MANDATORY UPDATES  
HAS BEEN APPLIED IN THIS ANALYSIS

THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

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*
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*
*      STEARNS - ROGER CORP
*
*      JOHN SOLAR PILOT PLANT - C-21700
*
*      DS DESUP TO TH STOR HTR(X-PS-05-A-1)
*
*
*
*      2/26/80
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*
*
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DYNAFLEX (000204F-100)

LAST UPDATE - 12/25/79

PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
JOB - NS DESUP TO TH STOR HTR(X-NS-05-A-1)

DATE 2/26/80

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	FRM	TO	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
1	000								STEARNS - ROGER CORP, 2/26/80, PROJ. 10MW SO LAR PILOT PLANT - C-2 1700, JOBS DESUP TO T H STOR HTR(X-NS-05-A- 1)	1
2	6L1								APPLY B31.1-1973, EXA	2
3			600	605			4-0	L	MAT=LCS, DO=8.625, WT=.500, UNIF=11.1, TEMP=650.	3
4				610		15-1-15716		L		4
5				620			16-9-5/16	L		5
6				625		-1-3-11716	1-3-11716	L		6
7				630		10-4				7
8				635		13-9			WLT	8
9				640		16-0		L		9
10				645		-2-16-374		L		10
11				650			-18-3	L		11
12				655		-10-11		L		12
13				660			-4-0			13
14				665		-2-16-374		L		14
15				670			-18-3	L		15
16				675		-10-11		L		16
17				680			-4-0			17
18				685						18
19				690						19
20				695						20
21				700		-1610				21
22				705		-1710				22
23				710		-1610				23
24				715		-1610				24
25				720		-1610				25
26				725		-1610				26
27				730		-1610				27
				735		-1610				
				740		-1610				
				745		-1610				
				750		-1610				
				755		-1610				
				760		-1610				
				765		-1610				
				770		-1610				
				775		-1610				
				780		-1610				
				785		-1610				
				790		-1610				
				795		-1610				
				800		-1610				
				805		-1610				
				810		-1610				
				815		-1610				
				820		-1610				
				825		-1610				
				830		-1610				
				835		-1610				
				840		-1610				
				845		-1610				
				850		-1610				
				855		-1610				
				860		-1610				
				865		-1610				
				870		-1610				
				875		-1610				
				880		-1610				
				885		-1610				
				890		-1610				
				895		-1610				
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				905		-1610				
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				940		-1610				
				945		-1610				
				950		-1610				
				955		-1610				
				960		-1610				
				965		-1610				
				970		-1610				
				975		-1610				
				980		-1610				
				985		-1610				
				990		-1610				
				995		-1610				

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SPEC 1 (FRE/G, .17.013,  
.25/.082, 2.5/.638,  
.9/.531, 33/.125),  
SPEC 2 (FRE/G, .17.009,  
.25/.055, 2.5/.638,

INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	FREQ	PH	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
28	DRG								:9./,531,33./,125), :SHOCK 1(X/1.0SPEC 1), :SHOCK 2(Z/1.0 SPEC 1), :SHOCK 3(Y/1.SPEC2), :SHOCK 4(X/1.SPEC1, :Y/1.SPEC2,Z/1.SPEC1), :LOG,1.92	28
29	OUT								:LUMP POINTS(AUTOMATIC :,MAX SPACING=5-0 :,EXCLUDE ALL P NODES) :,MODE CUTOFF=20, :FREQUENCY CUTOFF=43 :SHOCK 1,SHOCK 2, :SHOCK 3,SHOCK 4	29

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INPUT DATA - (CARD IMAGES)

LINE NUMBER	TYPE	LOC	PRG	CO	DELTA X	DELTA Y	DELTA Z	RADIUS	ADDITIONAL DATA	LINE NUMBER
28	000								:9./,531,33./,125), :SHOCK 1(X/1.0SPEC 1), :SHOCK 2(Z/1.0 SPEC 1), :,SHOCK 3(Y/1.SPEC2), :SHOCK 4(X/1.SPEC1, :Y/1.SPEC2,Z/1.SPEC1), :LOG,1.92	28
29	001								:LUMP POINTS(AUTOMATIC :,MAX SPACING=5-0 :,EXCLUDE ALL R NODES) :,MODE CUTOFF=20, :FREQUENCY CUTOFF=33 :SHOCK 1,SHOCK 2, :SHOCK 3,SHOCK 4	29

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DYNAFLEX

STEARNS - ROGER CORP

PAGE 3

PROJECT - 10MW SOLAR PILOT PLANT - C-21700  
JOB - MS DESUP TO TH STOR RTR(X-MS-05-A-1)

DATE 2/26/80

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\*  
\* EDITED PIPING SYSTEM DESCRIPTION \*  
\*-----\*  
\*  
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EXPLANATORY NOTES

IN THE EDITED PIPING SYSTEM DESCRIPTION:

- (1) POINT LOCATION NUMBER SUFFIXES N, M AND F IDENTIFY THE NEAR POINT, THE MID POINT AND THE FAR POINT RESPECTIVELY OF A PIPING ELEMENT IN THE DIRECTION IN WHICH THE LINE IS TRAVERSED. FRACTIONAL NODE NUMBERS APPEARING IN THE FOLLOWING REPORTS IDENTIFY THOSE NODES AT WHICH THE PROGRAM HAS INTRODUCED EVENLY SPACED LUMP POINTS AND ARE DETERMINED FROM THE MAXIMUM LUMP SPACING CRITERIA ENTERED BY THE USER. IF FRACTIONAL NODE NUMBERS ARE ASSIGNED TO MORE THAN ONE ELEMENT ORIGINATING AT THE SAME BRANCH POINT, A LETTER WILL PRECEDE THE FRACTIONAL PORTION OF THE NUMBER TO UNIQUELY IDENTIFY EACH POINT (E.G. 2.A). LETTERS ARE ASSIGNED TO ELEMENTS ALPHABETICALLY IN THE ORDER OF THEIR APPEARANCE IN THE INPUT DATA.
- (2) AMBIENT TEMPERATURE IS ASSUMED TO BE 70 DEGREES FAHRENHEIT.
- (3) THE MAXIMUM ALLOWABLE DIFFERENCE IN LOOP CLOSURE CALCULATIONS IS ASSUMED TO BE .25 INCHES.
- (4) THE MAXIMUM ALLOWABLE DIFFERENCE BETWEEN THE CALCULATED AND INPUT COORDINATES OF USER DESIGNATED CHECK POINTS IS ASSUMED TO BE .25 INCHES.
- (8) IN ORDER TO MAKE IT POSSIBLE IN CERTAIN SITUATIONS TO PERFORM INDEPENDENT WEIGHT AND THERMAL ANALYSES IN THE SAME RUN AND APPLY THE CODE STRESS CALCULATIONS CORRECTLY, THE PROGRAM MAKES CERTAIN ASSUMPTIONS. IF THE INPUT DATA SPECIFIES IMPOSED MOVEMENTS ON ANCHORS (ANC) OR DISPLACEMENTS OR ROTATIONS ON RIGID RESTRAINTS (RAD,RAR) THESE ARE ASSUMED BY THE PROGRAM TO BE THERMAL EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE THERMAL. ALTERNATELY, IF THE INPUT DATA SPECIFIES FORCES AND MOMENTS (FOR,MON) OR IMPOSED DISPLACEMENTS OR ROTATIONS ON FLEXIBLE RESTRAINTS (RAD,RAR) THESE ARE ASSUMED TO BE WEIGHT EFFECTS AND INCLUDED ONLY IN THOSE LOADING CASES WHICH INCLUDE WEIGHT. IF THESE ASSUMPTIONS ARE INVALID FOR A PARTICULAR SITUATION, THE USER CAN ACCURATELY MODEL THE MOST GENERAL SITUATION BY USING THE AUXILIARY SUPPORT MOVEMENTS (ASH) AND THE AUXILIARY CONCENTRATED LOADS (ACL) FEATURE OF THE PROGRAM AS AN ALTERNATE METHOD OF SPECIFYING THESE EFFECTS.
- (15) THE TABLE OF CONTENTS APPEARS AT THE END OF THIS REPORT.

GENERAL DATA

COORDINATES OF ALL POINTS OF THE SYSTEM ARE WITH RESPECT TO POINT 600, WHICH IS THE ORIGIN

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DYNAFLEX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 4

MATERIALS USED (TEMPERATURE DEPENDENT PROPERTIES ARE SHOWN AT POINT OF TEMPERATURE CHANGE)

LCS - LOW CARBON STEEL  
 YOUNGS MODULUS AT AMBIENT TEMPERATURE = 27,900,000. PSI  
 POISSONS RATIO = .30  
 DENSITY = 490. POUNDS PER CUBIC FOOT

SHOCK AND SPECTRUM DATA

MODE CUTOFF SET AT 20

FREQUENCY CUTOFF SET AT 33.00 CPS

SPECTRUM 1 (LOGARITHMIC INTERPOLATION)		SPECTRUM 2 (LOGARITHMIC INTERPOLATION)	
FREQUENCY ---CPS	G	FREQUENCY ---CPS	G
.10	.01	.10	.01
.25	.05	.25	.05
2.50	.64	2.50	.64
9.00	.53	9.00	.53
33.00	.13	33.00	.13

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SHOCK SPECIFICATIONS

SHOCK 1 CONSISTS OF :  
 100.00 PERCENT OF SPECTRUM 1 IN THE X DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

SHOCK 2 CONSISTS OF :  
 100.00 PERCENT OF SPECTRUM 1 IN THE Z DIRECTION.

THE GLOBAL COMPONENTS OF THIS SHOCK ARE ASSUMED TO ACT INDEPENDENTLY

THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE ARE COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. CLOSELY SPACED MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.



DYNALFX

EDITED PIPING SYSTEM DESCRIPTION (CONTINUED)

PAGE 6

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
								8.625	.500	LCS	650	43.4	11.1
	STRAIGHT	>605F	0.00	1.00	4.00		4.39						
	STRAIGHT	>605F.1	0.00	5.39	4.00		4.39						
	STRAIGHT	>605F.2	0.00	9.77	4.00		4.39						
		>610N	0.00	14.16	4.00								
	BEND	610N	0.00	14.87	4.29	1.00	1.57	90.000					
		>610F	0.00	15.16	5.00		3.84						
	STRAIGHT	>610F.1	0.00	15.16	8.84		3.84						
	STRAIGHT	>610F.2	0.00	15.16	12.68		3.84						
	STRAIGHT	>610F.3	0.00	15.16	16.52		3.84						
242		>620N	0.00	15.16	20.36								
		* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 620N IN DIRECTION Y											
	BEND	620N	0.00	15.09	20.74	1.00	.79	45.000					
		>620F	0.00	14.87	21.07		.43						
	STRAIGHT	>625N	0.00	14.56	21.38								
	BEND	625N	-2.29	14.06	21.88	1.00	1.57	90.000					
		>625F	-1.00	13.85	22.08		4.44						
	STRAIGHT	>625F.1	-5.44	13.85	22.08		4.44						
	STRAIGHT	>625F.2	-9.89	13.85	22.08		4.44						
	STRAIGHT	>630	-14.33	13.85	22.08								

\* NOTES PERTAINING TO POINT 630 APPEAR ON THE FOLLOWING PAGE



DYNAFLEX

EDITED PIPE SYSTEM DESCRIPTION (CONTINUED)

PAGE 8

POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (FAHR)	DIST. LOADS (LBS PER FT)	
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF
												43.4	11.1
	STRAIGHT	(	>-647F	-46.08	10.96	21.08							
							4.06						
	STRAIGHT	(	>-647F.1	-46.08	10.96	17.02							
							4.06						
	STRAIGHT	(	>-647F.2	-46.08	10.96	12.96							
							4.06						
	STRAIGHT	(	>-647F.3	-46.08	10.96	8.90							
							4.06						
			>-655F	-46.08	10.96	4.83							
	BEND	(	655B	-46.08	10.67	4.13	1.00	1.57	90.000				
	STRAIGHT	(	>-655F	-46.08	9.96	3.83							
							4.46						
	STRAIGHT	(	>-655F.1	-46.08	5.50	3.83							
							4.46						
			>-660F	-46.08	1.04	3.83							
244	BEND	(	660B	-46.08	.33	3.54	1.00	1.57	90.000				
	STRAIGHT	(	>-660F	-46.08	.04	2.83							
							3.00						
ANCHR			--665	-46.08	.04	-.17							
			* CONTROL COORDINATES OF POINT 665, IN FEET :										
			X = -46.08, Y = .04, Z = -.17										
			RESULTANT DIFFERENCE IS ZERO										
	BR PT		--635	-28.08	13.85	22.08							
			* WELDING IBE AT POINT 635										
			* RIGID RESTRAINT AGAINST DISPLACEMENT AT POINT 635										
			IN DIRECTION Y										
	STRAIGHT	(					1.90						
			>-670F	-28.08	11.96	22.08							
	BEND	(	670B	-28.08	11.25	21.77	1.00	1.57	90.000				
			>-670F	-28.08	10.96	21.08							

DYNAFLEX

ENTRER PIPING SYSTEM DESCRIPTION (CONTINUED)

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POINT TYPE	ELEMENT DESIGNATION	POINT LOC. NO.	COORDINATES (FT)			RADIUS (FT)	LENGTH (FT)	ANGLE (DEG)	OUTSIDE DIAMETER (IN)	WALL THICK (IN)	TEMP (DEG) (FAHR)	DIST. LOADS (LBS PER FT)		
			X	Y	Z							PRESS (PSI)	PIPE WT. UNIF	
		>-670F	-28.08	10.96	21.38				8.625	.500	LCS	650.	43.4	11.1
STRAIGHT	-C						4.06							
STRAIGHT	-C	>-670F.1	-28.08	10.96	17.02		4.06							
STRAIGHT	-C	>-670F.2	-28.08	10.96	12.96		4.06							
STRAIGHT	-C	>-670F.3	-28.08	10.96	8.90		4.06							
		>-680F	-28.08	10.96	4.83									
BEND	-C	680F	-28.08	10.67	4.13	1.00	1.57	90.000						
STRAIGHT	-C	>-680F	-28.08	9.96	3.83		4.46							
STRAIGHT	-C	>-680F.1	-28.08	5.50	3.83		4.46							
		>-685F	-28.08	1.04	3.83									
BEND	-C	685F	-28.08	.33	3.54	1.00	1.57	90.000						
STRAIGHT	-C	>-685F	-28.08	.04	2.83		3.00							
ANCHR		--690	-28.08	.04	-1.17									
285		* CONTROL COORDINATES OF POINT 690, IN FEET :												
		X = -28.08, Y = .04, Z = -1.17												
		RESULTANT DIFFERENCE IS ZERO												



DYNAFLEX

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*****
*
* LUMPED DYNAMIC MODEL
* -----
*
*****

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EXPLANATORY NOTES:

(1) THE MASS OF THE SYSTEM IS LUMPED AT THE POINTS SHOWN BELOW. FRACTIONAL LOCATION NUMBERS, IF ANY, RESULT FROM ADDITIONAL MASS POINTS LOCATED BY THE PROGRAM BASED ON THE MAXIMUM LUMP SPACING CRITERION GIVEN IN THE INPUT DATA.

(2) THE MASS AT A POINT IS A COMBINATION OF DISTRIBUTED MASS LUMPED IN A LINEAR FASHION PLUS ANY ADDITIONAL CONCENTRATED MASS INPUT BY THE USER (SHOWN IN PARENTHESES).

(3) ROTATIONAL DEGREES OF FREEDOM ARE NOT INCLUDED IN THE MODEL UNLESS SPECIFICALLY ADDED BY THE USER. ALL THREE DISPLACEMENT DEGREES OF FREEDOM ARE AUTOMATICALLY DELETED AT ANCHORS. THE DEGREE OF FREEDOM ASSOCIATED WITH A RIGID RESTRAINT OR SPRINGER ORIENTED IN THE X, Y OR Z DIRECTION IS ALSO DELETED.

DISPLACEMENT FREEDOM

ROTATIONAL FREEDOM

MOMENT OF INERTIA (LB - IN X IN)

LOC. NO.	EIGHT (POUNDS)	DIRECTION			MOMENT OF INERTIA (LB - IN X IN)		
		X	Y	Z	X	Y	Z
246 600	82.				0.	0.	0.
605A	125.	X	Y	Z	0.	0.	0.
605F	162.	X	Y	Z	0.	0.	0.
605F.1	239.	X	Y	Z	0.	0.	0.
605F.2	239.	Y	Y	Z	0.	0.	0.
610A	162.	X	Y	Z	0.	0.	0.
610F	140.	X	Y	Z	0.	0.	0.
610F.1	209.	X	Y	Z	0.	0.	0.
610F.2	209.	X	Y	Z	0.	0.	0.
610F.3	209.	X	Y	Z	0.	0.	0.
620A	120.	X		Z	0.	0.	0.
620F	35.	X	Y	Z	0.	0.	0.
625A	35.	X	Y	Z	0.	0.	0.

DYNAFLEX

Labeled Dynamic Model (Continued)

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LUG.	LUG. NO. (4-DIGIT)	DISPLACEMENT FREEDOM	ROTATIONAL FREEDOM			
			MOMENT OF INERTIA (LB - IN X IN)	X	Y	Z
623F	164.	X Y Z	0. 0. 0.	0.	0.	0.
623F.1	242.	X Y Z	0. 0. 0.	0.	0.	0.
623F.2	242.	X Y Z	0. 0. 0.	0.	0.	0.
633	246.	X Y Z	0. 0. 0.	0.	0.	0.
633.1	250.	X Y Z	0. 0. 0.	0.	0.	0.
633.2	250.	X Y Z	0. 0. 0.	0.	0.	0.
635	293.	X Y Z	0. 0. 0.	0.	0.	0.
635.1	252.	X Y Z	0. 0. 0.	0.	0.	0.
635.2	252.	X Y Z	0. 0. 0.	0.	0.	0.
635.3	252.	X Y Z	0. 0. 0.	0.	0.	0.
643U	159.	X Y Z	0. 0. 0.	0.	0.	0.
643F	67.	X Y Z	0. 0. 0.	0.	0.	0.
6474	67.	X Y Z	0. 0. 0.	0.	0.	0.
647F	154.	X Y Z	0. 0. 0.	0.	0.	0.
647F.1	222.	X Y Z	0. 0. 0.	0.	0.	0.
647F.2	222.	X Y Z	0. 0. 0.	0.	0.	0.
647F.3	222.	X Y Z	0. 0. 0.	0.	0.	0.
655H	154.	X Y Z	0. 0. 0.	0.	0.	0.
655F	164.	X Y Z	0. 0. 0.	0.	0.	0.
655F.1	243.	X Y Z	0. 0. 0.	0.	0.	0.
660I	164.	X Y Z	0. 0. 0.	0.	0.	0.
660F	125.	X Y Z	0. 0. 0.	0.	0.	0.
665	37.	X Y Z	0. 0. 0.	0.	0.	0.

LOADED DYNAMIC MODEL (CONTINUED)

DYNAPLEX

LOC. NO.	DISPLACEMENT FREEDOM			ROTATIONAL FREEDOM		
	X	Y	Z	X	Y	Z
6700	45.					
670F	159.					
670F.1	222.					
670F.2	222.					
670F.3	222.					
6800	154.					
680F	160.					
680F.1	245.					
6850	160.					
685F	125.					
690	82.					

MOMENT OF INERTIA (L<sup>2</sup> - IN X IN)

X Y Z

NUMBER OF DYNAMIC DEGREES OF FREEDOM = 128, TOTAL MASS OF SYSTEM = 8270. POUNDS

\*\*\*\*\*  
 \* FREQUENCIES, PERIODS OF VIBRATION AND MASS PARTICIPATION FACTORS \*  
 \*\*\*\*\*

MODE NO.	FREQUENCY (CPS)	PERIOD (SECONDS)	MASS PARTICIPATION FACTORS		
			X	Y	Z
1	1.52635	.65516	5.88951712	.01630599	.27545956
2	2.76933	.36102	.63427009	.19195063	3.30186102
3	3.73092	.26803	.42244403	.23227409	2.45824379
4	4.58370	.21816	.15246088	.12506582	.87394329
5	5.41246	.18476	.82814111	.06446611	.21491020
6	6.36071	.15576	.13441814	.00978226	.04569968
7	7.56594	.13045	1.28663528	.03953791	.04643485
8	11.38211	.08416	.04389784	.25333829	.13978470
9	15.06787	.06637	.01175935	1.60590947	.26273959
10	18.11271	.05521	.01160239	.21226547	.07721311
11	19.80632	.05100	.04154665	.13389753	.06830509
12	21.56634	.04615	.12628511	.55265724	.14247040
13	22.29184	.04486	.23922217	1.13469082	.15453501
14	24.57877	.04069	.28300172	.05131970	.10034304
15	24.83101	.04027	.24212571	1.27905980	.29040153
16	27.29943	.03663	.05264097	1.61575727	.27436827
17	27.45644	.03642	.04139636	.06307545	.13956712
18	30.64419	.03263	.24465225	.87840752	.28348253
19	31.97122	.03126	.26542495	.01197196	.02177659

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NOTE:

SECTION 3.7.2 OF THE U.S. NUCLEAR REGULATORY COMMISSION STANDARD REVIEW PLAN DATED JUNE, 1975 STATES THAT AN ADEQUATE NUMBER OF DEGREES OF FREEDOM IN DYNAMIC MODELING MAY BE TAKEN EQUAL TO FIVE THE NUMBER OF MODES WITH FREQUENCIES LESS THAN 33 CPS.

FOR THIS ANALYSIS THE RATIO OF DYNAMIC DEGREES OF FREEDOM TO FREQUENCIES LESS THAN 33 CPS = 6.74

THE COLD ELASTIC MODEL WAS USED IN CALCULATING THE NATURAL FREQUENCIES OF THE SYSTEM IN THIS ANALYSIS.

DYNAREX

MODE ORTHOGONALITY CHECK  
-----

ON DIAGONAL TERMS: MAX. = 1.60000 MIN. = 1.00000

OFF DIAGONAL TERMS: MAX. = 1.70010 x 10<sup>-13</sup> MIN. = -1.12708 x 10<sup>-13</sup>

DYNAFLÉX

LOADING - SHOCK LOADING NO. 1

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GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

FORCES (POUNDS)

MOMENTS (FOOT-POUNDS)

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)				
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT	
<b>ANCHORS:</b>									
600	749.	231.	624.	1002.	3880.	6478.	7661.	10757.	
665	776.	268.	682.	1067.	2725.	7992.	6544.	10683.	
690	925.	104.	384.	1007.	1762.	8657.	7447.	11555.	
<b>RESTRAINTS:</b>									
6200	0.	147.	0.	147.	0.	0.	0.	0.	
630	0.	128.	0.	128.	0.	0.	0.	0.	
635	0.	178.	0.	178.	0.	0.	0.	0.	

LOADING - SHOCK LOADING NO. 1

FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

TABLE NO. 1

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LIG. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
	X	Y	Z	X	Y	Z
RESTRAINTS (CONTINUED)						
BASE	0.	81.	0.	0.	0.	0.

DYNAFLEX

LOADING = SHOCK LOADING NO. 1

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 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 6414. OCCURS AT POINT 6054

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ASME B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, FITTERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 PORTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLEx

LOADING - SHOCK LOADING NO. 1

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS (PSI)	
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		STRESS	
ANCHP		-- 600	624.	784.	7661.	3860.	6478.	1.00	1.00	24.51	5266.	
	STR - (	-- 605N	624.	783.	7661.	4571.	4265.	1.77	1.77	24.51	6414.	**
		(										
	BEND - (	605N	605.	798.	2742.	4551.	7869.	1.77	1.77	24.51	6159.	**
		(										
	STR - (	-- 605F	231.	974.	3537.	4177.	6913.	1.77	1.77	24.51	5720.	
		-- 605F.1	245.	949.	3537.	3644.	1446.	1.00	1.00	24.51	2585.	
	STR - (	-- 605F.2	237.	907.	3537.	455.	1234.	1.00	1.00	24.51	1847.	
		-- 610N	258.	867.	3537.	3793.	2593.	1.77	1.77	24.51	3761.	
	BEND - (	610N	563.	701.	4508.	4118.	480.	1.77	1.77	24.51	3973.	
		(										
	STR - (	-- 610F	559.	704.	3254.	4114.	2882.	1.77	1.77	24.51	3282.	
		-- 610F.1	505.	625.	3254.	3192.	549.	1.00	1.00	24.51	2247.	
	STR - (	-- 610F.2	505.	625.	3254.	2254.	1775.	1.00	1.00	24.51	2124.	
	STR - (	-- 610F.3	476.	566.	3254.	1507.	3700.	1.00	1.00	24.51	2495.	
	STR - (	-- 620N	435.	426.	3254.	413.	5321.	1.77	1.77	24.51	4055.	
		(										
	BEND - (	620N	325.	513.	5047.	365.	3843.	1.77	1.77	24.51	4120.	
		(										
	STR - (	-- 620F	174.	512.	6143.	164.	1827.	1.77	1.77	24.51	4158.	
		-- 625N	166.	551.	6143.	1979.	124.	1.77	1.77	24.51	4187.	
		(										
	BEND - (	625N	346.	461.	4511.	2166.	3997.	1.77	1.77	24.51	4155.	
		(										
	STR - (	-- 625F	330.	472.	518.	2154.	5705.	1.77	1.77	24.51	3970.	
		-- 625F.1	159.	442.	518.	1692.	3734.	1.00	1.00	24.51	2023.	
	STR - (	-- 625F.2	96.	432.	518.	759.	2046.	1.00	1.00	24.51	1098.	
		-- 630	96.	432.	518.	226.	556.	1.00	1.00	24.51	388.	
	STR - (	-- 630.1	149.	376.	518.	595.	1382.	1.00	1.00	24.51	779.	

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DYLALEX

LOADING - SHOCK LOADING NO. 1

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION LENGTHS ( IN (4) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 630.1	199.	376.	516.	595.	1382.	1.00	1.00	24.51	779.
	STR - (	>- 630.2	252.	367.	516.	1001.	2948.	1.00	1.00	24.51	1545.
	BR pT	-- 635	366.	360.	518.	1410.	4504.	1.35	1.35	24.51	2362.
	BR pT	-- 635	47.	650.	294.	2511.	4259.	1.35	1.35	24.51	2463.
	STR - (	>- 635.1	87.	651.	294.	1229.	1880.	1.00	1.00	24.51	1109.
	STR - (	>- 635.2	87.	651.	294.	74.	883.	1.00	1.00	24.51	457.
	STR - (	>- 635.3	194.	653.	294.	1340.	3172.	1.00	1.00	24.51	1692.
		>- 645D	389.	652.	294.	2621.	5628.	1.77	1.77	24.51	4032.
	BEND - (	645E	460.	590.	4331.	2703.	4217.	1.77	1.77	24.51	4296.
	STR - (	>- 645F	279.	701.	6215.	2522.	362.	1.77	1.77	24.51	4357.
		>- 647N	280.	742.	6215.	876.	2167.	1.77	1.77	24.51	4307.
255	BEND - (	647H	614.	501.	3042.	1213.	5592.	1.77	1.77	24.51	4204.
	STR - (	>- 647F	595.	524.	1747.	1193.	5773.	1.77	1.77	24.51	3988.
	STR - (	>- 647F.1	603.	584.	1747.	252.	3706.	1.00	1.00	24.51	2009.
	STR - (	>- 647F.2	629.	712.	1747.	1123.	1340.	1.00	1.00	24.51	1210.
	STR - (	>- 647F.3	645.	754.	1747.	2222.	1432.	1.00	1.00	24.51	1551.
		>- 655H	656.	776.	1747.	3319.	4237.	1.77	1.77	24.51	3670.
	BEND - (	655H	656.	777.	4410.	3316.	2526.	1.77	1.77	24.51	3885.
	STR - (	>- 655F	271.	960.	4955.	2931.	1026.	1.77	1.77	24.51	3793.
	STR - (	>- 655F.1	269.	1030.	4955.	2532.	59.	1.00	1.00	24.51	2681.
		>- 660J	262.	1033.	4955.	3043.	5769.	1.77	1.77	24.51	5325.
	BEND - (	660J	269.	831.	1187.	3498.	8079.	1.77	1.77	24.51	5762.

DYNAFLEY

LOADING - SHOCK LOADING NO. 1

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EXTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	BEND	( 660F	669.	831.	1187.	3498.	8079.	1.77	1.77	24.51	5762.
		>- 660F	682.	821.	6549.	3511.	5707.	1.77	1.77	24.51	6075. *
	STR	(									
	ANCHR	-- 665	682.	821.	6540.	2725.	7992.	1.00	1.00	24.51	5230.
	BR PT	-- 635	124.	600.	8524.	3920.	768.	1.35	1.35	24.51	4682.
	STR	(									
		>- 670F	124.	642.	8524.	344.	2959.	1.77	1.77	24.51	5857. *
	BEND	( 670F	281.	590.	4184.	290.	7680.	1.77	1.77	24.51	5676.
		>- 670F	287.	587.	2423.	259.	7954.	1.77	1.77	24.51	5396.
	STR	(									
		>- 670F.1	308.	737.	2423.	461.	5359.	1.00	1.00	24.51	2888.
	STR	(									
		>- 670F.2	323.	799.	2423.	915.	2450.	1.00	1.00	24.51	1745.
	STR	(									
		>- 670F.3	342.	846.	2423.	1374.	928.	1.00	1.00	24.51	1437.
	STR	(									
		>- 680F	356.	872.	2423.	1818.	4204.	1.77	1.77	24.51	3361.
256	BEND	( 680F	328.	883.	4889.	1791.	1975.	1.77	1.77	24.51	3612.
		>- 680F	109.	936.	5055.	1571.	1565.	1.77	1.77	24.51	3580.
	STR	(									
		>- 680F.1	106.	996.	5055.	2431.	92.	1.00	1.00	24.51	2746.
	STR	(									
		>- 685F	105.	1000.	5055.	1786.	6524.	1.77	1.77	24.51	5478.
	BEND	( 685F	344.	945.	1692.	2027.	8768.	1.77	1.77	24.51	5940. *
		>- 685F	324.	930.	7447.	2067.	5944.	1.77	1.77	24.51	6325. *
	STR	(									
	ANCHR	-- 690	384.	931.	7447.	1762.	8657.	1.00	1.00	24.51	5656.

DYNAFLEX

LOADING - SHOCK LOADING NO. 1

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\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR	STRAIGHT -(	600	.000	.000	.000	.000	.000	.000
	>= 6059		.016	.011	.000	.035	.045	.084
	(	6059	(2.6)	(3.1)	(.0)			
	BEND -(	6059	.032	.019	.004	.081	.097	.124
	(		(4.5)	(4.7)	(.8)			
	>= 605F		.063	.025	.020	.125	.128	.194
	STRAIGHT -(		(8.3)	(5.7)	(3.6)			
	>= 605F.1		.274	.025	.153	.160	.184	.259
	STRAIGHT -(		(40.9)	(5.7)	(26.8)			
	>= 605F.2		.527	.025	.304	.161	.241	.284
	STRAIGHT -(		(79.1)	(5.7)	(52.1)			
	>= 610N		.735	.025	.440	.130	.297	.271
	(		(18.2)	(5.7)	(74.7)			
257	BEND -(	610N	.842	.031	.457	.090	.305	.254
	(		(24.1)	(6.6)	(77.5)			
	>= 610F		.964	.042	.461	.049	.325	.248
	STRAIGHT -(		(26.2)	(8.4)	(78.2)			
	>= 610F.1		1.171	.064	.461	.010	.343	.202
	STRAIGHT -(		(29.4)	(16.0)	(78.2)			
	>= 610F.2		1.443	.080	.461	.019	.336	.197
	STRAIGHT -(		(41.3)	(17.3)	(78.2)			
	>= 610F.3		1.702	.035	.462	.039	.307	.112
	STRAIGHT -(		(59.1)	(11.0)	(78.2)			
	>= 620N		1.929	.000	.462	.047	.258	.066
	(		(79.2)	(.0)	(78.2)			
	BEND -(	620N	1.937	.004	.462	.049	.235	.065
	(		(81.0)	(1.2)	(78.4)			
	>= 620F		1.966	.007	.465	.050	.219	.065
	STRAIGHT -(		(82.3)	(2.1)	(78.7)			
	>= 625N		1.970	.010	.468	.050	.210	.060

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
>-- 625.1	( )	625.1	1.970 (183.3)	.010 (3.1)	.068 (79.1)	.050	.210	.060
>-- 625.0	( )	625.0	1.984 (184.6)	.013 (4.6)	.063 (79.2)	.061	.179	.055
>-- 625F	( )	625F	1.989 (185.0)	.010 (6.1)	.045 (77.9)	.072	.130	.034
>-- 625F.1	( )	625F.1	1.909 (185.0)	.012 (14.6)	.356 (68.3)	.063	.077	.008
>-- 625F.2	( )	625F.2	1.989 (185.0)	.010 (11.8)	.307 (59.7)	.055	.049	.008
>-- 630	( )	630	1.909 (185.0)	.000 (.0)	.273 (54.7)	.047	.042	.011
>-- 630.1	( )	630.1	1.909 (185.0)	.009 (7.7)	.241 (55.7)	.038	.045	.006
>-- 630.2	( )	630.2	1.989 (185.0)	.011 (6.7)	.197 (57.1)	.030	.063	.004
-- 635	( )	635	1.989 (185.0)	.000 (.0)	.129 (53.9)	.021	.104	.019
-- 635	( )	635	1.989 (185.0)	.000 (.0)	.129 (53.9)	.021	.104	.019
>-- 635.1	( )	635.1	1.909 (185.0)	.006 (6.8)	.076 (45.5)	.025	.071	.003
>-- 635.2	( )	635.2	1.909 (185.3)	.001 (4.8)	.072 (34.7)	.028	.057	.010
>-- 635.3	( )	635.3	1.909 (185.3)	.006 (6.7)	.113 (36.9)	.032	.089	.002
>-- 645.0	( )	645.0	1.909 (185.3)	.000 (.0)	.202 (53.6)	.036	.139	.022
>-- 645.1	( )	645.1	1.907 (185.1)	.005 (1.5)	.224 (57.1)	.024	.188	.049
>-- 645F	( )	645F	1.977 (184.4)	.008 (2.2)	.234 (59.5)	.011	.213	.075
>-- 647.0	( )	647.0	1.909 (183.3)	.009 (2.2)	.233 (61.2)	.010	.233	.081
>-- 647.1	( )	647.1	1.933 (180.6)	.009 (2.2)	.233 (62.6)	.010	.262	.112
>-- 647F	( )	647F	1.909 (175.3)	.007 (2.0)	.233 (63.2)	.019	.317	.133
>-- 647F.1	( )	647F.1	1.909 (147.2)	.015 (11.7)	.233 (63.2)	.025	.371	.158
>-- 647F.2	( )	647F.2	1.900 (129.9)	.040 (17.0)	.233 (63.2)	.019	.399	.184
>-- 647F.3	( )	647F.3	.619 (128.0)	.043 (17.0)	.233 (63.2)	.003	.398	.210
>-- 650	( )	650	.594 ( )	.040 ( )	.233 ( )	.032	.367	.235

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LOADING - SHEAR LOADING NO. 1

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>- 655N		.544 (139.4)	.030 (7.6)	.233 (59.2)	.032	.367	.235
	>- 655B		.528 (134.3)	.023 (5.8)	.230 (58.4)	.065	.331	.232
	>- 655F		.474 (120.7)	.018 (4.6)	.218 (55.4)	.096	.314	.232
	>- 655F.1		.257 (65.8)	.019 (4.9)	.116 (29.4)	.114	.234	.224
	>- 660N		.068 (17.3)	.019 (4.9)	.015 (3.8)	.095	.154	.174
	>- 660N		.048 (11.7)	.014 (3.6)	.003 (0.8)	.061	.119	.110
	>- 660F		.019 (4.9)	.008 (2.0)	.000 (0.0)	.026	.057	.071
	>- 665		.000 (0.0)	.000 (0.0)	.000 (0.0)	.000	.000	.000
	>- 635		1.505 (385.4)	.000 (0.0)	.129 (32.8)	.021	.104	.019
	>- 670A		1.977 (503.4)	.000 (0.0)	.135 (34.3)	.019	.161	.038
	>- 670H		1.558 (395.2)	.000 (0.0)	.138 (35.0)	.018	.201	.080
	>- 670F		1.917 (488.0)	.000 (0.0)	.139 (35.3)	.018	.277	.109
	>- 670F.1		1.648 (418.0)	.014 (3.6)	.139 (35.3)	.016	.352	.144
	>- 670F.2		1.328 (337.0)	.029 (7.3)	.139 (35.3)	.008	.396	.180
	>- 670F.3		.916 (231.0)	.030 (7.6)	.139 (35.3)	.005	.405	.216
	>- 660A		.653 (165.8)	.019 (4.9)	.139 (35.3)	.023	.377	.251
	>- 640H		.595 (151.7)	.014 (3.6)	.136 (34.6)	.041	.343	.251
	>- 660F		.527 (134.3)	.011 (2.8)	.129 (32.8)	.058	.327	.256
	>- 665F.1		.246 (62.5)	.011 (2.8)	.068 (17.3)	.067	.245	.251
	>- 665A		.074 (18.8)	.011 (2.8)	.009 (2.3)	.056	.163	.195
	>- 665B		.041 (11.1)	.009 (2.3)	.002 (0.5)	.037	.126	.124
	>- 665F		.021 (5.6)	.005 (1.3)	.000 (0.0)	.016	.061	.081
	>- 690		.000 (0.0)	.000 (0.0)	.000 (0.0)	.000	.000	.000

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ANCHOR

GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSURED TO ACT INDEPENDENTLY
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
600	313.	565.	1532.	1663.	9113.	764.	3002.	9626.
665	159.	227.	863.	925.	3981.	856.	1331.	4284.
690	253.	401.	1330.	1412.	5795.	1131.	2111.	6270.
<b>RESTRAINTS:</b>								
260	0.	762.	0.	762.	0.	0.	0.	0.
630	0.	153.	0.	153.	0.	0.	0.	0.
635	0.	530.	0.	530.	0.	0.	0.	0.

LOADING - SHOCK LOADING NO. 2

FORCES AND MOMENTS IN MEMBERS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LINC. NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
	X	Z	RESULTANT	X	Z	RESULTANT
6450	0.	337.	337.	0.	0.	0.

RESTRAINTS (CONTINUED)



\*\*\*\*\*  
 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
 \*-----\*  
 \*  
 \*\*\*\*\*

THE MAXIMUM STRESS OF 7284. OCCURS AT POINT 605N

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ASME B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (16) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 BOYLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNALLEX

LOADING - SHOCK LOADING NO. 2

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHR		-- 600	1532.	646.	3002.	9113.	764.	1.00	1.00	24.51	4712.
	STR	<-- 605N	1532.	646.	3002.	10806.	554.	1.77	1.77	24.51	7284. **
	BEND	<-- 605M	1485.	751.	2460.	10759.	1709.	1.77	1.77	24.51	7244. *
		>-- 605F	566.	1564.	778.	9841.	2690.	1.77	1.77	24.51	6637. *
	STR	<-- 605F.1	575.	1557.	778.	1331.	3147.	1.00	1.00	24.51	1715.
	STR	<-- 605F.2	594.	1539.	778.	111.	3246.	1.00	1.00	24.51	1635.
		>-- 610N	605.	1199.	778.	9037.	889.	1.77	1.77	24.51	5912.
	BEND	<-- 610M	1267.	445.	473.	9698.	1203.	1.77	1.77	24.51	6347.
		>-- 610F	1187.	624.	1056.	9618.	895.	1.77	1.77	24.51	6303.
	STR	<-- 610F.1	1060.	629.	1056.	7251.	1219.	1.00	1.00	24.51	3636.
	STR	<-- 610F.2	481.	648.	1056.	4793.	1371.	1.00	1.00	24.51	2495.
263	STR	<-- 610F.3	794.	670.	1056.	2259.	1417.	1.00	1.00	24.51	1404.
		>-- 620N	429.	155.	1056.	470.	1451.	1.77	1.77	24.51	1203.
	BEND	<-- 620M	420.	177.	1463.	447.	1051.	1.77	1.77	24.51	1204.
		>-- 620F	359.	291.	1689.	470.	658.	1.77	1.77	24.51	1215.
	STR	<-- 625N	299.	253.	1689.	697.	528.	1.77	1.77	24.51	1234.
	BEND	<-- 625M	219.	325.	845.	600.	1614.	1.77	1.77	24.51	1291.
		>-- 625F	198.	377.	676.	948.	1769.	1.77	1.77	24.51	1374.
	STR	<-- 625F.1	101.	269.	676.	533.	2497.	1.00	1.00	24.51	1293.
	STR	<-- 625F.2	94.	188.	676.	197.	2630.	1.00	1.00	24.51	1333.
		>-- 630	90.	230.	676.	691.	2209.	1.00	1.00	24.51	1169.
	STR	<-- 630.1	92.	437.	676.	460.	1673.	1.00	1.00	24.51	912.

DYNAFLEX

LOADING - SHOCK LEADING NO. 2

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT FLEM. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS (PSI)	
		AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		STRESS	
STR - (	>- 630.1	72.	437.	676.	466.	1673.	1.00	1.00	24.51	:	912.
STR - (	>- 630.2	79.	535.	676.	340.	2325.	1.00	1.00	24.51	:	1197.
BR PT	-- 635	99.	535.	676.	246.	4247.	1.35	1.35	24.51	:	2143.
BR PT	-- 635	59.	199.	213.	620.	1625.	1.35	1.35	24.51	:	872.
STR - (	>- 635.1	32.	98.	213.	358.	1644.	1.00	1.00	24.51	:	830.
STR - (	>- 635.2	32.	130.	213.	123.	1604.	1.00	1.00	24.51	:	794.
STR - (	>- 635.3	41.	220.	213.	216.	1424.	1.00	1.00	24.51	:	713.
STR - (	>- 645E	50.	415.	213.	473.	1417.	1.77	1.77	24.51	:	974.
HEND - (	645F	231.	343.	1160.	424.	938.	1.77	1.77	24.51	:	1006.
STR - (	>- 645F	313.	277.	1506.	410.	337.	1.77	1.77	24.51	:	1036.
STR - (	>- 647E	313.	332.	1506.	574.	369.	1.77	1.77	24.51	:	1072.
HEND - (	647F	449.	79.	827.	706.	1285.	1.77	1.77	24.51	:	1092.
STR - (	>- 647F	527.	313.	322.	584.	1455.	1.77	1.77	24.51	:	1038.
STR - (	>- 647F.1	490.	311.	322.	714.	1210.	1.00	1.00	24.51	:	705.
STR - (	>- 647F.2	586.	293.	322.	1918.	900.	1.00	1.00	24.51	:	1049.
STR - (	>- 647F.3	637.	246.	322.	3050.	541.	1.00	1.00	24.51	:	1524.
STR - (	>- 655E	756.	283.	322.	4097.	323.	1.77	1.77	24.51	:	2674.
HEND - (	655F	110.	383.	370.	4051.	254.	1.77	1.77	24.51	:	2644.
STR - (	>- 655F	249.	767.	366.	3590.	189.	1.77	1.77	24.51	:	2344.
STR - (	>- 655F.1	235.	892.	366.	473.	86.	1.00	1.00	24.51	:	295.
STR - (	>- 660E	229.	397.	366.	4000.	1172.	1.77	1.77	24.51	:	2714.
HEND - (	660F	745.	490.	797.	4554.	1079.	1.77	1.77	24.51	:	3082.

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DYNAFLEX

LOADING - SHOCK LOADING NO. 2

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POTNT TYPE	FLEN TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESISTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE			
	HEND	(	660F	795.	490.	797.	4558.	1079.	1.77	1.77	24.51	3082.
		)	>- 660F	883.	278.	1531.	4656.	456.	1.77	1.77	24.51	3155.
	STR	(										
	ANCHP	--	665	883.	277.	1531.	3981.	856.	1.00	1.00	24.51	2097.
	BR PT	--	635	908.	723.	2893.	849.	508.	1.35	1.35	24.51	1520.
	STR	(										
		)	>- 670F	934.	763.	2893.	997.	650.	1.77	1.77	24.51	2029.
	HEND	(	6700	872.	245.	1627.	1374.	2420.	1.77	1.77	24.51	2091.
		)	>- 670F	754.	501.	540.	1250.	2781.	1.77	1.77	24.51	2008.
	STR	(										
		)	>- 670F.1	919.	497.	540.	754.	2284.	1.00	1.00	24.51	1207.
	STR	(										
		)	>- 670F.2	1020.	488.	540.	2668.	1700.	1.00	1.00	24.51	1571.
	STR	(										
		)	>- 670F.3	1123.	478.	540.	4513.	1025.	1.00	1.00	24.51	2281.
	STR	(										
		)	>- 680F	1198.	474.	540.	6272.	390.	1.77	1.77	24.51	4092.
265	HEND	(	680F	1145.	508.	433.	6222.	412.	1.77	1.77	24.51	4054.
		)	>- 680F	921.	1216.	363.	5501.	322.	1.77	1.77	24.51	3582.
	STR	(										
		)	>- 680F.1	409.	1309.	368.	741.	156.	1.00	1.00	24.51	412.
	STR	(										
		)	>- 685F	403.	1354.	368.	6061.	1859.	1.77	1.77	24.51	4119.
	HEND	(	685F	1224.	765.	1571.	6884.	1556.	1.77	1.77	24.51	4664.
		)	>- 685F	1350.	675.	2111.	6990.	479.	1.77	1.77	24.51	4747.
	STR	(										
	ANCHP	--	686	1340.	470.	2111.	5795.	1131.	1.00	1.00	24.51	3069.

\*\*\*\*\*  
 \* DISPLACEMENTS AND ROTATIONS \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )  
 DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT			
			X	Y	Z	X AXIS	Y AXIS	Z AXIS	
ANCHOR	STRAIGHT	600	.000	.000	.000	.000	.000	.000	
		>- 605A	(.0)	(.0)	(.0)				
		>- 605B	.002	.025	.000	.084	.004	.033	
		( 605C	(1.3)	(9.2)	(.1)				
		>- 605D	.004	.045	.010	.192	.009	.044	
		( 605E	(2.6)	(15.6)	(3.1)				
		>- 605F	.013	.059	.046	.295	.013	.067	
		( 605F.1	(7.1)	(19.9)	(14.8)				
		>- 605F.2	.047	.059	.362	.375	.017	.092	
		( 605F.3	(9.9)	(19.9)	(113.5)				
		>- 610A	.176	.059	.713	.374	.027	.100	
		( 610B	(101.2)	(19.9)	(222.7)				
		>- 610C	.266	.059	1.029	.299	.038	.096	
		( 610D	(153.5)	(19.9)	(320.9)				
266	STRAIGHT	610E	.280	.073	1.067	.205	.043	.086	
		>- 610F	(159.9)	(24.0)	(352.7)				
		( 610F.1	.286	.096	1.077	.108	.051	.080	
		STRAIGHT	610F.2	(158.6)	(30.7)	(336.0)			
		>- 610F.3	.252	.105	1.077	.017	.061	.066	
		( 610F.4	(132.5)	(46.2)	(336.0)				
		>- 610F.5	.221	.140	1.078	.047	.072	.052	
		STRAIGHT	610F.6	(101.6)	(42.4)	(336.1)			
		>- 620A	.189	.075	1.078	.085	.083	.038	
		( 620B	(66.5)	(24.7)	(336.1)				
		>- 620C	.167	.000	1.078	.095	.093	.025	
		( 620D	(34.0)	(.0)	(336.1)				
		>- 620E	.165	.008	1.079	.094	.097	.024	
		( 620F	(31.9)	(2.5)	(336.6)				
	STRAIGHT	620G	.169	.010	1.083	.092	.099	.023	
	>- 620H	(35.1)	(2.6)	(337.9)					
	( 620I	.165	.020	1.089	.092	.100	.021		

DYWAFLEX

LOADING - SHOCK LOADING NO. 2

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DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>= 625	( )	.163	.020	1.089	.092	.100	.021
	(	(28.1)	(8.6)	(339.6)				
	BEAD - (	625	.161	.029	1.095	.084	.104	.016
	(	(28.7)	(9.8)	(340.7)				
	>= 625F	( )	.161	.032	1.088	.078	.115	.011
	STRAIGHT - (	625F	(28.9)	(11.2)	(337.3)			
	>= 625F.1	( )	.161	.023	1.010	.068	.132	.012
	STRAIGHT - (	625F.1	(28.9)	(12.3)	(307.7)			
	>= 625F.2	( )	.161	.011	.913	.057	.151	.013
	STRAIGHT - (	625F.2	(28.9)	(8.3)	(277.8)			
	>= 630	( )	.161	.000	.797	.047	.168	.009
	STRAIGHT - (	630	(28.9)	(.0)	(248.3)			
	>= 630.1	( )	.161	.005	.668	.036	.173	.002
	STRAIGHT - (	630.1	(28.9)	(5.5)	(220.8)			
	>= 630.2	( )	.161	.005	.546	.027	.159	.003
	STRAIGHT - (	630.2	(28.9)	(5.2)	(199.5)			
BRANCH PT.	-- 635	( )	.161	.000	.453	.020	.121	.006
	(	(28.9)	(.0)	(186.8)				
BRANCH PT.	-- 635	( )	.161	.000	.453	.020	.121	.006
	STRAIGHT - (	635	(28.9)	(.0)	(186.8)			
	>= 635.1	( )	.161	.003	.396	.018	.106	.001
	STRAIGHT - (	635.1	(28.9)	(8.8)	(181.4)			
267	>= 635.2	( )	.161	.003	.353	.018	.094	.002
	STRAIGHT - (	635.2	(28.9)	(10.0)	(177.5)			
	>= 635.3	( )	.161	.001	.321	.018	.082	.002
	STRAIGHT - (	635.3	(28.9)	(7.0)	(172.7)			
	>= 645	( )	.161	.000	.300	.018	.069	.003
	(	(28.9)	(.0)	(168.3)				
	BEAD - (	645	.161	.001	.299	.019	.058	.007
	(	(28.8)	(1.3)	(168.4)				
	>= 645F	( )	.160	.001	.301	.022	.053	.011
	STRAIGHT - (	645F	(28.5)	(1.9)	(169.9)			
	>= 647	( )	.158	.001	.305	.022	.049	.012
	(	(28.7)	(1.9)	(172.1)				
	BEAD - (	647	.156	.002	.308	.029	.044	.018
	(	(28.3)	(2.8)	(174.3)				
	>= 647F	( )	.155	.007	.310	.035	.035	.022
	STRAIGHT - (	647F	(28.1)	(5.5)	(175.3)			
	>= 647F.1	( )	.147	.037	.310	.034	.030	.027
	STRAIGHT - (	647F.1	(26.4)	(24.7)	(175.3)			
	>= 647F.2	( )	.134	.061	.310	.019	.029	.032
	STRAIGHT - (	647F.2	(20.1)	(36.0)	(175.3)			
	>= 647F.3	( )	.118	.066	.310	.009	.030	.036
	STRAIGHT - (	647F.3	(20.9)	(34.3)	(175.2)			
	>= 655	( )	.097	.042	.310	.050	.029	.041

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEVATION	LOG. DIST.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
STRAIGHT - ( 29.8)	6550	0.97	0.42	0.30	0.50	0.29	0.041	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 30.7)	6550	0.92	0.52	0.50	0.91	0.27	0.042	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 35.9)	6554	0.94	0.26	0.29	0.29	0.26	0.044	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 41.5)	6606	0.92	0.11	0.00	0.151	0.20	0.042	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 46.6)	6665	0.90	0.00	0.00	0.126	0.15	0.032	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 51.9)	6703	0.94	0.00	0.04	0.083	0.11	0.021	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 57.2)	6703	0.92	0.02	0.00	0.036	0.05	0.015	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 62.5)	6703	0.90	0.00	0.00	0.000	0.00	0.000	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 67.8)	6703	0.91	0.00	0.53	0.020	0.121	0.006	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 73.1)	6703	0.91	0.00	0.60	0.021	0.102	0.010	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 78.4)	6703	0.92	0.02	0.64	0.033	0.089	0.022	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 83.7)	6703	0.91	0.08	0.66	0.047	0.066	0.031	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 89.0)	6703	0.91	0.06	0.66	0.049	0.043	0.039	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 94.3)	6703	0.91	0.06	0.66	0.030	0.031	0.047	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 99.6)	6703	0.91	0.06	0.66	0.011	0.029	0.055	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 104.9)	6703	0.91	0.06	0.66	0.072	0.029	0.062	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 110.2)	6703	0.91	0.06	0.66	0.135	0.028	0.065	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 115.5)	6703	0.91	0.06	0.66	0.198	0.027	0.068	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 120.8)	6703	0.91	0.06	0.66	0.227	0.023	0.065	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 126.1)	6703	0.91	0.06	0.66	0.188	0.019	0.049	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 131.4)	6703	0.91	0.06	0.66	0.124	0.014	0.032	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 136.7)	6703	0.91	0.06	0.66	0.054	0.007	0.023	
ANCHOR								
BRANCH PT.								
STRAIGHT - ( 142.0)	6703	0.91	0.06	0.66	0.000	0.000	0.000	
ANCHOR								
BRANCH PT.								

GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED PIPING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSUMED TO ACT INDEPENDENTLY.
- (3) THE ASSUMED VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

FORCES (POUNDS)

MOMENTS (FOOT-POUNDS)

LOC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
600	50.	140.	100.	179.	717.	114.	320.	793.
665	18.	105.	90.	140.	505.	61.	120.	523.
670	20.	121.	102.	159.	581.	74.	149.	604.
<b>RESTRAINTS:</b>								
269	0.	60.	0.	60.	0.	0.	0.	0.
640	0.	30.	0.	30.	0.	0.	0.	0.
635	0.	71.	0.	71.	0.	0.	0.	0.



FORCES AND MOMENTS

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM

AT THE ANCHORS AND RESTRAINTS.

FORCES (POUNDS)

LOC.	Y	Z	RESULTANT	X	Y	Z	RESULTANT
605	9.	69.	0.	69.	0.	0.	0.

RESTRAINTS (CONTINUED)

LOC.	Y	Z	RESULTANT	X	Y	Z	RESULTANT
605	9.	69.	0.	69.	0.	0.	0.

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 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
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 \*  
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THE MAXIMUM STRESS OF 471. OCCURS AT POINT 605N

EXPLANATORY NOTES:

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ANSI B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE COLE MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (7) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 ORIFICE CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESIDUAL SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		:	:
ANCHR		600	100.	149.	520.	717.	114.	1.00	1.00	24.51	:	388.
	STR - (	>- 6050	100.	142.	520.	648.	64.	1.77	1.77	24.51	:	471. ***
		6050	122.	124.	278.	643.	161.	1.77	1.77	24.51	:	466. **
	STR - (	>- 605F	155.	111.	105.	592.	280.	1.77	1.77	24.51	:	431. **
		>- 605F.1	118.	108.	105.	168.	259.	1.00	1.00	24.51	:	160.
	STR - (	>- 605F.2	70.	93.	105.	134.	255.	1.00	1.00	24.51	:	150.
		>- 6100	69.	88.	105.	547.	95.	1.77	1.77	24.51	:	366.
	HEND - (	6100	91.	61.	37.	584.	135.	1.77	1.77	24.51	:	390.
		>- 610F	91.	78.	97.	575.	104.	1.77	1.77	24.51	:	384.
	STR - (	>- 610F.1	72.	75.	97.	456.	177.	1.00	1.00	24.51	:	244.
		>- 610F.2	61.	50.	97.	367.	239.	1.00	1.00	24.51	:	220.
272	STR - (	>- 610F.3	51.	47.	97.	233.	202.	1.00	1.00	24.51	:	158.
		>- 6200	40.	56.	97.	150.	145.	1.77	1.77	24.51	:	140.
	HEND - (	6200	35.	59.	121.	123.	128.	1.77	1.77	24.51	:	139.
		>- 620F	29.	63.	142.	116.	115.	1.77	1.77	24.51	:	140.
	STR - (	>- 6250	27.	63.	142.	126.	110.	1.77	1.77	24.51	:	142.
		>- 625F	42.	55.	197.	150.	137.	1.77	1.77	24.51	:	149.
	HEND - (	625F	53.	44.	103.	160.	137.	1.77	1.77	24.51	:	152.
		>- 625F.1	52.	54.	103.	121.	197.	1.00	1.00	24.51	:	124.
	STR - (	>- 625F.2	50.	29.	103.	75.	232.	1.00	1.00	24.51	:	129.
		>- 6300	41.	50.	103.	76.	234.	1.00	1.00	24.51	:	130.
	STR - (	>- 6300.1	47.	50.	103.	87.	213.	1.00	1.00	24.51	:	123.

LONGITUDINAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT FILE TYPE TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI
		AXIAL	RESOLVANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	>- 630.1	47.	38.	103.	87.	213.	1.00	1.00	24.51	123.
STR - (	>- 630.2	46.	51.	103.	99.	210.	1.00	1.00	24.51	128.
BR PT	-- 635	46.	51.	103.	91.	339.	1.35	1.35	24.51	182.
BR PT	-- 635	25.	60.	35.	75.	204.	1.35	1.35	24.51	147.
STR - (	>- 635.1	25.	60.	35.	84.	211.	1.00	1.00	24.51	113.
STR - (	>- 635.2	23.	25.	35.	128.	335.	1.00	1.00	24.51	177.
STR - (	>- 635.3	23.	25.	35.	98.	269.	1.00	1.00	24.51	141.
	>- 645h	23.	103.	35.	88.	98.	1.77	1.77	24.51	88.
BEND - (	645h	36.	99.	94.	60.	97.	1.77	1.77	24.51	96.
	>- 645f	56.	89.	147.	45.	94.	1.77	1.77	24.51	117.
STR - (	>- 647h	56.	93.	147.	172.	37.	1.77	1.77	24.51	148.
273 BEND - (	647h	60.	90.	106.	241.	105.	1.77	1.77	24.51	184.
	>- 647f	90.	60.	40.	287.	140.	1.77	1.77	24.51	209.
STR - (	>- 647f.1	79.	33.	40.	421.	144.	1.00	1.00	24.51	219.
STR - (	>- 647f.2	77.	57.	40.	448.	130.	1.00	1.00	24.51	229.
STR - (	>- 647f.3	74.	87.	40.	367.	72.	1.00	1.00	24.51	184.
	>- 655f	80.	92.	40.	415.	36.	1.77	1.77	24.51	271.
BEND - (	655f	109.	55.	32.	428.	44.	1.77	1.77	24.51	280.
	>- 655f	91.	81.	42.	588.	35.	1.77	1.77	24.51	254.
STR - (	>- 655f.1	100.	51.	42.	42.	49.	1.00	1.00	24.51	38.
STR - (	>- 655h	104.	91.	42.	396.	103.	1.77	1.77	24.51	267.
BEND - (	655h	117.	73.	92.	448.	79.	1.77	1.77	24.51	301.

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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## INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT FLEX. TYPE	FLEX. TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL STRESS PSI	
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		:	:
	BEND	-( 660M	117.	73.	92.	448.	79.	1.77	1.77	24.51	:	301.
		>- 660F	90.	105.	120.	455.	35.	1.77	1.77	24.51	:	306.
	STR	-(									:	
	ANCHR	-- 665	90.	107.	120.	505.	61.	1.00	1.00	24.51	:	256.
	BP DT	-- 635	65.	141.	213.	79.	111.	1.35	1.35	24.51	:	126.
	STR	-(									:	
		>- 670N	65.	139.	213.	172.	46.	1.77	1.77	24.51	:	180.
	BEND	-( 670M	91.	123.	137.	271.	164.	1.77	1.77	24.51	:	224.
		>- 670F	136.	70.	46.	327.	200.	1.77	1.77	24.51	:	251.
	STR	-(									:	
		>- 670F.1	112.	40.	46.	467.	169.	1.00	1.00	24.51	:	244.
	STR	-(									:	
		>- 670F.2	102.	63.	46.	503.	136.	1.00	1.00	24.51	:	256.
	STR	-(									:	
		>- 670F.3	102.	63.	46.	420.	80.	1.00	1.00	24.51	:	211.
	STR	-(									:	
		>- 680N	96.	104.	46.	477.	36.	1.77	1.77	24.51	:	312.
274	BEND	-( 680M	127.	61.	34.	490.	44.	1.77	1.77	24.51	:	320.
		>- 680F	103.	97.	37.	442.	35.	1.77	1.77	24.51	:	288.
	STR	-(									:	
		>- 680F.1	114.	103.	37.	50.	47.	1.00	1.00	24.51	:	38.
	STR	-(									:	
		>- 685N	118.	104.	37.	453.	129.	1.77	1.77	24.51	:	307.
	BEND	-( 685M	132.	85.	107.	513.	101.	1.77	1.77	24.51	:	346.
		>- 685F	102.	120.	149.	521.	35.	1.77	1.77	24.51	:	352.
	STR	-(									:	
	ANCHR	-- 690	102.	122.	149.	581.	74.	1.00	1.00	24.51	:	296.

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 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X AXIS	Y AXIS	Z AXIS	X AXIS	Y AXIS	Z AXIS
ANCHOR	STRAIGHT - (	600	.000	.000	.000	.000	.000	.000
	>= 605D		(.0)	(.0)	(.0)	.005	.000	.003
	(	605D	.000	(26.4)	(.1)	.012	.001	.005
	STRAIGHT - (	605F	(7.2)	(37.2)	(4.7)	.018	.001	.007
	>= 605F		(.0)	(.0)	(.0)	.022	.002	.009
	(	605F.1	(11.6)	(41.7)	(15.3)	.022	.003	.010
	STRAIGHT - (	610A	(59.6)	(42.1)	(60.1)	.022	.003	.010
	>= 605F.2		(.0)	(.0)	(.0)	.018	.005	.010
	(	610A	(45.1)	(42.4)	(55.3)	.012	.005	.009
	STRAIGHT - (	610F	(37.1)	(42.6)	(23.7)	.007	.006	.008
	>= 610F		(.0)	(.0)	(.0)	.002	.007	.007
	(	610F.1	(52.2)	(49.0)	(24.6)	.003	.008	.006
	STRAIGHT - (	620A	(25.5)	(37.2)	(25.3)	.006	.009	.005
	>= 610F.2		(.0)	(.0)	(.0)	.007	.010	.004
	(	620A	(37.5)	(55.4)	(25.4)	.007	.010	.004
	STRAIGHT - (	620B	(60.5)	(64.7)	(25.4)	.007	.010	.004
	>= 610F.3		(.0)	(.0)	(.0)	.007	.010	.004
	(	620B	(53.4)	(44.3)	(25.5)	.007	.010	.004
	STRAIGHT - (	620C	(25.1)	(.0)	(25.5)	.007	.010	.004
	>= 620C		(.0)	(.0)	(.0)	.007	.010	.004
	(	620C	(15.2)	(5.0)	(25.4)	.007	.010	.004
	STRAIGHT - (	625A	(14.6)	(5.4)	(25.5)	.007	.010	.004
	>= 625A		(.0)	(.0)	(.0)	.007	.010	.003

DYNAFLEX

LOADING - SHOCK LOADING NO. 3

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DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>- 6250		.009	.001	.065	.007	.010	.003
	(		(9.8)	(13.7)	(26.2)			
	FLAT - ( 6250		.009	.002	.065	.007	.010	.003
	(		(5.0)	(20.2)	(28.9)			
	>- 625F		.009	.002	.064	.007	.011	.002
	STRAIGHT - (		(5.1)	(22.7)	(30.8)			
	>- 625F.1		.009	.002	.059	.008	.011	.001
	STRAIGHT - (		(5.2)	(32.6)	(34.7)			
	>- 625F.2		.009	.001	.053	.008	.012	.001
	STRAIGHT - (		(5.2)	(25.0)	(35.2)			
	>- 630		.009	.000	.047	.010	.012	.001
	STRAIGHT - (		(5.3)	(.0)	(32.5)			
	>- 630.1		.009	.001	.042	.011	.012	.001
	STRAIGHT - (		(5.3)	(19.4)	(36.3)			
	>- 630.2		.009	.001	.037	.012	.010	.001
	STRAIGHT - (		(5.3)	(19.9)	(37.3)			
BRANCH PT.	-- 635		.009	.000	.034	.014	.008	.002
	(		(5.4)	(.0)	(43.0)			
BRANCH PT.	-- 635		.009	.000	.034	.014	.008	.002
	STRAIGHT - (		(5.4)	(.0)	(43.0)			
	>- 635.1		.009	.001	.033	.013	.008	.001
	STRAIGHT - (		(5.4)	(26.5)	(70.7)			
276	>- 635.2		.009	.002	.032	.013	.007	.000
	STRAIGHT - (		(5.4)	(38.5)	(92.5)			
	>- 635.3		.009	.001	.031	.013	.007	.001
	STRAIGHT - (		(5.4)	(26.4)	(83.7)			
	>- 6450		.009	.000	.030	.013	.007	.002
	(		(5.4)	(.0)	(54.4)			
	FLAT - ( 6450		.009	.000	.029	.013	.006	.002
	(		(6.1)	(9.3)	(44.1)			
	>- 645F		.009	.000	.029	.013	.005	.002
	STRAIGHT - (		(8.5)	(6.0)	(29.4)			
	>- 6470		.009	.000	.030	.012	.005	.002
	(		(12.4)	(6.0)	(20.6)			
	FLAT - ( 6470		.007	.001	.030	.011	.004	.002
	(		(14.8)	(9.6)	(25.1)			
	>- 647F		.009	.002	.030	.008	.003	.003
	STRAIGHT - (		(15.2)	(29.4)	(28.2)			
	>- 647F.1		.009	.008	.030	.005	.002	.003
	STRAIGHT - (		(25.6)	(67.7)	(28.3)			
	>- 647F.2		.009	.010	.030	.002	.002	.003
	STRAIGHT - (		(26.1)	(81.8)	(28.4)			
	>- 647F.3		.008	.009	.030	.004	.002	.003
	STRAIGHT - (		(11.3)	(59.7)	(28.4)			
	>- 6550		.000	.000	.030	.007	.002	.003

LOADING - SHOCK LOADING NO. 3

DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT IDENTIFICATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR	STRAIGHT - (	655	.006	.004	.030	.007	.002	.003
			(21.4)	(17.7)	(28.4)			
ANCHOR	STRAIGHT - (	656	.006	.003	.030	.010	.002	.003
			(25.2)	(11.4)	(27.1)			
ANCHOR	STRAIGHT - (	657	.007	.003	.028	.013	.002	.004
			(25.5)	(9.6)	(24.8)			
ANCHOR	STRAIGHT - (	658	.004	.003	.015	.015	.001	.004
			(19.5)	(9.5)	(13.2)			
ANCHOR	STRAIGHT - (	660	.001	.003	.002	.012	.001	.003
			(2.0)	(9.3)	(2.2)			
ANCHOR	STRAIGHT - (	660A	.000	.002	.000	.008	.001	.002
			(1.0)	(0.8)	(.8)			
ANCHOR	STRAIGHT - (	660F	.000	.001	.000	.004	.000	.001
			(.6)	(.6)	(.0)			
ANCHOR	STRAIGHT - (	665	.000	.000	.000	.000	.000	.000
			(.0)	(.0)	(.0)			
BRANCH PT.	STRAIGHT - (	675	.009	.000	.034	.014	.008	.002
			(5.4)	(.0)	(43.0)			
BRANCH PT.	STRAIGHT - (	676	.009	.000	.034	.013	.007	.002
			(12.2)	(.1)	(27.9)			
BRANCH PT.	STRAIGHT - (	677	.009	.001	.035	.012	.006	.002
			(17.2)	(6.9)	(34.6)			
BRANCH PT.	STRAIGHT - (	678	.009	.002	.035	.009	.004	.003
			(17.9)	(20.4)	(39.3)			
BRANCH PT.	STRAIGHT - (	678F.1	.010	.009	.035	.006	.003	.003
			(23.0)	(70.8)	(39.5)			
BRANCH PT.	STRAIGHT - (	678F.2	.011	.012	.035	.002	.002	.003
			(22.6)	(92.2)	(39.6)			
BRANCH PT.	STRAIGHT - (	679	.010	.010	.035	.005	.002	.004
			(11.5)	(60.6)	(39.7)			
BRANCH PT.	STRAIGHT - (	680	.010	.005	.035	.008	.002	.004
			(17.6)	(21.4)	(39.7)			
BRANCH PT.	STRAIGHT - (	680A	.010	.004	.034	.011	.002	.004
			(20.5)	(19.1)	(37.9)			
BRANCH PT.	STRAIGHT - (	680F	.009	.003	.032	.015	.002	.005
			(21.1)	(12.0)	(34.7)			
BRANCH PT.	STRAIGHT - (	680F.1	.009	.003	.017	.017	.002	.004
			(10.7)	(11.8)	(18.7)			
BRANCH PT.	STRAIGHT - (	675	.001	.003	.002	.010	.001	.003
			(1.7)	(11.7)	(3.2)			
ANCHOR	STRAIGHT - (	685	.000	.002	.000	.009	.001	.002
			(1.3)	(10.9)	(1.1)			
ANCHOR	STRAIGHT - (	685F	.000	.001	.000	.004	.000	.002
			(.8)	(0.2)	(.0)			
ANCHOR	STRAIGHT - (	690	.000	.000	.000	.000	.000	.000
			(.0)	(.0)	(.0)			



GENERAL NOTES FOR THIS LOADING:

- (1) THE SPECIFICATIONS FOR THE SHOCK LOAD INCLUDING SPECTRUM INTERPOLATION RULES APPEAR IN THE EDITED FIRING SYSTEM DESCRIPTION.
- (2) THE GLOBAL COMPONENTS OF THE SHOCK HAVE BEEN ASSIGNED TO ACT INDEPENDENTLY
- (3) THE ABSOLUTE VALUE OF THE RESPONSE QUANTITIES ASSOCIATED WITH EACH FUNDAMENTAL MODE HAVE BEEN COMBINED USING THE CLOSELY SPACED FREQUENCY (CSF) METHOD. MODES WITHIN 10 PERCENT OF EACH OTHER ARE DIRECTLY SUMMED AND THEN COMBINED WITH THE REMAINING MODAL QUANTITIES BY SRSS.

\*\*\*\*\*  
 \* FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS \*  
 \*\*\*\*\*

LIC. NO.	FORCES (POUNDS)				MOMENTS (FOOT-POUNDS)			
	X	Y	Z	RESULTANT	X	Y	Z	RESULTANT
<b>ANCHORS:</b>								
606	813.	627.	1658.	1950.	9931.	6524.	8235.	14457.
665	792.	366.	1119.	1419.	4851.	8038.	6679.	11522.
690	959.	431.	1388.	1741.	6085.	8731.	7742.	13160.
<b>RESTRAINTS:</b>								
6204	0.	779.	0.	779.	0.	0.	0.	0.
630	0.	201.	0.	201.	0.	0.	0.	0.
635	0.	563.	0.	563.	0.	0.	0.	0.

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FORCES AND MOMENTS ON ANCHORS AND RESTRAINTS (CONTINUED)

EXPLANATORY NOTES

THE FORCES AND MOMENTS SHOWN BELOW ARE THE ACTIONS OF THE PIPING SYSTEM ON THE ANCHORS AND RESTRAINTS.

LINE NO.	FORCES (POUNDS)			MOMENTS (FOOT-POUNDS)		
	X	Y	Z	X	Y	Z
5450	0.	353.	0.	0.	0.	0.

RESTRAINTS (CONTINUED)

DYNAFLEX

LOADING - SHOCK LOADING NO. 4

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 \*  
 \* INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS \*  
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THE MAXIMUM STRESS OF 9717. OCCURS AT POINT 605N

EXPLANATORY NOTES

- (1) THE FORCES AND MOMENTS SHOWN BELOW ARE WITH REFERENCE TO THE ELEMENT LOCAL AXES.
- (2) THE STRESSES SHOWN BELOW ARE CALCULATED IN ACCORDANCE WITH THE  
 ASTM B31.1 - 1977  
 PIPING CODE (INCLUDING ALL MANDATORY UPDATES)
- (3) A (\*\*) FOLLOWING A STRESS VALUE INDICATES THE MAXIMUM STRESS FOR THIS LOADING CONDITION.
- (4) A (\*) FOLLOWING A STRESS VALUE INDICATES A STRESS WITHIN 10 PER CENT OF THE MAXIMUM.
- (5) THE GILD MODULUS HAS BEEN USED IN THIS ANALYSIS.
- (6) THE STRESS INTENSIFICATION FACTORS (ABBREVIATED AS SIF IN THE TABLE HEADING BELOW) ARE  
 AUTOMATICALLY CALCULATED AND APPLIED AT BENDS, ELBOWS, MITERS AND ADMISSIBLE TYPES OF TEES  
 (AS SPECIFICALLY IDENTIFIED IN THE INPUT DESCRIPTION OF THE PIPING SYSTEM). USER ENTERED  
 STRESS INTENSIFICATION FACTORS TAKE PRECEDENCE OVER INTERNALLY CALCULATED VALUES.
- (14) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS  
 UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY  
 REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE  
 MODULUS IS USED FOR THE BRANCH LEG AT REDUCED  
 PORTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK.  
 THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.

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DYNAFLEX

LOADING - SHEAR LOADING NO. 4

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOC. ID.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS ( IN (3) )	LONGITUDINAL: STRESS PSI
			AXIAL	RESUL TANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
ANCHR		-- 600	1658.	1076.	8235.	9931.	6524.	1.00	1.00	24.51	7077.
	STR - (	>- 605D	1658.	1076.	8235.	11751.	4301.	1.77	1.77	24.51	9717. **
	BEND - (	605H	1608.	1103.	3695.	11699.	8054.	1.77	1.77	24.51	9520. *
		>- 605F	627.	1346.	3624.	10708.	7423.	1.77	1.77	24.51	8773. *
	STR - (	>- 605F.1	632.	1839.	3624.	3883.	3473.	1.00	1.00	24.51	3106.
	STR - (	>- 605F.2	648.	1620.	3624.	487.	3482.	1.00	1.00	24.51	2472.
	STR - (	>- 610H	654.	1422.	3624.	9816.	2743.	1.77	1.77	24.51	7017.
	BEND - (	610F	1390.	832.	4533.	10553.	1302.	1.77	1.77	24.51	7498.
		>- 610F	1314.	947.	3422.	10477.	3020.	1.77	1.77	24.51	7413.
	STR - (	>- 610F.1	1190.	926.	3422.	7936.	1349.	1.00	1.00	24.51	4282.
	STR - (	>- 610F.2	1018.	902.	3422.	5310.	2256.	1.00	1.00	24.51	3284.
281	STR - (	>- 610F.3	851.	878.	3422.	2620.	3967.	1.00	1.00	24.51	2868.
	STR - (	>- 620H	611.	456.	3422.	639.	5517.	1.77	1.77	24.51	4232.
	BEND - (	620H	532.	546.	5256.	555.	3986.	1.77	1.77	24.51	4294.
		>- 620F	392.	634.	6372.	511.	1945.	1.77	1.77	24.51	4334.
	STR - (	>- 625H	343.	610.	6372.	2102.	553.	1.77	1.77	24.51	4367.
	BEND - (	625H	411.	567.	4591.	2314.	4313.	1.77	1.77	24.51	4353.
		>- 625F	351.	606.	858.	2359.	5974.	1.77	1.77	24.51	4204.
	STR - (	>- 625F.1	188.	481.	858.	1778.	4496.	1.00	1.00	24.51	2404.
	STR - (	>- 625F.2	188.	471.	858.	787.	3340.	1.00	1.00	24.51	1732.
	STR - (	>- 630	100.	494.	858.	647.	2290.	1.00	1.00	24.51	1236.
	STR - (	>- 630.1	100.	497.	858.	761.	2180.	1.00	1.00	24.51	1206.

DYNAFLEX

LOADING - SHEAR LOADING NO. 4

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	ELEM TYPE	LOG. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION MODULUS (IN (3))	LONGITUDINAL STRESS (PSI)
			AXIAL	RESULTANT SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	STR - (	>- 630.1	180.	497.	858.	761.	2180.	1.00	1.00	24.51	1206.
	STR - (	>- 630.2	382.	647.	858.	1062.	3761.	1.00	1.00	24.51	1959.
BR PT		-- 635	382.	647.	858.	1435.	6200.	1.35	1.35	24.51	3194.
BR PT	STR - (	-- 635	66.	672.	365.	2588.	4567.	1.35	1.35	24.51	2617.
	STR - (	>- 635.1	95.	640.	365.	1283.	2506.	1.00	1.00	24.51	1390.
	STR - (	>- 635.2	198.	668.	365.	192.	1861.	1.00	1.00	24.51	933.
	STR - (	>- 635.3	198.	668.	365.	1361.	3488.	1.00	1.00	24.51	1841.
		>- 645H	344.	779.	365.	2665.	5805.	1.77	1.77	24.51	4150.
	BEND - (	645F	516.	699.	4485.	2737.	4321.	1.77	1.77	24.51	4413.
	STR - (	>- 645F	424.	759.	6396.	2555.	503.	1.77	1.77	24.51	4480.
		>- 647H	424.	818.	6396.	1061.	2198.	1.77	1.77	24.51	4441.
282	BEND - (	647H	743.	516.	3155.	1424.	5738.	1.77	1.77	24.51	4347.
	STR - (	>- 647F	685.	616.	1777.	1358.	5956.	1.77	1.77	24.51	4127.
	STR - (	>- 647F.1	724.	667.	1777.	866.	3901.	1.00	1.00	24.51	2141.
	STR - (	>- 647F.2	865.	774.	1777.	2267.	1619.	1.00	1.00	24.51	1617.
	STR - (	>- 647F.3	906.	811.	1777.	3791.	1532.	1.00	1.00	24.51	2183.
		>- 655H	1065.	831.	1777.	5289.	4249.	1.77	1.77	24.51	4549.
	BEND - (	655F	973.	862.	4426.	5253.	2341.	1.77	1.77	24.51	4707.
	STR - (	>- 655F	379.	1208.	4966.	4651.	1044.	1.77	1.77	24.51	4466.
	STR - (	>- 655F.1	379.	1365.	4966.	2380.	115.	1.00	1.00	24.51	2697.
		>- 660H	347.	1471.	4966.	5072.	5088.	1.77	1.77	24.51	5983.
	BEND - (	660F	1039.	967.	1433.	5763.	8151.	1.77	1.77	24.51	6542.

DYNAFLX

LOADING - SHOCK LOADING III. 4

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INTERNAL FORCES, MOMENTS AND STRESS CALCULATIONS (CONTINUED)

POINT TYPE	FLEM TYPE	LOC. NO.	FORCES (POUNDS)		MOMENTS (FOOT-POUNDS)			SIF		SECTION ( IN (3) )	LONGITUDINAL STRESS PSI
			AXIAL	RESIDUAL SHEAR	TORSION	IN-PLANE BENDING	OUT-OF-PLANE BENDING	IN PLANE	OUT-OF PLANE		
	BEND - (	660F	1559.	967.	1433.	5763.	8151.	1.77	1.77	24.51	6542.
		>- 660F	1119.	873.	6679.	5849.	5725.	1.77	1.77	24.51	6853.
	STR - (										
	ANCHR	-- 665	1119.	873.	6679.	4851.	8038.	1.00	1.00	24.51	5640.
	BR PT	-- 665	507.	950.	9004.	4011.	927.	1.35	1.35	24.51	4924.
	STR - (										
		>- 6700	507.	1006.	9004.	1068.	3029.	1.77	1.77	24.51	6201.
	BEND - (	670F	929.	650.	4491.	1430.	8054.	1.77	1.77	24.51	6053.
		>- 670F	618.	775.	2483.	1317.	8428.	1.77	1.77	24.51	5763.
	STR - (										
		>- 670F.1	976.	490.	2483.	999.	5828.	1.00	1.00	24.51	3139.
	STR - (										
		>- 670F.2	1075.	939.	2483.	2865.	2985.	1.00	1.00	24.51	2362.
	STR - (										
		>- 670F.3	1178.	977.	2483.	4736.	1385.	1.00	1.00	24.51	2704.
	STR - (										
		>- 680E	1251.	1000.	2483.	6548.	4222.	1.77	1.77	24.51	5304.
283	BEND - (	680F	1198.	1663.	4908.	6493.	2017.	1.77	1.77	24.51	5439.
		>- 680F	459.	1537.	5068.	5738.	1598.	1.77	1.77	24.51	5073.
	STR - (										
		>- 680F.1	434.	1680.	5068.	2542.	187.	1.00	1.00	24.51	2777.
	STR - (										
		>- 685N	433.	1687.	5068.	6335.	6785.	1.77	1.77	24.51	6860.
	BEND - (	685F	1279.	1182.	2181.	7195.	8906.	1.77	1.77	24.51	7560.
		>- 685F	1388.	1051.	7742.	7308.	5963.	1.77	1.77	24.51	7916.
	STR - (										
	ANCHR	-- 690	1388.	1051.	7742.	6085.	8731.	1.00	1.00	24.51	6442.

DYNAPLEX

LOADING - SHOCK LOADING NO. 4

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 \*  
 \* DISPLACEMENTS AND ROTATIONS \*  
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EXPLANATORY NOTES:

VALUES IN PARENTHESES ARE ACCELERATIONS ( IN PER SECOND SQUARED )

DISPLACEMENTS (INCHES)

ROTATIONS (DEGREES) ABOUT

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
ANCHOR		-- 600	.000	.000	.000	.000	.000	.000
	STRAIGHT	-(	(.0)	(.0)	(.0)			
		>- 605F	.016	.027	.000	.091	.045	.090
		(	(5.3)	(28.1)	(.1)			
	BEND	-(	.033	.049	.011	.209	.097	.131
		(	(8.9)	(40.6)	(5.7)			
		>- 605F	.064	.064	.051	.321	.129	.205
	STRAIGHT	-(	(15.9)	(46.5)	(21.5)			
		>- 605F.1	.288	.394	.394	.408	.185	.275
	STRAIGHT	-(	(75.7)	(46.9)	(131.2)			
		>- 605F.2	.556	.064	.776	.408	.242	.301
	STRAIGHT	-(	(137.6)	(47.2)	(235.3)			
		>- 610F	.830	.064	1.121	.326	.299	.287
		(	(197.3)	(47.3)	(330.3)			
	284	BEND	-(	.888	.080	1.162	.224	.308
		(	(205.6)	(47.1)	(342.5)			
		>- 610F	.947	.105	1.174	.119	.329	.261
STRAIGHT		-(	(208.4)	(48.9)	(345.9)			
		>- 610F.1	1.198	.159	1.174	.020	.348	.213
STRAIGHT		-(	(189.0)	(72.7)	(306.0)			
		>- 610F.2	1.460	.144	1.174	.051	.343	.165
STRAIGHT		-(	(180.2)	(79.2)	(346.0)			
		>- 610F.3	1.712	.083	1.174	.094	.318	.118
STRAIGHT		-(	(130.6)	(51.9)	(306.0)			
		>- 620F	1.536	.000	1.174	.107	.275	.071
		(	(183.9)	(.0)	(346.0)			
BEND		-(	1.555	.009	1.176	.104	.254	.069
		(	(184.8)	(5.7)	(346.5)			
		>- 620F	1.907	.016	1.181	.105	.240	.069
STRAIGHT	-(	(185.3)	(10.7)	(347.8)				
	>- 620F	1.976	.022	1.187	.104	.233	.064	

DYNAFLEX

LOADING - SHOCK LOADING NO. 4

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## DISPLACEMENTS AND ROTATIONS (CONTINUED)

POINT DESIGNATION	ELEMENT DESIGNATION	LOC. NO.	DISPLACEMENTS (INCHES)			ROTATIONS (DEGREES) ABOUT		
			X	Y	Z	X AXIS	Y AXIS	Z AXIS
	>- 6251		1.976	.022	1.187	.104	.233	.064
	(		(185.8)	(15.5)	(349.7)			
	BEND -(	6251	1.991	.032	1.190	.104	.208	.058
	(		(187.1)	(22.9)	(351.0)			
	>- 625F		1.996	.034	1.177	.106	.174	.036
	STRAIGHT -(		(187.7)	(26.0)	(347.6)			
	>- 625F.1		1.996	.026	1.073	.093	.153	.015
	STRAIGHT -(		(187.7)	(37.8)	(317.1)			
	>- 625F.2		1.996	.015	.965	.080	.160	.016
	STRAIGHT -(		(187.7)	(26.8)	(286.4)			
	>- 630		1.996	.000	.844	.067	.173	.015
	STRAIGHT -(		(187.7)	(.0)	(256.3)			
	>- 630.1		1.996	.010	.711	.054	.179	.007
	STRAIGHT -(		(187.7)	(21.5)	(230.6)			
	>- 630.2		1.996	.012	.581	.042	.172	.005
	STRAIGHT -(		(187.7)	(21.7)	(210.8)			
BRANCH PT.	-- 635		1.996	.000	.472	.032	.159	.020
	(		(187.7)	(.0)	(199.1)			
BRANCH PT.	-- 635		1.996	.000	.472	.032	.159	.020
	STRAIGHT -(		(187.7)	(.0)	(199.1)			
	>- 635.1		1.996	.007	.405	.033	.128	.003
	STRAIGHT -(		(187.7)	(28.2)	(200.0)			
285	>- 635.2		1.996	.004	.362	.036	.115	.010
	STRAIGHT -(		(187.7)	(41.0)	(203.1)			
	>- 635.3		1.996	.008	.342	.039	.121	.003
	STRAIGHT -(		(187.7)	(28.1)	(195.4)			
	>- 645F		1.995	.000	.363	.043	.155	.022
	(		(187.7)	(.0)	(184.8)			
	BEND -(	645F	1.993	.005	.374	.033	.197	.049
	(		(187.5)	(4.7)	(183.2)			
	>- 645F		1.984	.009	.382	.027	.219	.076
	STRAIGHT -(		(186.8)	(6.7)	(182.4)			
	>- 647F		1.989	.009	.385	.027	.238	.082
	(		(185.9)	(6.7)	(183.9)			
	BEND -(	647F	1.980	.009	.386	.032	.266	.113
	(		(183.4)	(10.2)	(186.9)			
	>- 647F		1.989	.010	.389	.040	.319	.135
	STRAIGHT -(		(177.5)	(21.4)	(188.5)			
	>- 647F.1		1.986	.041	.389	.042	.372	.161
	STRAIGHT -(		(153.7)	(73.0)	(188.5)			
	>- 647F.2		1.967	.071	.389	.027	.400	.187
	STRAIGHT -(		(137.9)	(91.9)	(188.5)			
	>- 647F.3		.927	.079	.389	.011	.400	.213
	STRAIGHT -(		(137.7)	(73.5)	(188.4)			
	>- 655F		.902	.052	.389	.059	.368	.239





DYNAFLEX

STEARNS - ROGER CORP

PAGE 51

PROJECT 100W SOLAR PILOT PLANT - C-21700  
 JOB BS DESUP TO THE STOR HTR(X-MS-05-A-1)  
 DATE 2/26/80

\*\*\*\*\*  
 \* STRESS SUMMARY \*  
 \* \*\*\*\*\* \*  
 \*  
 \*\*\*\*\*

EXPLANATORY NOTES:

- (1) EQUATION NUMBERS REFER TO ARTICLE 104.8 OF THE ANSI B31.1-1977 PIPING CODE.
- (7) THE EXACT SECTION MODULUS IS USED IN THE STRESS CALCULATIONS UNLESS THE SPECIFIC PIPING CODE OR THE USER EXPLICITLY REQUESTS OTHERWISE. IN THIS ANALYSIS THE SO-CALLED EFFECTIVE MODULUS IS USED FOR THE BRANCH LEG AT REDUCED OUTLET CONNECTIONS. THESE ARE IDENTIFIED WITH AN ASTERISK. THE EXACT SECTION MODULUS IS USED FOR ALL OTHER POINTS.
- (9) THE COLD MODULUS WAS USED IN ANALYZING THE OCCASIONAL LOADS.

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DYNALLEX

STRESS SUMMARY (CONTINUED)

PAGE 52

STRESSES ARE IN PSI

E1 (12) E2 (12) E3 (12) E4 (12)

SHOCK 1 SHOCK 2 SHOCK 3 SHOCK 4

LOC.  
NO.

LOC. NO.	SHOCK 1	SHOCK 2	SHOCK 3	SHOCK 4
600	5266	4712	388	7077
605N	6414	7284	471	9717
605M	6159	7244	466	9520
605F	5720	6637	431	8773
605F.1	2505	1715	160	3106
605F.2	1047	1635	150	2472
610N	3761	5912	366	7017
610M	3973	6347	390	7498
610F	3882	6303	384	7413
610F.1	2247	3636	244	4282
610F.2	2124	2495	220	3264
610F.3	2495	1404	158	2868
620N	4055	1293	140	4232
620M	4120	1204	139	4294
620F	4158	1215	140	4334
625N	4187	1234	142	4367
625M	4155	1251	140	4353
625F	3970	1374	152	4204
625F.1	2023	1293	124	2464
625F.2	1098	1333	122	1732
630	308	1109	130	1238

STRESS SUMMARY (CONTINUED)

STRESSS ARE IN PSI

EQ (12) EQ (12) EQ (12) EQ (12)

SUPPORTS 1 2 3 4

LOC. NO.

630.1	779	912	123	1206
630.2	1545	1147	129	1959
635	2362	2143	182	3194
635	2063	872	147	2617
635.1	1109	630	113	1596
635.2	457	794	177	933
635.3	1692	713	141	1841
645N	4032	979	28	4150
645M	4296	1006	56	4613
645F	4357	1036	117	4680
647N	4307	1072	148	4241
647M	4204	1092	184	4347
647F	3968	1036	209	4127
647F-1	2009	705	219	2141
647F-2	1210	1649	229	1617
647F-3	1551	1524	184	2183
655M	3670	2674	271	4549
655M	3685	2644	249	4707
655F	3793	2304	254	4666
655F-1	2681	205	38	2697
660M	5325	2714	267	5983

DYNAFLEX

STRESS SUMMARY (CONTINUED)

STRESSES ARE IN PSI

EQ (12) EQ (12) EQ (12) EQ (12)

SHOCK 1 SHOCK 2 SHOCK 3 SHOCK 4

LOC.  
30.

LOC.	SHOCK 1	SHOCK 2	SHOCK 3	SHOCK 4
660M	5762	3082	301	6542
660F	6975	3155	306	6853
665	5230	2097	256	5640
635	4682	1520	126	4924
670M	5857	2029	180	6201
670M	5676	2091	224	6053
670F	5396	2008	251	5763
670F.1	2208	1207	244	3139
670F.2	1745	1571	256	2362
670F.3	1437	2281	211	2704
680M	3361	4092	312	5304
680M	3612	4054	320	5435
680F	3580	3542	288	5073
680F.1	2746	412	38	2777
685M	5478	4119	307	6860
685M	5940	4664	306	7560
685F	6325	4747	352	7916
690	5656	3769	296	6442

200

DYNALOX

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STMPO 190

Department of Energy  
San Francisco Operations Office  
1333 Broadway  
Oakland, California 94612

Reply To: DOE Solar One Project Office  
P.O. Box 366  
Daggett, CA 92327

OCT 11 1984

Mr. Robert L. Gervais  
Solar One Project Office  
McDonnell Douglas Astronautics Corp.  
P.O. Box 366  
Daggett, CA 92327

Subject: Contractor Clearance of Contract DE-AC03-79SF10499  
Solar One Reports for DOE/TIC Inclusion.

Dear Bob:

Enclosed are copies of covers and title pages of eight reports prepared by McDonnell Douglas Astronautics Corporation for the Solar One Project under the above referenced contract. In preparation for delivery of these documents to DOE/TIC, I have prepared a SAN form 70 "Request for Patent Clearance" and a DOE form RA-426 "Recommendations for Announcement and Distribution of Documents" for each document.

Please have the appropriate MDAC personnel complete and sign these forms. As agreed, SAN form 70 should be forwarded to SAN/OPC by your office with copies of the completed SAN form 70 and the transmittal letter being sent to me. The completed DOE form RA-426 should be sent directly back to me.

The documents covered by this letter are:

<u>Primary Document No.</u>	<u>Secondary No.</u>	<u>Brief Title</u>
DOE/SF/10499-T100	STMPO 185	PSS Final Design Calculations (Book 10 of 26)
DOE/SF/10499-T101	STMPO 186	PSS Final Design Calculations (Book 11 of 26)
DOE/SF/10499-T102	STMPO 187	PSS Final Design Calculations (Book 12 of 26)
DOE/SF/10499-T103	STMPO 188	PSS Final Design Calculations (Book 13 of 26)
DOE/SF/10499-T104	STMPO 189	PSS Final Design Calculations (Book 14 of 26)

DOE/SF/10499-T105	STMPO 190	PSS Final Design Calculations (Book 15 of 26)
DOE/SF/10499-T106	STMPO 191	PSS Final Design Calculations (Book 16 of 26)
DOE/SF/10499-T107	STMPO 192	PSS Final Design Calculations (Book 17 of 26)

If you should have any questions or concerns please do not hesitate to contact me by telephone at, (619) 254-2672.

Sincerely,



S.D. Elliott, Jr., Director  
DOE Solar One Project Office

SDE/aks  
Project File: CCC009.RNO(SA3:)

Encl: Eight Document Covers W/forms 70 and RA-426

cc: Roger Gaither, SAN/OPC  
W.D. Matheny, DOE/TIC  
Mike Lopez, DOE/SAN (FGS)  
Mary Soderstrum, B&MCD





DEPARTMENT OF ENERGY  
SAN FRANCISCO OPERATIONS OFFICE

CONTRACTOR REQUEST FOR PATENT CLEARANCE  
FOR RELEASE OF UNCLASSIFIED DOCUMENT

Prime Contract No. DE-AC03-79SF10499
Subcontract No. (N/A)
Report No. (STMP0 190) DOE/SF/10499-T105
Date of Report September 1980
Name & Phone No. of DOE Technical Representative S.D. Elliott, Jr. (619) 254-2672

TO: Roger S. Gaither, Asst. Chief for Prosecution  
Office of Patent Counsel/Livermore Office  
P.O. Box 808, L-376  
Livermore, California 94550

FROM: McDonnell Douglas Corporation  
3855 Lakewood Blvd.  
Long Beach, CA 90846

- Document Title: Plant Support Subsystems Final Design Calculations  
(Book 15 of 26)
- Type of Document:  Technical Report,  Conference Paper,  Journal Article,  Abstract or Summary,  
 Copy of Oral Presentation,  Other (please specify): \_\_\_\_\_  
(Routine)
- In order to meet a publication schedule or submission deadline, patent clearance by \_\_\_\_\_  
would be desired.

SENDER IS TO CHECK BOX #4 OR #5 BELOW.

4. I have reviewed (or have had reviewed by technically knowledgeable personnel) this document for possible inventive subject matter (Subject Inventions) and that no inventions or discoveries (Subject Inventions) are deemed to be disclosed in this document except as stated below:
- Attention should be directed to pages \_\_\_\_\_ of this document.
  - This document describes matter relating to an invention:
    - Contractor Invention Docket No. \_\_\_\_\_
    - A disclosure of the invention was submitted to DOE on \_\_\_\_\_ (date)
    - A disclosure of the invention will be submitted shortly \_\_\_\_\_ (approximate date)
    - A waiver of DOE's patent rights to the contractor:
 

has been granted,  has been applied for; or  will be applied for \_\_\_\_\_ (date)
5. This document is being submitted, but no review has been made of this document for possible inventive subject matter.  
Provide copy of clearance to: Solar One Project Office  
P.O. Box 366, Daggett, CA 92327
6. Remarks:

Reviewing/Submitting Official: Name (Print/Type) \_\_\_\_\_  
Title \_\_\_\_\_  
Signature \_\_\_\_\_ Date \_\_\_\_\_

TO: INITIATOR OF REQUEST  
FROM: ASSISTANT CHIEF FOR PROSECUTION  
Office of Patent Counsel/Livermore Office

- No patent objection to above-identified release.
- Please defer release until advised by this office.

Signed \_\_\_\_\_ Date Mailed \_\_\_\_\_

U.S. DEPARTMENT OF ENERGY

DOE AND MAJOR CONTRACTOR RECOMMENDATIONS FOR  
ANNOUNCEMENT AND DISTRIBUTION OF DOCUMENTS

See Instructions on Reverse Side

1. DOE Report No. <u>DOE/SF/10499-T105 (STMPO 190)</u>	2. Contract No. <u>DE-AC03-79SF10499</u>	3. Subject Category No. <u>UC-62, 62c, 62d</u>
4. Title <u>Plant Support Subsystems Final Design Calculations (Book 15 of 26)</u>		
5. Type of Document ("x" one) <input checked="" type="checkbox"/> a. Scientific and technical report <input type="checkbox"/> b. Conference paper: Title of conference _____  Date of conference _____  Exact location of conference _____ Sponsoring organization _____ <input type="checkbox"/> c. Other (specify planning, educational, impact, market, social, economic, thesis, translations, journal article manuscript, etc.) _____		
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DOE/SF/10499-T105  
STMPO 190  
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10 MWe Solar Thermal  
Central Receiver Pilot Plant

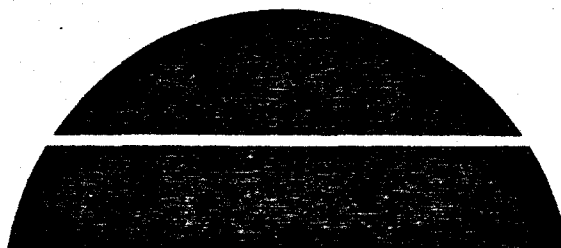
SOLAR FACILITIES DESIGN INTEGRATION

PSS FINAL DESIGN CALCULATIONS  
BOOK 15 OF 26--MAIN STEAM PIPING  
CONSTRUCTION PACKAGE 9 (RADL ITEM 7-8)

September 1980

WORK PERFORMED UNDER CONTRACT  
DE-AC03-79SF10499

STEARNS-ROGER ENGINEERING CORP  
4500 CHERRY CREEK DRIVE  
P.O. BOX 5888  
DENVER, CO 80217



## U.S. Department of Energy



Solar Energy

**10 MWe Solar Thermal  
Central Receiver Pilot Plant  
Solar Facilities Design Integration**

---

**PSS FINAL DESIGN CALCULATIONS  
BOOK 15 OF 26--MAIN STEAM PIPING  
CONSTRUCTION PACKAGE 9 (RADL 7-8)**

---

**September 1980**

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**STEARNS-ROGER ENGINEERING CORP  
4500 CHERRY DRIVE  
P.O. BOX 5888  
DENVER, CO 80217**

**PREPARED FOR THE  
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SOLAR ENERGY  
UNDER CONTRACT DE-AC-03-79SF10499**

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4. Title  
Plant Support Subsystems Final Design Calculations (Book 15 of 26)

5. Type of Document ("x" one)  
 a. Scientific and technical report  
 b. Conference paper: Title of conference \_\_\_\_\_

\_\_\_\_\_ Date of conference \_\_\_\_\_

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S.D. Elliott, Jr., Director, DOE Solar One Project Office  
Organization

P.O. Box 366, Daggett, CA 92327 (619) 254-2672  
Signature

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DEPARTMENT OF ENERGY  
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Prime Contract No. DE-AC03-79SF10499
Subcontract No. (N/A)
Report No. (STMPD 190) DOE/SF/10499-T105
Date of Report September 1980
Name & Phone No. of DOE Technical Representative S.D. Elliott, Jr. (619) 254-2672

TO: Roger S. Gaither, Asst. Chief for Prosecution  
Office of Patent Counsel/Livermore Office  
P.O. Box 808, L-376  
Livermore, California 94550

FROM: McDonnell Douglas Corporation  
3855 Lakewood Blvd.  
Long Beach, CA 90846

- Document Title: Plant Support Subsystems Final Design Calculations  
(Book 15 of 26)
- Type of Document:  Technical Report,  Conference Paper,  Journal Article,  Abstract or Summary,  
 Copy of Oral Presentation,  Other (please specify): \_\_\_\_\_  
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  - This document describes matter relating to an invention:
    - Contractor Invention Docket No. \_\_\_\_\_.
    - A disclosure of the invention was submitted to DOE on \_\_\_\_\_ (date)
    - A disclosure of the invention will be submitted shortly \_\_\_\_\_ (approximate date)
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6. Remarks:

Reviewing/Submitting Official: Name (Print/Type) John P. Scholz  
Title Asst. Chief Patent Counsel, MDC (MS 122-23)  
Signature [Signature] Date 8 Nov 84

TO: INITIATOR OF REQUEST  
FROM: ASSISTANT CHIEF FOR PROSECUTION  
Office of Patent Counsel/Livermore Office

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